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Blast and Impact Engineering on Structures and Materials

Edited by
Ricardo Castedo, Lina M. López and Anastasio P. Santos

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Preface

This Special Issue aimed to collect and present all breakthrough research on all intentional or unintentional explosions and impact problems. During the last decade, the investigation of these phenomena has been an active area of research in different fields (i.e., civil, defense, mining, aeronautical, naval, etc.), including experimental studies, analytical models, or numerical simulations; and this Special Issue is a faithful reflection of this trend.

A total of twenty-six papers (twenty-five research papers and one review paper) in various fields of blast and impact engineering including blast loading issues over structures, beams, walls; penetration and impact; explosives safety and security; blasting effects on rocks and tunnels; are presented in this Special Issue.

In the topic related with blast loading over structures, seven papers have been published. These articles include papers by Santos et al. [1], Mollaie et al. [2], Shunck and Eckenfels [3], Lim et al. [4], Li et al. [5], Hung et al. [6] and Lukić and Draganić [7] dealing with the effects of explosives (some as improvised explosives devices) on full-scale structures, masonry walls, protective barriers, or the safety of personnel inside tunnel structures.

Eight articles have been published in the topic of penetration and impact. These papers are published by: Yang et al. [8], Imran Latif et al. [9], Pu et al. [10], Yuan et al. [11], Wang et al. [12], Wang et al. [13], Fowler and Teixeira-Dias [14] and Malesa et al. [15]. They deal with different materials such as concrete, ultra-high-performance concrete, recycled aggregate concrete pavements or different phenomena such as penetration of liquid cabin for warships, or the impact of fragments at high speed on wings or on unmanned spacecraft. Most of the papers include tests and numerical simulations.

Six papers published in this issue (Sánchez-Monreal et al. [16], Taylor [17], Marín et al. [18], Chen et al. [19], Filice et al. [20] and Traná et al. [21]) have focused on the development of numerical or empirical models that predict the effects of the shock waves of explosives. The papers are varied, such as the development of computational tools to approximate the effects over structural elements, estimation of attenuation laws, or peak pressure predictions.

Finally, five papers (Alsabhan et al. [22], Ko et al. [23], Dong et al. [24], Choi and Lee [25], Huo et al. [26]) have been published on experiments and numerical models focused on the effects of explosives on rocks. These include topics as varied as the effect of surface impact of projectiles on tunnels, the use of different stemming materials, or models for predicting the peak particle velocity from blasting-induced vibrations.

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Ricardo Castedo, Lina M. López, and Anastasio P. Santos

Editors

Article

Reinforced Concrete Building with IED Detonation: Test and Simulation

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Abstract: There is growing concern about the possibility of a suicide bomber being immolated when the army forces or the law enforcement agencies discover the place where they prepare their material or simply find themselves inside a building. To study the possible effects that these improvised explosive devices (IEDs) would have on the structures, eight tests were carried out with various configurations of IEDs with vest bombs inside a reinforced concrete (including walls and roof) building constructed ad hoc for these tests. These vests were made with different explosives (black powder, ANFO, AN/AL, PG2). For the characterization of these tests, a high-speed camera and pressure and acceleration sensors were used. The structure behaved surprisingly well, as it withstood all the first seven detonations without apparent structural damage. In the last detonation, located on the ground and with a significant explosive charge, the structural integrity of the roof and some of the walls was compromised. The simulation of the building was carried out with the LS-DYNA software with a Lagrangian formulation for the walls, using the LBE (based on CONWEP) module for the application of the charge. Despite the difficulty of this simulation, the results obtained, in terms of applied pressures and measured accelerations, are acceptable with differences of about 20%.

Keywords: numerical modeling; LS-DYNA; IEDs; field test; reinforced concrete

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1. Introduction

The risk of an attack in the operating area or in the neutral zone has increased in the last decades. Many of these attacks are carried out using improvised explosive devices (IEDs) which are unconventional weapons that can be easily fabricated. Access to the products and knowledge necessary for the use and creation of IEDs has risen in recent years. As an example, terrorist attacks such as Flight 9268 which covered the Egypt–Russia route (2015), Paris (2015), Belgium (2016), Germany (2016), England (2017) and Spain (2017), all of which resulted in fatalities, demonstrate the urgent need to better understand the possible effects of these devices on people and/or structures [1–3]. Most of the IED attacks over the past 15 years involved small bombs of less than 5 kg [4] or a person-borne improvised explosive device (PBIED) usually containing less than 10 kg of explosives [5,6]. Moreover, terrorist actions may most often be carried out in crowded areas, in urban environments, near critical infrastructure or even inside buildings. For this reason, there has been considerable research on structural damage and blast effects on buildings, and therefore, much literature has been published on blast mitigation and retrofit methods [7,8]. However, many of these works are not based on experimental tests and use numerical modeling to predict the structural response in different scenarios by comparing it with empirical equations [9,10].

Other times, numerical modeling results are validated with experimental data found in the literature [11,12]. In these scenarios, comparisons can only be made with the available data which in some cases are deficient. Numerical modeling is a good alternative and a very useful tool, but in the case of blast loading, it must be calibrated and validated by corresponding field tests.

Since concrete is a construction material widely used in many building structures, its behavior has been extensively studied through experimental tests and numerical simulation. Experimental data are essential to understand the explosive phenomenon and predict the structural response, but this kind of experiment is very difficult to implement and has a high cost. For these reasons, many of the experiments are based on single structural elements such as beams [13,14] or slabs [15–17] which are easier to handle and monitor. The data obtained in this type of trial can be used to calibrate numerical models as well as to check different laws of materials' behavior. However, these results cannot be used to analyze and predict the structural response of a whole building, as the element failure causes loads to be redistributed to the neighboring supporting elements. The failure of individual structural elements can have a decisive influence on whether or not the structure collapses. Progressive collapse of structures has also been studied by numerous researchers, although not many have conducted experimental tests at full scale [18–20], and there are even fewer cases in which, in addition to the structure, non-structural elements such as masonry walls or the roof are represented [21,22].

However, in the last decade, most casualties of terrorism have been caused by shootings, vehicle impacts or PBIED attacks [23]. In these scenarios, there is no need to protect any kind of structure. In addition, existing infrastructure has proven to be highly resilient and robust against blast loadings. On the other hand, there is a research gap related to primary and secondary blast injuries, even though they are the main source of fatalities. Primary blast injuries are caused by the blast pressure wave and generally affect gas-containing organs, usually the eardrums and lungs. The secondary blast injuries result from the direct impact of airborne debris due to the blast wind [24,25]. Therefore, more research is needed to understand casualty risks from bomb fragmentation and blast overpressure hazards, especially from IEDs and PBIEDs produced inside buildings. In this situation, the overpressure is amplified by the reflection of the blast wave on the enclosure walls, and the explosion yield can be increased up to eight times.

In this research, eight tests were carried out with different IED configurations simulating a PBIED inside a building using vest bombs. The building consisted of a small concrete structure of 6.80×5 m with a corridor and an inner room. This work focuses on the analysis of the high-speed video, pressures and accelerations recorded during the tests and the development of a suitable numerical model capable of reproducing the behavior of the blast effects inside the building.

2. Test Description and Instrumentation

In this context, the German Federal Office of Criminal Investigation (namely BKA), which is part of the Federal Ministry of the Interior, started a project in 2017 on the effects of IEDs on state security forces personnel. This project consisted of numerous tests carried out in Germany, with different types of explosive charges, with and without shrapnel, and at different targets. During the project, the BKA had the collaboration of the Centre of Excellence against Improvised Explosive Devices (C-IED COE), a member of the NATO Centre of Excellence community, to advise on the creation of the IEDs.

The last phase of testing consisted of creating a reinforced concrete structure that reproduces a possible location where terrorists prepare their material and detonate them before counter-terrorism police can get in and arrest them. This phase was carried out at the Sierra del Retín maneuvering and firing range, Barbate, Cádiz, from 18 to 20 September 2018. The concrete structure was designed by the “*Subdirección General de Proyectos y Obras—DIGENIN*” (part of the Spanish Ministry of Defense), while the instrumentation, measurements and modeling were carried out by the staff of the E.T.S.I. Minas y

Energía (Universidad Politécnica de Madrid—UPM). Finally, the explosive charges and their detonations were prepared by the Explosive Ordnance Disposal Section (SEDEX) of the Amphibious Mobility Group (GRUMA) of the Third Army (TEAR)—Marines in collaboration with personnel from NATO's Counter Improvised Explosive Devices Centre of Excellence (C-IED COE).

The structure built ad hoc for the tests was made of reinforced concrete and consisted of a perimeter corridor and an interior room in which the IEDs were placed. The design of the structure was based on the project requirements suggested by the BKA. The original idea was to have brick enclosures, but these were going to be destroyed after each trial making the project unfeasible in terms of time and money. For this reason, the structure was redesigned with reinforced concrete walls, with greater thicknesses in the exterior walls than in the interior walls. Doors and windows were also aligned to improve the venting of the shock wave. CYPECAD code was used for the design, and it was developed under the Spanish Technical Building Code (CTE) based on Eurocodes required by the European Union. The ground plan dimensions of the structure are shown in Figure 1, with the height between the floor and ceiling of the structure equal to 3 m. The outer walls were built with a thickness of 40 cm while the inner walls were 30 cm thick and the roof slab 25 cm. The concrete used for both the walls and the roof slab had a nominal compressive strength of 40 MPa, a density equal to 2300 kg/m³, a tensile strength of 3.5 MPa, an elastic modulus equal to 30.9 GPa and a 20 mm maximum aggregate size. The reinforcement of the structure, made of B-500 C corrugated steel, was equally distributed on both sides of the walls and the roof slab in both directions (vertical and horizontal). The vertical reinforced steel of outer walls was constructed with a 12 mm diameter rebar evenly spaced at 300 mm, while in the inner walls, the diameter of the rebar used was 10 mm spaced at 200 mm. The horizontal steel of the outer walls had a diameter equal to 8 mm spaced at 150 mm; although in the inner walls the diameter used was the same, the spacing was increased to 200 mm. The reinforcement of the roof was made symmetrically on both sides, the inside of the cubicle where the detonation was located and the outside, using 12 mm diameter rebars with a square mesh of 150 mm on each side. In height, both reinforcements were 180 mm apart, with the thickness of the slab equal to 250 mm; therefore, a sufficient concrete layer was ensured on both sides. Finally, there was a perimeter reinforcement in all the joints between the walls and the roof, with 16 mm diameter rebars separated in height by 160 mm. The steel was assumed to have a density of 7850 kg/m³, Young's modulus equal to 200 GPa, yield strength of 500 MPa, Poisson's ratio of 0.3 and tangent modulus of 20 GPa, following the EN 1992-1-1:2004 [26] and EN 1998-2:2:2005 [27]. Finally, the floor of the structure was covered with a concrete-reinforced layer with a steel mesh of #15 × 15 × 6 of 15 cm thick.

Eight tests were carried out (Table 1), and a previous (test) shot was performed to verify the operation of the measurement and recording equipment deployed in the area. The explosives used in the tests were black powder, ANFO, AN/AL and PG-2 (like the US C-4). The black powder used has a composition of potassium nitrate (75%), sulfur (10%) and carbon (15%) and is always granulated and graphitized, with particle sizes ranging from 0.1 to 4 mm. ANFO (ammonium nitrate and fuel oil) is the stoichiometric mixture of ammonium nitrate and fuel oil. AN/AL consists of a mixture of ammonium nitrate and aluminum powder. Finally, PG-2 is a military explosive whose composition is mainly RDX embedded in plastic additives.

The IEDs created for these trials were attached to different types of personal vests and in some cases were confined to steel tubes. The design of the explosive charges used in each test was based on the quantities of each of the explosives that can be included in a typical suicide vest configuration: in the case of tests T1 to T6 (black powder, ANFO and AN/AL), explosives inside steel tubes, and in the case of tests T7 and T8 (plastic explosive—PG2), packages directly attached to the inside of the vest. In all vest and tube tests (tests T1 to T6), 0.7 m of 15 g/m detonating cord was used to initiate the main charge. Instead, 3.7 m of detonating cord was used in tests where the explosive was directly stuck to the

vest (without tubes—T7 and T8). Note that the explosive mass in test T8 is higher than the PG2 equivalent as the remaining charges were included.

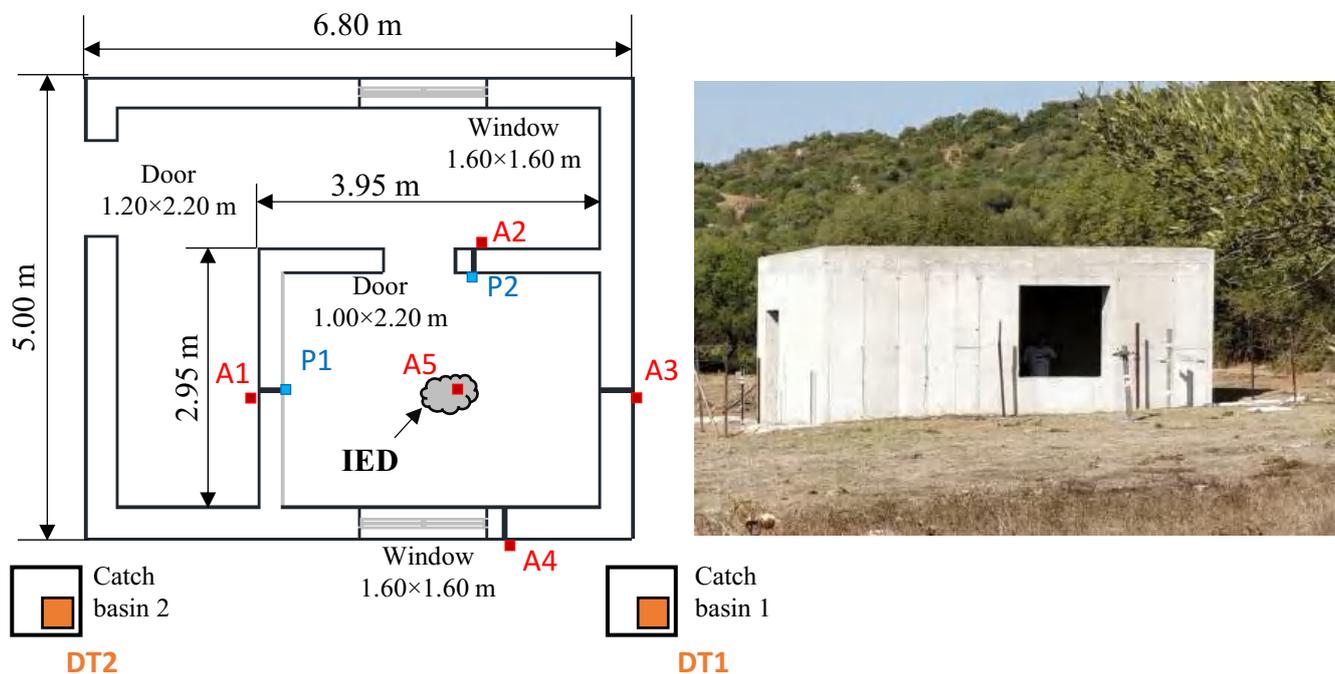


Figure 1. Details of the structure, location of the measuring equipment and photograph of the structure. The letters P refer to pressure sensors, the letters A are accelerometers, and the DTs are Datatrap II recording equipment.

Table 1. Load characteristics during the tests carried out.

Test	Day	Explosive Type	Charge (kg)	PETN (g)	TNT Equivalent Mass (kg)	Confinement
T0	18 September 2018	PG-2	0.10	0	0.14	–
T1	18 September 2018	Black Powder	3.37	10.5	0.79	Steel tubes
T2	18 September 2018	Black Powder	3.27	10.5	0.77	Steel tubes
T3	18 September 2018	ANFO	2.29	10.5	1.48	Steel tubes
T4	19 September 2018	ANFO	2.20	10.5	1.42	Steel tubes
T5	19 September 2018	AN/AL	2.16	10.5	1.88	Steel tubes
T6	19 September 2018	AN/AL	2.25	10.5	1.95	Steel tubes
T7	20 September 2018	PG-2	7.00	55	9.87	Vest
T8	20 September 2018	PG-2	8.20	55	14.21	Vest

The instrumentation of the tests consisted of accelerometers, pressure sensors, recording equipment and a high-speed camera. Figure 1 shows the location of the pressure sensors (P1 and P2), the accelerometers (A1–A5) and the two pieces of recording equipment used (DT1 and DT2).

The two pressure sensors used were 5000 PSI (344.7 MPa) PCB model 102B with ablative protection for the fireball. The sensors were placed with a passing tube on the concrete wall so that the sensing surface was normal to the main direction of the impact. In this way, the first wave registered would be the one reflected by the wall where the sensor is located. These sensors were at a height of 1.51 m and 1.55 m in the case of P1 and P2, respectively. Piezoelectric shock PCB accelerometers located on the opposite side of the wall from the explosive charge of 5000 or 10,000 g measurement limit were used (Table 2). In the accelerometer position called A1, one sensor was used for tests T0 to T4 and a different one for the last three tests (T5 to T7), due to the breakage of the sensor during the T4 test.

No measuring equipment was used in the last test (T8) for fear of complete destruction of the structure or, at least, of compromising its structural stability. Note that the sensor located at position A5 was placed on the roof of the structure on the outside of the structure (Figure 1). Two Datatrap II recorders from MREL were used for data acquisition. This system has up to eight recording channels, with a sampling rate in each channel of 10 MHz with a resolution of 14 bits. It is a portable and very robust piece of equipment prepared to work outdoors, in dust, rain and a wide range of temperatures. Figure 1 shows the location of the data acquisition equipment, inside interconnected chambers (catch basin). Signal conditioners PCB 480E09 were used, necessary to feed and condition both the pressure sensors and the accelerometers. See Figure 2 for more details of the measuring equipment and positions.

Table 2. Characteristics of the accelerometers used and the height at which they were positioned.

#Sensor	Model	Measurement Range (g)	Test	Height (m)
A1	350C23	±10,000	T0–T4	1.375
A1	350C04	±5000	T5–T7	1.375
A2	350C04	±5000	T0–T7	1.370
A3	350C23	±10,000	T0–T7	1.395
A4	350B04	±5000	T0–T7	1.370
A5	350B04	±5000	T0–T7	3.300

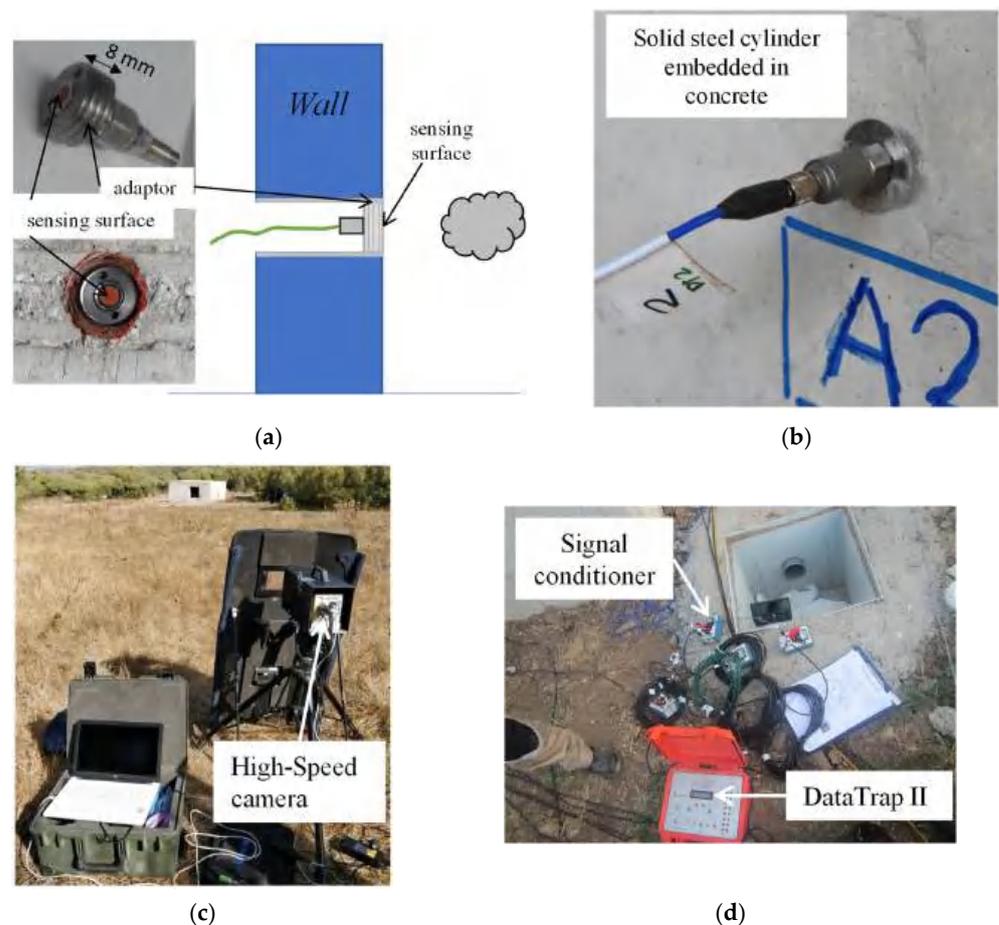


Figure 2. (a) Pressure sensors (PCB 102B), details of the P1. (b) Accelerometer in position A2. (c) Location of the high-speed camera in one of the tests. (d) Signal conditioners and DataTrap II recording equipment.

Finally, the high-speed camera (CAV) used was a Photron Fastcam SA3-120k, adapted for explosion testing with a steel case. It reaches a recording speed of 5000 images per second for a resolution of 512×512 pixels, reaching up to 120,000 fps for a resolution of 128×16 pixels.

3. Numerical Model

The 3D numerical models were made using the LS-DYNA Version 971-R11 software [28], which is based on explicit numerical methods that are suitable for solving problems associated with large deformations subjected to blasting. The destructive effect of these kinds of blast tests, along with the fast structures' reaction and short duration of the explosive event, makes the detailed study of these events very complex.

3.1. Finite Element Model

This model was made of two main critical parts: concrete and steel rebar. In addition, the ground was also introduced into the model but only for visualization purposes (Figure 3). The functionality of "Constrained Lagrange in Solid" was used for the correct operation of both materials as a single assembly. This option can be used as the interaction between parts (steel and concrete) can be presumed to be ideal as the event is almost instantaneous [29,30]. Moreover, the structure was fixed into the ground by using the single point constraint (SPC), canceling displacements and rotations in all directions of space.



Figure 3. Details of the complete 3D model made with LS-DYNA and details of the steel armor.

The concrete was defined with 3D Lagrangian solid elements with reduced integration to decrease the computational time. The element size used for the concrete was 20 mm, based on previous studies of the concrete blasting response under similar conditions and charges [30–32]. The reinforcement was modeled using beam-type elements with a size of 50 mm in length. The number of solids elements was 5,150,187 while the number of beams was 64,266.

To solve the model, LS-DYNA offers two parallel programming methods: symmetric multi-processing (SMP) and massively parallel processing (MPP). SMP runs on a computer with multiple identical cores with the cores and memory connected via a shared data bus, being scalable up to 8 CPUs. MPP uses various separate CPUs running in parallel, each with its own memory to execute a single analysis, performing a domain decomposition of the problem and then distributing the sub-domains to different cores. This solver is scalable over a wide range of CPUs. Although the MPP method allows a reduction in computation time, the size of the model, the lack of symmetries and the complexity of the problem to be solved are considerable. The simulation time was lengthened to 2.5 s, which in SMP resulted in 306 h and 53 min, while in MPP, this time can be reduced to 151 h and 25 min. The computer used for these simulations has two Intel XEON E5-2630 v4 processors at 2.20 GHz (10 cores each, 2 threads per core), with 64 GB of RAM and a Windows 10 operating system.

Moreover, dynamic relaxation (DR) was included in the model. DR is the recommended way to preload a model before the application of dynamic loads in the subsequent transient analysis. This technique makes it possible to achieve a steady-state preload condition free of dynamic oscillations (or nearly free). It is important in cases like this work to apply gravity before the transient analysis (detonation) to avoid an unstressed state at the beginning of the analysis. If the gravitational load is suddenly applied, dynamic

oscillations could be enhanced that would invalidate the utility of any calculation. The application of gravity is performed by the Load Body command (see [28] for more details). In this command, a first curve is defined for the quasi-static analysis (dynamic relaxation) of gravity. For this purpose, a curve is created where the acceleration rises linearly from zero to the constant value (gravity) for a short period of time and then remains constant. Then, a second curve of constant value (gravity) is created which will be used for the rest of the simulation time. The Control Dynamic Relaxation card is also used but with the LS-DYNA default values.

Acceleration in LS-DYNA can be measured with the use of sensors at certain coordinates [28]. The processing of these data is sometimes complex and does not usually work well when the blast processing is performed with tabulated values such as the Load_Blast_Enhanced command. Another alternative is using the *Database_History_Node command to explicitly track the history of features of that node (i.e., acceleration, velocity, displacement, etc.). A shorter time interval between acceleration data in those nodes (i.e., 1×10^{-6} s) than the normal one determined between drawings (0.01 s on the D3Plot card) can be used which improves performance, computation time and hard disk space occupied.

3.2. Blast Implementation

Explosive charge implementation can be handled from two different approaches: using the parameters of the explosive material and its equation of state [17]; or by using a TNT equivalent for the load and its implementation with the load blast enhanced (LBE) function. This last option is usually simpler to implement, as well as computationally faster, producing very good results [33–35].

As for the application of the load, the LBE was used here, which is the way LS-DYNA introduces the CONWEP [36]. This can be used assuming that the steel tubes would have a potentially lethal effect on people but are quite harmless to the structure. With the LBE instructions, the necessary input parameters to calculate and apply the generated pressure (incident and reflected) on the concrete elements are the type of shock wave, the equivalent mass of TNT, the coordinates of the load center and the concrete sides where the pressure wave will be applied. The software applies the pressures following Friedlander's equation to calculate the pressure curve, including the negative phase. It should be noted that with this methodology only the pressure peak set by the measured signal is reproduced, and it is not possible to reproduce reflections of the wave or shrapnel produced during the explosive detonation inside the tubes. In addition, the pressures were only recorded in the room where the IEDs were located. This makes that the pressures were only applied with LBE on all faces of this room (including the ceiling) and not outside of it such as the corridor.

3.3. Materials

LS-DYNA offers more than 25 models that can be used to describe the concrete, some require many input parameters while others work with reduced data, but not all of them perform well under blasting events [37–39].

In this research, the continuous surface cap model (CSCM) concrete was used to describe the concrete behavior. The automatic generation of parameters was based on introducing the values of the compressive strength, the aggregate maximum size (maximum) and the density. The model is plasticity based with the implementation of shear failure surfaces corresponding to the elastic limit, residual strength and failure. This model works based on an isotropic elastic behavior before cracking to move to a plastic behavior limited by the failure surfaces. This model implements an internal calculation of the damage that allows the erosion of the elements when they reach 99% of the damage limit [28,40] and the maximum principal strain in the element exceeds a value defined by the user, known as ERODE. The default value of 1.05 was used for the ERODE parameter [40]. The CSCM includes a dynamic increase factor (DIF), governed by specific data from the CEB-FIP

design code using the Duvaut–Lions overstress formulation based on time rather than strain rate [41]. The material properties used in this model are shown in Table 3.

Table 3. Concrete and steel properties used in the numerical modeling.

Property	Concrete	Steel
Density (kg/m ³)	2300	7850
Uniaxial compressive strength (MPa)	40	-
Maximum aggregate size (m)	0.02	-
Young modulus (MPa)	-	2×10^5
Poisson's ratio	-	0.3
Yield stress (MPa)	-	5×10^2
Tangent modulus (MPa)	-	2×10^4

The steel used in the reinforcement was the classic B-500 S, introduced in the model as the material model “Piecewise Linear Plasticity”. In this work, the option of defining the rupture based on the effective plastic deformation was chosen, as opposed to the rupture based on the time step of the numerical model achieved by the convergence of the method. The value entered was equal to 0.075 [30], i.e., when the plastic strain reaches this value, the element is deleted from the calculation. In addition, to define the stress–strain behavior, a bilinear stress–strain curve was applied by using the tangent modulus. Moreover, the strain rate effects were included in the steel model based on the scale yield stress by using the Cowper–Symonds model (C equal to 25.36 s^{-1} and P equal to 2.52) [28,37,42].

4. Results and Discussion

4.1. High-Speed Camera

The images were captured at a speed ranging from 3000 to 5000 fps. Figure 4 shows a sequence of images of the video obtained in the T2 test with black powder in steel tubes. It shows the extension of the fireball that reaches up about two meters outside the cubicle in the vicinity of the window. The powder generated a significant volume of gases that are expelled practically simultaneously through the two openings to the outside of the cubicle: the window directly to the charge and the window at the rear of the image connected to the main room by a door. If the images of the gunpowder test are compared with those recorded in the ANFO test (T4) shown in Figure 5, important differences are observed in terms of the extension of the fireball and the volume of gases generated. The fireball did not reach the outside in the case of the ANFO test, and the volume of gases was clearly lower. Figure 6 shows a sequence of 12 images obtained in a test with AN/AL (T6). The fireball extended considerably more than in the ANFO test due to the aluminum in its composition.

Figure 7 shows a similar sequence of images for test T7 in which a vest without steel tubes was fired with 7 kg of PG2 plastic explosive. The extension of the fireball reaches the facade completely on both the front and rear faces of the cubicle. The first images just after the start of the detonation show a large white glow indicating very high temperatures. The escape of gases that comes at the same time as the fireball takes place through the openings mentioned above (windows) and gas escape can be seen in the upper part of the cubicle, probably due to the displacement of the upper slab.

4.2. Pressure Signals

In some cases, the registration of the different pressure–time signals presented an important level of noise that can mask the real signal. In these cases, filtering the signal is necessary to obtain the parameters of the shock wave. Details of the procedure followed can be found in the work published by Chiquito et al., 2019 [43]. In the first test with ANFO (T1) there was a problem with the trigger of one of the DT2 recording systems, and therefore, no data were obtained from the P2 sensor. The P1 sensor in that test suffered the

impact of fragments, and therefore, it did not record anything either. In the following tests (T2 onwards), no more sensors were placed.



Figure 4. Sequence of images obtained with HSC in trial T2: black powder. Time in milliseconds; the reference time is the first frame of the video where the detonation is observed.

Pressure measurements were recorded with some reliability in tests T0 to T2 (see Table 4). The simulation values are compared with the average (field) values when there is more than one signal. See Figure 8 as an example of the pressure application. The sequence shows how the wave first reaches the inner wall of the door (area closest to the IED) and then the ceiling. It then reaches the rest of the surfaces and expands in a very similar manner. The expansion pattern is the classic one in a shock wave of this type from the center to the sides, ending at the corners. Once the pressure front passes a point, the pressure decays to the initial pressure. As shown in Table 4, the values simulated with the LBE card are quite similar to those obtained in the field, with relatively low errors given the nature of the phenomenon. As mentioned in Section 3.2, LBE only reproduces the first pressure peak, and this is what is compared in Table 4. This is obviously an important limitation of the simulation, but the other available techniques (i.e., SPH, ALE or PBM), which might be able to reproduce the behavior more realistically, become unfeasible due to the resources required (meshing, number of elements, computational time, etc.). The results show differences of about 11%. This shows that the simulation is relatively reliable and that the TNT equivalent used in the description of the explosives was quite accurate.



Figure 5. Sequence of images obtained with HSC in trial T4: ANFO. Time in milliseconds; the reference time is the first frame of the video where the detonation is observed.

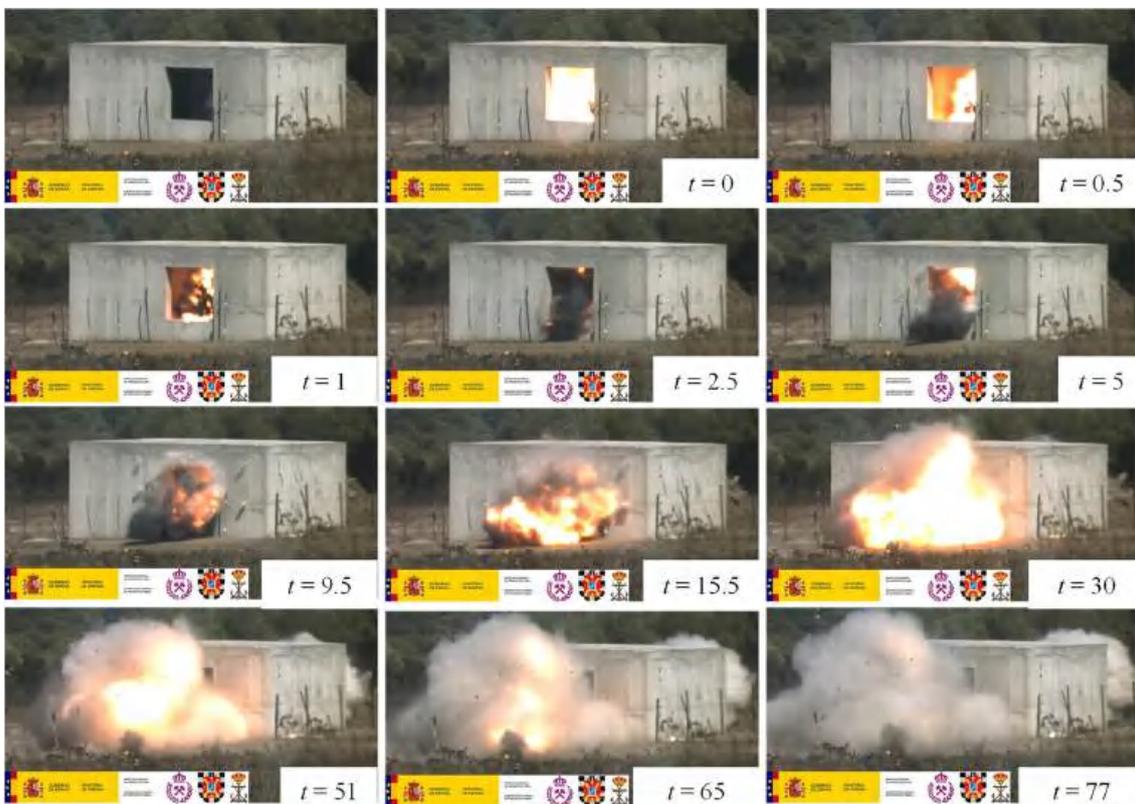


Figure 6. Sequence of images obtained with HSC in trial T6: AN/AL. Time in milliseconds; the reference time is the first frame of the video where the detonation is observed.



Figure 7. Sequence of images obtained with HSC in trial T7: PG2. Time in milliseconds; the reference time is the first frame of the video where the detonation is observed.

Table 4. Pressure sensor results. Friedlander adjustment.

Test	Explosive	Sensor	P_r (kPa)	P_r (LS-DYNA) (kPa)	Relative Dif. (%)
T0	PG-2	P1	88.72	99.67	−12.34
T0	PG-2	P2	129.80	136.02	−4.79
T1	Black Powder	P1	193.11	181.21	6.71
T2	Black Powder	P1	195.38	181.21	6.71
T1	Black Powder	P2	372.27	309.64	17.93
T2	Black Powder	P2	382.36	309.64	17.93

All logs show multiple reflections on the various walls, but no sustained gas pressure is observed due to the large vent provided by the window and access door, as shown in Figure 4. Figure 9 shows how the simulation with LBE only reproduces the first peak of the signal recorded by the sensors. It can also be seen how well the model is able to reproduce the shape (duration and impulse) of the recorded shock wave. Moreover, the pressures recorded in sensor P1 are much lower, almost half, than those of sensor P2. This may be due to the orientation of the explosive device focused more directly toward P2, not having a direct “view” of the sensor located at P1. However, the different reflections to which the sensor is subjected are greater in P1, which makes sense, since it is farther away from the large vents that are the doors and windows.

4.3. Acceleration Signals

Peak acceleration values comparing all the simulations performed and the measurements with the different sensors (A1–A4, Figure 1 and Table 2) can be found in Table 5. A filter was applied to the acceleration signals to eliminate noise and electrical peaks. The filtering applied was the Butterworth low-pass type of order 4; in addition, if the signal

presented an offset, it was also corrected. Sensors that were not measured in the field are not reflected in Table 5, which is used to show the differences between measured and simulated values. Measurement failures were sometimes due to poor sensor coupling, failures in the trigger system or measurements that did not make sense because of the extreme (high/low) values obtained.

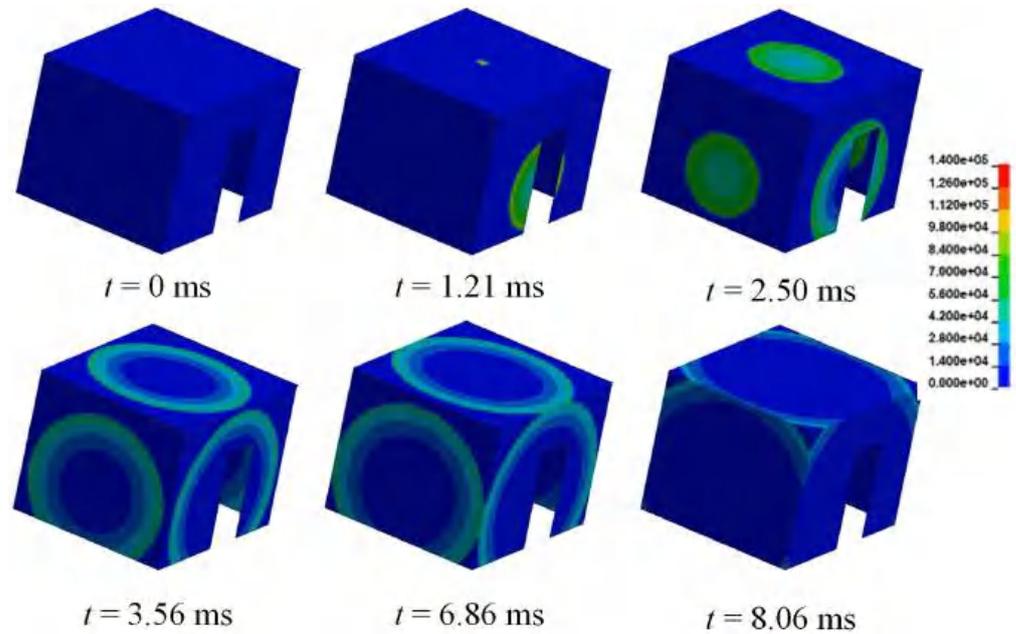


Figure 8. Pressure isovalues (view from the inner door) on the interior faces of the room where the detonations were carried out for the T0 test. The color scale is the pressure values in Pa.

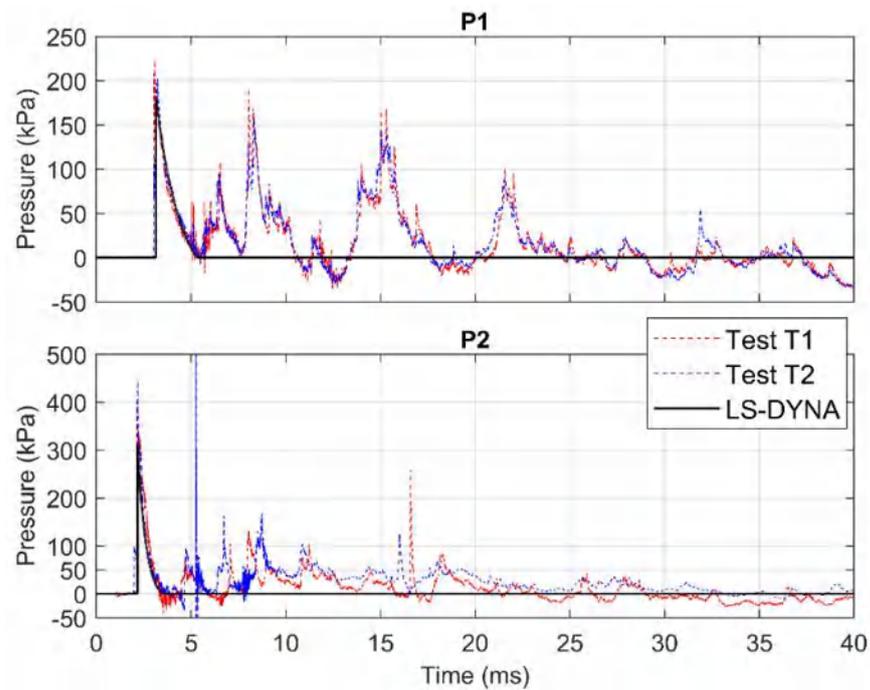


Figure 9. Pressure signals (P1 and P2) for tests T1 and T2: black powder, and the simulated signal with LS-DYNA.

Table 5. Peak acceleration values for different sensors and trials. BP means black powder.

Test	Explosive	#Sensor	Measured Acceleration (g)	Model Acceleration (g)	Difference (%)
T0	PG2	A2	44.4	26.3	40.77
		A3	77.1	57.8	25.03
		A4	29.6	30.2	−1.94
T1	BP	A1	207.2	152.8	26.25
		A3	699.3	416.5	40.44
		A4	69.2	79.9	−15.49
T2	BP	A1	105.2	75.4	28.30
		A3	430.3	403.6	6.21
		A5	298.8	354.8	−18.76
T3	ANFO	A1	1048.9	852.4	18.73
T4	ANFO	A1	1113.6	928.3	16.64
		A3	1048.2	809.9	22.74
		A4	259.8	349.4	−34.50
		A5	1646.4	1056.0	35.86
T5	AN/AL	A1	998.9	1115.2	−11.64
		A3	3786.5	4192.0	−10.71
		A4	4685.5	4896.0	−4.49
		A5	2683.10	1691.5	36.96
T6	AN/AL	A1	903.4	1126.5	−28.68
		A3	5600.6	4305.0	23.13
		A4	4828.6	5094.0	−5.50
		A5	1996.25	1518.4	23.94
T7	PG2	A1	902.7	652.0	27.77
		A3	1507.90	1521.8	3.12
		A4	5171.8	3464.4	33.01
		A5	1342.3	1505.0	−12.12

As can be seen in Table 5 for sensors A1 to A4, the acceleration data obtained are quite large for the ANFO, AN/AL and PG2 tests, with values between 1000 and 5000 g, while in the black powder tests, the acceleration values are around 400 g. This clearly indicates that accelerations increase with the use of more powerful charges, as expected. Given the non-linear nature of the phenomenon, as well as the limitations of the simulation itself, the differences between the model and the real data are quite good with an average absolute value of 20%. Therefore, the model can reproduce with some reliability the acceleration peaks. The highest difference is found in the only useful measurement of sensor A2 and sensor A3 in test T1. On the other hand, the lowest value is also found at sensor A3 in test T7, followed by sensor A4 in tests T5 and T6. The lowest values are generally found in sensor A4, but it is also the sensor that shows the largest deviations from the mean value. The highest mean values are found in sensor A1 but with the smallest deviations. Considering the results obtained, it can be deduced that the differences between measured and simulated values on sensor A1 are the most important. This may be since the behavior of this interior wall is not well reproduced, being more rigid in the model than it should be. The opposite is true for the outer wall where sensor A4 is located, where the model reproduces quite faithfully the behavior of sensor A4.

However, sensor A5, located on the roof of the structure (see Figure 1), shows somewhat different results (see Table 5 and Figure 10). In general, the model can reproduce the results measured in the field with errors averaging (and in absolute value) around 25%. Figure 10 shows how the accelerations look similar in all cases. In the T2 test, the peak accelerations were not recorded with the first arrival of the wave as in the other cases, as always happens in the modeling. Therefore, in this test, the acceleration peaks between LS-DYNA and the tests do not coincide in time, although they do coincide in peak values.

Despite the increase in the explosive load, the maximum accelerations recorded on the roof slightly decreased, contrary to the rest of the sensors. This was especially noticeable in the case of test T7, with 7 kg of PG2. This may be a consequence of the decrease in the stiffness of the structure due to the accumulated damage after performing the tests consecutively without intermediate reinforcement or support actions on the structure.

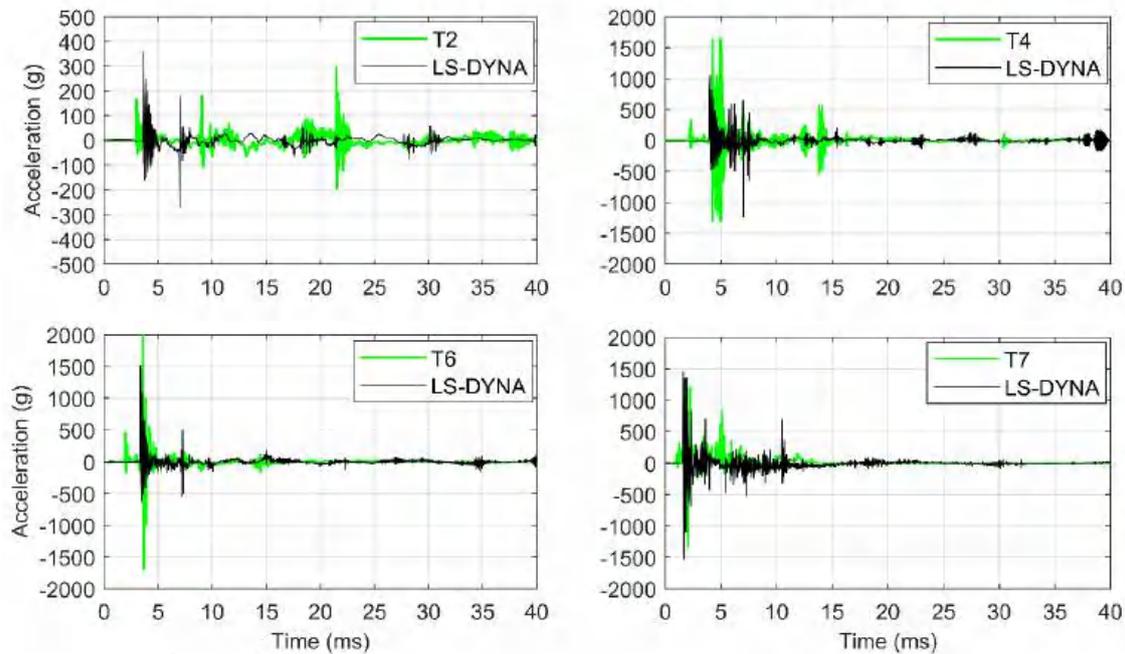


Figure 10. Accelerations from tests T2, T4, T6 and T7, in position A5 (see Figure 1) and their comparison with LS-DYNA.

The extra stiffness provided by the double reinforcement used in the roof slab may have lost its effect after so many tests, but it was probably the one that prevented an earlier collapse of the structure. To reinforce this idea, in the model (Figure 11), it is easy to observe how stress accumulations occur in the window and in the joints of the roof slab with the walls, and therefore, these are the areas that withstood the most stresses throughout the tests. The sequence of images shows how in two seconds the stresses produced in the structure have already stabilized. It can be seen in Figure 8 how the effect of the detonation is only 4 ms when the sequence of images in Figure 11 is every second. It can also be observed that after the T7 test, the stress state of the structure is considerably higher than before.

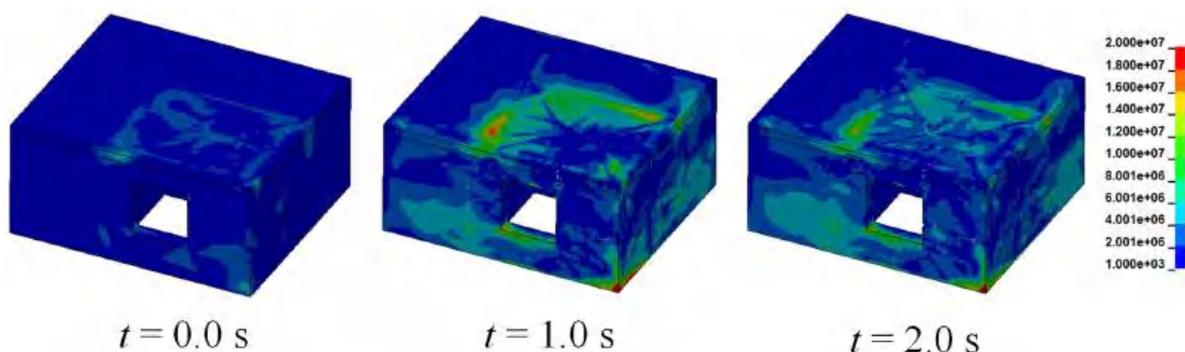


Figure 11. Effective stress (scale in Pa) of trial T7. The first image with time zero corresponds to the stress state when the T7 test load is detonated in the model.

4.4. Final Test E8

In this test, the explosive charge was placed on the ground in the corner between the walls of the access door and where the P1 sensor was located (see Figure 1). Figure 12 shows the result of the structure after the last test (T8). It can be seen how the structure is destroyed on the window wall and on the adjoining one on the DT1 side (see Figure 1). The gases try to exit through that area (as seen in the previous test T7, Figure 7), projecting most of the shock wave energy on this side of the already weakened structure. Consequently, the walls and part of the roof collapse, leaving the reinforcement exposed. In the model, something similar happens: although the roof seems to be somewhat more damaged than the real structure, the side wall shows significant damage as it happened in the test.

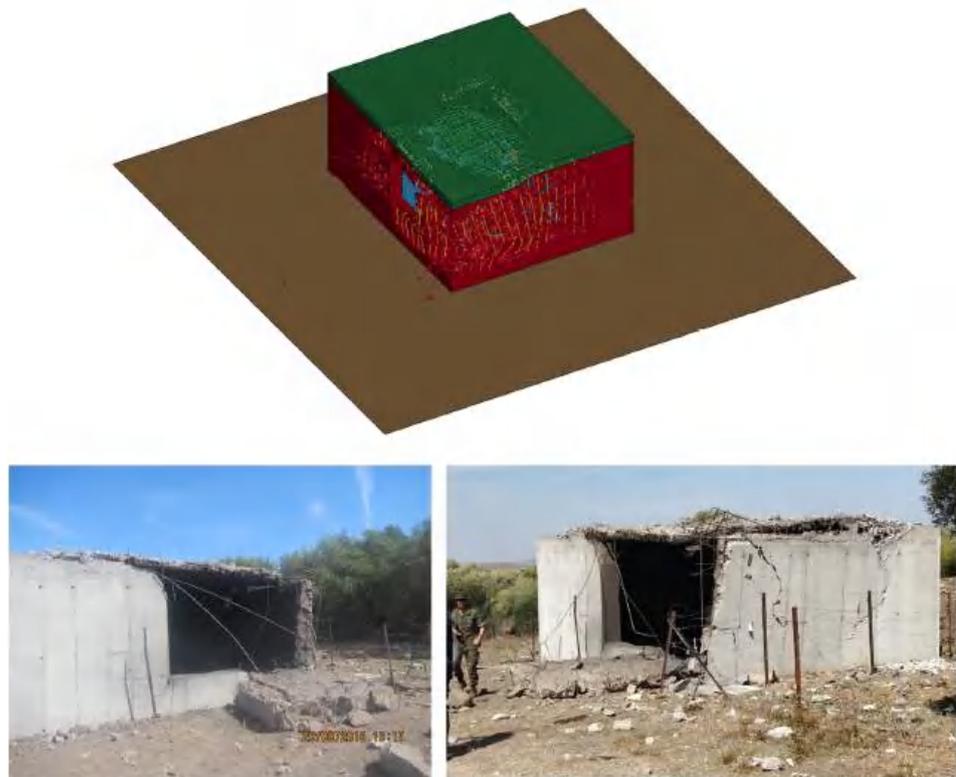


Figure 12. Final test (T8) results: numerical model and photographs.

4.5. Effects of the IED Type

In general, the type of IED used can have a major impact on human casualties related to the air blast wave or shrapnel impact. However, it is not very common that they can affect entire structures or parts of them, especially in cases of small charges. This work, although with repeated explosive charges on a structure accumulating some damage, can serve as a small case study.

The results show that the IEDs used in the tests (T1–T6), where explosives that are not too powerful and with low charge are confined in tubes, can produce high accelerations. These pressures are surely produced by the confinement of the explosive in a steel tube since its attempted detonation in air would probably produce milder effects. This fact results in greater damage to the structure given the high accelerations. See Figure 13 for details of the interior parts of the structure affected by successive detonations. It can be seen how the black powder tests hardly affect the structure or the concrete (Figure 13A). In the case of ANFO, whose charge is more powerful, some spalling of the concrete near the interior door can be observed (Figure 13B). The same happens in the case of AN/AL, leaving even the first reinforcements of the structure visible, indicating that the erosion has already been significant (Figure 13C). The case of the plastic explosive (T7–T8) is slightly

different because it is a very powerful explosive that does not need confinement to improve its performance. It is known that these explosives are used by security forces and corps for the destruction or demolition of parts of a structure or even the full structure. In this case, it is no different, producing significant damage to the structure as seen in the cracks generated in the interior wall (Figure 13D), which although in principle does not have to compromise the stability of the structure, does leave it very damaged. As discussed above, in the last test (T8), the load was appreciably higher, and this caused the roof to decouple from the walls (Figure 13E), in addition to the obvious collapse occurring in the area of the exterior window.



Figure 13. Different effects of IEDs on the interior structure: (A) after the two black powder tests (T1–T2); (B) after the ANFO tests (T3–T4); (C) after the AN/AL tests (T5–T6); (D) after the first PG2 test (T7); (E) after the second PG2 test (T8).

5. Conclusions

A total of eight tests were carried out with different types of IEDs on the same reinforced concrete structure, simulating a scenario where charges are detonated at the entrance of the State security forces and bodies. Some conclusions can be extracted as follows:

- The high-speed camera images allow us to see the correct detonation of the explosive, while the pressure log allows us to validate the model input data.
- The acceleration recorded at the roof of the structure decreases as more tests are performed due to the loss of stiffness of the structure.
- IEDs of relatively low power (with homemade explosives or low-TNT equivalent), although they cause significant accelerations in the structure, do not compromise its structural stability, while more powerful IEDs (plastic explosives), although with similar accelerations, do put the structural stability of the building at risk.
- A solid element model using LBE offers, even in complex cases such as this one, a reasonable reproduction of the behavior of a structure reducing testing costs by being able to reproduce with some certainty different scenarios.

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preparation, R.C., M.C. and E.C.; supervision, R.C., L.M.L. and J.I.Y.; project administration, R.C.; funding acquisition, R.C., A.P.S. and J.I.Y. All authors have read and agreed to the published version of the manuscript.

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Article

Investigation of Behavior of Masonry Walls Constructed with Autoclaved Aerated Concrete Blocks under Blast Loading

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Abstract: Autoclaved aerated concrete (AAC) blocks have widespread popularity in the construction industry. In addition to lightness, these materials have other advantages, including fire resistance, low acoustic and thermal conductivity, ease of cutting and grooving, and simple transportation. Since the behavior of AAC under severe dynamic loading conditions such as blast loads has not been adequately studied in the literature, in the current paper, the behavior of masonry walls constructed with AAC blocks was evaluated under blast loading. In this study, after performing experimental testing on materials and obtaining their compressive, tensile, and shear strength values, the finite element (FE) models of AAC-based masonry walls were created in the ABAQUS/Explicit nonlinear platform. Three different wall thicknesses of 15, 20, and 25 cm were simulated, and the models were analyzed under a lateral explosion caused by 5 and 7 kg of TNT at the stand-off distances of 2, 5, and 10 m from the wall face. The stress distributions, displacement responses, adsorbed energy, and crack propagation pattern were investigated in each case. The results showed the inappropriate behavior of these materials against explosion loads, especially at shorter distances and on walls with less thickness. The outcome gives valuable information to prioritize these walls for possible blast strengthening.

Keywords: AAC block; blast loads; masonry wall; finite element; strengthening; ABAQUS

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1. Introduction

The existing challenges in the construction industry mainly include increasing the speed of the construction process, increasing the useful lifetime of buildings, retrofitting, cost reduction, reducing thermal and acoustic conductivity, reducing the weight of the building, and environmental issues. Efforts to meet these needs led to the invention of autoclaved aerated concrete (AAC) products. AAC is a relatively modern material with a favorable strength-to-density ratio, thermal insulation properties, and other advantages such as lightness, fire resistance, and ease of cutting and application [1–3]. Today, AAC is widely used in the United States, Europe, and many other countries [4]. These materials are considered environmentally friendly construction materials [5]. AAC products are commonly made of cement, water, lime, silica-based materials (silica sand, ash, or silica fume), porosity-generating materials (aluminum powder), and additives [6].

During their service life, buildings may be exposed to several dynamic load conditions, such as earthquakes, explosions, impacts, and wind loads. Explosions caused by terrorist attacks or accidental incidents in urban areas can cause severe human and financial losses. Blast loading experiments and strengthening the structures to reduce the damage caused by explosions are among the most critical topics for researchers and structural engineers [7–10]. In this regard, multiple studies have modeled the effect of blasts on various building materials and structures.

Previous studies on the behavior of structural and nonstructural components made of AAC materials were limited to static loading conditions [11]. A small number of studies

have considered the seismic or impulse load conditions. Yankelevsky and Avnon [12] tested AAC exterior walls under impact loading and evaluated the damages patterns [12]. Tanner et al. [13,14] conducted extensive studies on shear walls made of AAC; many of the requirements of ACI regulations on AAC materials [15] are derived from their research. Uddin et al. [16] introduced a new type of sandwich panel using AAC and FRP composite materials. The behavior of the panel was evaluated under low-speed impulse tests [16]. According to the results, the failure patterns and energy absorption of AAC–FRP panels were improved compared to simple AAC units. Moreover, Tomažević and Gams [17] performed compressive, tensile, and shear strength tests on masonry walls made of AAC blocks. They also tested several reduced-scale structures with AAC walls on the shaking table [17]. In a study conducted by Bayat et al. [18], the behavior of AAC blocks under severe impulse loads was analyzed. They also investigated the ballistic limit velocity of AAC targets under the influence of rigid projectiles. The results showed that the introduced analytical model was in good agreement with the experimental results [18].

TM 5-855-1 [19] can be considered as one of the first instructions provided for nonatomic explosion-resistant structures. In addition, TM 5-1300 [20] instruction was widely used to design explosion-proof structures; TM 5-1300 was more comprehensive than TM 5-855-1 and based on many subsequent theoretical studies. Finally, UFC 3-340-02 [21] guidelines, as an updated version of TM 5-1300, were developed by the US Department of Defense (DOD) and have been widely used as the primary basis for design and research works in this area.

Historically, some studies were conducted on the behavior of masonry building materials under blast loading. In a series of studies, Hao and Wu [22] and Wu and Hao [23] investigated the effect of infilled walls on RC building behavior under explosion loading. Using explicit finite element modeling in LS-DYNA hydrocode, Wei and Stewart [24] reported that increasing the masonry wall thickness reduces the explosion damage to the buildings. In an experimental study, Ahmad et al. [25] tested a cantilever masonry wall consisting of clay bricks under blast loads. Pandey and Bisht [26] and Pereira et al. [27] investigated the dynamic performance of the brick masonry walls against blast loading. Shi et al. [28] studied local and global damage to a reinforced masonry wall under the close-in explosion scenario. According to the results, instead of bending or shear failure of the wall, the close-in explosion caused local damage by punching [28]. Parisi et al. [29] reported the explosive resistance of a stone wall. Keys and Clubley [30] and Badshah et al. [31] investigated failure patterns of masonry walls through real blast loading tests. Zeng et al. [32] applied 3D finite element models to simulate the out-of-plane behavior of un-reinforced masonry walls constructed with bricks under static and dynamic loadings.

According to ASCE 51-11 [33], the fragmentation of building elements and thrown fireballs have the most dangerous impact in an explosion event. Strengthening methods to prevent the destructive effects of explosions have been an area of interest for some researchers. The most common explosives strengthening techniques in masonry walls include the use of fiber-reinforced polymers (FRP), polyurea, and polyurethane coatings, using steel sheets, aluminum foam, and engineered cementitious composites [34–41].

Previous studies on the behavior of AAC materials under blast loads are limited. In particular, studies investigating the effect of blast loads on structural elements made with AAC are scarce. Xu et al. [42] numerically modeled infilled walls constructed with AAC blocks under gas explosion in LS-DYNA. Li et al. [43] investigated the performance of an autoclaved masonry wall under methane explosion. This study was performed using field tests and numerical simulations [43]. In an experimental study, Wang et al. [44] evaluated retrofitted masonry walls consisting of clay bricks and autoclaved aerated concrete blocks under explosion. They used polyurea layers to increase the explosion resistance of the considered walls [44]. In addition, Liu et al. [45] studied the effect of high strain loading conditions on the properties of AAC materials. Sovják et al. [46] determined the ballistic resistance of AAC against projectile penetration.

AAC lightweight concrete blocks are considered among the first alternatives in construction, especially in the reconstruction of urban areas damaged in the Middle East wars. There is a knowledge gap in the assessment methods and priorities of masonry components [47]. This matter is even more notable in the building constructions with AAC units. Since very few studies have been performed in this field so far, the material properties of autoclaved aerated concrete are not comprehensively known, especially under severe loading conditions. Therefore, investigation of the behavior of building elements constructed with AAC units under blast loading seems necessary. Understanding the behavior of these blocks under explosion and providing solutions to increase their explosive capacity can be an interesting topic for researchers in this field.

The present study aimed to identify, investigate, and analyze the behavior of masonry walls made of AAC lightweight concrete units under the effect of blast loading. The crack growth, displacements, stress distribution, and energy absorption of different models of this type of wall were investigated using FE modeling in the ABAQUS/Explicit package. The main goal of this study was to implement an effective FE procedure in the analysis of masonry models under lateral blast pressure considering different wall thicknesses, since the autoclaved aerated concrete units can be produced with various dimensions.

2. Materials and Methods

2.1. Blast Loading

When an explosion occurs in the open air, a shock wave containing very dense air is propagated radially outwards from the source center at supersonic speeds [48]. Figure 1 shows the schematic time variations of blast pressure. The time history of the pressure is mainly divided into positive and negative phases. The positive phase begins from the moment the blast wave reaches the structure (point B in Figure 1). At this point, the pressure suddenly reaches its highest value and then gradually decreases to the atmospheric pressure during the positive phase. Then, as it decreases relative to atmospheric pressure, it creates a negative or suction state (point C in Figure 1). The magnitude of the overpressure in the positive phase is much higher than that in the negative phase, and except for lightweight structures, the reverse pressure effects in the negative phase zone are assumed to be negligible [20]. Points A and D in Figure 1 represent the normal atmospheric pressure.

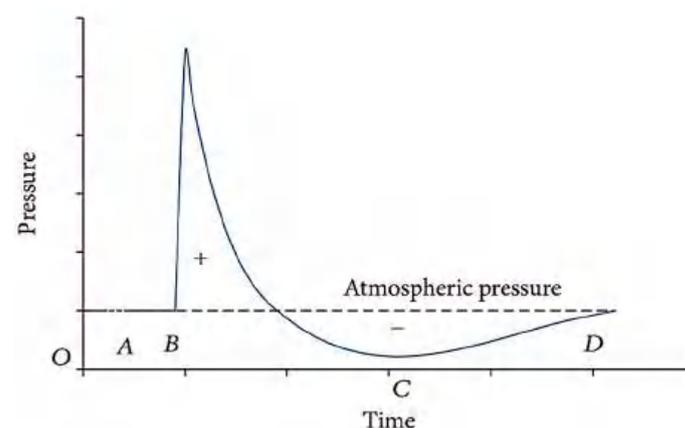


Figure 1. Pressure time history for free explosion [9].

In general, the distance from the source of the explosion (stand-off distance), R , and the explosive charge weight, W , are two crucial factors in determining the specifications of the blast wave. For two different weights of the explosives, if the ratio of the distances from the structure is equal to the ratio of one-third of the power of the charge weight, then

the resulting pressure is identical in both cases. This is known as the Hopkinson–Cranz (cube root) scaling law and is expressed according to Equation (1) [49].

$$\frac{R_1}{R_2} = \left(\frac{W_1}{W_2} \right)^{1/3}, \quad (1)$$

where R is the distance from the center of the explosives, and W is the charge weight for two different cases. The scaled distance (Z) is a basis for evaluating the explosion intensity variations (Equation (2)). Scaled distance is one of the most important characteristics that affects all explosion wave parameters.

$$Z = \frac{R}{W^{1/3}} \quad (2)$$

In general, there are three types of explosions, based on the measured distance: contact, close-in, and far-field explosions [20]. In the contact state, blast load usually causes a non-uniform pressure distribution on the face of the structure, and the intensified local pressure causes cracks and ruptures. In the close-in state, blast waves are generated in a high impulse area on the face of the structure. A far-field blast is a state in which the waves reaching the outside of the building are planar due to the great distance from the structure, and the load distribution can be assumed to be linear or uniform. In this study, the close-in explosion scenario was considered for all the models.

Various experimental relations have been presented in different studies to calculate the explosion wave parameters using the parameter Z [20,21,50–53]. The Conwep module was developed by the US Army Ground Forces Strategic Research Institute following the requirements of TM 5-855-1 Code [54]. The primary purpose of this software is to estimate and apply explosion and impulse loads on the external surface of structures. In this study, the capabilities of this sub-program in ABAQUS were used to calculate the blast load specifications.

2.2. AAC and Grout Materials

Autoclaved masonry walls are defined in MSJC Code [55] as masonry AAC units placed and connected with suitable mortar or adhesives. These walls may be made with or without reinforcement. Equations (3)–(7) estimate the AAC material specifications [55].

$$E = 6500(f'_{AAC})^{0.6} \quad (\text{Mpa}), \quad (3)$$

$$f_{t \text{ AAC}} = 0.2\sqrt{f'_{AAC}} \quad (\text{Mpa}), \quad (4)$$

$$f_v = 0.15\sqrt{f'_{AAC}} \quad (\text{Mpa}), \quad (5)$$

$$E_v = 0.4 E, \quad (6)$$

$$E_g = 500 f'_g, \quad (7)$$

where f'_{AAC} is the compressive strength, E is the modulus of elasticity, $f_{t \text{ AAC}}$ is the tensile strength, f_v is the direct shear strength, E_v is the shear modulus of AAC materials, f'_g is the compressive strength of adhesive or grout, and E_g is the elastic modulus of adhesive or grout. In this study, the compressive strength of the considered AAC materials and the compressive and tensile strengths of mortar (adhesive) were determined experimentally in the laboratory. Other properties needed for finite element modeling of the materials were estimated using the equations proposed in MSJC. Here, in defining the constitutive

behavior of AAC material, the equation proposed by Entezari and Esmaili [56] was used, which is given in Equations (8) and (9).

$$f_c = f'_c \left[\frac{n^{pq} \left(\frac{\epsilon_c}{\epsilon_0}\right)}{\left(\frac{\epsilon_c}{\epsilon_0}\right)^{npq} + n^{pq-1}} \right], \tag{8}$$

$$q = 1.25 + 0.009f'_c, \tag{9}$$

For the ascending region, p and q are constants assumed to be 3 and 1, respectively. The quantity of n^{pq} is determined based on the properties of concrete, such as compressive strength, modulus of elasticity, and strain corresponding to the maximum stress. The values of n and p for the descending region are the same as those for the ascending part, and the value of q is determined using the return point of the descending curve. According to the experiments, the stress–strain curve obtained for this study is plotted in Figure 2.

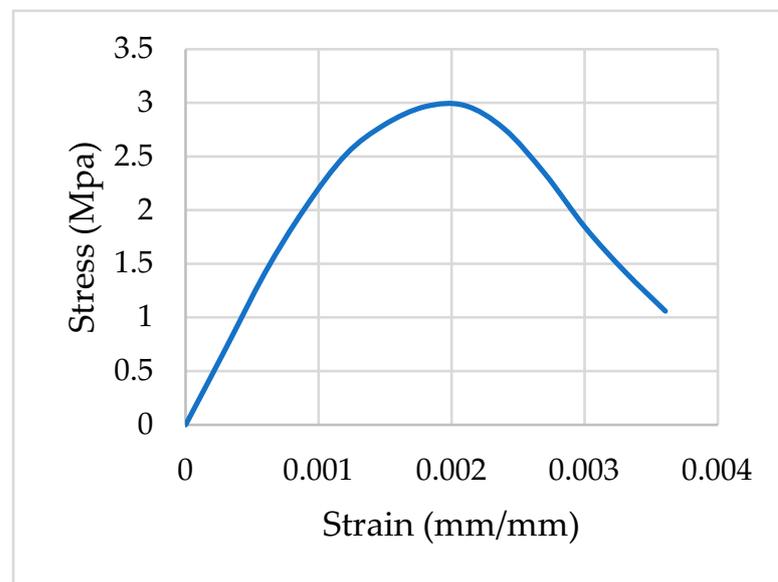


Figure 2. Stress–strain curve of AAC materials with a compressive strength of 3 MPa.

The modulus of elasticity of AAC materials was 1716 MPa, and the Poisson’s ratio was 0.2. To define the nonlinear properties of AAC concrete, the concrete damage plastic (CDP) model was used here, the specifications of which are given in Table 1.

Table 1. Parameters of concrete damage plastic (CDP) model used here.

Parameter	Dilation Angle	Eccentricity	Biaxial-to-Uniaxial Compressive Strength Ratio	Shape Factor Function	Viscoelastic Parameter
Quantity	20	0.1	1.16	0.66	0.001

2.3. Numerical Modeling

ABAQUS/Explicit [57] is a finite element package based on an explicit integration approach used to solve extreme nonlinear systems such as high strain rate loadings. In high-velocity dynamic phenomena such as explosive and impulse loads, which apply an intense load in a very short time, it is practically impossible for the FE solution to converge in the implicit approach. Therefore, in this study, the Explicit solver was used to analyze AAC masonry models under explosion loads. ABAQUS includes an extensive library of continuum three-dimensional solid elements which are suitable for modeling solid objects. In this study, C3D20 and C3D8 were used in modeling masonry components made with

AAC units. Based on a mesh sensitivity analysis, a solid elements' meshing size of up to 15 mm was selected.

In ABAQUS, Conwep subroutine can calculate the blast pressure distribution in various structures [57]. In this study, the Conwep feature was used for blast loading. Thus, by entering the explosive charge weight and the stand-off distance, the program automatically calculates the spatial and temporal distribution of the blast pressure on the interaction surface. In this study, the blast event was defined as an air blast in Conwep subroutine, and one side of the wall was considered the blast wave interaction zone.

2.4. FE modeling of Masonry Walls

In addition to the analytical approaches [58] and discrete-element analysis [59], numerical methods can be successfully applied to modeling and analysis of masonry walls. Generally, there are three main methods for developing the FE model of infilled frame walls [60,61], including detailed micro modeling, simplified micro modeling, and macro modeling (Figure 3).

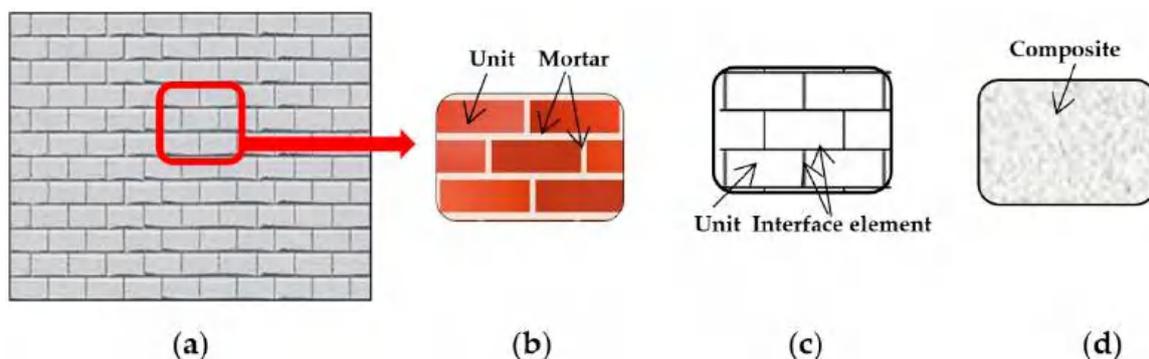


Figure 3. Masonry wall modeling approaches: (a) a real masonry wall; (b) accurate detailed micro modeling; (c) simplified micro modeling; (d) macro modeling.

Micro modeling can be performed in accurate or simplified manners. The accurate detailed modeling provides the most realistic state or representation of a masonry wall composite. In the accurate approach, the construction units and joints of the mortar layers are modeled, and the properties of each material are assigned separately. The requirements related to the aspect ratio of the elements in meshing, the low thickness, and the long mortar joints mean the accurate detailed micro model needs a very fine mesh. For this reason, most complex calculations require considerable time [43,62–64].

In simplified micro modeling, bricks (blocks) and mortar are not modeled separately. The mortar is bonded to homogeneous construction units and added to the unit by a zero-thickness interface element. Mortar joints are added to the intermediate elements representing crack and slip surfaces [32,64]. Using this modeling approach, accuracy is expected to decrease to some extent [65]. In this study, a simplified micro modeling approach was used to prepare masonry wall models with AAC units. Therefore, the mortar was not modeled, and its behavior was considered by adding contact elements between the construction units.

In the macro modeling methodology, the whole infilled wall is modeled as a homogeneous material with equivalent properties regardless of its constituent units. The accuracy of this modeling approach is lower, and the analysis speed is much higher than that of the micro models. It should be noted that the mechanical properties of materials have different values for various conditions, i.e., the arrangement of bricks and horizontal/vertical joints of the mortar in the wall cause the varied stiffness values in different directions [66,67].

2.5. Properties of Interface Elements

The elastic properties of mortar joints are determined by normal stiffness (K_{nn}) and shear stiffness values (K_{tt} and K_{ss}). If the interaction between the two pieces is similar to

that of adhesive, then adhesive elements can be used. In terms of elastic properties, the relationships between stress and vertical and shear strains can be defined as coupled or uncoupled. The stress–strain relationship for uncoupled and coupled states is in the form of Equations (10) and (11), respectively.

$$\begin{Bmatrix} t_n \\ t_s \\ t_t \end{Bmatrix} = \begin{bmatrix} K_{nn} & 0 & 0 \\ 0 & K_{ss} & 0 \\ 0 & 0 & K_{tt} \end{bmatrix} \cdot \begin{Bmatrix} \varepsilon_n \\ \varepsilon_s \\ \varepsilon_t \end{Bmatrix}, \quad (10)$$

$$\begin{Bmatrix} t_n \\ t_s \\ t_t \end{Bmatrix} = \begin{bmatrix} K_{nn} & K_{ns} & K_{nt} \\ K_{ns} & K_{ss} & K_{st} \\ K_{nt} & K_{st} & K_{tt} \end{bmatrix} \cdot \begin{Bmatrix} \varepsilon_n \\ \varepsilon_s \\ \varepsilon_t \end{Bmatrix}, \quad (11)$$

where t_n , t_s , and t_t are the vertical and shear stresses in two directions, the matrix k is the corresponding stiffness, and ε is the vector of strains for the interface plane. The entries of the main diagonal of the stiffness matrix are the normal and shear stiffness in the main directions of the interface. Here, normal (K_n) and shear ($K_s = K_t$) stiffness were used to define the adhesive behavior according to Equations (12) and (13).

$$K_n = \frac{E_u E_m}{h_m (E_u - E_m)}, \quad (12)$$

$$K_s = K_t = \frac{G_u G_m}{h_m (G_u - G_m)}, \quad (13)$$

where h_m , G_m , and E_m are the thickness, shear modulus, and modulus of elasticity of the mortar, and G_u and E_u are the shear modulus and modulus of elasticity of the block. The coefficient of friction of the layer was also defined as 0.7.

2.6. Mechanical Properties of AAC Materials and Grout

The AAC material mixture considered in this study is summarized in Table 2. With the lack of reliable data related to the mechanical properties of AAC materials, experimental tests were performed on AAC block samples prepared from Aranshahr Aran Polymer Concrete Plant (East Azerbaijan, Iran). The requirements of the ASTM C495 [68] code were used to measure the compressive strength of cubic specimens with the dimensions of 10 cm (Figure 4a). The ASTM C109 [69] code was also used here to determine the compressive strength of adhesive materials (Figure 4b). The dimensions of the cube molds in this experiment were 50 mm, and the samples were treated in water for seven days. The tensile strength of the briquette samples was also determined according to the requirements of the ASTM C 307-3 [70] code (Figure 4c). According to the results, the average compressive strength of the AAC block was about 3 MPa. Moreover, the average compressive strength of the mortar (adhesive) was 10 MPa, and its tensile strength was 1.3 MPa.

Table 2. AAC mix design with a density of 500 kg/m³.

Materials	Amounts (kg/m ³)
silica sand	350
lime	100
cement	25
aluminum powder	0.5
water	330

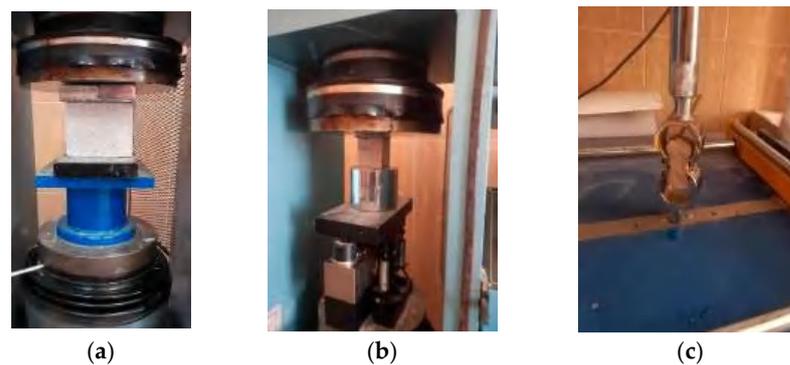


Figure 4. Test setup for (a) compressive strength of AAC materials; (b) compressive strength of adhesive materials; (c) tensile strength of adhesive materials.

In this study, the blast loading was assumed to have occurred as a result of typical suitcase bombs at reasonable distances from the wall face. In addition, the internal pores modeling of the AAC structure was ignored. Experimental data collection and available information from the manufacturer and previous studies were used here to estimate the mechanical properties of AAC material. Moreover, the compressive and tensile strength tests were performed on the standard AAC and the special mortar specimens. The explicit finite element software ABAQUS/Explicit was used for modeling and analysis of masonry walls under explosion loads. Using suitable material models, numerical modeling of the masonry walls made of AAC block units was created and analyzed under various blast loading scenarios perpendicular to the wall face. The cracking, displacement responses, stress distribution, and energy absorption patterns in AAC wall models were investigated and compared.

2.7. Considered Models

In the modeling stage, according to Figure 5, the height and width of the wall were 3 and 2 m, respectively, and the dimensions of AAC blocks were 600×250 mm with thicknesses of 15, 20, and 25 cm. The thickness of the mortar layer was also considered to be 10 mm. According to Figure 6, the boundary conditions of the wall were in three different states. The models studied here were subject to the explosions caused by 5 and 7 kg of TNT at the stand-off distances of 2, 5, and 10 m. In all models, the distance of the blast center from the ground (its height from the base of the wall) was considered to be 500 mm. Therefore, in general, 18 models of AAC masonry walls were considered here, with the specifications summarized in Table 3. It should be noted that no axial loading on the walls was taken into account since the considered walls were not assumed to be load-bearing structural components.

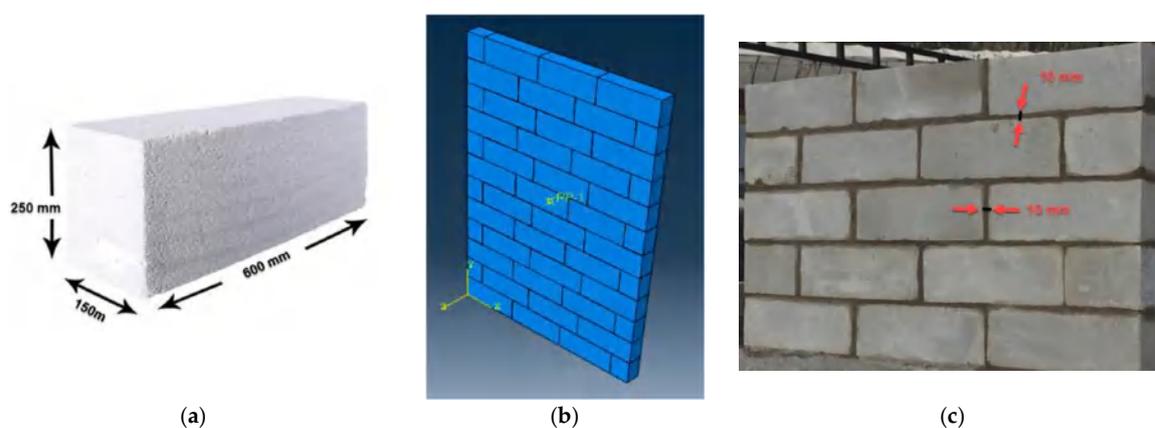


Figure 5. Specifications of the AAC wall and block model: (a) AAC block dimensions; (b) FE model of the wall; (c) assumed thickness of grout layer.

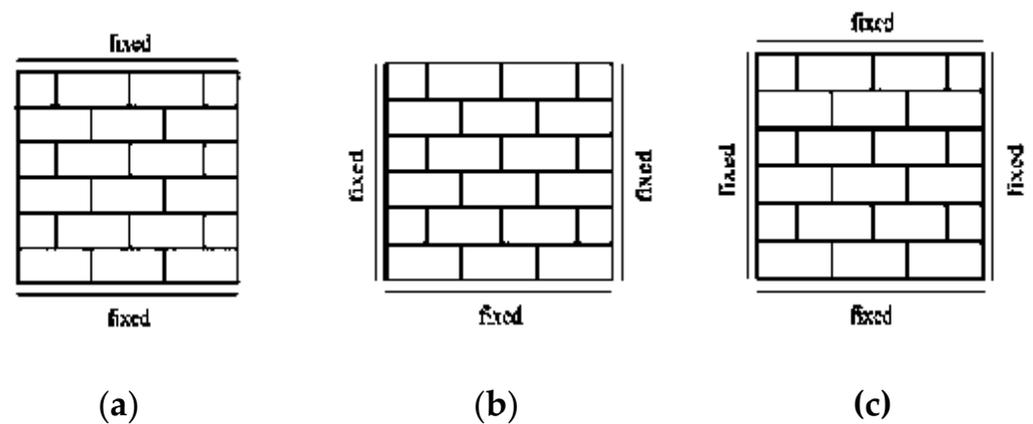


Figure 6. Boundary conditions of the walls: (a) fixed top and bottom of the wall; (b) fixed sides and bottom of the wall; (c) fully fixed BCs.

Table 3. Details of the studied models.

Model	Thickness (cm)	Stand-Off Distance (m)	Charge Weight (kg)	Scaled Distance ($m/kg^{1/3}$)
Model-1	15	2	5	1.170
Model-2	15	5	5	2.924
Model-3	15	10	5	5.848
Model-4	15	2	7	1.046
Model-5	15	5	7	2.614
Model-6	15	10	7	5.228
Model-7	20	2	5	1.170
Model-8	20	5	5	2.924
Model-9	20	10	5	5.848
Model-10	20	2	7	1.046
Model-11	20	5	7	2.614
Model-12	20	10	7	5.228
Model-13	25	2	5	1.170
Model-14	25	5	5	2.924
Model-15	25	10	5	5.848
Model-16	25	2	7	1.046
Model-17	25	5	7	2.614
Model-18	25	10	7	5.228

3. Results and Discussion

3.1. Validation

Kumar et al. [71] conducted a study on the behavior of RC slabs against blast loading. The slab with dimensions of $1000 \times 1000 \times 100$ mm was exposed to explosions with a scaled distance of $0.079\text{--}0.527$ $m/kg^{1/3}$. Here, to validate the process of blast load calculations in ABAQUS, this concrete slab was modeled, and the blast pressure distribution was compared to the original reference. The *slab-32* model in reference [71] was modeled under the effect of a blast of 2 kg TNT at a distance of 0.5 m ($Z = 0.3968$ $m/kg^{1/3}$). Here, due to the symmetry of the structure and loading, only one-fourth of the concrete slab was modeled and analyzed. The considered RC plate was a square with 500 mm length of side and 100 mm thickness. The finite element model of the RC slab with its support structure prepared here and the definition of the explosive charge are given in Figure 7.

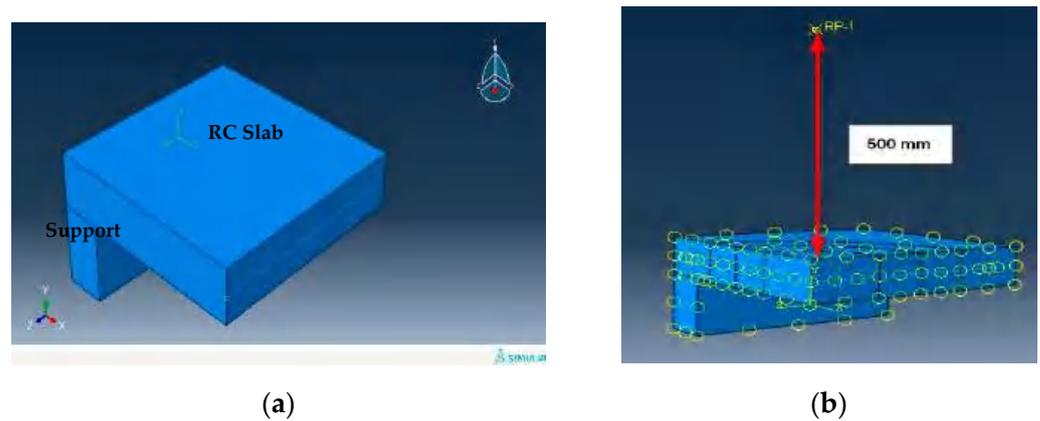


Figure 7. FE modeling of the slab: (a) modeling in Abaqus for validation; (b) blast loading in Conwep.

Figure 8 shows the time history of the calculated pressure in the middle of the slab in this study compared to the reference [71]. It can be observed that the maximum overpressure difference is equal to $100\% - \frac{42.36}{44.0} \times 100\% = 3.72\%$, which is in an acceptable range.

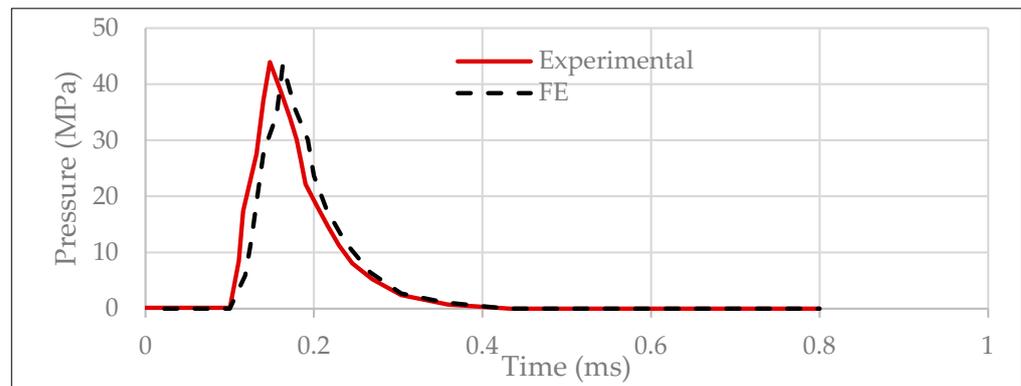


Figure 8. Pressure–time diagram obtained in the present study for validation process.

3.2. Von Mises Stress Distribution

After verifying the modeling procedure, the wall models were simulated, and structural responses were monitored and investigated. Figure 9 shows the maximum von Mises stress distribution in the studied wall models under blast loading. Table 4 also shows the maximum stresses in each model. It can be observed that with a reduction in the stand-off distance, the stress level was increased and distributed over a wider area of the wall. For example, in model-1 with a stand-off distance of 2 m, the maximum stress was 118.56 kN/m^2 higher than that in model-2 with a stand-off distance of 5 m. As the amount of TNT increased, the stress also increased, and more significant damages were observed in the models. For example, in model-3, with an explosive charge weight of 5 kg, the maximum stress was 29.32 kN/m^2 lower than that in model-6, with an explosive charge weight of 7 kg. In addition, as the thickness of the walls increased, the stress generally decreased. For example, in model-13 with a thickness of 25 cm, the maximum stress was 73.55 kN/m^2 lower compared to model-7 with the wall thickness of 20 cm.

Since the considered boundary conditions in these models were fixed at both ends (Figure 6a), the one-way behavior of the wall is obvious in Figure 9. According to Figure 9, shear failure was the dominant failure mode, since the grout shear strength was less than its compressive strength.

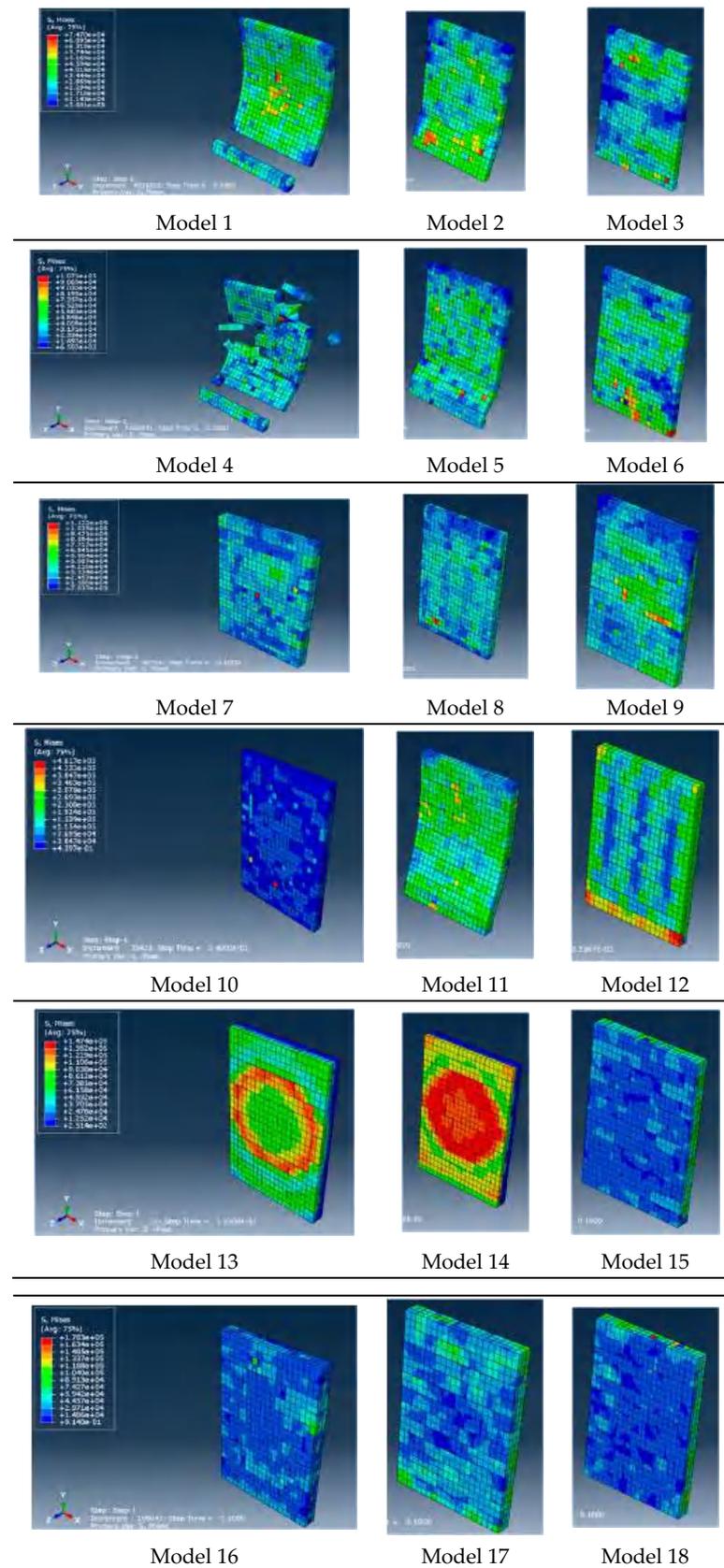


Figure 9. The von Mises stress distribution in some of the simulated wall models.

Table 4. Maximum stress values in the models.

Model	Thickness (cm)	Stand-Off Distance (m)	Charge Weight (kg)	Maximum Stress (kgf/m ²)
Model-1	15	2	5	7.470×10^4
Model-2	15	5	5	6.261×10^4
Model-3	15	10	5	5.494×10^4
Model-4	15	2	7	1.071×10^5
Model-5	15	5	7	9.522×10^4
Model-6	15	10	7	5.793×10^4
Model-7	20	2	5	1.122×10^5
Model-8	20	5	5	6.234×10^4
Model-9	20	10	5	9.829×10^4
Model-10	20	2	7	4.617×10^5
Model-11	20	5	7	6.715×10^4
Model-12	20	10	7	1.367×10^4
Model-13	25	2	5	1.474×10^5
Model-14	25	5	5	3.933×10^4
Model-15	25	10	5	6.075×10^4
Model-16	25	2	7	1.783×10^5
Model-17	25	5	7	5.638×10^4
Model-18	25	10	7	6.591×10^4

3.3. Displacement Responses

According to Figure 10, the displacement time history at the center of the wall was obtained for different thicknesses. It can be seen that the closer the explosives to the wall, the greater the displacement. With an explosive charge weight of 7 kg, for all wall thicknesses, the displacement was more significant than that for the other charge weights. For example, Models 4 and 6 had higher displacements than Models 1 and 3, respectively. According to the results, it can be observed that at a stand-off distance of 2 m (such as in Model-1), the wall models practically failed and had a larger displacement compared to the other models. Therefore, it can be concluded that at short stand-off distances, AAC-based masonry walls do not have enough resistance against blast loading, and a retrofitting scheme is required.

3.4. The Influence of the Boundary Conditions

There were three different cases for the support conditions of the walls. In the first case, the base of the wall was assumed to be restrained. In the second case, the whole perimeter of the wall was restrained, and in the third case, the three sides of the wall (bottom and sides) were restrained. Figure 11 shows the maximum stress distribution for different support conditions in the wall models. Clearly, the greater the restraints of the sides of the wall (case b), i.e., the better the restraining of the wall to the structural components of the building, the lower the stress in the wall. As such, in case "a" with the base of the wall constrained, a large surface of the wall had stress ranging from 2.3×10^4 to 5.2×10^4 kgf/m². In case "c" with the base of the wall and the sides restrained, a large surface of the wall had the stress of 1.2×10^4 to 4.8×10^4 kgf/m². Finally, in case "b" with the whole sides of the wall restrained, a large surface of the wall had the stress of 3.2×10^3 to 3.3×10^4 kgf/m².

According to Figure 11a, shear failure was obvious in the wall with one-way behavior. Two-way performance of the wall, as in Figure 11b, led to a decrease in the deformations compared to the other BCs. However, it increased the induced stress to the elements. As was anticipated, the boundary conditions had significant effects on the response of the walls made with AAC, similar to other masonry walls [22–24].

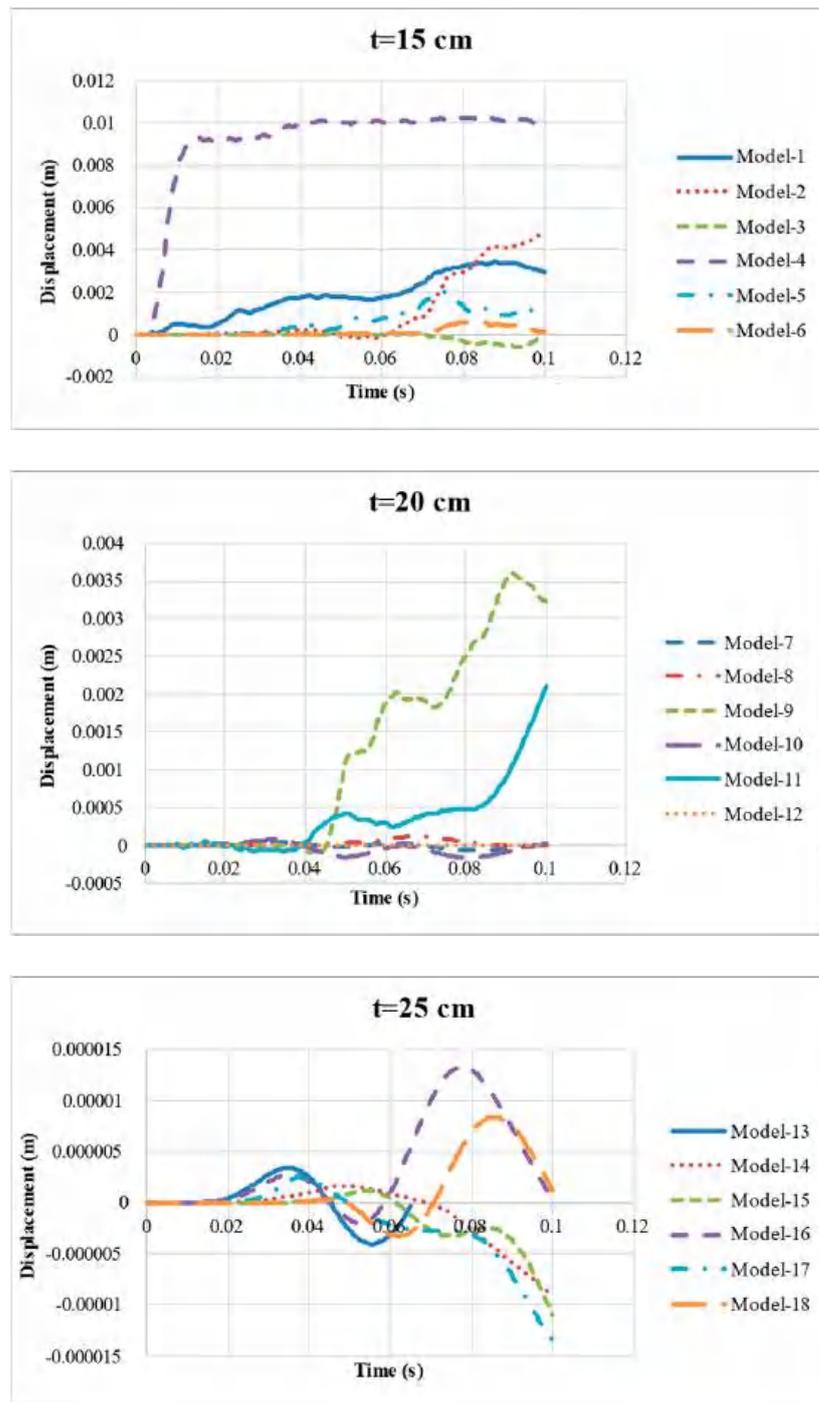


Figure 10. Time–displacement diagrams for wall thicknesses of 15, 20, and 25 cm.

3.5. Base Shear Force

Figure 12 indicates the temporal variations of the base shear in wall models with different thicknesses. It can be seen that with an increase in wall thickness, the force incurred on the base of the wall due to blast loading was significantly reduced. Therefore, the thickness of the walls was very effective in reducing the explosive demand on the AAC-based masonry walls. For example, for a wall with a thickness of 15 cm, the values of the base shear were about ten times higher compared those for a wall with a thickness of 25 cm.

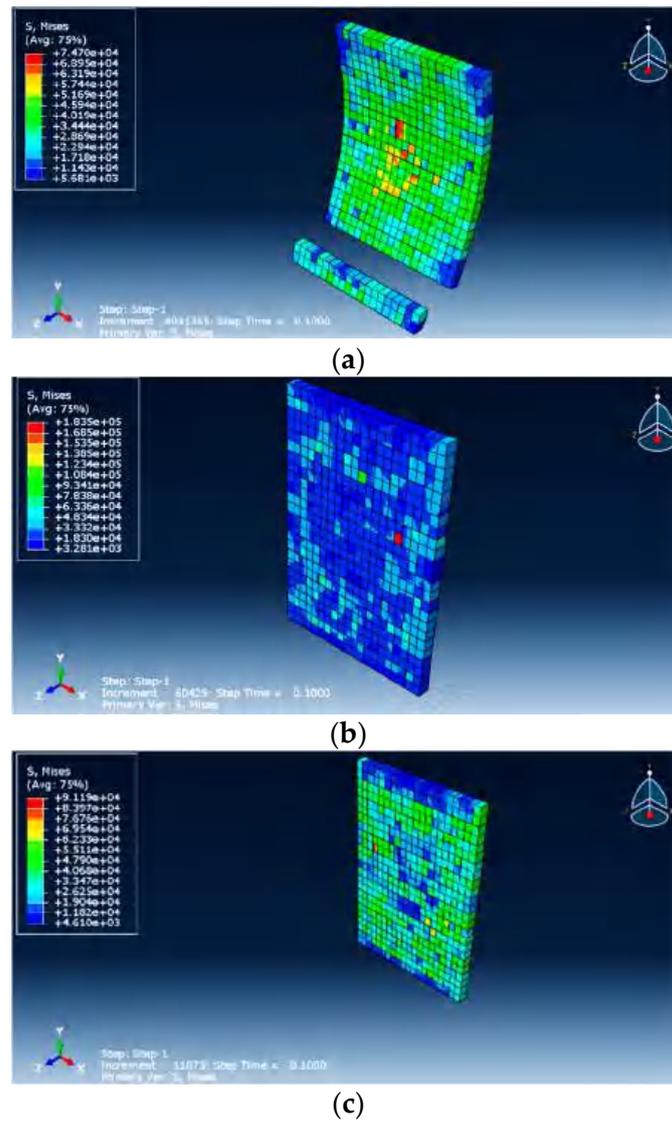


Figure 11. Distribution of stress in different support conditions: (a) restrained base; (b) restrained whole sides; (c) three-sided restrained.

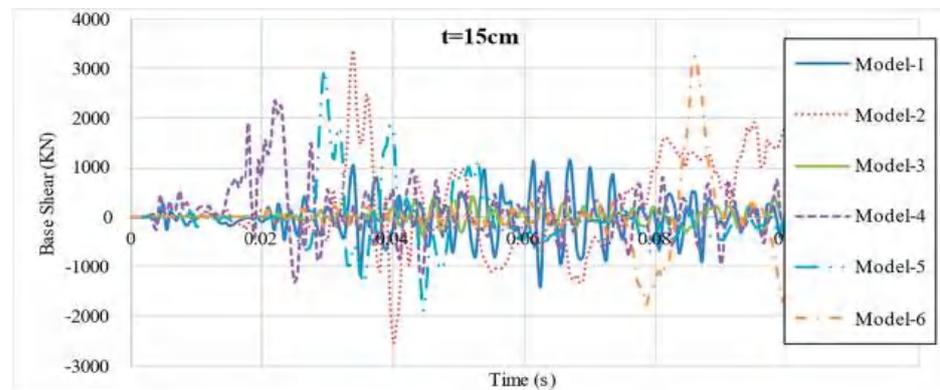


Figure 12. Cont.

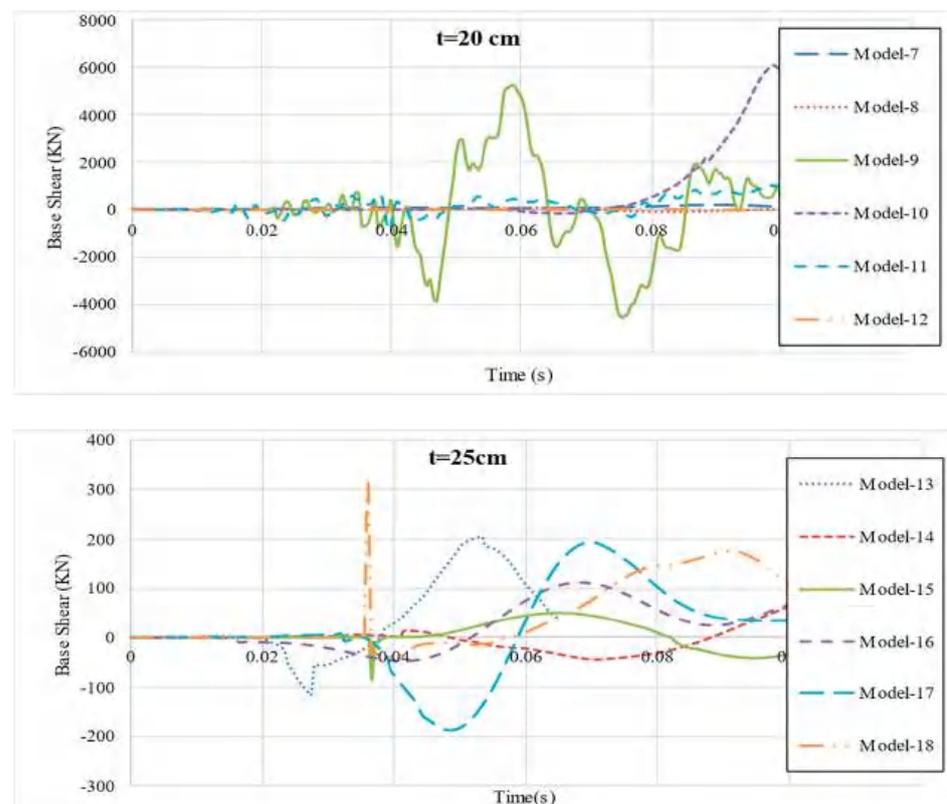


Figure 12. Base shear of the wall with thicknesses of 15, 20, and 25 cm.

4. Conclusions

Autoclaved aerated concrete (AAC) block is used in the construction of load-bearing and masonry walls due to its low thermal expansion coefficient, high fire resistance, and low weight. However, the low strength of materials and the heterogeneity of the material lead to the vulnerability of AAC masonry walls under external loads. To reduce the potential hazards in the structure and enhance the safety level, it is necessary to investigate the dynamic responses and failure of AAC masonry walls under explosive loads. This study aimed to investigate the behavior of AAC walls under blast loading. Therefore, the specifications of the materials required for modeling were first determined through experimental tests. Then, by modeling and analysis of the AAC walls in ABAQUS/Explicit, the behavior of these walls was investigated under blast loads. The main outcomes are as follows:

- Considering the weight of TNT used in a short distance ($R = 2$ m), it was observed that very large local stresses were created in the wall, which caused the wall to collapse in a very short time. It should be noted that at distances of less than 2 m, the wall models diverged at the very first moments. Therefore, the analysis and presentation of their results were avoided here.
- With the increasing charge weight, wall performance degraded. The stress level in the case of an explosive charge weight of 7 kg TNT increased by about 10% compared to that for 5 kg TNT. It is important to note that the walls modeled in this study under a charge larger than 7 kg TNT had a rapid failure in the initial moments. Therefore, considering the typical values of charge weight in the explosion events of hand grenades and suitcase bombs (about 20 kg-TNT) [72], it can be stated that masonry walls made with AAC do not have a good explosion resistance and would need retrofitting.
- Some retrofitting methods in masonry walls could involve using CFRP coating, steel wire mesh, and laminating. In addition, polyurea and polyurethane coatings, using

steel sheets, aluminum foam, and engineered cementitious composites, are suggested for masonry units that can be applied to walls made with AAC.

- With the increasing charge weight and decreasing stand-off distance, the wall displacement increased significantly, so that at a distance of 2 m, the displacement was several times that for the 5- and 10-m distances. In the walls with thicknesses of 15, 20, and 25 cm, the performance was also observed to be the same. As the amount of TNT increased, the stress values increased, and more damage was observed in the walls.
- The thickness of the walls was very effective in reducing the explosive demand force. For example, for a wall with a thickness of 15 cm, compared to that with a thickness of 25 cm, the base shear values induced by the same explosion were about 10 times higher.

To complete this study and achieve practical findings, complementary studies will be executed on construction units made of AAC blocks, including AAC walls reinforced by various methods such as using horizontal and vertical rebar meshes. The effect of various characteristics of mortar and adhesive on the behavior of AAC walls under blast loads will also be investigated.

It should be noted that the modeling of the walls built with AAC blocks in ABAQUS finite element software requires more extensive data and more detailed experiments. In particular, dynamic properties under high strain rates require further experimental and laboratory studies.

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Article

Experimental Study of Explosion Mitigation by Deployed Metal Combined with Water Curtain

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Abstract: In this paper, protective barriers made of perforated plates with or without a water cover were investigated. In urban areas, such barriers could be envisaged for the protection of facades. An explosive-driven shock tube, combined with a retroreflective shadowgraph technique, was used to visualize the interaction of a blast wave profile with one or two plates made of expanded metal. Free-field air blast experiments were performed in order to evaluate the solution under real conditions. Configurations with either one or two grids were investigated. The transmitted pressure was measured on a wall placed behind the plate(s). It was observed that the overpressure and the impulse downstream of the plate(s) were reduced and that the mitigation performance increased with the number of plates. Adding a water layer on one grid contributed to enhance its mitigation capacity. In the setup with two plates, the addition of a water cover on the first grid induced only a modest improvement. This blast mitigation solution seems interesting for protection purposes.

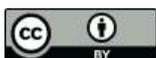
Keywords: blast; mitigation; grid; water curtain

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1. Introduction

The protection of people and structures against the effects of blast waves from terrorist attacks or industrial hazards is of significant interest. It is well known that protective barriers are an effective way to reduce blast loads and to mitigate the adverse effects. Protective solid barriers are usually made of reinforced concrete, concrete masonry unit or steel-concrete-steel composite materials. These barriers are very rigid and have negligible deformation. In urban areas, other protective barriers could be envisaged for facades. The blast wave would be absorbed, deflected or disrupted to be ultimately reduced before it reaches its intended target. Perforated plates disrupt blast waves: it has been shown that grids or perforated plates modify the flow field by introducing new shock waves, regions of vortices and considerable turbulence in which the energy of the incident shock wave can be dissipated [1–3]. Moreover, grids or perforated plates could be modern architectural design elements which can be used outside buildings. The use of water walls for mitigating the damages from blast waves generated by an explosion has also been described [4], although there is only limited literature on this topic. It seems that the mitigation stems from the fact that the blast wave is obstructed, reflected and diffracted by the water wall. Consequently, it could be an advantage to add a water cover to the grids or perforated plates in order to protect building façades or walls.

Recently, the shock wave attenuation performance of protective barriers made of woven wire mesh was investigated [5]. The woven wire mesh was located about 5 m in front of a wall and no obvious mitigation was observed behind the barrier. The woven wire mesh had a very high porosity and this probably explains such results. Two previous studies have been published regarding metal ring meshes, grids or perforated plates combined with a downward-streaming water curtain for blast mitigation [6,7]. Gebbeken et al. [6] tested a stainless steel ring mesh in combination with a flowing water layer. The charges were detonated 5 m in front of the ring mesh with or without a water cover and the side-on

overpressure was measured on the shock wave path. The reflected overpressure was also measured on a wall located 5 m behind the ring mesh. For the ring mesh alone, they described an initial side-on overpressure reduction of approximately 17% at 50 cm behind the grid and of around 1–6% at 1.5–2.5 m behind the grid. When the ring mesh was covered by water, the initial side-on overpressure was reduced by 56% close behind the ring mesh and by 20% 5 m behind the mesh. As regards the positive impulse, the ring mesh itself caused no reduction. However, a reduction of about 17–31% was obtained behind the mesh by adding a water curtain. The reflected pressure on the wall was slightly decreased by the ring mesh but the addition of a water cover did not improve the attenuation performance. In the second study [7], a transonic shock tube was used to visualize the interaction of a blast wave profile with a metallic perforated plate or with a metallic perforated plate covered by a layer of water. Free-field air blast experiments were also performed. Three grid types with different porosities were tested. The highest attenuation was obtained with the grid having the lowest porosity. The attenuation was of the order of 17–25% 25 cm downstream from the grid and about 25–30% 3 m downstream from the grid. When the grids were covered by a water layer, the initial overpressure behind the plate was reduced for all grid types. The blast mitigation was improved, especially with the grids having a high porosity. Again, the most pronounced reduction (about 35 to 48%) was obtained with the grid having the lowest porosity. The initial overpressure 3 m downstream was also reduced by about 20–30%. The impulse was also reduced by the grids with or without a water film cover. Consequently, this blast mitigation method appears promising but still needs some further work and improvement. The porosity of the plate is an important factor to take into account in the blast mitigation performance and a water cover on the plate increases even more the blast attenuation. The plate should have a relatively low porosity. The use of a configuration with two plates could also represent an option for improvement. Indeed, it has been shown that shock wave trapping between two perforated plates enhanced the shock wave attenuation downstream from the grids [8].

In this paper, the assessment of this blast mitigation solution made of perforated plates with or without a water cover was further investigated in order to have a better understanding of its mechanism and to improve its performance. Grids made of expanded metal, with or without a water film, served as an obstacle. Expanded metal has an interesting geometry that enhanced the reflection of the shock wave, and a low porosity. First, an explosive-driven shock tube (EDST) was used to visualize the interaction of a blast wave profile with one or two plates made of expanded metal, using a retroreflective shadowgraph technique. Indeed, the propagation of blast waves in complex media is an important topic of shock wave research and there is a need to study wave phenomena in complex environments. EDSTs generates high dynamic loadings compared to conventional shock tubes. Their mitigation capacity could be assessed under a load comparable to that produced by several kilos of high explosives, located several meters from a target. Secondly, free-field air blast experiments were performed in order to evaluate the protection system under real conditions. Configurations with either one or two grids were investigated. The case in which a film of water was added on the grid, or on the first grid in the case of a two-grid configuration, was also studied. The transmitted pressure was measured on a wall placed behind the plate(s) and the shock wave reflection by the plate was assessed using a sensor located on the ground in front of the first grid.

2. Materials and Methods

2.1. Samples

Grids made of expanded metal were investigated (Figure 1). Expanded metal is a metal sheet that has been cut and stretched to form a regular pattern. Expanded metal is stronger than an equivalent weight of wire mesh because the material is deformed, allowing the metal to remain as a single piece. The expanded metal used in the present study had a hexagonal mesh. The characteristics were an open area of 24%, a hole size

of 45 mm × 13 mm, a thickness of 3 mm, an apparent thickness of 9 mm and a strand of 5 mm.

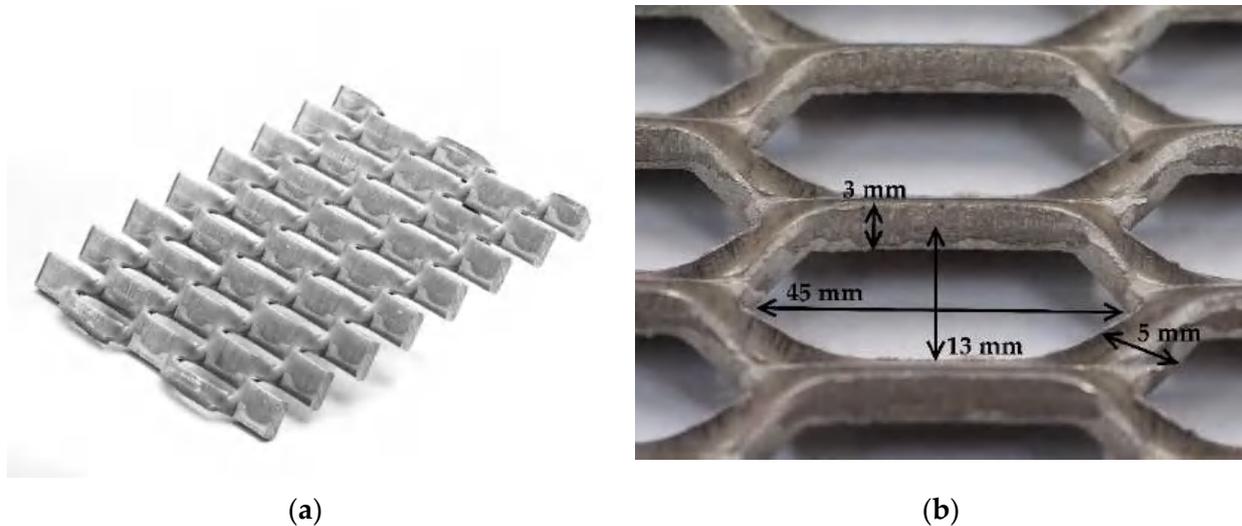


Figure 1. (a) Piece of expanded metal; (b) detail: open area of 24%, hole size of 45 mm × 13 mm, thickness of 3 mm, apparent thickness of 9 mm, strand of 5 mm.

2.2. Explosive Driven Shock Tube

The EDST was based on previously published works [9,10]. The shock tube has a square external section of 100 mm × 100 mm, a square internal section of 80 mm × 80 mm and a total length of 1750 mm (Figure 2). A pressure sensor (Kulite HKS 375) was used to measure the reflected pressure at a wall behind the plates. A spherical charge of C4 was used ($m = 15$ g) to produce a planar blast wave. All the charges were cast and detonated, without any container, 50 mm from the shock tube inlet (Figure 2). A part of the initial spherical blast wave enters in the tube. At the tube inlet, the surface of the incident shock has a square curved shape which will be flattened as the shock travels through the tube. The initial spherical blast wave becomes almost completely planar after about 1.5 m of propagation, leading to an initial uniform loading at the outlet of the shock tube. The blast profile obtained in this way is realistic and in line with real threats.

The distance between the outlet of the EDST and the wall was 180 mm. The plate, or the last plate in the case of two plates, was positioned 50 mm in front of the wall. Where two plates were used, the spacing between the plates was 40 mm. The grid holes were always aligned directionally.

2.3. Imaging for EDST

A high speed Photron SA-Z camera was used to record images of the propagation and of the interaction of the shock wave with the plate(s) using a retroreflective shadowgraph technique [11]. An extreme high power light emitting diode (LED) (XHP70.2, CREE), located on an axis with the center of the camera lens, illuminated the outlet of the EDST and a panel covered by a retroreflective material (3M Reflexfolie 4090) which was placed in the background. Figure 3 shows a photograph of the camera with the LED and the EDST and the Figure 2 shows the position of the retroreflective panel relative to the EDST and the camera. A power supply and a purpose-built trigger unit allowed the LED to be pulsed for about 10 ms. The reflection and the transmission patterns of the shock wave through the obstacle could be visualized via their shadow on the panel. Moreover, a piece of retroreflective material was plastered on the wall and a 45-degree inclined mirror was attached to the end of the EDST (Figure 4). The camera filmed the panel placed in the background but also the mirror, allowing for the visualization of the shock wave in the

axial direction. The videos were recorded with a frame rate of 100,000 fps at a resolution of 408×384 pixels.

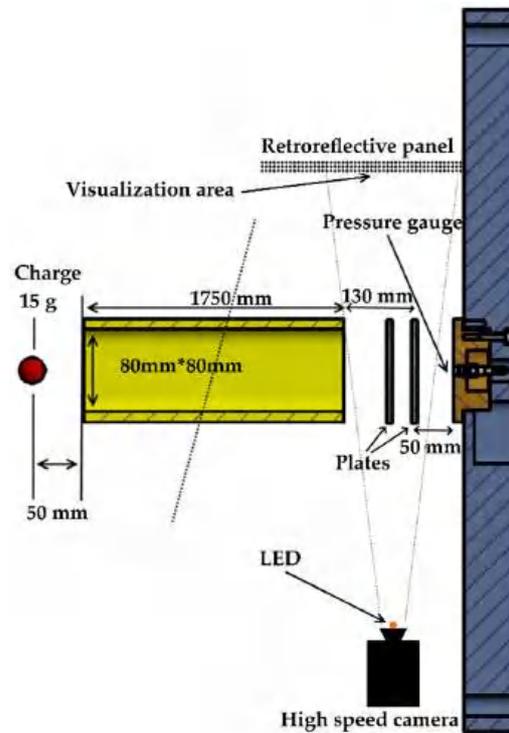


Figure 2. Schematic view of the explosive-driven shock tube and positioning of the pressure sensor and plates. The position of the high speed camera and of the retroreflective panel is also shown.



Figure 3. View of the experimental setup, showing the explosive driven shock tube and the high speed camera with the LED.

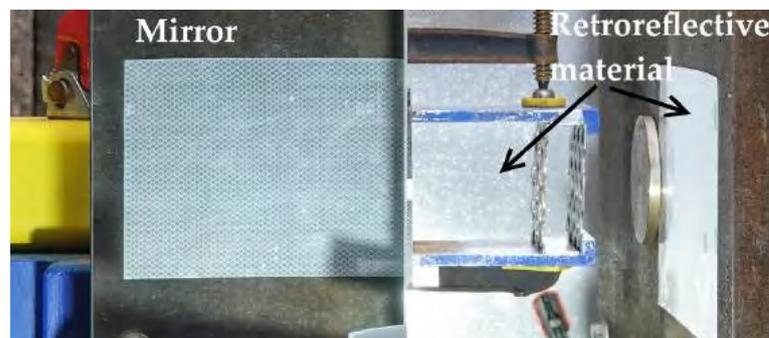


Figure 4. View of the EDST end showing the panel in the background, the piece of retroreflective material plastered on the wall and the 45-degree inclined mirror.

2.4. Free Field

Explosion tests were conducted with spheres of 2 kg C4. The charges were raised by 25 cm and ignited by a high voltage cap (RP 501) (Figure 5). Configurations with either one or two grids were investigated. The addition of a film of water on the grid, or on the first grid in the case of a two-grid configuration, was also tested. For tests with one grid, the charge was positioned 3.8 m from the grid and 4.8 m from the wall (Figure 6). For tests with two grids, the charge was positioned 3 m from the first grid, 3.8 m from the second one and 4.8 m from the wall (Figure 5). When paired, grids were spaced 0.8 m apart. The grid size was 2 m × 3 m and concrete blocks placed at the left and right of the grids were used to mount them (Figures 5 and 6). Each had the following dimensions: length 160 cm, height 40 cm and width 80 cm. One side-on pressure gauge (PCB137A23) (Figure 5), located at a right angle to the shock wave's propagation toward the wall, allowed for verification of reproducibility. One PCB sensor (M102A) was used to evaluate the effect of the grids or, of the grids covered by a film of water, of the reflected pressure on the wall located behind the grid (Figure 7, left). The sensor was positioned at a height of 50 cm. One pressure gauge (M102A) was positioned on the ground in front of the first grid position (Figures 5–7, left). This gauge was located 19.5 cm from the first grid. The height of this gauge was 5 cm. The water layer was generated by a pool fountain (VidalXL) (Figure 8).

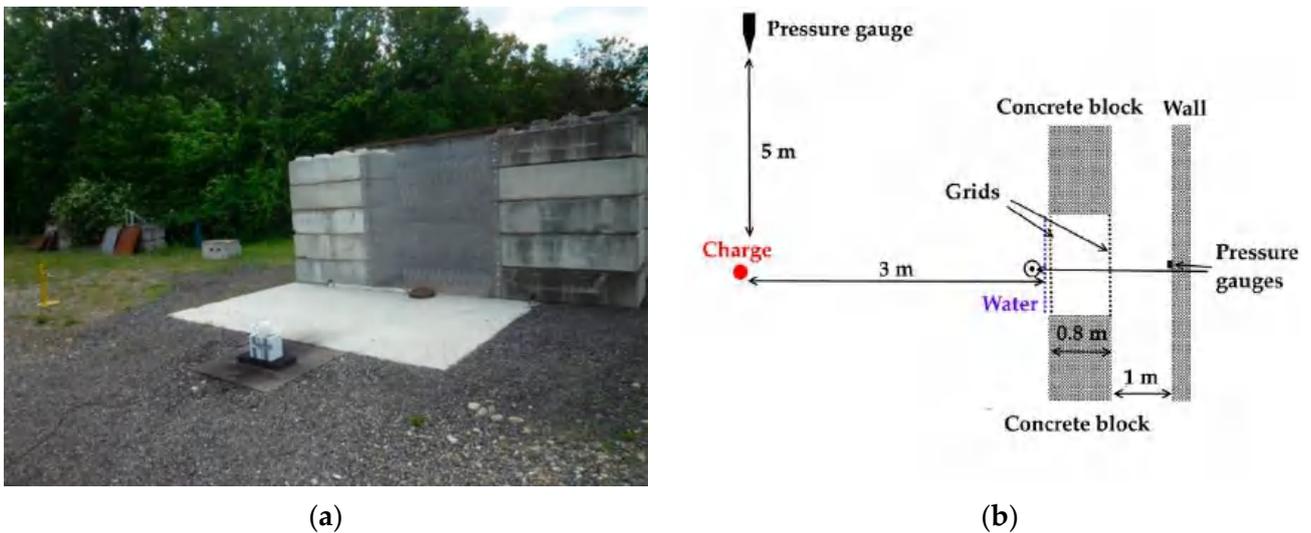


Figure 5. (a) Photography of the experimental setup; (b) Schematic diagram showing the charge, grids, concrete blocks, gauge in front of the grid and one side-on pressure gauge, placed at a right angle.



Figure 6. View of the experimental setup showing the one-grid configuration: the grid position, concrete blocks and gauge in front of the grid.



Figure 7. (a) Reflected pressure gauge located on the wall behind the grids; (b) pressure gauge located in front of the first grid position.



Figure 8. First grid equipped with a pool fountain.

3. Results

3.1. Explosive Driven Shock Tube

Figure 9 shows the shock wave propagating at the outlet of the EDST when there was no perforated plate. The photographs were taken with high-speed imaging. Thanks to the mirror, set at 45° in front of the camera, imagery could be obtained from the axial direction simultaneously with the direct view. Consequently, events happening axially to the EDST outlet could be observed. The shock wave was not totally planar at the outlet of the tube; indeed, the two metallic plates, fixed parallel to the EDST at the end and which were used to clamp the perforated plates, reflected the shock wave ($t = 1440 \mu\text{s}$). These two plates had a shoulder at their ends which provided additional thrust for the fixation of the perforated plates. These shoulders can be seen in Figure 4 and the shock wave visibly interacts with them ($t = 1520 \mu\text{s}$). The shock wave emerging laterally from the EDST was visible in the mirror. It should be noted that the shock wave exited the EDST outlet immediately, but in the beginning the optical setup did not allow the observation of this expansion.

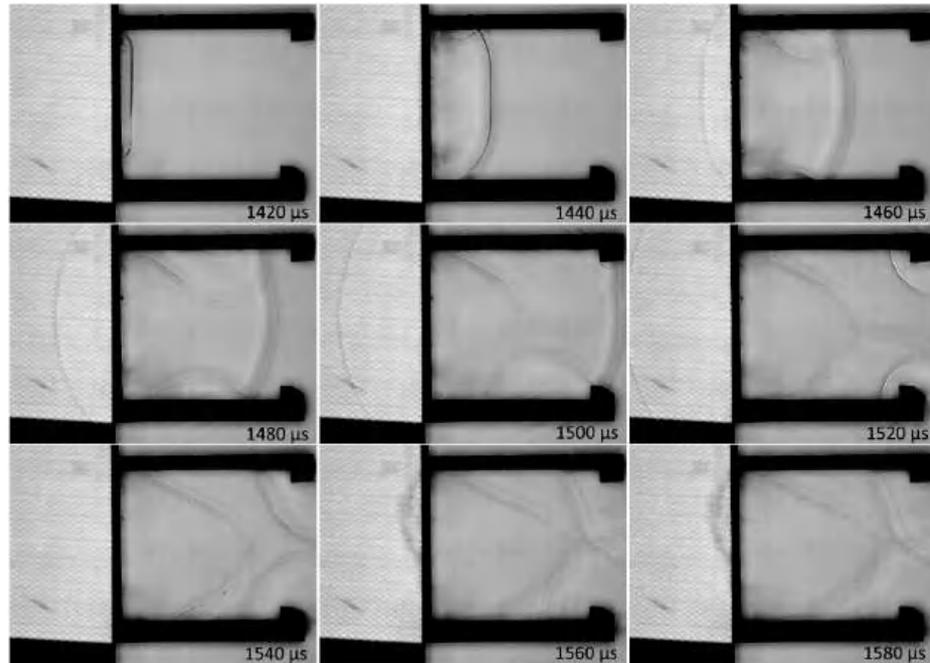


Figure 9. Photographic results of high-speed video recording, showing shock wave propagation at the outlet of the EDST (20 μ s between each photograph).

Figure 10 shows shock wave propagation through one deployed metal plate. The plate was positioned at a distance of 130 mm from the tube outlet and at a distance of 50 mm in front of the wall. At $t = 1470 \mu$ s, the shock wave when passing through the plate apertures split into several shock waves, one for each aperture, and these shock waves recombined further down. At $t = 1530 \mu$ s, the shock wave behind the plate became almost planar again. A complex structure of turbulence appeared just behind the grid and it persisted for some time; at $t = 1510 \mu$ s turbulence began to form along the backside the grid, and at $t = 1790 \mu$ s it was still observable. The recombined shock wave behind the perforated plate hit the wall and was reflected at $t = 1570 \mu$ s. The remaining shock wave, which did not pass through the grid and was reflected by it, became almost planar at $t = 1550 \mu$ s.

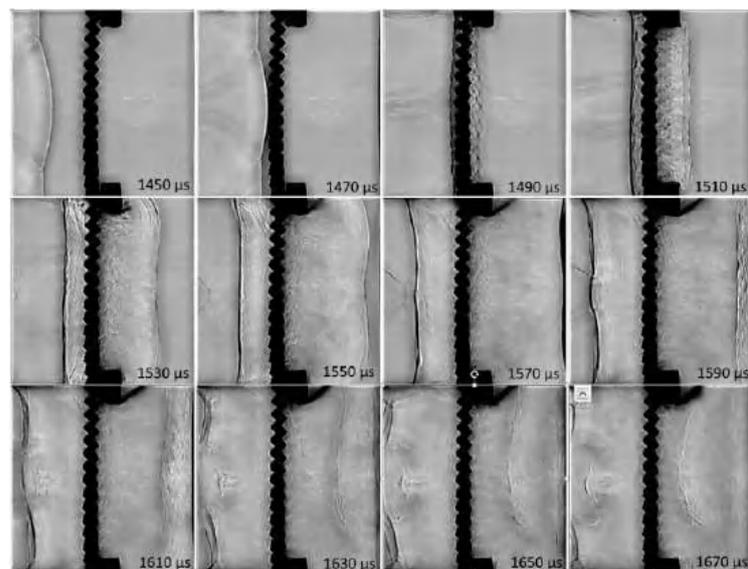


Figure 10. Photographic results of high-speed video recording, showing shock wave propagation through one plate of deployed metal (20 μ s between each photograph).

Figure 11 shows shock wave propagation through two deployed metal plates. The first plate was positioned at a distance of 90 mm from the tube outlet and at a distance of 90 mm in front of the wall. Grids were paired 40 mm apart. When passing through the apertures of the first plate ($t = 1450 \mu\text{s}$), the shock wave split into several shock waves, one for each aperture, and these shock waves recombined further down. The transmitted shock wave was similar to the incident shock wave at $t = 1490 \mu\text{s}$. At the back of the shock wave, between the two plates, lasting turbulence could be observed, especially immediately behind the first plate. The remaining shock wave, which did not pass through the first plate, was reflected, leading to the pattern captured at the back of this plate. Thereafter, the shock wave transmitted by the first plate impacted the second plate ($t = 1510 \mu\text{s}$) and, once again, a part of the shock wave was transmitted and part was reflected. The wave reflected by the second plate propagated toward the first plate and, as it arrived near the plate, the shock front and the turbulence became less obvious. We could also see turbulence behind the second plate.

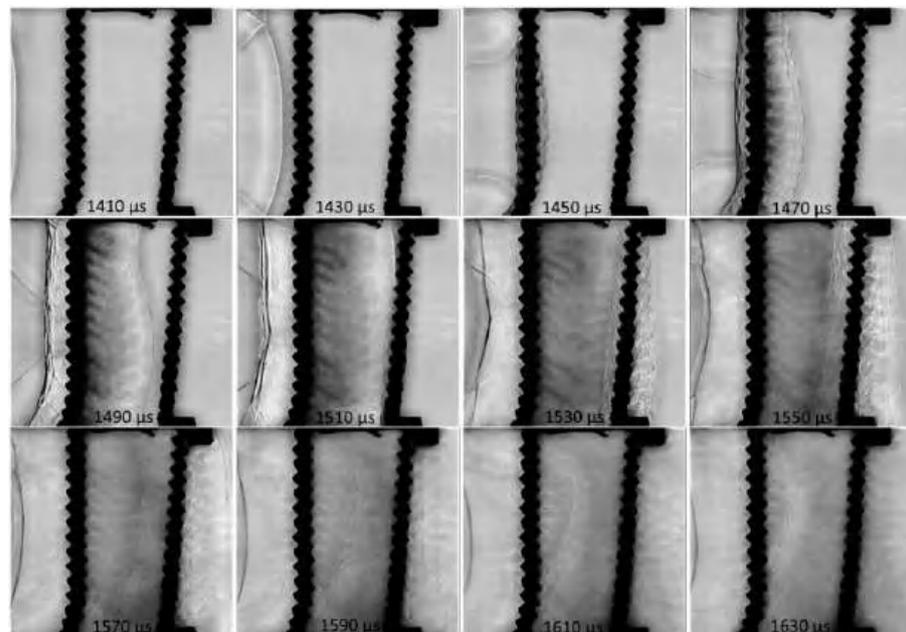


Figure 11. Photographic results of high-speed video recording, showing shock wave propagation through one plate of deployed metal ($20 \mu\text{s}$ between each photograph).

Five reference tests were conducted without any plates and seven experiments were conducted with either one or two grids positioned at the outlet of the EDST. The reflected overpressure versus time was collected by the sensor inserted in the wall behind the plate. The blast wave propagation through the shock tube was computed with Autodyn (ANSYS) and the overpressure at the outlet of the shock tube was obtained. The calculated initial overpressure was 32 bar. The measured value at a distance of 180 mm was ~ 18 bar, which was consistent with simulation. The impulse, which is the pressure signal integrated over time, as a function of time, was also computed. Figure 12a presents the reflected pressure as a function of time, obtained in a reference test and in one test each using one or two expanded metal plates. When the number of plates increased, the reduction of the initial reflected overpressure increased. The initial overpressure reflected on the wall was reduced by 46% and 72%, respectively, when either one or two plates were positioned in front of the EDST (Table 1). Correspondingly, the maximum impulse was reduced by 68 and 89% (Table 1).

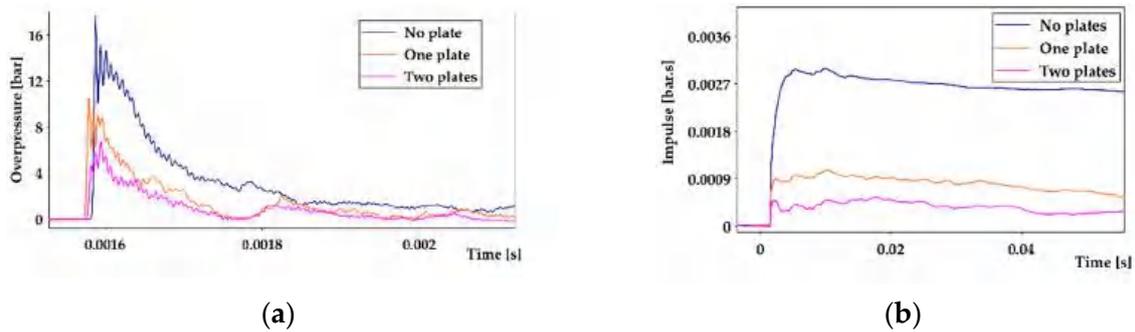


Figure 12. Reflected pressure (a) and impulse (b) measured on the wall positioned 50 mm behind either one or two expanded metal plates. One reference test, which was performed without plates, is also shown.

Table 1. Initial reflected overpressure and maximum impulse collected by the sensor inserted in the wall and located 50 mm downstream from the perforated plates for all experiments. The difference from the mean value obtained with the reference tests (no plate) is also given.

Type of Plate	Number of Plates	Overpressure (Bar)	Attenuation (%)	Impulse (Bar.s)	Attenuation (%)
no plate	-	22.82	-	0.0032	-
no plate	-	19.48	-	0.0040	-
no plate	-	15.61	-	0.0035	-
no plate	-	15.65	-	0.0037	-
no plate	-	15.73	-	0.0034	-
expanded metal	1	9.36	48	0.0011	70
expanded metal	1	7.71	57	0.0012	65
expanded metal	1	10.89	39	0.0013	63
expanded metal	1	10.30	42	0.0015	59
expanded metal	2	4.46	75	0.0003	93
expanded metal	2	4.50	75	0.0005	87
expanded metal	2	5.86	67	0.0005	87
expanded metal	2	4.46	75	0.0003	93

3.2. Free Field

The initial overpressure and the maximum impulse obtained through the control gauge for all experiments are presented in Tables 2 and 3, respectively. These measurements showed that there was rather good reproducibility. The side-on overpressure was estimated with Kingery’s and Bulmash’s formula [12]. The detonation of 2 kg of C4 generates an overpressure of 0.8 bar at a distance of 5 m. The measured values were consistent with this value. Figure 13 presents the overpressure measured on the wall for one test of each configuration. Two or three experiments were conducted for each configuration. Table 2 gives the initial overpressure value obtained in all the tests. According to Kingery’s and Bulmash’s formula [12], the detonation of 2 kg of C4 generates a reflected overpressure of 2.6 bar at a distance of 4.8 m. The measured values were consistent with this value. The overpressure was reduced when one plate was located in front of the wall, in the order of 32–47%. Adding a second plate on the shock wave path led to a stronger attenuation. The initial overpressure was reduced by about 62–66%. When a water film was used, the reflected overpressure on the wall was reduced even more, especially in the case of the one-plate configuration. The reduction of overpressure was about 69–71% and 66–74% for one-plate configurations and two-plate configurations, respectively. The setup using a water wall alone was also evaluated and the reflected overpressure on the wall was not modified. Figure 14 presents the impulse measured on the wall for one test of each configuration and Table 3 gives the maximum impulse values obtained for all tests. The results were similar to those observed for overpressure.

Table 2. Initial overpressure collected by three sensors (control, inserted into the wall and laid on the ground) for all experiments. The attenuation of the overpressure measured on the wall relative to the mean value obtained by the reference tests (no plate) is also given. Overpressure ground 1 and overpressure ground 2 correspond to the maximum value of the first and the second peak, respectively.

Experiment Type	Overpressure Control (Bar)	Overpressure Wall (Bar)	Attenuation (%)	Overpressure Ground 1 (Bar)	Overpressure Ground 2 (Bar)
no plate	0.90	2.53	-	3.20	-
no plate	0.89	2.61	-	3.22	-
no plate	0.89	2.70	-	-	-
2 expanded metal	0.90	1.05	60	3.08	2.12
2 expanded metal	0.86	0.88	66	2.26	1.96
2 expanded metal	0.89	0.99	62	3.31	2.14
2 expanded metal & water	0.89	0.90	66	1.88	1.89
2 expanded metal & water	0.78	0.68	74	3.17	1.83
1 expanded metal	0.89	1.42	46	3.16	-
1 expanded metal	0.82	1.78	32	3.52	-
1 expanded metal	0.84	1.40	47	3.62	-
1 expanded metal & water	0.80	0.82	69	2.80	-
1 expanded metal & water	0.76	0.75	71	3.08	-

Table 3. Maximum impulse collected by the three sensors (control, inserted into the wall and laid on the ground) for all experiments. The attenuation of impulse measured on the wall relative to the mean value obtained by the reference tests (no plate) is also given. Impulse ground 1 and impulse ground 2 correspond to the maximum value of the first and the second peak, respectively.

Experiment Type	Impulse Control (Bar-s)	Impulse Wall (Bar-s)	Attenuation (%)	Impulse Ground 1 (Bar-s)	Impulse Ground 2 (Bar-s)
no plate	0.00103	0.00247	-	0.001427	-
no plate	0.00105	0.00275	-	0.001388	-
no plate	0.00097	0.00261	-	-	-
2 expanded metal	0.00108	0.00105	60	0.001384	0.002338
2 expanded metal	0.00108	0.00105	60	0.001287	0.002258
2 expanded metal	0.00105	0.00095	64	0.001127	0.001849
2 expanded metal & water	0.00104	0.00077	71	0.001312	0.002546
2 expanded metal & water	0.00098	0.00071	73	0.001143	0.002234
1 expanded metal	0.00107	0.00155	41	0.001411	-
1 expanded metal	0.00100	0.00144	45	0.001395	-
1 expanded metal	0.00096	0.001437	45	0.001289	-
1 expanded metal & water	0.00094	0.00108	58	0.001418	-
1 expanded metal & water	0.00093	0.00110	58	0.001402	-

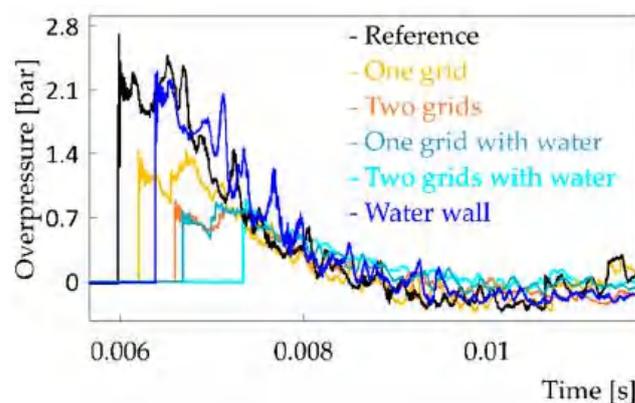


Figure 13. Reflected pressure measured on the wall for one test of each configuration. One reference test is also shown.

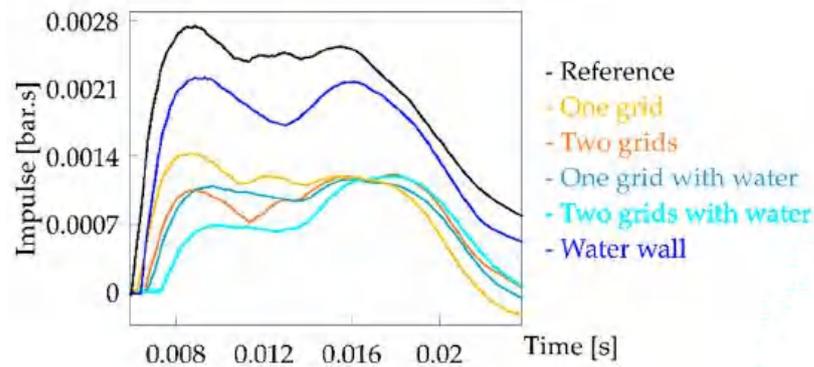


Figure 14. Reflected impulse measured on the wall for one test of each configuration. One reference test is also shown.

Figure 15 presents the overpressure measured by the gauge positioned on the ground at the first plate's position. The reflection of the shock wave, when a plate was placed near this sensor (two-plate configuration), could be observed (see the arrows in the Figure 15). In case of one-plate configurations, the sensor was placed further off the plate and the shock wave reflection was not perceptible. In the case of two-plate configurations, the reflection of the shock wave on the second plate was clearly manifested as a huge overpressure second peak.

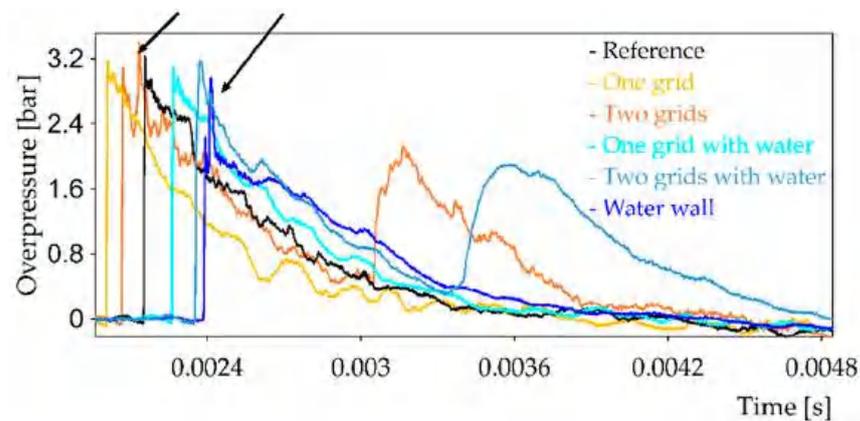


Figure 15. Overpressure obtained by the gauge laid on the ground for one test of each configuration. One reference test is also shown.

4. Discussion

Shock wave propagation through one or two deployed metal plate(s) was observed thanks to the EDST and a retroreflective shadowgraph technique. This experimental set-up was very well suited to this research, since the shock wave interaction with perforated plates could be assessed with high loading and visualized at the same time. This is not possible when using a conventional shock tube, the blast loading being rather moderate. In any event, it is not possible to use high Mach numbers when the air flow in the shock tube is especially blocked, due to risk of damage. Understanding of the complex flow field induced by blast that passes through a complex media is an important aspect in blast mitigation research and could help to design new devices for protection against blast loading. The shock wave, when it passed through the apertures of a deployed metal plate, split into several shock waves, one for each aperture, which recombined further on. A complex structure of turbulence appeared just behind the grid and it persisted for some time. The remaining shock wave, which did not pass through the grid, was reflected by the grid. In the case of two plate configurations, the shock wave had time to reform between the two plates, which impacted the second plate. Once again, a part of the shock wave was transmitted and part was reflected. Turbulence behind the second plate was also visible.

The shock wave reflection and the creation of turbulence led to a blast wave attenuation. Consequently, when two plates served as an obstacle, these phenomena occurred twice and the mitigation was greater. This was confirmed by the reflected overpressure and impulse measured by the sensor inserted into the wall behind the plate(s). When the number of plates increased, these values were reduced. Similar observations were made by O. Ram et al. [13], who assessed the propagation of shock waves through an array of perforated plates in a conventional shock tube. Thus, we can conclude that similar phenomena occurred at low (3 bar) and high loading (20 bar).

The results obtained in free field were consistent with the results obtained by means of the EDST. The reflected overpressure and impulse measured behind one plate on a wall was reduced on the order of 32–47%. The addition of a second plate on the shock wave path led to a stronger attenuation (62–66%). When a water film was used, the reflected overpressure on the wall was reduced still more, especially in the case of one-plate configurations—the mitigation approached that obtained from two-plate configurations. The water cover had only a small effect on blast mitigation when two plates were used. We can conclude that the water layer's contribution mostly enhanced the reflection of the shock wave by filling the apertures with water. When one perforated plate was covered with a water film, its capacity to reflect the blast wave was enhanced, its performance to mitigate the blast increased and approached that obtained from two-plate configurations. In cases with two plates, the obstruction of the blast wave was rather high, and the addition of a water cover induced only a modest improvement. In [7], thanks to a transonic shock tube, the interaction of a blast wave with a perforated plate with a water cover was imaged and it was observed that the water film disintegrated into droplets significantly after the shock wave front had passed through it. The fragmentation of the water film had little effect on the attenuation, as it broke long after the passage of the shock wave front and there is little extraction of energy from the shock front from water layer fragmentation. In this work, the results have also shown that a water wall alone had almost no impact on the reflected overpressure on the wall. Moreover, the study [4] on blast mitigation using a water wall, in which walls made of plastic bags, filled with water and having a thickness of 5 to 8 cm, it was shown that the mitigation was obtained by obstruction, reflection and diffraction of the blast wave. The mitigation mechanism was comparable to that of a rigid wall, thus the mitigating effect of energy exchange with water was not primarily responsible for the effect. The results obtained here could also be compared to those obtained by Gebbeken et al. [6] and Xiao et al. [5]. In these two studies, blast mitigation observed when using a single grid was very low and this could be explained by the high porosity of the grids used. Indeed, the first study [6] used a stainless steel ring mesh with a porosity of 63% and, in the second, a woven wire mesh having a relative opening fraction of 60.2%. In the setup using ring mesh, adding a water curtain also enhanced the attenuation of both peak overpressure and positive impulse. The authors also claimed that when a blast wave hits ring mesh covered by water, the water layer forms a closed surface that reflects the blast wave to a greater extent.

In the case of a higher-charge explosive, the phenomenon of shock wave transmission/reflection would likely be the same and equivalent mitigation performance would be achieved. However, if the grids are not sufficiently resistant to a high loading, the grids could deform and tear. Some debris could impact the structure behind such grids, and consequently the use of gridded plates could have prejudicial effects. The grids and their attachment system must be sized to guarantee their resistance and their structural integrity with respect to anticipated blast size.

5. Conclusions

In this paper, we have assessed a blast mitigation solution made of perforated plates with or without a water cover. The mitigation of a blast wave after its passage through one or two plate(s) made of deployed metal, covered or not by a water film, was investigated. First, we imaged the interaction of a blast wave with the grids at high loading. Secondly,

free-field air blast experiments were performed in order to evaluate the protection system under real conditions. It was observed that the overpressure and the impulse downstream of the grids were reduced and that the mitigation performance increased with the number of plates. Adding a water layer to one grid contributed to its mitigation capacity. However in setups with two plates, the addition of a water cover on the first grid induced only a modest improvement. All in all, this method seems to warrant interest for protection purposes.

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Article

Numerical Simulation on Dynamic Behavior of Slab–Column Connections Subjected to Blast Loads

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Abstract: Although many studies on the blast-resistant performance of structures have focused mainly on single members such as beams and columns, there is little research on the behavior of joints that are subjected to blast loads. In this study, the structural behavior of a slab–column connection subjected to blast load was investigated using a numerical analysis method. LS-DYNA was used as a finite element analysis program, and in order to improve the accuracy of numerical analysis, mesh size, material model, and simulation method of blast load were determined through preliminary analysis. The effect of different restraints of the joints, depending on the position of the columns in the slab, on the blast resistance performance was investigated. As a result, the highly confined slab-interior column connection showed better behavior than other edge and corner columns. The drop panel installed between the lower column and the slab was effective in improving the blast-resistance performance of the slab–column connection. For a more accurate evaluation of blast resistance performance, it was suggested that various evaluation factors such as ductility ratio, reinforcing stress, and concrete fracture area can be considered along with the support rotation, which is an important evaluation factor suggested by many standards.

Keywords: blast loads; slab; column; connections; numerical analysis

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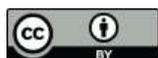
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1. Introduction

As explosive terrorism and explosion accidents continue to occur around the world, research on the behavior of structures under such extreme situations is increasing [1–3]. Most of these studies focus on how single members such as beams, columns, and slabs behave under explosive loads. Since the failure or large deformation of the joint can directly lead to the collapse or malfunction of the entire structural system, the study on the joint behavior is no less important than the study on single members. Although many studies have been conducted on the structural behavior of joints under static and dynamic loads, further studies are still needed to improve the understanding of joint behavior under explosive loads [4–6].

In this study, the joints of the columns and slabs were investigated. In particular, the behavior of the joints under blast load was investigated according to the position of the column. In other words, the effect of different restraints of the joints on the blast-resistance performance was observed as the columns were located on the inside, edge, and corner of the slab. In addition, the behavior of slab–column joint with drop panel, which is a square portion provided above the lower column and below the slab, was investigated. Drop panels that provide increased shear strength and moment resistance are expected to be effective in improving the blast-resistance performance of slab–column joints. The experimental approach with explosives is really challenging, expensive, and difficult. Therefore, in this study, numerical analysis, one of the best options to discuss this

phenomenon, was performed using LS-DYNA, a general-purpose finite element analysis program whose reliability has been verified through many previous studies [7,8].

2. Literature Review

There are some studies on the structural behavior of the slab–column connections according to the position of columns under static loads. Bianchini et al. (1960) performed tests on a total of 45 specimens of the interior, edge, corner, and isolated columns [9]. As a result, the effective strength of the interior column–slab joint was 75% of the column strength and 1.5 times the slab strength. However, when the column strength exceeds the slab strength by 1.4 times, the effective strength of the column–slab joint at the edge and corner columns is not significantly increased by the restraint of the surrounding slab [9]. McHarg et al. (2000) performed a total of 12 test specimens consisting of column–slab specimens and isolated column specimens [10]. The interior column showed greater axial compressive strength than the isolated column due to the restraint effect of the slab, and also showed ductile behavior [10]. Lee et al. (2007) performed column load transmission experiments depending on the position of the column. As a result, the interior column test specimen has improved ultimate load capacity than the isolated column specimen due to the slab restraint effect [11].

There are also many studies on the structural behavior of slab–column connections under dynamic loads. Some of them experimentally confirmed that the amount of flexural reinforcement affects the seismic behavior of the slab–column joint [12–16]. Some studies have investigated the seismic response of slab–column joints with high-strength concrete (HSC) applied to the slab, and they showed that the specimens with HSC had superior performance in terms of ductility and strength, compared to specimens with conventional normal strength concrete (NSC) [17–20]. Scotta and Giorgi (2016) performed cyclic experiments on four full-scale exterior slab–column connections made of normal concrete and fiber-reinforced lightweight aggregate concrete [21]. They reported that the addition of steel fibers to the concrete mix improved the hysteretic behavior of slab–column connections [21]. Several researchers have studied the progressive collapse and robustness of building structures due to column or joint collapse [22–26]. Setiawan et al. (2019) performed numerical analysis on slab–column connections subjected to cyclic loading and captured the characteristics of cyclic degradation observed in experiments with nonlinear finite analysis and suggested a simplified design method for punching shear [27].

As shown above, studies dealing with the structural behavior of joints under dynamic loads mainly focus on seismic loads. While there are relatively many experimental and numerical investigations of reinforced concrete (RC) slabs subjected to blast loading [28–37], there are few studies of slab–column connections. Shahriari et al. (2021) numerically investigated the blast response and progressive collapse of RC structures equipped with viscoelastic dampers [38]. They found that viscoelastic dampers designed for seismic loads resulted in a suitable performance for reducing the response of structures to blast loads [38]. Krauthammer (1999) reported that plastic hinge control through diagonal reinforcing bars can contribute to the improvement of the blast resistance performance of the connections [39]. Lim et al. (2016) have reported the blast-resistance performance of joints of slab-interior and slab-edge columns [40]. However, there is no study on the blast-resistance behavior for the slab-corner column connection and for the relatively large explosive load. Additionally, the method of transmission of the explosive load has not been verified sufficiently in the numerical analysis. Therefore, it is necessary to investigate the blast-resistance performance of slab–column connections depending on the type of the column and the amount of explosive load.

According to the ASCE/SEI (2011), the connection should be designed to resist shear force, axial force, bending moment, and torsion [41]. The effects of rebound are also considered for all connections. The reinforcements in beam–column connections are supposed to comply with details of earthquake-resistant structures according to ASCE/SEI (2011) [41]. There are no other guidelines for the design of blast-resistant slab–column

connections. Especially, the material properties in blast events are different in earthquake conditions because of the difference in strain rate. The strain rate is typically over 10^0 s^{-1} in blast events and 10^{-5} s^{-1} in earthquakes [42]. Therefore, more accurate material properties are needed to design structures subjected to explosive loads comparing to structures subjected to seismic loads.

According to UFC 3-340-02 issued by the US Department of Defense (DoD), support rotation and ductility are selected as criteria for evaluating the structural performance of RC structural members under explosive loads [43]. This criterion assumes that the structural member is effectively resisting the blast load when the support rotation is 2 degrees or less. In addition, ASCE/SEI 59-11 proposes the limit of support rotation for blast-resistant RC structures and also provides a level of protection (LOP) [41]. It is noteworthy that both DoD and ASCE are proposing support rotation as an evaluation factor for the behavior of blast-resistant structures.

3. Numerical Analysis

3.1. Details of Specimens

Details of all slab–column connections are shown in Figure 1. The specimen IC is a statically designed slab–interior column connection. The specimen IC-D has the same reinforcement and shape details as specimen IC, but the drop panel was additionally placed according to ACI 318 (2011) [44] and ACI 352.1R [45]. The specimens EC and CC were designed based on the specimen IC, but the column was located at the edge and corner of the slab, respectively. Accordingly, the specimens EC and CC are confined on three sides and two sides of the connection, respectively, while the specimen IC is confined on all four sides. In general, the effective strength of connections can be improved in static loads when the column in connection is confined by slab [9,46]. To verify the confinement effects, comparative studies were conducted on the behavior of specimens IC, EC, and CC.

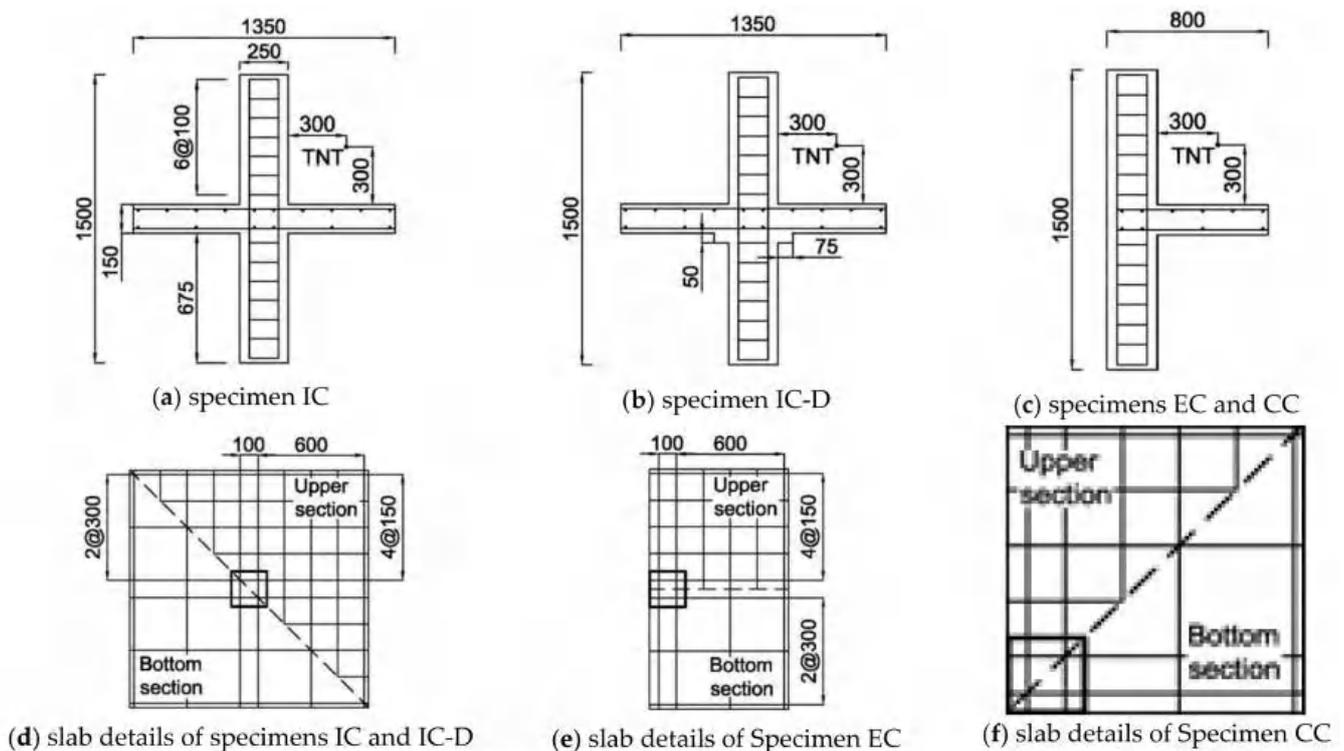


Figure 1. Details of specimens: (a) specimen IC; (b) specimen IC-D; (c) specimens EC and CC; (d) slab details of specimens IC and IC-D; (e) slab details of specimen EC; (f) slab details of specimen CC.

3.2. Modeling of Specimens

In order to obtain accurate numerical analysis results, it is necessary to establish an accurate material model. In this study, Mat_072R3 was selected from the concrete material models provided by the analysis program LS-DYNA. This material model reflects the strain-rate effect and has already been found in several studies in the literature to be suitable for analyzing concrete structures under high strain-rate [47–49]. However, Mat_072R3 was unable to exhibit local damage caused by explosions, such as crater spalls, which are associated with structural failure and erosion [50]. Therefore, to simulate these characteristics, LS-DYNA’s “Add_Erosion keyword” option was applied to the concrete material model. To model the reinforcing bars, LS-DYNA’s Mat_024 was applied, which is defined as an elastic-plastic material with arbitrary stress–strain curve and an arbitrary strain-rate dependency. The fracture of Mat_024 is based on plastic deformation [49].

Numerical analysis results may vary depending on the mesh size of the element [51,52]. According to the previous studies, when simulating a structure subjected to an explosive load, a mesh size of 25 to 30 mm led to the analysis results most similar to the experimental results [33,53]. In this study, before the main analysis was conducted, various mesh sizes were evaluated in terms of accuracy and efficiency of analysis. In the preliminary analysis, the displacement, stress of reinforcing bars, and fracture shape were investigated in the same way as in the main analysis. Considering the analysis results and the time required for analysis, a mesh size of 20 to 25 mm is considered to be the most reasonable. Therefore, in this study, the concrete mesh is composed of 25 mm cubic, 8-node solid elements.

The interaction between concrete and reinforcing bars has a great influence on the behavior of RC structures. In particular, interactions such as bond–slip are very difficult to simulate. A method of tying nodes was recommended to simulate the structure’s actual behavior and to provide the simplicity of analysis [48,54]. In this study, the nodes of the reinforcing bar and concrete are connected to each other to provide accurate structural performance.

The one-point integration method used in this analysis is effectively applied to dynamic analysis due to its relatively short analysis time. However, there is a risk of creating a zero-energy state that causes a negative volume or creating an element that behaves differently from the actual behavior. In the numerical analysis, the volume of solid elements is generally reduced when subjected to the compressive pressure, but when zero energy is generated inside the solid elements, the negative volume occurs due to the abnormal operation of the element, resulting in an increase in volume, as shown in Figure 2. In this case, the amount of internal energy loss is called hourglass energy. When the hourglass energy is largely generated, it is difficult to ensure the accuracy of the analysis [55–57]. Therefore, LS-DYNA’s “Hourglass” option, which can control the accuracy of analysis due to this phenomenon, was applied to the material model [48,58].

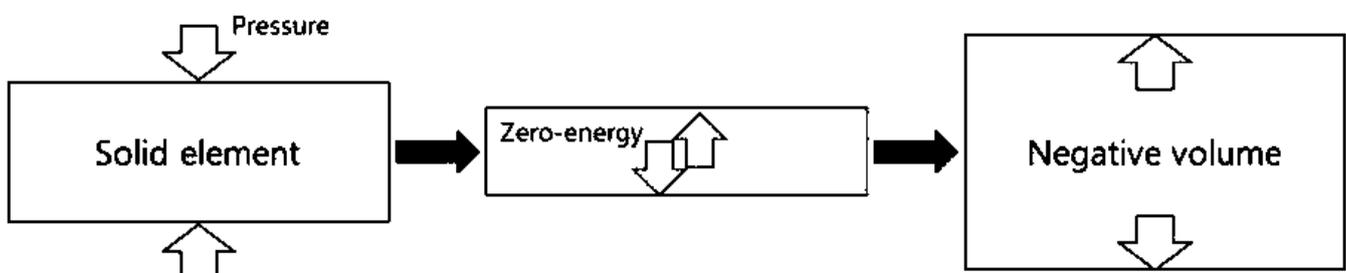


Figure 2. Process of the negative volume.

3.3. Modeling of Blast Loads

In this study, a preliminary analysis was performed to select the analysis method between multi-material arbitrary Lagrangian–Eulerian (MME) and load-blast enhanced (LBE). The concrete walls, which have the same characteristics as the main analysis including

materials and element size, were analyzed. As in the main analysis, 4 kg of TNT was placed at a vertical distance of 300 mm from the center of the structure surface. The surfaces for LBE were defined as the front of a wall that was directly affected by the explosion. Table 1 and Figure 3 show the analysis results from both methods of MME and LBE. The maximum pressures of MME and LBE were 3.97×10^{-7} MPa and 3.33×10^{-7} MPa, respectively. The area of the pressure curve of MME was larger than that of LBE, as shown in Figure 3. For the LBE method, the explosive pressure was directly applied to the element surfaces. For the MME, however, since the explosion load at the origin was transmitted through the atmosphere elements, the residual pressure was transmitted through the atmosphere after the maximum explosion pressure. Therefore, as shown in Figure 3, the pressure curve area of the MME was larger than the pressure curve area of the LBE, although the maximum pressure did not show a large difference. Table 1 compared the duration time of both analysis methods. The LBE method could be regarded as a more efficient explosion analysis method since the analysis time of MME was about 90 times longer than that of LBE. As a result, it is considered that the MME method is suitable for understanding the flow and progress of the explosive pressure, and the LBE method is suitable for understanding the effect of the maximum pressure on the structure under the explosive load. Therefore, in this study, the LBE method was chosen considering that the maximum pressure is similar to MME and it is more efficient in terms of analysis time. Moreover, many research studies showed that the LBE method is more efficient than the MME method considering analysis results and time [59–61].

Table 1. Comparison of analysis time of MME and LBE.

Variables	Number of Elements			Analysis Time	Duration of Analysis
	Structures	Air	TNT		
LBE	800	-	-	129 s	100 ms
MME	800	56075	125	11,075 s	

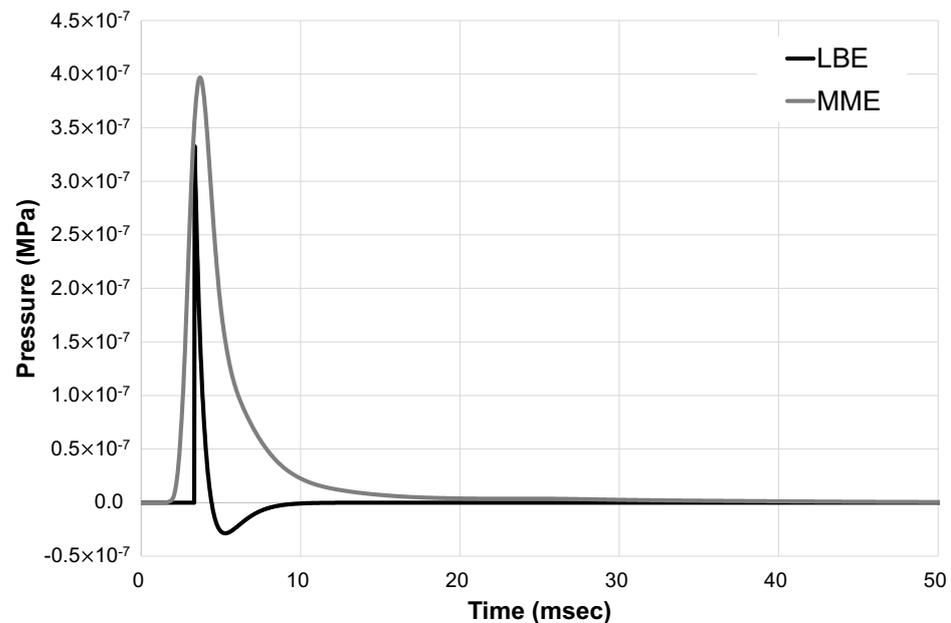


Figure 3. Comparison of pressure histories of MME and LBE.

Based on empirical formulas of blast loads, the Protective Design Center (PDC) of US ARMY releases the Conventional Weapons Effects (ConWep), which could perform a variety of conventional weapons effects from TM 5-855-1 [58,62]. The LS-DYNA applies the ConWep system to LBE. The blast loads of TNTs were located 300 mm from the column and

slab. The variables for the amount of TNT were 4 kg and 12 kg. The mass of TNT used in this analysis represents the small and large briefcase bomb, as shown in Table 2 [63]. When 4 kg of TNT was used, which is equal to the amount used in a small briefcase, the blast resistance behavior could be well observed because the slab was not completely destroyed. On the other hand, when 12 kg of TNT was used, which is equal to the amount used in a large briefcase, a significant portion of the slab where the explosive load was placed was destroyed. Table 3 summarized the descriptions of the specimens including blast loads.

Table 2. Typical example of terrorist explosive materials [63].

Explosion Method	Material Type	Loaded Weight
Small briefcase	Military commercial bomb such as TNT	2~4 kg
Large briefcase		4~12 kg
Suitcase		12~22 kg
Bicycle		30 kg

Table 3. Specimen descriptions.

Specimen	Description	Charged Weight of TNT
IC4	Slab-interior—column connection	4 kg
IC12		12 kg
IC-D4	Slab-interior—column connection reinforcing with drop panel	4 kg
IC-D12		12 kg
EC4	Slab-edge—column connection	4 kg
EC12		12 kg
CC4	Slab-corner—column connection	4 kg
CC12		12 kg

4. Analysis Results

From the analysis results, typical forms of pressure distribution were commonly observed in every specimen, as shown in Figure 4. When the explosive load was applied, high compressive forces were generated in the slabs and columns directly affected by the explosion load. Then, the overpressure spread spherically through the slab–column connection. The analysis end time was set to 3000 ms, which is the time at which deformation of all specimens was found stable. As shown in Figure 5a–d, for specimens subjected to 4 kg of TNT, spalling on the rear face of the slab, was severer than that on the front face. These phenomena of pressure development and spalling are quite similar to previous researches [33,43,53]. Looking at the fracture pattern of the specimen under 12 kg of TNT, the part of the slab where the explosion load was placed was completely lost, as shown in Figure 5e–h.

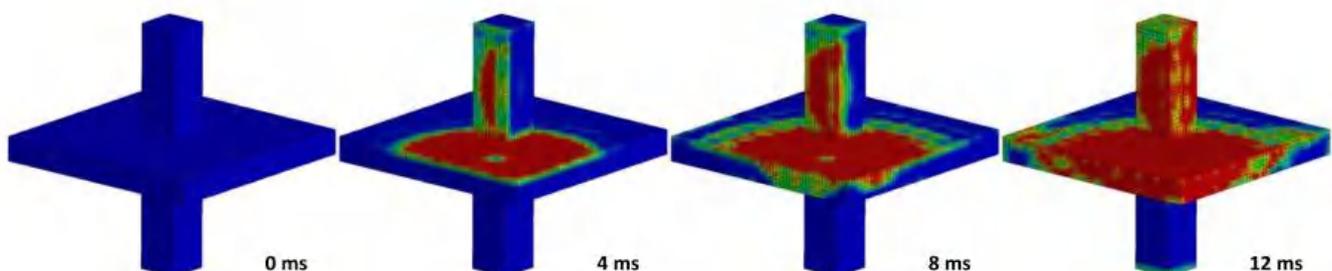


Figure 4. Typical forms of pressure distribution.

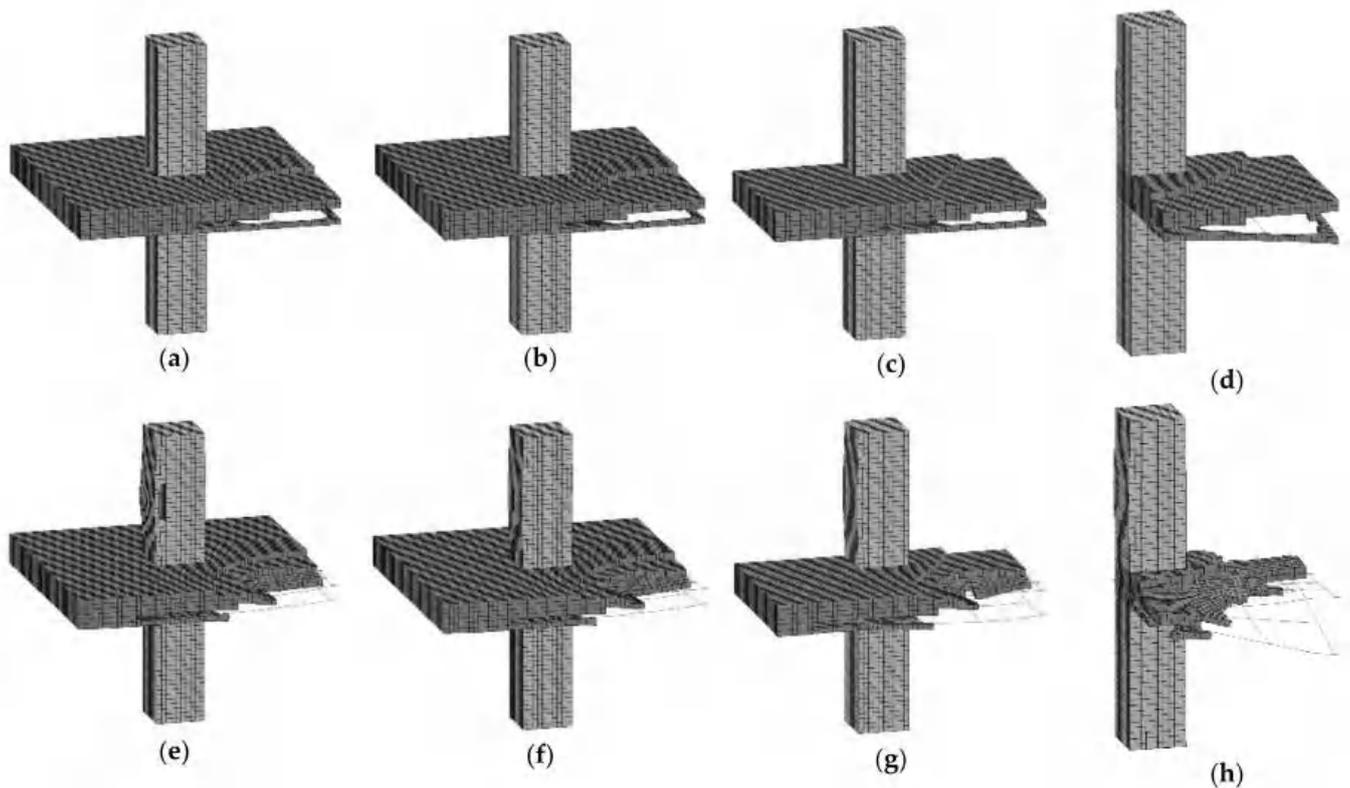


Figure 5. Failure shape of slab–column connections: (a) specimen IC4; (b) specimen IC-D4; (c) specimen EC4; (d) specimen CC4; (e) specimen IC12; (f) specimen IC-D12; (g) specimen EC12; (h) specimen CC12.

4.1. Slab Deflection

Figure 6 shows the deflection of the slab subjected to TNT 4 kg along the diagonal distance away from the corner of the column. The maximum deflections of the slab occurred similarly in every specimen with TNT 4 kg. The deflection of the slab increased rapidly from the point about 350 mm away from the corner of the column. This point is similar to the point where the fracture of the slab occurred. Figure 7 shows the slab deflections for TNT 12 kg. Large deflections were observed at about 150 mm away from the corner of the column, and beyond that point, fracture of the slab was observed. As the larger explosive load was applied, the fracture area was much larger than that of 4 kg TNT applied specimens.

When comparing the effective deflection of the unbroken part of the slab, it was confirmed that the drop panel slightly reduced deflection. However, for all specimens with TNT of 4 kg and 12 kg, comparing the specimens IC, EC, and CC, the slab deflection according to the position of the column showed no significant difference.

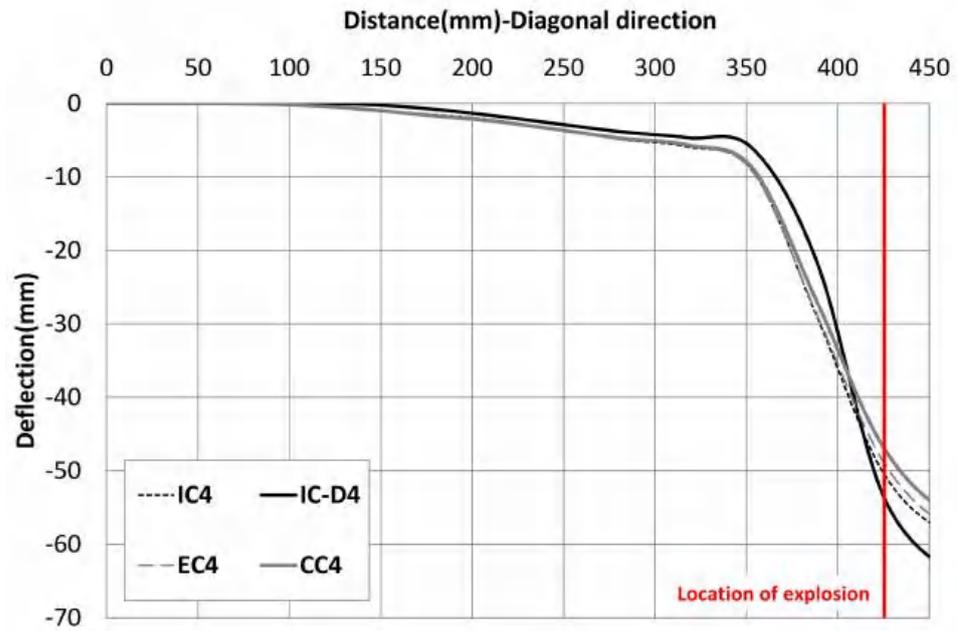


Figure 6. Slab deflection of specimens with TNT 4 kg.

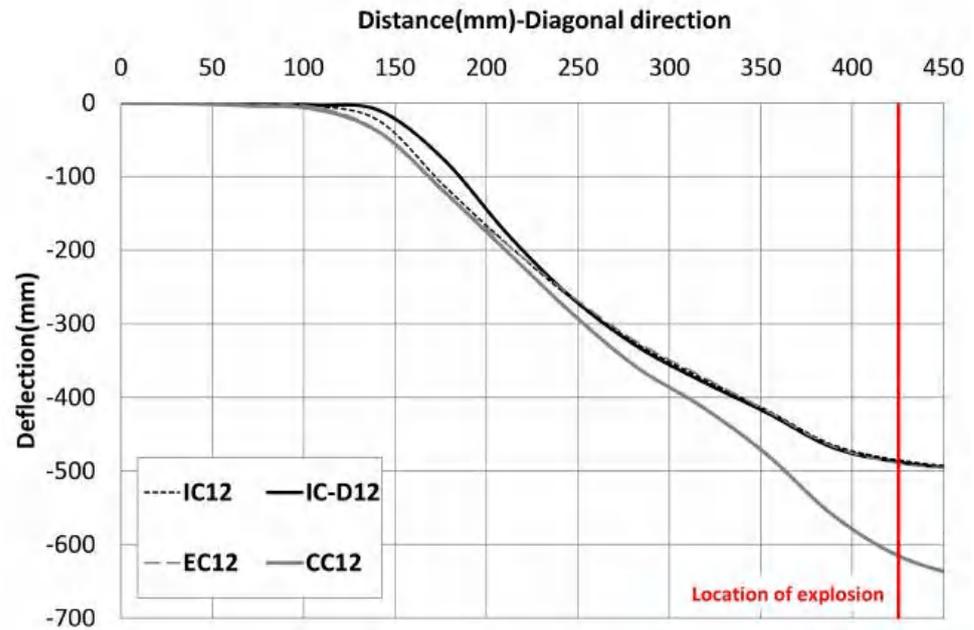


Figure 7. Slab deflection of specimens with TNT 12 kg.

4.2. Critical Section in Slab

In slab–column connections, a section that is 1/2 of the effective depth, d , from the outer surface of the column is regarded as a critical section [45]. Sufficient safety must be ensured for critical sections to prevent the collapse of the entire structure due to large damage of the joints [44,64]. Figure 8 shows failure shapes of slabs for specimens subjected to 12 kg of TNT. For specimens EC and CC, spalling due to the blast load occurred over the critical section, but specimens IC and IC-D showed a more positive structural behavior in which spalling did not spread to the critical section. Table 4 shows the deformation and support rotation at the critical section for all specimens. The deformations in the critical section of the specimens IC-D4 and IC-D12 were 0.040 mm and 3.284 mm, which were the least deformations among the specimens subjected to the same blast load. Figures 9 and 10 show the deflections in the critical section. In Figure 9, the specimens, except for specimen

IC-D4, show similar behavior. The inflection points of deflection curves occurred within the critical sections for all specimens except the specimens reinforced with drop panels. In other words, the safety of the critical section was enhanced by the drop panel. Therefore, the drop panel can be considered as a method to effectively resist blast loads.

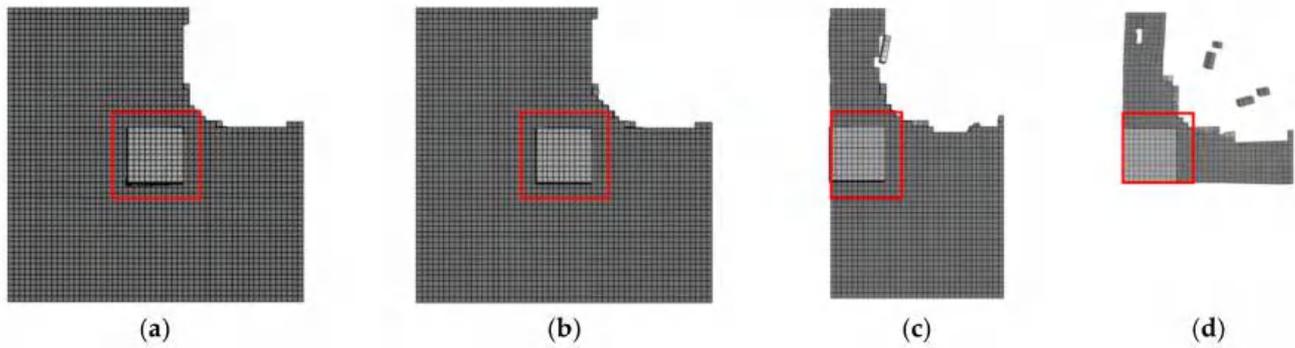


Figure 8. Critical section and slab failure shapes of specimens with TNT 12 kg: (a) IC12; (b) IC-D12; (c) EC12; (d) CC12.

Table 4. Maximum deflection and support rotation at critical section.

Specimen	Deflection (mm)	Support Rotation (°)
IC4	0.294	0.159
IC-D4	0.040	0.021
EC4	0.326	0.176
CC4	0.244	0.132
IC12	3.941	1.277
IC-D12	3.284	1.064
EC12	5.190	1.682
CC12	8.297	2.687

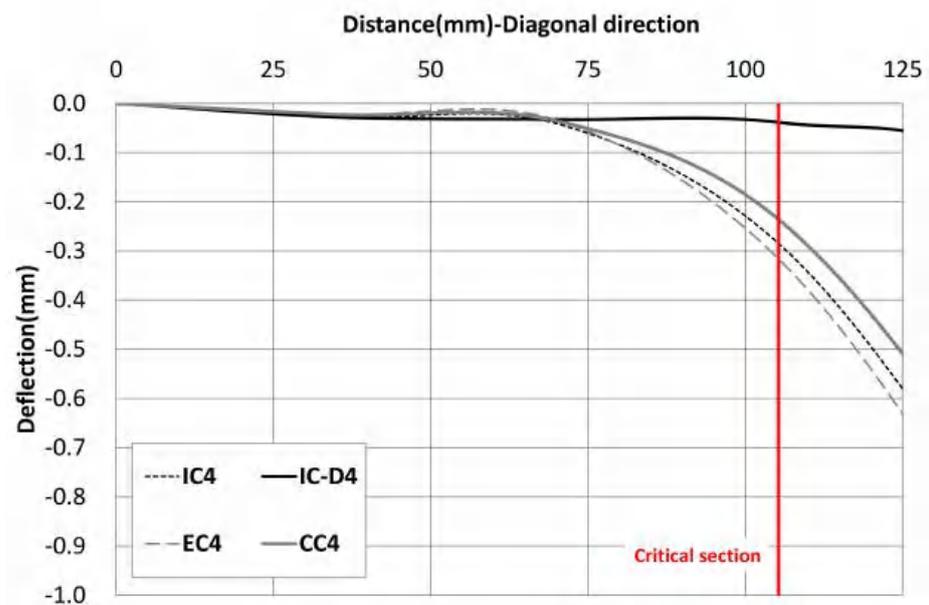


Figure 9. Slab deflection at the critical section with TNT 4 kg.

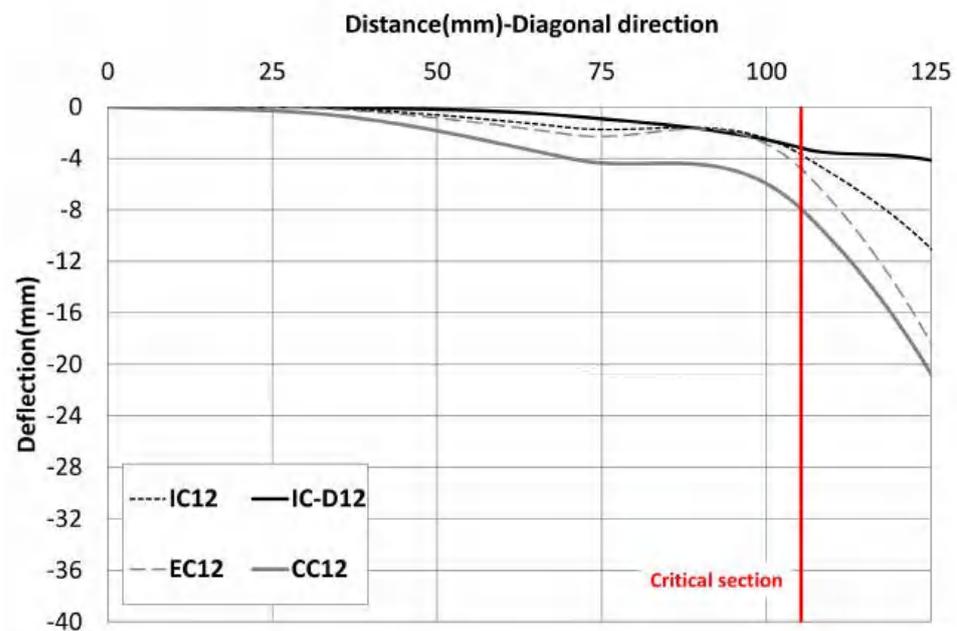


Figure 10. Slab deflection at the critical section with TNT 12 kg.

According to the criteria, the limit of support rotation to effectively resist the moment is two degrees [41,43]. However, according to the analysis results, it was found that this evaluation criterion alone was insufficient to adequately represent the blast-resistant performance of the member. This is because the support rotation in the critical section of all specimens was below the criteria limit of two degrees, but in reality, most specimens showed that the slab was destroyed. Therefore, in order to more accurately and reasonably evaluate blast-resistant performance, it is necessary to consider various evaluation factors in addition to support rotation.

4.3. Steel Stresses in Slab

The peak stresses in the reinforcement of the slab are shown in Table 5. In every specimen with TNT 4 kg and 12 kg, the peak rebar stress occurred near the explosive load. Although a large fracture occurred in the slab, the peak stresses of all reinforcing bars did not reach the maximum strength to which dynamic increased factor was applied. This phenomenon is believed to occur because the concrete is destroyed by the explosive energy at a moment and the energy is not sufficiently transmitted to the reinforcing bar. However, in the previous study, it has been confirmed that the reinforcing bars affect the blast-resistance capacities in the beam–column connections [65].

Table 5. Peak stresses in reinforcements of the slab.

Specimen	Peak Stress (MPa)	
	Top Reinforcing Bar	Bottom Reinforcing Bar
IC4	32.40	157.04
IC-D4	33.93	164.08
EC4	29.52	162.08
CC4	31.17	137.43
IC12	122.62	298.52
IC-D12	128.38	279.72
EC12	176.12	312.99
CC12	440.92	440.48

For most specimens, the stresses of bottom reinforcements of the slab were larger than those of top slab reinforcements. This phenomenon is due to the failure mode in which

the rear fracture was more severe than the fracture of the front face to which the explosive load was applied. The effect of the drop panel on the reinforcement stress was not clearly observed, considering that the slab reinforcement stresses of IC and IC-D were similar.

4.4. Column Behavior

Table 6 shows the maximum horizontal deformation of columns. When comparing column behaviors of specimens IC and IC-D, the drop panel was found to be effective in controlling the horizontal displacement of columns under explosive load. Comparing CC, EC, and IC specimens, the larger the constraint of the column by the surrounding slab was, the less horizontal displacement of the column was observed. In the case of CC12, a very large displacement occurred at the end of the analysis, and the column of specimen CC12 seems to have been destroyed as displacement shows a continuous trend of increase. Therefore, it is necessary to pay special attention to the blast-resistance performance for the relatively large explosive load in the case of a corner column having a low constraint by a surrounding slab.

Table 6. Maximum horizontal deformation of the column.

Specimen	Column Deformation (mm)	
	TNT 4 kg	TNT 12 kg
IC	0.034	0.104
IC-D	0.033	0.099
EC	0.035	0.171
CC	0.038	12.91

5. Conclusions

The blast resistance of slab–column connection was numerically analyzed. The confinement effect of connection on the blast resistance was investigated through a comparison of the slab-interior column, slab-edge column, and slab-corner column. In addition, the effect of the drop panel on the blast resistance performance was investigated. The conclusions from this numerical study are as follows:

- (1) Analysis results showed that the slab-interior column had a better performance than the slab-edge column and slab-corner column in terms of slab failure at critical section and column deformation. The confinement effect seems to be considered in the design of blast-resistant structures. However, the effect of the position of the column on the behavior of the slab such as slab deflection and support rotation under explosive load was not apparent. Further research is needed with the location of the explosive load and the dimensions of columns and slabs as variables.
- (2) The drop panel was observed to contribute to the improvement of the blast-resistance performance. For 4 kg and 12 kg of TNT, the drop panel reduced the maximum deflection of the slab at the critical section by approximately 86% and 17%, and the column deformation by approximately 2.9% and 4.8%, respectively.
- (3) Although significant concrete fracture occurred in the slab, the maximum stress of the reinforcing bar did not reach the tensile strength. This phenomenon occurs because the concrete is momentarily destroyed by the explosive energy and the energy is not sufficiently transmitted to the reinforcing bars. Further research is needed to ensure that the blast energy can be sufficiently transmitted to the rebar through the concrete.
- (4) For most design criteria, the support rotation has been considered as a major criterion for blast-resistant capacities. It is a very simple and good evaluation factor representing the critical behavior of the joint. However, in this study, considerable failure occurred in the slab member even though the support rotations at the critical section were satisfied with the criteria. Therefore, for a more accurate evaluation of blast resistance performance, various evaluation factors such as ductility ratio, reinforcing stress, and concrete fracture area can be considered along with the support rotation.

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Article

Corner Convergence Effect of Enclosed Blast Shock Wave and High-Pressure Range

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Abstract: An explosion inside a cabin will converge at the corners to form high-pressure areas, significantly impacting the destruction of a bulkhead structure. This paper investigates shock wave convergence characteristics at the corners when the explosive detonates at the center of the cabin, based on a combination of the wall reflection law for shock waves and a numerical simulation method. The parameter K represents the aspect ratio of the cabin structure. This study shows that when $1 \leq K \leq 1.19$, the high pressure at the corner is caused by the superposition of Mach waves along both wall surfaces. However, for the initial shock wave, when $1.2 < K \leq 2$, the high pressure is caused by the superposition of Mach waves along the longer wall surface and regular reflected waves on the shorter wall surface; when $2 < K$, the cause are Mach waves along the longer wall surface and the corresponding positive reflection on the shorter wall surface. The influence of K on the range for the high-pressure region at the corner is also analyzed, the functional relationship between the range of the high-pressure area and K is given, and the universality is verified.

Keywords: internal explosion; shock wave; corner; structural dimensions; Mach waves

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1. Introduction

Internal explosions can cause more significant destruction to structures than air explosions due to the combined effect of the reflection, superposition, and convergence of shock waves [1]. High-pressure shock waves can cause structural damage [2]. The pressure peak resulting from the superposition and convergence effect of a shock wave at the corner of a cabin during implosion is significantly higher than the reflected shock wave at the same distance from the wall in an open environment, which can cause a local tear in the corner of the cabin structure first and then expands to the destruction of the entire bulkhead. Therefore, studying the convergence effect of the shock wave at the corner of the cabin during implosion is critical. It is necessary to understand the high-pressure formation rules and the factors influencing the convergence effect at the corner of the implosion shock wave to guide the design of the protection of the cabin structure against internal explosion, and it is also of importance for shock-wave experiments to determine the Hugoniot and melting curves of metals [3,4].

Explosions inside chambers have been a hot topic of research [5–9]. There are a few specific reports on internal blast wave loading [10]. Shock waves have a significant convergence effect at the corners under internal blast conditions [11]. A combined experimental and numerical simulation study [12,13] of the characteristics and typical destruction modes of cabin structures under implosion loads showed that the intensity of the converging shock waves at the corner of two-wall and three-wall surfaces was, respectively, 5- and 12-times greater than the reflected shock waves on the same region of the wall and that the primary failure mechanism of the bulkhead structure during the implosion of the cabin was tearing failure along the corner. Another numerical simulation study of the load situation under implosion conditions showed that the peak pressure at the corner of the three-wall surface was 9–12 times greater than the pressure at the center of the bulkhead, and the

peak pressure at the corner of the two-wall surface was 3–5 times greater than the pressure at the center of the bulkhead [14]. An experimental study measuring the pressure at the corner of a two-wall surface and the peak pressure at the center of the bulkhead showed that the peak pressure at the two walls was smaller than the pressure at the center of the bulkhead [15]. Another study measuring the peak value of the shock wave for three-corner structures (flat plate, concave plate, and convex plate transition connections) at different doses and the peak value of the initial shock wave showed that the corner structure could retard the convergence effect of shock waves at a low dosage [16]; however, when the dosage was higher, the corner structure did not significantly retard the convergence effect. The maximum ratio of the corner converging shock waves to the initial shock waves was 1.24. In addition, another study was reported using an imaging method to explain the convergence effect of corner shock waves [17], whose angle of incidence was the same as the angle of reflection. The actual reflection of the shock wave in the cabin includes the regular oblique reflection and Mach reflection, which was also considered significant. The analysis confirmed the convergence effect of the shock waves at the corner. However, compared to the studies mentioned above, there was a difference between the peak pressures of the convergence of the shock wave at the corner. The study further suggests that the corner convergence occurs at a specific corner area and that the difference in the results is owing to differences in experimental and simulation measurement points. Therefore, it is necessary to investigate the problem of defining the corner convergence area. In addition, the formation of high-pressure areas at the corners and the associated factors have not yet been determined and must be studied in detail.

The characteristics of the explosion load in the enclosed space depend mainly on the spatial dimensions of the structure [18,19]. This paper determines the causes of the convergence phenomenon of shock waves at the corner by 2D cross-sectional analysis with aspect ratio variation. Furthermore, the peak pressure contour map of the corner area at different aspect ratios was plotted through extensive simulation calculations to make a preliminary determination of the range of the high-pressure area at the corner and obtain the functional relationship between the high-pressure area and the size of the structure.

The research in this paper is based on the following three points:

- (1) The explosives are in the center of the cabin;
- (2) The structure is assumed to be a rigid wall;
- (3) The focus is on the peak of the shock wave only.

2. Simulation Model

2.1. Model Design

When the explosive detonates in the center of the cabin, an arbitrary surface is chosen through the location of the explosion point to intersect, giving the 2D rectangular cross-sectional diagram shown in Figure 1. The dual study of the spread of shock waves on the cross-section and the convergence effect at the corners simplifies the calculation and has a general character. Another study [20] adopted the same method in determining the influential factors for implosion loads.

Figure 2 shows the 2D schematic diagram used in the simulation model. Set wall A as the long side and wall B as the short side. The convergence phenomenon at the corners is studied on half of the cross-section, where a and b are halves of the long and short sides, respectively. The red line area in Figure 2 is the corner area, a square with side length b , and the angle between the shock front and the wall surface is Φ . Each side of the corner area is equally divided into 10 parts, and pressure measurement points are set at the intersections, giving a total of 121 side points arranged as shown in Figure 3.

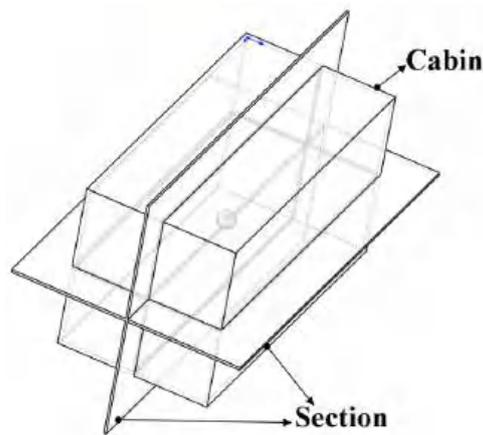


Figure 1. Cross-section through the center of the cabin.

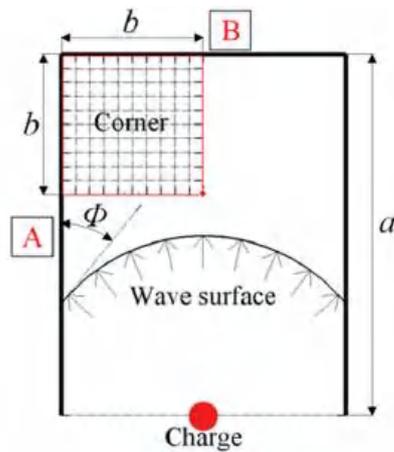


Figure 2. Two-dimensional schematic diagram of the calculation model.

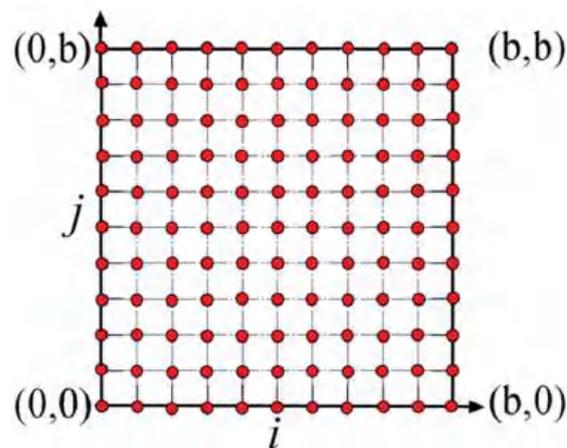


Figure 3. Coordinate distribution of measurement points.

The simulation model was run using AUTODYN-2D. The eulerian unit was used for air and the explosives were packed into the air unit. The initial rigid boundary conditions in AUTODYN are adopted for the air boundary to establish a 2D symmetrical model, as shown in Figure 4. The finite element model uses $0.5 \text{ mm} \times 0.5 \text{ mm}$ mesh. Simulations were carried out using 0.5 mm, 1 mm, 2 mm, and 4 mm meshes for the shock wave of a

100 g charge at 1 m, indicating that the simulation results converged when the mesh size was 0.5 mm, as shown in Figure 5. The ideal gas equation of state is used for air:

$$P = (\gamma - 1)\rho e \tag{1}$$

where γ , ρ , and e are the specific heat capacity, density, and internal energy of the air, respectively, and the values used for the simulation are $\gamma = 1.4$, $\rho = 1.225 \times 10^{-3} \text{ g/cm}^3$, and $e = 2.068 \times 10^5 \text{ J}$.

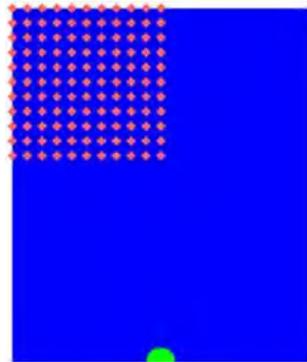


Figure 4. Finite element simulation model diagram.

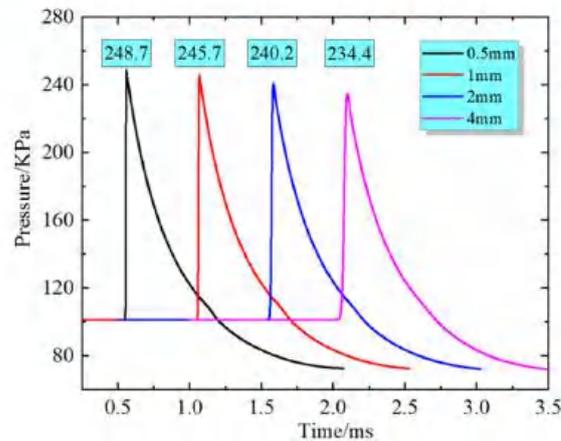


Figure 5. Grid convergence checking.

The Jones–Wilkins–Lee (JWL) equation of state is used for the explosive:

$$P_T = C_1 \left(1 - \frac{\omega}{r_1 v}\right) e^{-r_1 v} + C_2 \left(1 - \frac{\omega}{r_2 v}\right) e^{-r_2 v} + \frac{\omega e}{v} \tag{2}$$

where C_1 , C_2 , r_1 , r_2 , and ω are constants, P_T , v , and e are the pressure, relative volume, and initial energy, respectively. The specific parameters of trinitrotoluene (TNT) are shown in Table 1.

Table 1. Parameters of TNT in the JWL equation of state.

Density, ρ (kg/m ³)	Detonation Velocity, D (m/s)	C-J Pressure (Pa)	C_1 (Pa)
1630	6800	2.10×10^{10}	3.74×10^{11}
C_2	r_1	r_2	ω
3.75×10^9	4.15	0.9	0.35

2.2. Simulation Model Verification

A related study by Isabelle Sochet [21] investigated an explosion in a partially confined space under different boundary conditions using the 0.106 g equivalent of TNT using gas explosives and obtained a time-history curve of shock wave pressure at each measurement point. The experimental arrangement diagram is shown in Figure 6. This paper uses some of these experimental results to verify the simulation model. The simulation determines the time-history curve of the pressure at measurement points A, B, and C when only one, two, and three walls are available. The model parameters and grid size used in the simulation are identical to those used in Section 2.1. A comparison of the simulation results with the experimental results is shown in Figure 7.

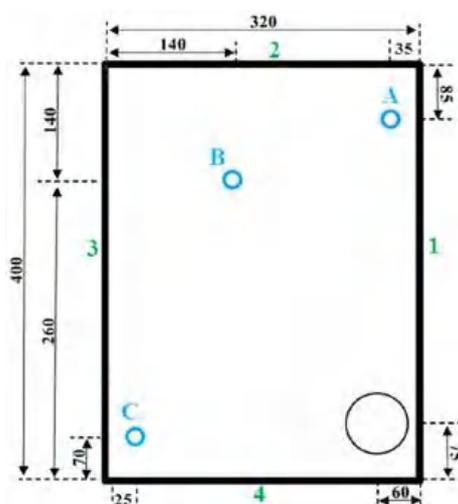


Figure 6. Isabelle Sochet experimental layout diagram.

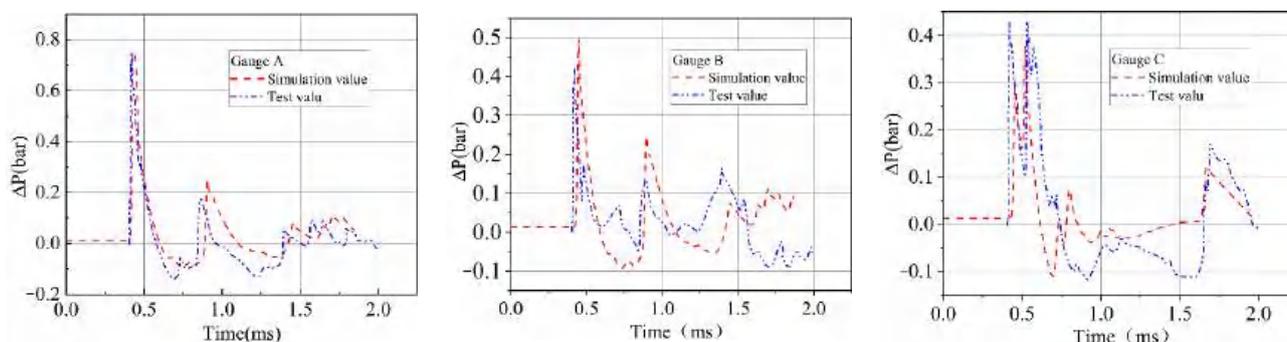


Figure 7. Comparison of numerical simulations with experimental results.

From the comparison of the simulation model and experimental results, the pressure peak and the curve change trend are generally consistent, the pressure peak error at B is larger, and the maximum error is 12% which is within the acceptable range and thus, can verify the reliability of the simulation model.

2.3. Simulation Working Arrangement

The parameter K represents the ratio of the half of the long side, a, to the half of the short side, b, in Figure 2, viz., $K = a/b$, which is a dimensionless number used to represent the change in size of the structure. In this study, K is in the range of 1 to 5, and the particular values 500 mm and 1000 mm are used for b. Furthermore, the explosive equivalents of 100 g, 200 g, 500 g, and 1000 g TNT are used. Table 2 gives the specific working conditions.

Table 2. Simulation working conditions.

Serial Number	K	a (mm)	b (mm)	W (g)
1	1	500	500	
2	1.2	600	500	
3	1.4	700	500	
4	1.6	800	500	
5	1.8	900	500	
6	2.0	1000	500	
7	2.2	1100	500	
8	2.4	1200	500	100
9	2.6	1300	500	
10	2.8	1400	500	
11	3.0	1500	500	
12	3.4	1700	500	
13	4	2000	500	
14	4.4	2200	500	
15	5	2500	500	
16	1.1	550	500	
17	1.3	650	500	
18	1.5	750	500	
19	1.7	850	500	
20	1.9	950	500	
21	2.1	1050	500	200
22	2.3	1150	500	
23	2.5	1250	500	
24	2.7	1350	500	
25	2.9	1450	500	
26	1.1	1100	1000	
27	1.3	1300	1000	
28	1.5	1500	1000	
29	1.7	1700	1000	
30	1.9	1900	1000	
31	2.1	2100	1000	1000
32	2.3	2300	1000	
33	2.5	2500	1000	
34	2.7	2700	1000	
35	2.9	2900	1000	

3. Mechanism of High-Pressure Formation at Corners

3.1. Theoretical Analysis of Convergence Effects at Corners

The spread of an explosive shock wave inside the cabin is complex, characterized by multiple reflections and superpositions, and follows the wall reflection principle. The shock wave reflection at the wall comprises positive and oblique reflections, with the oblique reflections including both regular and Mach reflections [22]. Figure 8 is a schematic diagram of the wall reflection during an air explosion, where d is the vertical distance from the explosive to the wall, c is the distance from the projection point of the explosive on the wall to the intersection of the shock wave front and the wall, Φ is the angle of incidence of the shock wave on the wall, and θ is the included angle in the vertical direction between the shock wave front and the wall intersection line. The geometric relationship shows that $\theta = \Phi$, thus $\tan\theta = c/d$, $c = d\tan\theta = d\tan\Phi$, and the Mach angle tends to a limiting value of 39.97° [23]. Therefore, when $c/d \geq 0.838$, Mach reflection occurs.

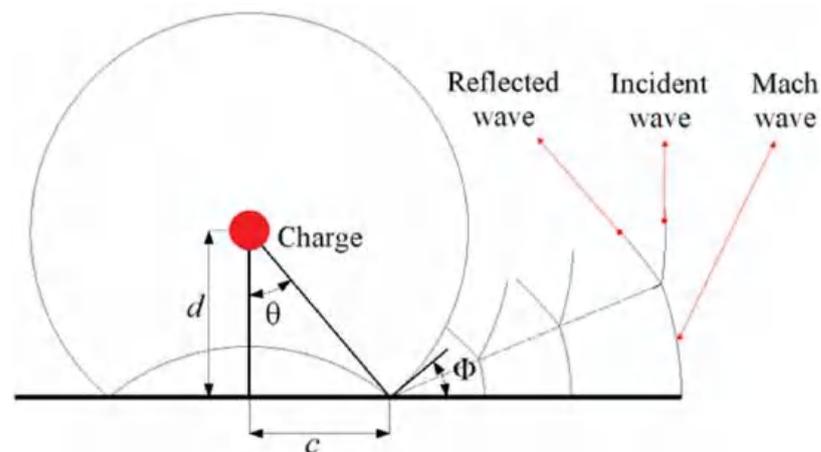


Figure 8. Schematic diagram of shock wave wall reflection.

In analogy to Figure 2, the distance from the explosive to bulkhead A is b , and the distance from the explosive to bulkhead B is a . When $K = a/b = 1 > 0.838$, the reflected shock waves on walls A and B will form Mach reflections before they reach the point $(0, b)$. At point $(0, b)$, the initial shock wave and the Mach reflected waves from walls A and B converge, forming a converging wave. When $K > 1$, the reflected waves from wall A also form Mach waves before they reach the point $(0, b)$. However, for the reflected waves from wall B, the Mach reflection is formed when $b/a \geq 0.838$, i.e., when $a/b \leq 1/0.838 = 1.193$. Thus, when $1 \leq K \leq 1.193$, the initial shock wave at point $(0, b)$ and the Mach reflected waves from walls A and B converge to form a high-pressure region. When $1.193 < K$, there is no Mach reflection on wall B. Therefore, the convergence at the point $(0, b)$ is owing to the initial shock wave, the Mach reflection wave from wall A, and the regular reflection wave from wall B. The Mach wavefront gradually widens during its spread, as shown in Figure 8, thus a value exists for n . When $K \geq n$, the Mach reflection wave along wall A reaches point $(0, b)$ first, while the initial shock wave superimposes with the Mach wave from wall A in its spread towards point $(0, b)$ and spreads along the three-wave line to wall B without converging at the corner. The simulations in the next section were used to verify the above inference and determine the value of n .

3.2. Simulation of Convergence Effects at Corners

Several simulations were conducted with variations in K , as shown in Table 2. As the spread of waves is mainly related to the size of the structure, the convergence clouds of waves at the corner for $K = 1, 2, 3, 4$, and 5 are exemplified in cases when $b = 500$ mm and $W = 100$ g of explosive, as shown in Figure 9.

As observed in the diagram, when $K = 1$, the converging waves at the corner from the waves reflected at walls A and B and the initial shock waves do not form a noticeable Mach rod phenomenon owing to the short distance between the walls; however, as K increases, the Mach wave on the surface of wall A gradually widens, and a clear Mach rod is observable. As the Mach wave speed is faster than the initial shock wave speed, its wavefront surface gradually flushes with the initial shock wave and surpasses it. Therefore, the high pressure formed near the corner $(0, b)$ comes from the reflection of the Mach wave at the surface of wall B. The above deductions support the theoretical analysis in Section 3.1. For the value of n given in Section 3.1, the simulation shows that when $K \geq 2$, the high pressure at the corner $(0, b)$ mainly comes from the positive reflection of Mach waves from wall A to wall B. An example of the spread of the Mach wave and the initial shock wave, when $K = 2.4$, is shown in Figure 10, where the black dashed line depicts the three-wave line. The high pressure at the corner $(0, b)$ is formed by the positive reflection of the Mach wave.

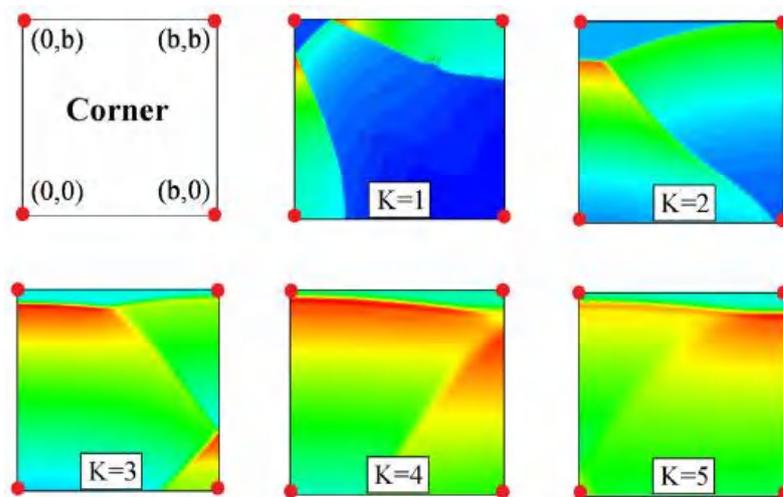


Figure 9. The cloud diagram of the convergence effect of shock waves at the corner at different K values when $b = 500$ mm and $W = 100$ g.

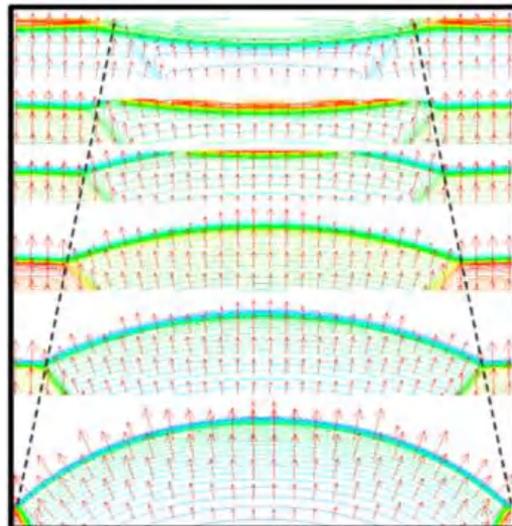


Figure 10. The formation process of the convergence effect of shock waves at the corner when $K = 2.4$.

In previous experiments [2–5], the peak pressure at points $(0, b)$ and (b, b) was frequently used for comparison. The relationship between the peak pressure at the two measuring points and K when $b = 500$ mm and $W = 100$ g and the peak pressure ratio at $(0, b)$ and (b, b) is shown in Figure 11. Figure 11 shows that the value for $P(b, b)/P(0, b)$ tends to be stable when K is in the range of 1 to 2.5, but as K increases, the value for $P(b, b)/P(0, b)$ drops linearly, indicating that the corner convergence effect is weakening.

To better demonstrate the changes in the corner high-pressure area with K, the peak pressure distribution in the corner at $b = 500$ mm and $W = 100$ g when $K = 2$, $K = 3$, $K = 4$, and $K = 5$ is shown in Figure 12. The graph shows that K significantly influences the corner’s high-pressure area. As K increases, the high-pressure area at the corner gradually expands and moves towards the vicinity of the short side center (b, b) and finally disappears.

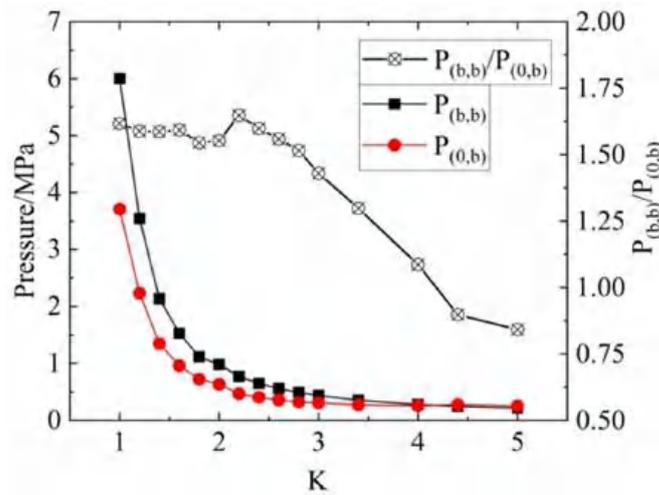


Figure 11. Changes in the pressure and pressure ratio at two lateral points with K when $b = 500$ mm and $W = 100$ g.

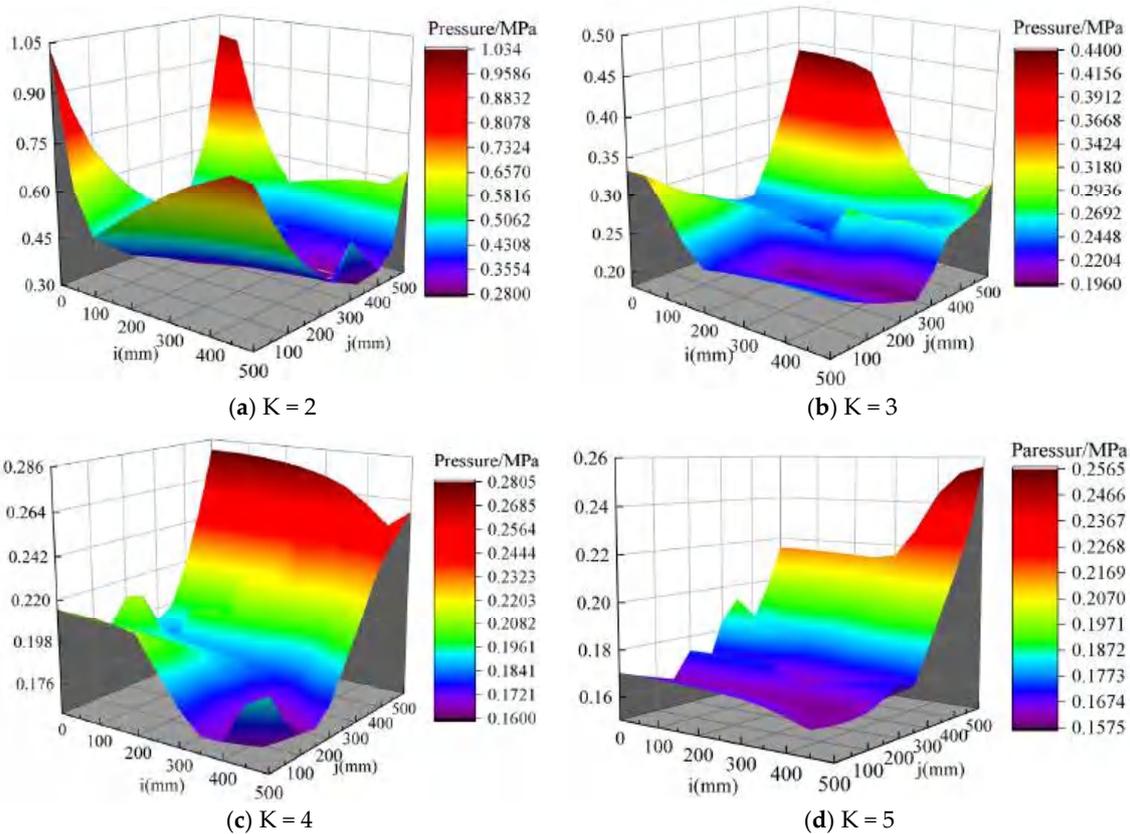


Figure 12. Distribution of peak pressure in the corner area at different values of K.

4. Determination of the Range of the High-Pressure Area at the Corner

The formation of high pressure at the corner has been studied previously but further study into the range of high-pressure areas is also crucial to understanding the convergence effect of shock waves at the corners. Therefore, the pressure peaks at each measurement point in Figure 3 were recorded and pressure contour maps were plotted. Figure 13 shows the pressure contour map when $b = 500$ mm, $W = 100$ g, and $K = 2, 3, 4,$ and 5 .

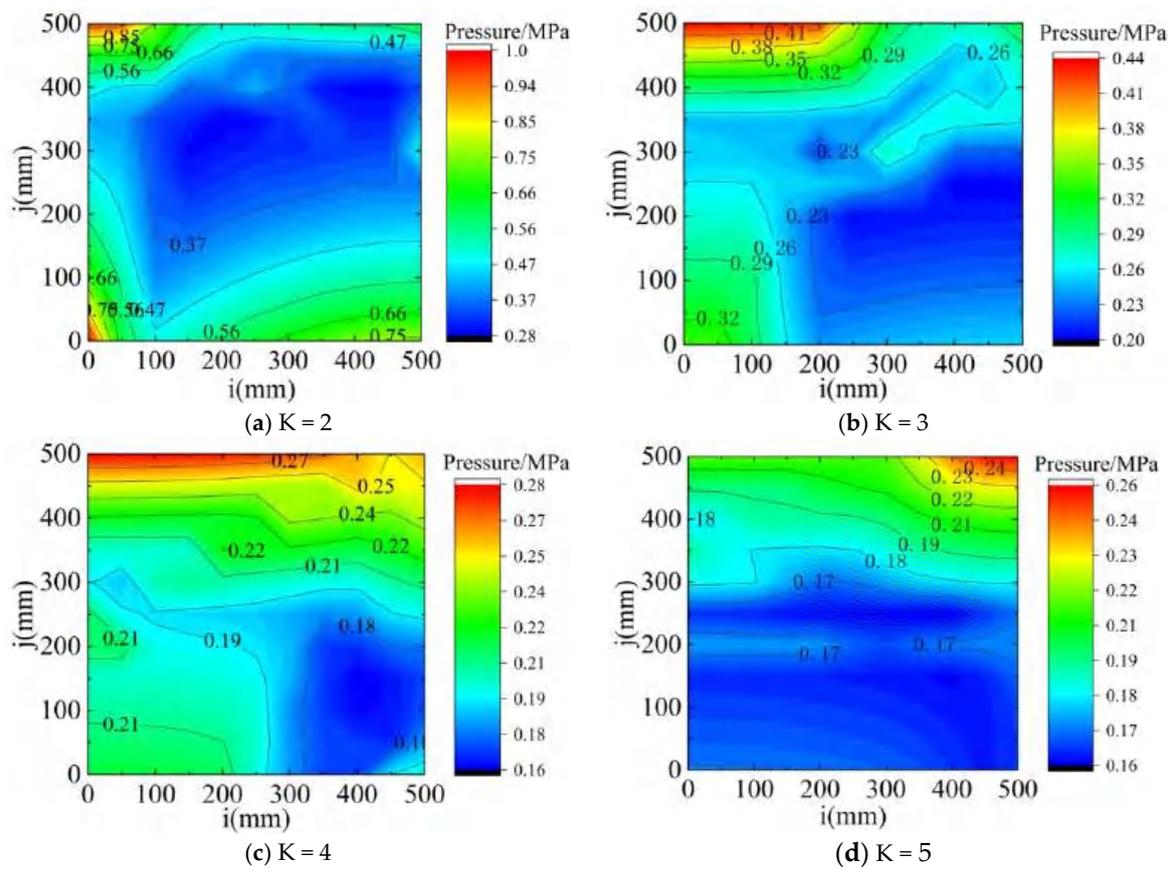


Figure 13. Pressure contour distribution diagram in the corner area.

Figure 13 shows that as K increases, the high-pressure area at the corner gradually widens with little change in height. In addition, the high-pressure area gradually moves towards the center of the surface of wall B and is no longer noticeable at the corner but a high-pressure area is formed at the center of the B wall surface. Figure 13 also shows that when $K > 3$, the high-pressure area at the corner is not apparent; thus, the K range from 1 to 3 will be discussed next. The high-pressure areas at the corners are individually intercepted along the boundary line, as shown in Figure 14, where the numerical relationship between the size of the area boundary and the value of b is indicated.

Figure 14 clearly shows that the high-pressure area appears triangular when K is small and as isosceles triangles when $K = 1$. As K increases, the shape of the high-pressure area gradually approximates to a rectangular form; this result is consistent with the high-pressure area formation rule at the corner that has already been discussed. The high-pressure area gradually widens primarily as the Mach-reflected wavefront formed at the surface of wall A widens. The relationship between the range of areas obtained in Figure 14 was represented in a coordinate system with the height and width of the corner high-pressure area a_p and b_p , respectively. Figure 15 shows the data points in the coordinate system with K , with a_p/b and b_p/b as the horizontal and vertical coordinates, respectively.

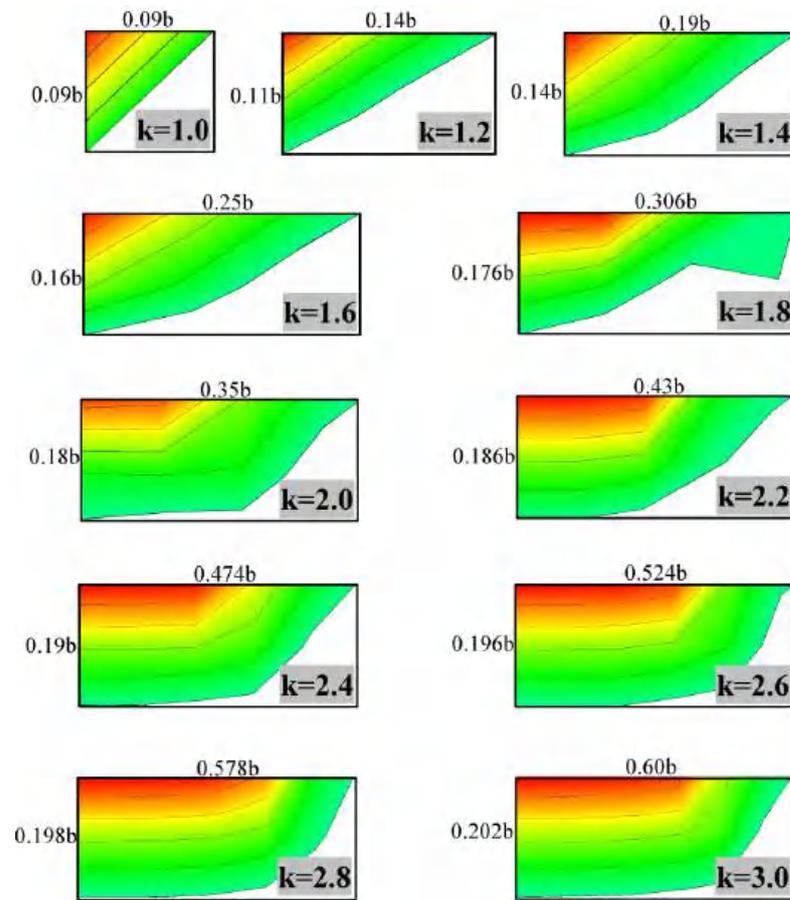


Figure 14. Change of the high-pressure area at the corners with the value of K.

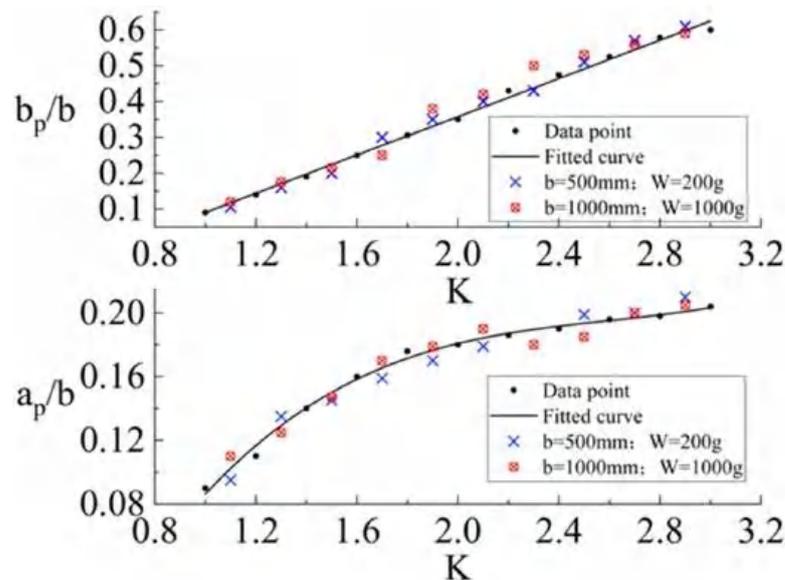


Figure 15. Relationship between the size of the high-pressure area and K.

The polynomial fit performed on the data points in Figure 15 shows that b_p/b increases linearly as K increases. Equation (3) gives the functional relationship between b_p/b and K based on the polynomial fit.

$$b_p/b = -0.17673 + 0.26709K \quad (3)$$

The functional relationship between a_p/b and K is also polynomial, as shown by Equation (4):

$$a_p/b = 0.02015K^3 - 0.15693K^2 + 0.42428K - 0.20127 \quad (4)$$

As the fitted data points were obtained under the conditions $b = 500$ mm and $W = 100$ g, simulations were conducted considering the generality of the functional relationship given by Equations (3) and (4). Two sets of data points were obtained at $b = 500$ mm and $W = 200$ g, and $b = 1000$ mm and $W = 1000$ g, as represented in Figure 15. Both data sets satisfy the functional relationship obtained and show that K is the main factor influencing the range of high-pressure areas.

5. Conclusions

This study investigated the convergence effect of shock waves at the corners of a cabin under implosion conditions using a 2D cross-sectional method. The high-pressure area formation mechanism during implosion shock wave convergence at the corners and the associated change law were determined with the aspect ratio (K). The specific conclusions are as follows:

1. The aspect ratio, K , significantly influences implosion shock wave convergence at the corner and the associated high-pressure area formation mechanism. When $1 \leq K \leq 1.193$, the convergence of the initial shock wave and Mach reflected waves from the surfaces of walls A and B occurs at the corner, creating a high-pressure region. However, when $1.193 < K < 2$, the convergence at the corner comes from the initial shock wave, the Mach reflection wave on wall A, and the regular reflection wave on wall B. When $2 \leq K$, the high pressure at the corner mainly originates from the positive reflection of Mach waves from the surface of wall A to wall B;
2. As K increases, the convergence effect of the shock waves at the corner is no longer noticeable, and the high-pressure region moves towards the center of the short side;
3. The functional relationship between K and the range of the high-pressure region at the corner was obtained when $K = 1$ to 3 and its universality was verified.

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Article

Numerical Study of Pressure Attenuation Effect on Tunnel Structures Subjected to Blast Loads

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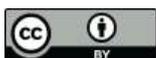
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Abstract: This study used experimental and numerical simulation methods to discuss the attenuation mechanism of a blast inside a tunnel for different forms of a tunnel pressure reduction module under the condition of a tunnel near-field explosion. In terms of the experiment, a small-scale model was used for the explosion experiments of a tunnel pressure reduction module (expansion chamber, one-pressure relief orifice plate, double-pressure relief orifice plate). In the numerical simulation, the pressure transfer effect was evaluated using the ALE fluid–solid coupling and mapping technique. The findings showed that the pressure attenuation model changed the tunnel section to diffuse, reduce, or detour the pressure transfer, indicating the blast attenuation effect. In terms of the effect of blast attenuation, the double-pressure relief orifice plate was better than the one-pressure relief orifice plate, and the single-pressure relief orifice plate was better than the expansion chamber. The expansion chamber attenuated the blast by 30%, the one-pressure relief orifice plate attenuated the blast by 51%, and the double-pressure relief orifice plate attenuated the blast by 82%. The blast attenuation trend of the numerical simulation result generally matched that of the experimental result. The results of this study can provide a reference for future protective designs and reinforce the U.S. Force regulations.

Keywords: blast; tunnel; pressure reduction module; LS-DYNA

1. Introduction

Tunnels are usually concealed and sheltered by landforms and ground objects to prevent a direct hit from enemy weapons, meaning the transfer of a blast is obstructed and attenuated by orifice plate attenuators, expansion chambers, explosion doors, and tunnel branches.

Studying the dynamic response of structures subjected to air blast loading has received a lot of attention in the last few decades [1–10]. In terms of studies regarding tunnel explosion protection, in 1992, Song et al. [11] used a reduced specimen of a steel ammunition storage magazine, with the internal dimensions of 100 × 50 × 23 cm and loading density of 16.7 kg/m³; detonated 1.9 kg of C-4 explosives inside the specimen; and then discussed the influence of Straight, Elbow, and Dead-End channels on the blast transfer. In 1993, Scheklinski-Glück [12] used a round-section of a full-scaled tunnel with a diameter of 3.6 m,

and 4000 kg, 2000 kg, and 1000 kg cylindrical RDX explosives in a model scale tunnel with a diameter of 9 cm and cylindrical RDX charge weights of 64, 32, and 16 g. The explosives stand outside the entrance in distances from one to five times the tunnel diameters. The direction is from 0° (tunnel axis) to 90° (charge touching the wall) in steps of 30°. The result showed that the blast inside the tunnel attenuated as the distance increased. In 2004, McMahon et al. [13] used a circular tunnel with a diameter of 0.298 m and 54.3 m in length and placed 0.177 kg and 1.77 kg spherical B explosives at the tunnel portal, as well as 60, 30, and 15 cm outside the tunnel portal, in order to perform explosion experiments. The result showed that the blast inside the tunnel attenuated as the distance increased, and the detonation wave impulse inside the tunnel could be regarded as a constant. In the WES (TM 5-855-1, 1998) [5] equation, according to the position of the explosive source, explosions outside a tunnel are divided into end-on and side-on. In the EMI equation (TM 5-855-1, 1998) [14], the proposed empirical equation can be used to estimate the blast inside a tunnel from an explosion outside the tunnel. As proposed by Welch et al., in 2005 [15], the empirical equation can be used for estimating the blast inside a tunnel, as resulted from an explosion outside the tunnel.

In 2006, Cheng et al. [16] used LS-DYNA software to simulate a strip and a bent channel type ammunition storage magazine and analyzed the internal explosion. The simulation result showed that the bent channel was more effective at attenuating the blast than the strip channel. An appropriate channel design could reduce the lethal area of an explosion inside the ammunition storage magazine. In 2007, Ishikawa and Beppu [17] compiled the protective structure explosion experiment results of Johoji et al. from 1965 to 1981. They analyzed the blast transfer attenuation in vertical bar, branch, and mesh tunnels. According to the experimental document review, the aforesaid experiments mainly discussed detonation waves inside the tunnel after an explosion outside the tunnel. This paper discusses the transfer mechanism of a blast resulting from a near-field explosion inside a tunnel. The near-ground and variable tunnel explosion experiments were performed, the numerical simulations and U.S. Force empirical equations were used for analysis and validation, and related empirical equations were established, which are intended to establish a tunnel blast protection evaluation and improvement mechanism to provide a reference for subsequent tunnel building and renovation.

2. Experiment

The aim was to reduce and avoid explosion pressure directly jeopardizing the safety of personnel inside a tunnel structure. This study designed three pressure attenuation models by changing the tunnel's cross-section, namely, an expansion chamber, single-orifice-plate attenuator, and double-orifice-plate attenuator, to investigate the attenuation effect, wave propagation pattern, and pressure distribution. When a blast wave passes through the tunnel, the pressure is expected to be attenuated by diffusion and detour due to the tunnels' cross section change.

In this study, a small-sized rectangular section tunnel specimen was used to demonstrate an underground tunnel structure subjected to external explosions. The tunnel specimen was made of steel plate with a thickness of 0.5 cm, and the size of its cross-section was 30 × 30 cm. The charge used in the explosion test was C-4 explosive. Its appearance is gray to light yellow. The density was between 1.59 and 1.60 g/cm², and the detonation speed can reach 8193 m/s.

Two types of pressure transducer produced by PCB company were used in the field test. The first type was pencil type sensor (models: 137A21 and 137A23), and the measuring range was from 345 to 345 MPa. This type of sensor is used to measure the explosion pressure near the ground in the free field; the second type is high-frequency pressure gauge (models: 113B23, 113B27, and 113B28), and the measuring range was from 345 kPa to 69 MPa, which were used to measure the pressure in the rectangular tunnel specimens. The maximum bandwidth of the oscilloscope was 100 MHz, and the maximum sampling rate was $2 \times 10^9 \text{ s}^{-1}$.

2.1. Explosion Experiment on the Pressure Reduction Module Effect

2.1.1. Linear Tunnel with Expansion Chamber

A linear tunnel 140 cm long with a square cross-section of 30 × 30 cm was combined with a 60 × 60 × 60 cm expansion chamber for an explosion experiment. The cross-section dimension of the expansion chamber was four times the section of the linear tunnel. The pressure transducers were mounted on the specimen sidewall at 2, 30, 90, and 170 cm away from the tunnel portal. In order to investigate the pressure reduction effects under different quantities of explosives, we used five quantities of C-4 explosives (100, 150, 200, 250, and 350 g), and the C-4 explosive was hung at 30 cm aboveground and detonated at 60 cm away from the tunnel portal. The experimental configuration is shown in Figure 1. In order to know the blast attenuation characteristic of the expansion chamber, as designed by expanding the cross-section, we analyzed and discussed the transfer of the blast inside the tunnel and the pure linear tunnel explosion experiment.

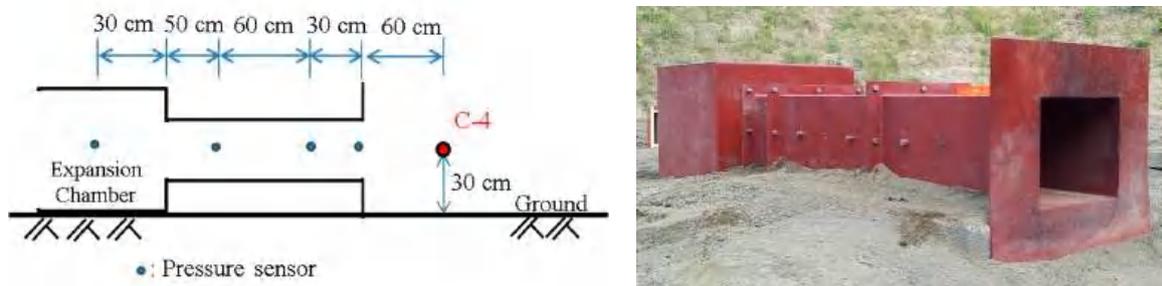


Figure 1. Experimental configuration and specimen of the linear tunnel with expansion chamber.

2.1.2. Linear Tunnel with One-Pressure Relief Orifice Plate

The linear tunnel was a square-section tunnel with a side length of 30 cm—the total length was 200 cm, and the orifice plate (circular orifice in diameter of 12 cm) was mounted at 127 cm away from the tunnel portal. The orifice plate specimen was designed by reducing the scale of U.S. Force regulation UFC 3-340-01 [13] by 2.5 times. The pressure transducers were mounted on the specimen sidewall and at 2, 30, 90, and 170 cm away from the tunnel portal. In order to know the pressure reduction effects under different quantities of explosives, we used five quantities of C-4 explosive (100, 150, 200, 250, and 350 g), and the explosive was hung at 30 cm aboveground and detonated at 60 cm away from the tunnel portal. The experimental configuration and specimen are shown in Figure 2.

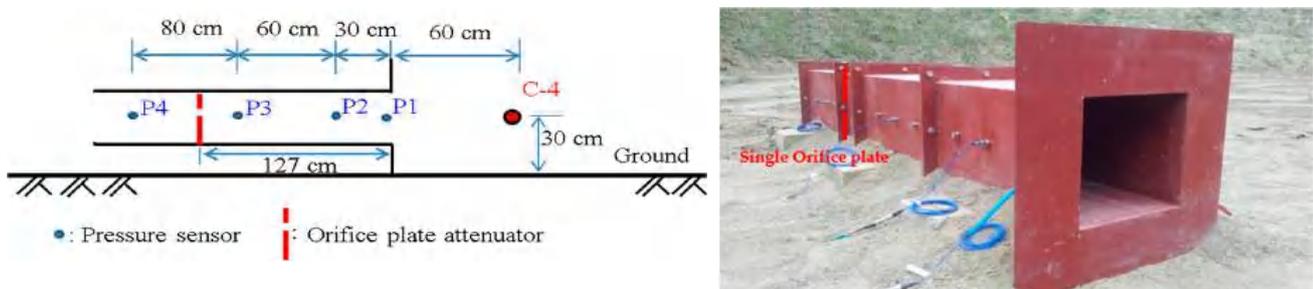


Figure 2. Experimental configuration and specimen of linear tunnel with single-pressure relief orifice plate.

2.1.3. Linear Tunnel with Double-Pressure Relief Orifice Plate

The linear tunnel was a square-section tunnel with a side length of 30 cm—the total length was 300 cm, and the orifice plates (circular orifice in diameter of 12 cm) were diagonally mounted at 127 cm and 134 cm away from the tunnel portal inside the linear tunnel. The pressure transducers were mounted on the specimen sidewall at 2, 30, 90, and 170 cm away from the tunnel portal. In order to know the pressure reduction effects under

different quantities of explosives, we used five quantities of C-4 explosive (100, 150, 200, 250, and 350 g), and the explosive was hung at 30 cm aboveground and detonated at 60 cm away from the tunnel portal. The experimental configuration and specimen are shown in Figure 3.

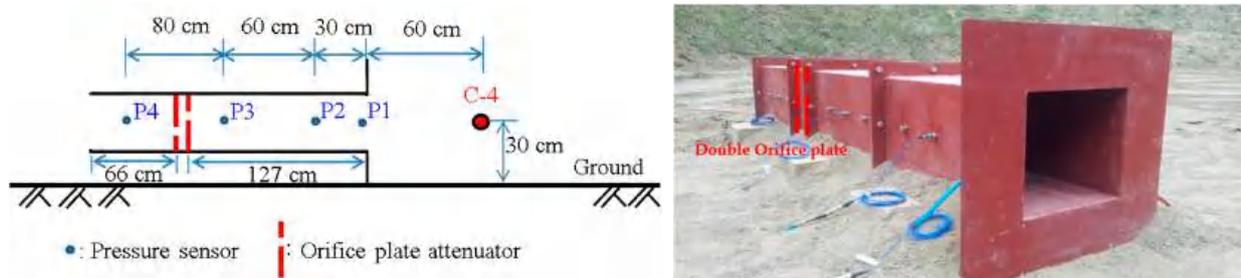


Figure 3. Experimental configuration and specimen of the linear tunnel with double-pressure relief orifice plate.

3. Numerical Simulation

There are three main numerical models in the LS-DYNA program: the Lagrangian numerical model, the Eulerian numerical model, and the ALE (arbitrary Lagrangian–Eulerian) numerical model. As the ALE numerical model has the characteristics of the Lagrangian and Eulerian numerical models, it was used for numerical simulation in this study. It can overcome the problem in that the operation stops as the numerical calculation becomes difficult when the mesh element deformation is too large compared to the Lagrangian system. Eulerian describes the fluid and Lagrangian describes the solid, and it can effectively control and track the motion behavior of the structural boundary. Thus, it is applicable to the dynamic real-time analysis of fluid–solid coupling and it has better computational accuracy than the Eulerian system. However, as the number of grids increases, the analysis model and grid size are limited. In order to solve this problem, we used the LS-DYNA mapping technology to break through the limit.

3.1. Numerical Models

Regarding the building methods of the various pressure reduction modules, as the linear tunnel with expansion chamber was symmetrical, the 1/2 symmetrical simplified numerical model was used for analysis. The linear tunnel with a single-pressure relief orifice plate and the linear tunnel with double-pressure relief orifice plate models were analyzed using full models. The models are shown in Figure 4. Regarding the orifice plate model, as the shell element had coupling directivity in the fluid–solid coupling, this study considered the vortex of the blast through the orifice plate or reflected blast, and in order to avoid analytical errors, the orifice plate was built using entity elements.

3.2. Constitutive Models and Equation of State

Numerical simulation was performed to investigate the pressure attenuation effect on the tunnel models. The constitutive model and material parameters of the air, explosives, and steel plates are described as follows:

3.2.1. Air

Regarding the air part of the numerical model, MAT_NULL material model was provided with the EOS_LINEAR_POLYNOMIAL condition equation, as shown in the following equation:

$$P = C_0 + C_1\mu + C_2\mu^2 + C_3\mu^3 + (C_4 + C_5\mu + C_6\mu^2)E_0 \quad (1)$$

where P is the pressure composed of initial internal energy, and E_0 is the ratio of current density to initial density, μ , and material parameters, C_0 to C_6 . In the present study, because

air was assumed to be an ideal gas, C_1 , C_2 , C_3 , and C_6 were set to zero, and C_4 and C_5 were set to 0.4.

3.2.2. Explosive

For explosive material, MAT_HIGH_EXPLOSIVE_BURN material model was applied with the JWL (Jones–Wikens–Lee) equation of state to model TNT explosive with the pressure defined as

$$P = A \left(1 - \frac{\omega}{R_1 V_r} \right) e^{-R_1 V_r} + B \left(1 - \frac{\omega}{R_2 V_r} \right) e^{-R_2 V_r} + \frac{\omega E_0}{V_r} \quad (2)$$

where P is the hydrostatic pressure, and V_r is the relative volume. A , B , R_1 , R_2 , and ω are material parameters used for the explosives, which can be experimentally determined.

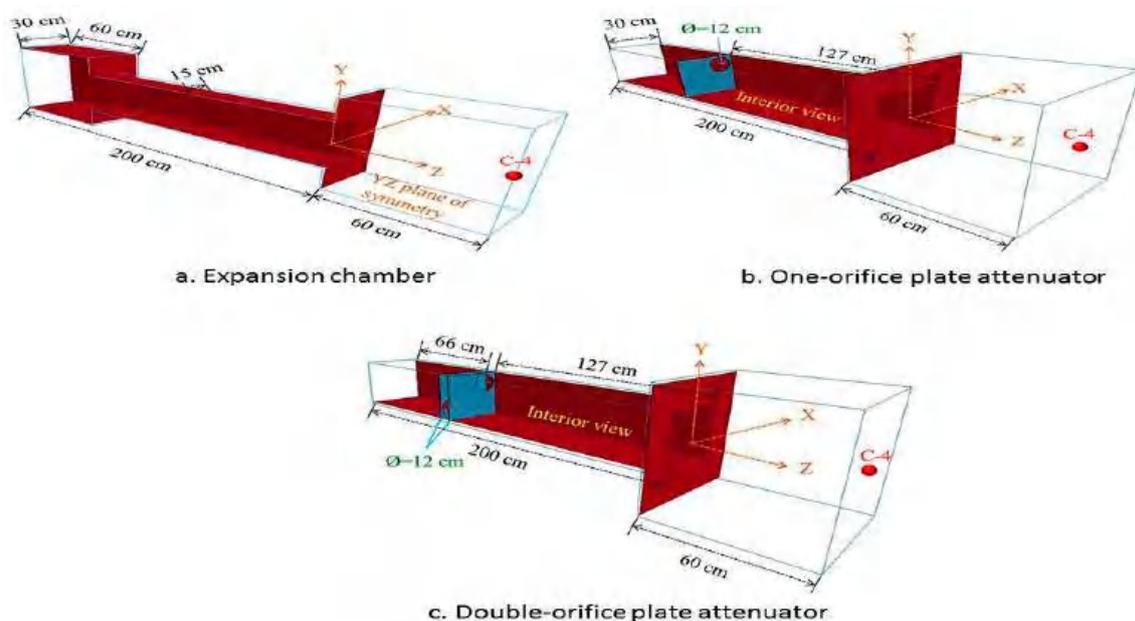


Figure 4. Schematic diagram of numerical models of pressure reduction modules.

3.2.3. Steel Plate

The MAT_PLASTIC_KINEMATIC material model was used to simulate the steel plate structure, which was the tunnel part of the numerical model. For simplified and conservative operations, the idealized stress–strain curve was used, and the strain hardening behavior of the material after plasticization can be controlled by parameter β . If $\beta = 0$, it represents a dynamic plastic hardening material. If $\beta = 1$, it is an isotropic strain hardening material. When unloading occurs, the dynamic plastic hardening curve and isotropic plastic hardening curve unload according to the original slope, the yield stress value of the isotropic plastic hardening curve will increase during reverse loading, and the yield point of the dynamic plastic hardening curve remains. The isotropic plastic hardening curve ($\beta = 1$) is more suitable for large deformation of the material, as resulted from the explosion. The present study assumed $\beta = 0$.

4. Results

4.1. Pressure Reduction Module Effect Analysis

4.1.1. Linear Tunnel with Expansion Chamber Explosion

The variation of blast attenuation inside the tunnel of this experiment is shown in Figure 5. Due to the nature of explosion characteristics, the pressure decayed extremely

rapidly with time and space. As a result, for different charge weights, a larger scattering of the pressures could be found at the measurement point P1 compared to P4. According to the experimental results, the transfer of the blast inside the tunnel decreased as the distance increased. In order to know the blast attenuation characteristic of the expansion chamber, as designed by enlarging the section, we analyzed and discussed the blast transfer rate and variation rate of the expansion chamber. In terms of the blast transfer rate, the linear tunnel with expansion chamber was tested, and when the blast was transferred from the smaller tunnel section (pressure transducer P3 position) to the expansion chamber with a larger section (pressure transducer P4 position), the blast transfer attenuation in P3 and P4 was analyzed. In terms of pressure attenuation rate, the linear tunnel with expansion chamber and the pure linear tunnel was tested, and the pressure attenuation in the P4 position was analyzed and discussed.

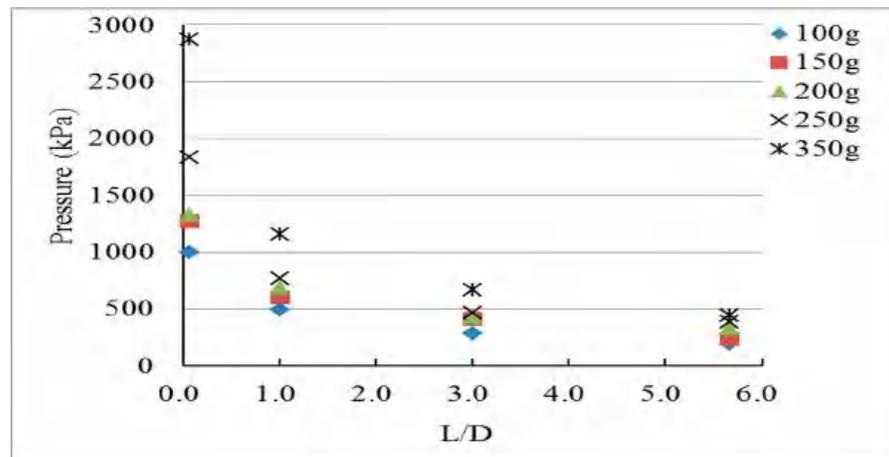


Figure 5. Comparison diagram of explosion experiment results of linear tunnel with expansion chamber.

The effect of the expansion chamber on the pressure transfer rate was described using the pressure transducers in positions P3 and P4. When the quantity of C-4 explosive was 100 g, and the blast had not been transferred to the expansion chamber, the measured blast value (P3) was 282.81 kPa. When the blast was transferred to the expansion chamber (P4), the measured blast value was 189.52 kPa. Therefore, the blast transfer rate in pressure transducer position P4 was 0.67; in other words, when the blast was transferred from position P3 with the smaller tunnel section (side length 30 cm) to position P4 with the larger tunnel section (side length 60 cm), the blast was diffusively attenuated by enlarging the section, and the blast attenuation amplitude was 33%. The blast transfer rate ($R_{transfer}$) was expressed as follows:

$$R_{transfer_expansion} = \frac{P4_{expansion}}{P3_{expansion}} \tag{3}$$

where $P3_{expansion}$ and $P4_{expansion}$ are the measured blast at the pressure transducer positions, P3 and P4, respectively.

When the quantity was changed (150~350 g), the blast was transferred from the smaller tunnel section (position P3) to the expansion chamber (position P4), and the range of blast transfer rate was 0.57 to 0.82.

When the quantity of the explosive was ≤ 350 g, the expansion chamber design mode could reduce the blast transfer rate to 0.7, meaning the blast was transferred from P3 to P4, and the blast could be attenuated by 30% by enlarging the tunnel section.

The effect of the expansion chamber on the pressure attenuation rate was tested using a pure linear tunnel and the linear tunnel with an expansion chamber. The pressure attenuation in position P4 was analyzed and discussed. When the quantity of C-4 explosive

was 100 kg, the measured blast value of the expansion chamber (P4) was 189.52 kPa, and the measured blast value of the pure linear tunnel (P4) was 271.77 kPa. Therefore, in pressure transducer position P4, the pressure attenuation rate of the expansion chamber was 0.70, as compared with the pure linear tunnel. The blast attenuation rate ($R_{attenuate}$) is expressed as follows:

$$R_{attenuate_expansion} = \frac{P4_{expansion}}{P4_{linear}} \tag{4}$$

where $P4_{linear}$ is the measured blast at the pressure transducer position, P4, in the linear tunnel.

When the quantity of the explosive was ≤ 350 g, the pressure attenuation rate in P4 was 0.77, meaning with the expansion chamber, the blast in P4 was attenuated by 23%, as compared with the pure linear tunnel (without an expansion chamber). In addition, according to the comprehensive comparison of pressure transducer positions P1 to P3, before the blast was transferred to the expansion chamber, as the tunnel specimen model was consistent, the blast transfer of the pure linear tunnel was approximate to that of the linear tunnel with an expansion chamber (pressure attenuation rate was 0.95 to 1.06), which matched the estimated result.

Generally speaking, the expansion chamber designed by enlarging the section was very effective on blast attenuation. In terms of the blast transfer rate, the blast was transferred from the smaller section (P3) to the expansion chamber with a larger section (P4), and the transfer rate was 0.70; thus, the blast can be attenuated by 30% by enlarging the section. In terms of the pressure attenuation rate, the pressure attenuation in P4 was discussed according to the experiments on the pure linear tunnel and the linear tunnel with expansion chamber. The findings show that the pressure attenuation rate was 0.77, meaning with the expansion chamber, the blast in P4 could be attenuated by 23%, as compared with the pure linear tunnel (without an expansion chamber).

In terms of numerical simulation, the numerical simulation result of the linear tunnel with expansion chamber and the experimental blast are compared in Table 1. The simulation result shows that the blast inside the tunnel attenuated as the transfer distance increased, and the blast attenuation trend of numerical simulation was similar to that of the experiment; however, the experimental result was a little lower than the numerical simulation. The transfer of the blast inside the tunnel is shown in Figure 6.

Table 1. Comparison of experiment and numerical results of linear tunnel with expansion chamber.

		Weight of C-4 (g)	100	150	200	250	350	
Position	P1 (L/D = 0.07)	Experiment	999.27	1276.14	1327.74	1834.18	2873.28	
		Simulation	889.32	1207.76	1553.81	1725.41	2281.28	
	P2 (L/D = 1.00)	Experiment	492.06	600.50	593.31	766.60	1158.69	
		Simulation	591.04	747.45	872.07	967.59	1133.06	
	P3 (L/D = 3.00)	Experiment	282.81	417.29	427.71	468.46	664.85	
		Simulation	449.52	554.88	639.48	700.86	816.89	
	P4 (L/D = 5.67)	Experiment	Explosion pressure (kPa)	189.52	237.99	332.31	384.24	443.45
			Blast transfer rate ($R_{transfer}$)	0.67	0.57	0.78	0.82	0.67
		Simulation	Explosion pressure (kPa)	244.00	304.74	354.77	391.55	462.05
			Blast transfer rate ($R_{transfer}$)	0.54	0.55	0.55	0.56	0.57
				Average: 0.55				

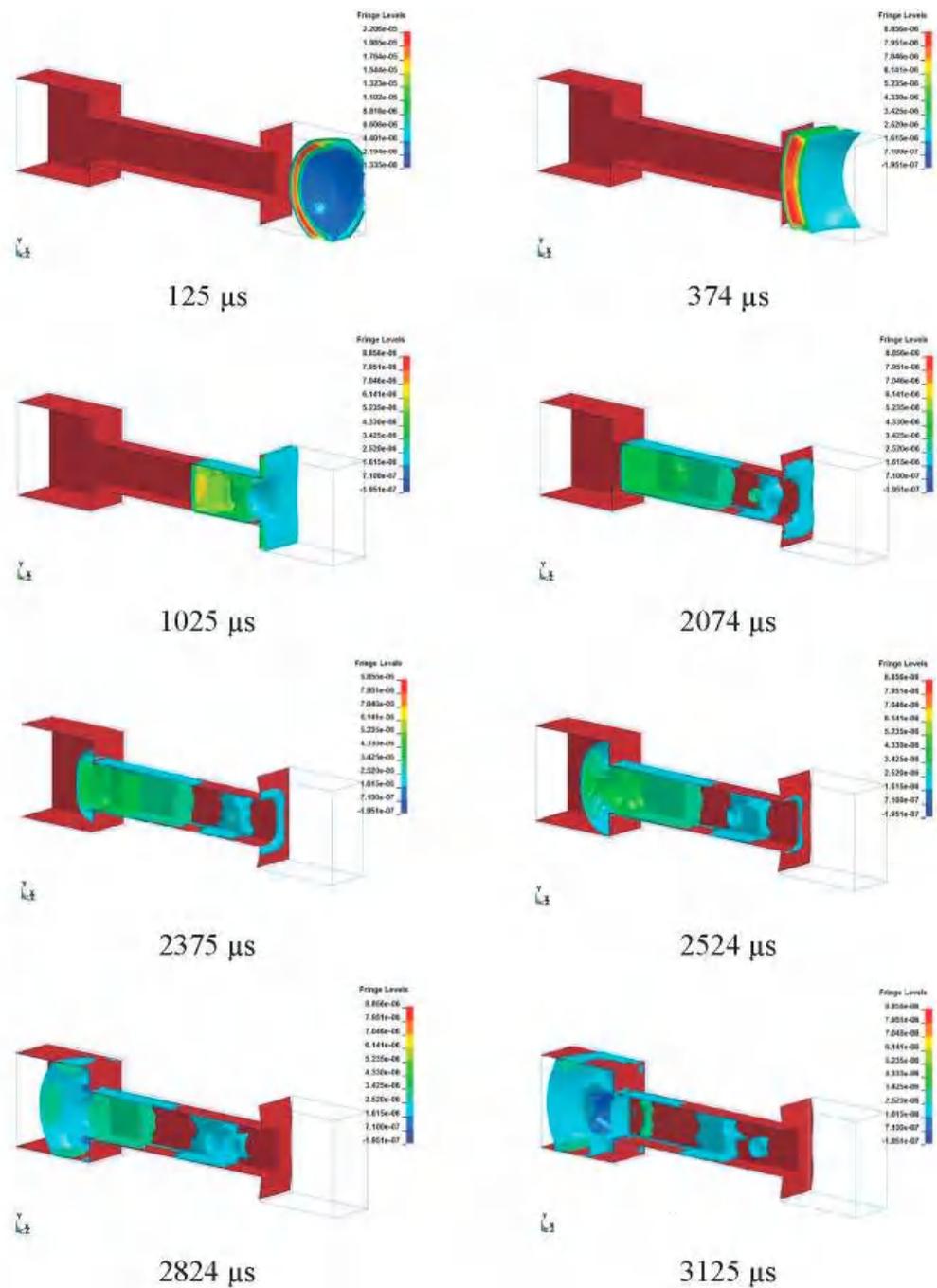


Figure 6. Blast transfer in numerical simulation of linear tunnel with expansion chamber explosion (100 g C-4).

4.1.2. Linear Tunnel with Single-Pressure Relief Orifice Plate Explosion

The variation of blast attenuation inside the tunnel of this experiment is shown in Figure 7. According to the experimental results, the transfer of the blast inside the tunnel will decrease as the distance increases. In order to know the blast attenuation characteristic of the pressure relief orifice plate, as designed by reducing the section, we analyzed and discussed the blast transfer rate and variation rate of the pressure relief orifice plate. In terms of the blast transfer rate, the linear tunnel with a single-pressure relief orifice plate was tested; when the blast was transferred from the larger tunnel section (pressure transducer position P3) to the pressure relief orifice plate with the smaller section (pressure transducer position P4), the blast transfer attenuation in P3 and P4 was analyzed. In terms

of the pressure attenuation rate, the linear tunnel with a single-pressure relief orifice plate and the pure linear tunnel were tested, and the pressure attenuation in position P4 was analyzed and discussed. The blast transfer rate ($R_{transfer}$) is expressed as follows:

$$R_{transfer_single_orifice} = \frac{P4_{single_orifice}}{P3_{single_orifice}} \quad (5)$$

where $P3_{single_orifice}$ and $P4_{single_orifice}$ are the measured blast at the pressure transducer position, P3 and P4, in the linear tunnel with single orifice plate attenuator, respectively.

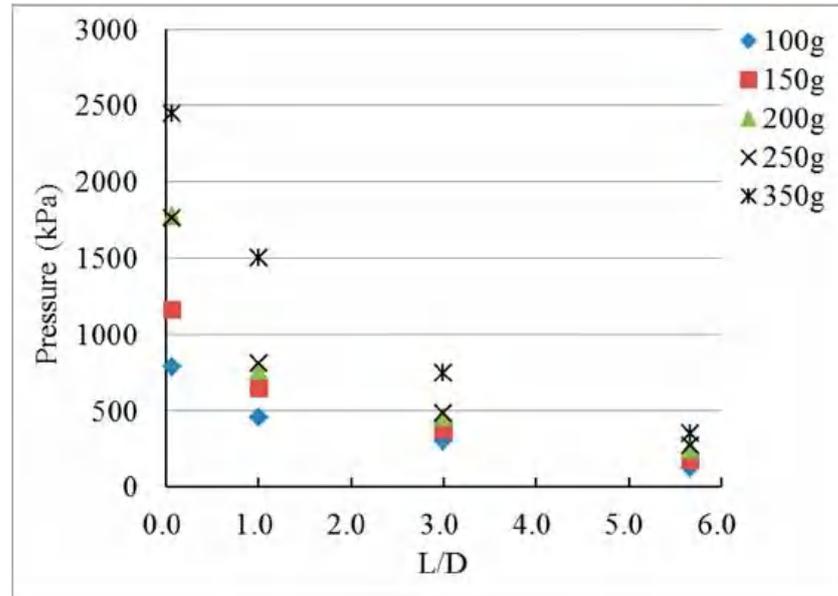


Figure 7. Comparison diagram of explosion experiment results of linear tunnel with single-pressure relief orifice plate.

When the quantity of the explosive was ≤ 350 g, the one-pressure relief orifice plate design mode can reduce the blast transfer rate to 0.49, meaning the blast was transferred from P3 to P4, and the blast could be attenuated by 51% by reducing the tunnel section.

The effect of the one-pressure relief orifice plate on the pressure attenuation rate was tested using the pure linear tunnel and the linear tunnel with one-pressure relief orifice plate, and the pressure attenuation in position P4 was analyzed and discussed. When the quantity of C-4 explosive was 100 g, the measured blast value of the one-pressure relief orifice plate (P4) was 123.89 kPa, and the measured blast value of the pure linear tunnel (P4) was 271.77 kPa. Therefore, in pressure transducer position P4, the pressure attenuation rate of the one-pressure relief orifice plate was 0.46, as compared with the pure linear tunnel. The blast attenuation rate ($R_{attenuate}$) is expressed as follows:

$$R_{attenuate_single_orifice} = \frac{P4_{single_orifice}}{P4_{linear}} \quad (6)$$

According to Table 2, when the quantity of explosive was ≤ 350 g, the pressure attenuation rate in P4 was 0.56, meaning with the one-pressure relief orifice plate, the blast in P4 was attenuated by 44%, as compared with the pure linear tunnel (without pressure relief orifice plate). In addition, according to a comprehensive comparison of pressure transducer positions P1 to P3, before the blast was transferred to the single-pressure relief orifice plate, as the cross-section was consistent, the blast transfer of the pure linear tunnel was approximate to that of the linear tunnel with one-pressure relief orifice plate (pressure attenuation rate was 0.98 to 1.08), which matched the estimated result.

Table 2. Comparison of experiment and numerical results of steel tunnel with single-pressure relief orifice plate.

		Weight of C-4 (g)	100	150	200	250	350	
Position	P1 (L/D = 0.07)	Experiment	784.67	1164.00	1777.00	1763.02	2443.99	
		Simulation	Explosion pressure (kPa)	792.96	1006.73	1623.96	1721.41	2175.51
	P2 (L/D = 1.00)	Experiment	457.43	642.93	760.66	908.14	1502.18	
		Simulation	Explosion pressure (kPa)	584.15	739.17	863.49	960.17	1126.15
	P3 (L/D = 3.00)	Experiment	300.49	367.77	454.65	482.50	747.77	
		Simulation	Explosion pressure (kPa)	568.95	617.27	655.25	702.23	837.73
	P4 (L/D = 5.67)	Experiment	Explosion pressure (kPa)	123.89	170.62	242.00	272.76	348.09
			Blast transfer rate ($R_{transfer}$)	0.41	0.46	0.53	0.57	0.47
		Simulation	Average: 0.49					
			Explosion pressure (kPa)	117.91	144.36	166.38	182.05	211.96
			Blast transfer rate ($R_{transfer}$)	0.21	0.23	0.25	0.26	0.25
			Average: 0.24					

In terms of numerical simulation, the numerical simulation result of the linear tunnel with a single-pressure relief orifice plate and the experimental blast are compared in Table 2. According to this numerical simulation, when the quantity of explosive was ≤ 350 g, the blast was transferred from position P3 to position P4, and the blast transfer rate of the one-pressure relief orifice plate was 0.24. Therefore, as predicted by numerical simulation using tunnel section reduction, the blast attenuation amplitude can be 76%. In terms of the blast transfer rates of the experiment and numerical simulation, the blast transfer rate of the one-pressure relief orifice plate (position P4) obtained by experimental analysis was 0.49 (blast was attenuated by 51%). Thus, the blast transfer rate predicted by numerical simulation was 0.24 (blast is attenuated by 76%). In contrast, the blast attenuation predicted by numerical simulation was larger. Both the experimental and numerical simulation results showed that the one-pressure relief orifice plate, as designed by reducing the tunnel section, was surely effective on blast attenuation. The transfer of the blast inside the tunnel is shown in Figure 8.

4.1.3. Linear Tunnel with Double-Pressure Relief Orifice Plate Explosion

The variation of blast attenuation inside the tunnel of this experiment is shown in Figure 9. According to the experimental results, the transfer of the blast inside the tunnel will decrease as the distance increases. In order to know the blast attenuation characteristic of the double-pressure relief orifice plate, as designed by reducing the section, we analyzed and discussed the blast transfer rate and variation rate of the double-pressure relief orifice plate. In terms of the blast transfer rate, the linear tunnel with a double-pressure relief orifice plate was tested, and when the blast was transferred from the larger tunnel section (pressure transducer position P3) to the double-pressure relief orifice plate (pressure transducer position P4), the blast transfer attenuation in P3 and P4 was analyzed. In terms of the pressure attenuation rate, the linear tunnel with a single-pressure relief orifice plate and the pure linear tunnel were tested, and the pressure attenuation in position P4 was analyzed and discussed.

The pressure transducer positions P3 and P4 were taken as examples to describe the effect of a double-pressure relief orifice plate on the blast transfer rate. When the quantity of C-4 explosive was 100 g, before the blast was transferred to the double-pressure relief orifice plate, the measured blast value in pressure transducer position P3 was 272.79 kPa. When the blast was transferred through the double-pressure relief orifice plate (pressure transducer position P4), the measured blast value was 38.18 kPa. Therefore, the blast transfer rate in pressure transducer position P4 was 0.14. In other words, when the blast was transferred from position P3 of the larger tunnel section (side length 30 cm) to position P4 of the smaller tunnel section after the path was changed, the double-pressure relief

orifice plate reflected the blast, reducing and detouring the throughput, and the blast attenuation amplitude was 86%. The blast transfer rate ($R_{transfer}$) is expressed as follows:

$$R_{transfer_double_orifice} = \frac{P_{4_double_orifice}}{P_{3_double_orifice}} \quad (7)$$

where $P_{3_double_orifice}$ and $P_{4_double_orifice}$ are the measured blast at the pressure transducer position, P_3 and P_4 , in the linear tunnel with single orifice plate attenuator, respectively.

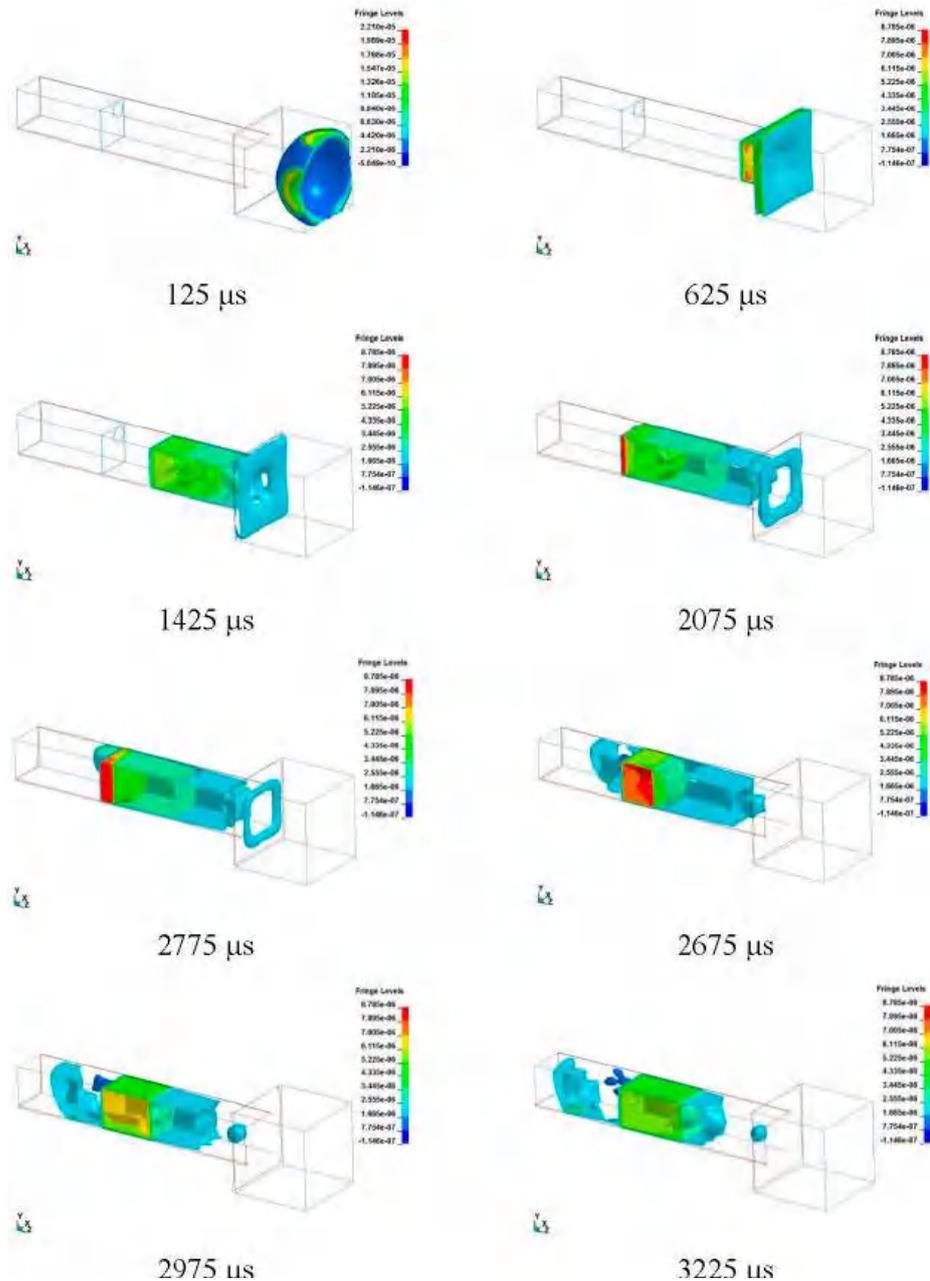


Figure 8. Blast transfer of numerical simulation of linear tunnel with single-pressure relief orifice plate explosion (100 g C-4).

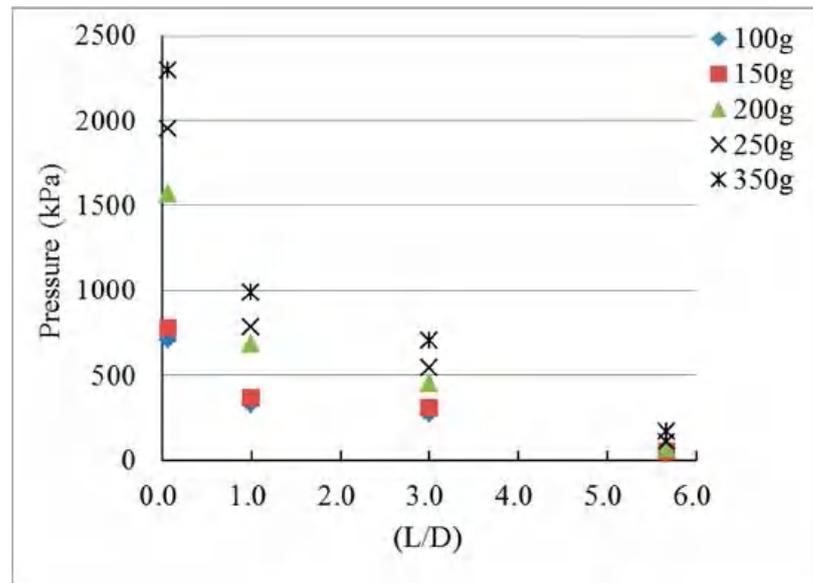


Figure 9. Comparison diagram of double-pressure relief orifice plate explosion experiment results.

When the quantity was changed (150–350 g), the blast was transferred from the larger tunnel section (P3) and through the double-pressure relief orifice plate (P4); thus, the range of the blast transfer rate of the double-pressure relief orifice plate was 0.16 to 0.24.

According to this study, when the quantity of explosive was ≤ 350 g, the double-pressure relief orifice plate design mode could reduce the blast transfer rate to 0.18, meaning the blast was transferred from P3 to P4, and the blast could be attenuated by 82% by reducing the tunnel section and changing the path.

The effect of the double-pressure relief orifice plate on the pressure attenuation rate was tested by using the pure linear tunnel and the linear tunnel with a double-pressure relief orifice plate. The pressure attenuation in position P4 was analyzed and discussed. When the quantity of C-4 explosive was 100 g, the measured blast value of the double-pressure relief orifice plate (P4) was 38.18 kPa, and the measured blast value of the pure linear tunnel was 271.77 kPa. Therefore, in pressure transducer position P4, the pressure attenuation rate of the double-pressure relief orifice plate was 0.14, as compared with the pure linear tunnel. The blast attenuation rate ($R_{attenuate}$) is expressed as follows:

$$R_{attenuate_double_orifice} = \frac{P4_{double_orifice}}{P4_{linear}} \tag{8}$$

According to Table 3, when the quantity of explosive was ≤ 350 g, the pressure attenuation rate in P4 was 0.20; in other words, when the pure linear tunnel was equipped with the double-pressure relief orifice plate, the blast in P4 was attenuated by 80%, as compared with the pure linear tunnel (without the pressure relief orifice plate). In addition, according to a comprehensive comparison of pressure transducer positions P1 to P3, before the blast was transferred to the double-pressure relief orifice plate, as the tunnel specimen model was consistent, the blast transfer of the pure linear tunnel was approximate to that of the linear tunnel with double-pressure relief orifice plate (pressure attenuation rate was 0.86 to 0.94), which matched the estimated result.

In terms of numerical simulation, the numerical simulation result of the linear tunnel with a double-pressure relief orifice plate and the experimental blast are compared in Table 3. According to this numerical simulation, when the quantity of explosive was ≤ 350 g, the blast was transferred from position P3 to position P4, and the blast transfer rate of the double-pressure relief orifice plate was 0.15. Therefore, the blast attenuation amplitude, as predicted by numerical simulation using tunnel section reduction and blast transfer path detour, could be 85%. In terms of the blast transfer rates of the experiment

and numerical simulation, the blast transfer rate of the double-pressure relief orifice plate obtained by experimental analysis was 0.18 (blast was attenuated by 82%), and the blast transfer rate predicted by numerical simulation was 0.15 (blast is attenuated by 85%); thus, the blast predicted by numerical simulation was approximate to the experimental result. The transfer of the blast inside the tunnel is shown in Figure 10.

Table 3. Comparison of experiment and numerical results of steel tunnel with double-pressure relief orifice plate.

			Weight of C-4 (g)	100	150	200	250	350	
Position	P1 (L/D = 0.07)	Experiment	Explosion pressure (kPa)	711.49	779.04	1568.8	1950.37	2294.61	
		Simulation		792.96	1006.73	1523.11	1786.34	2204.64	
	P2 (L/D = 1.00)	Experiment	Explosion pressure (kPa)	379.33	454.14	686.29	782.28	986.88	
		Simulation		584.15	739.17	863.49	960.17	1126.15	
	P3 (L/D = 3.00)	Experiment	Explosion pressure (kPa)	272.79	306.57	450.01	544.88	703.04	
		Simulation		651.38	780.52	739.47	787.03	843.57	
	P4 (L/D = 5.67)	Experiment	Explosion pressure (kPa)	38.18	52.33	71.64	108.96	166.3	
				Blast transfer rate ($R_{transfer}$)	0.14	0.17	0.16	0.20	0.24
		Simulation	Explosion pressure (kPa)		65.40	101.19	116.50	128.89	151.12
				Blast transfer rate ($R_{transfer}$)	0.10	0.13	0.16	0.16	0.18
					Average: 0.18				
					Average: 0.15				

Table 4 shows the percentage error between the experimental and numerical results of the three models. Although significant difference was observed in some cases, especially the double-orifice plate at P3, most cases agreed with the test results. It is worth noting that it is challenging to simulate a perfect match with the field test, especially in an explosion test, where highly nonlinear dynamic loading exists. The quality and density variation of explosive charge could also cause inconsistency. In addition, the inherent limitation of the continuum FE model may also cause variation. The results might be improved by upgrading the data acquisition system or further investigating the material parameters used in the model. It can be concluded that the agreement of the trend of the pressure attenuation rate is good, and the model gives reasonable predictions for different tunnel blast attenuation designs.

Table 4. Percentage error of different pressure attenuation models.

		Weight of C-4	100 g	150 g	200 g	250 g	350 g
Expansion chamber	P1		11%	5.4%	17%	5.9%	20.6%
	P2		20%	24%	46%	26.2%	2.21%
	P3		59%	33%	49%	49.6%	22.8%
	P4		29%	28%	6.8%	1.9%	4.2%
Single-orifice plate	P1		1%	13.5%	8.6%	2.3%	10.9%
	P2		27.7%	14.9%	13.5%	5.7%	25%
	P3		89.3%	67.9%	44.1%	45.5%	12%
	P4		4.8%	15.4%	31.2%	33.3%	39.1%
Double-orifice plate	P1		11.4%	29.3%	2.9%	8.4%	3.9%
	P2		53.9%	62.7%	25.8%	22.7%	14.1%
	P3		138%	154%	64.3%	44.4%	19%
	P4		71.2%	93.3%	62.6%	18.3%	9.12%

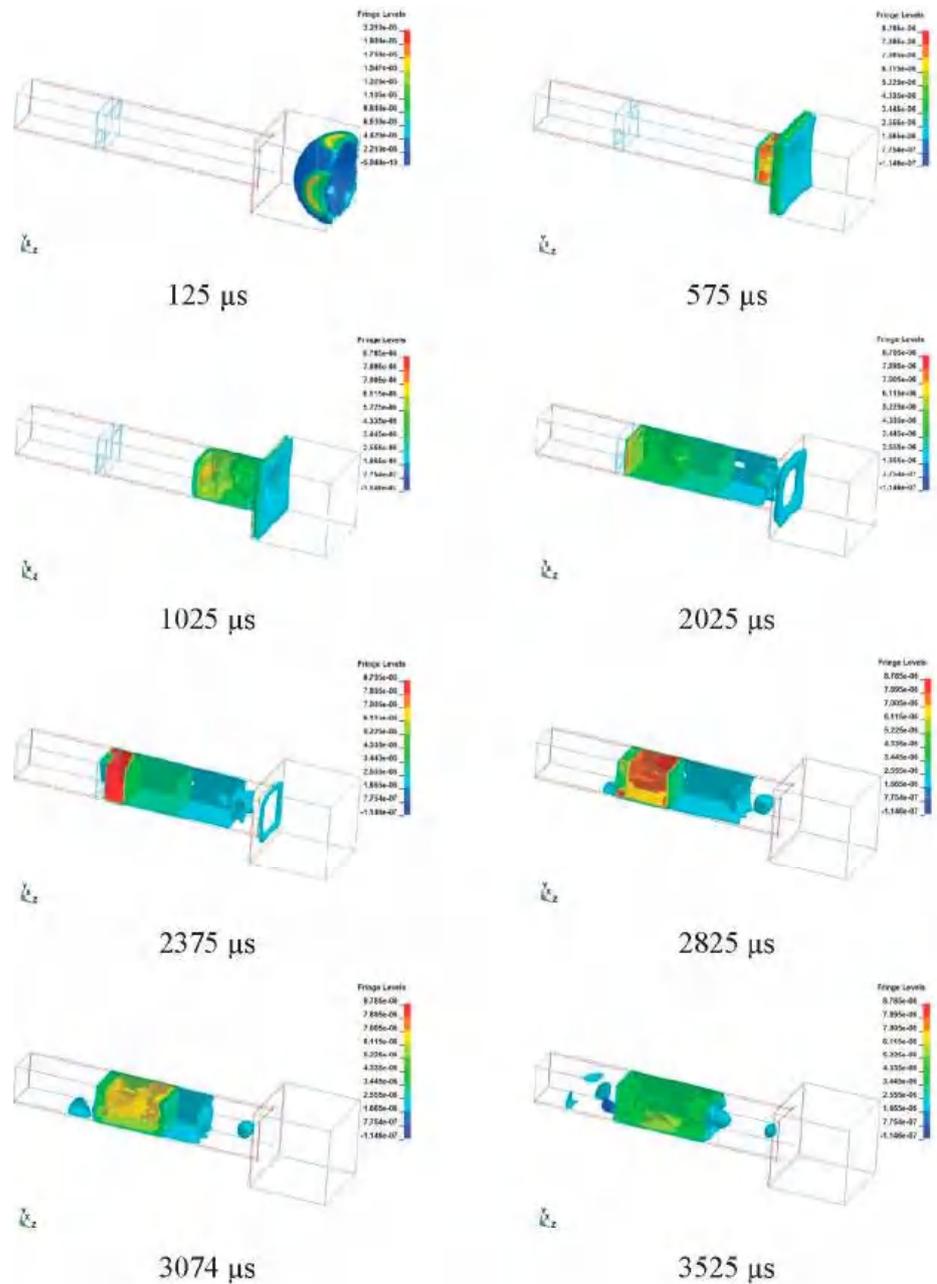


Figure 10. Blast transfer in numerical simulation of linear tunnel with double-pressure relief orifice plate explosion (100 g C-4).

5. Conclusions

- (1) The pressure reduction module (expansion chamber, one-pressure relief orifice plate, double-pressure relief orifice plate) changes the tunnel section to diffuse the blast and reduce or detour the transfer; thus, the aforesaid design modes have a blast attenuation effect.
- (2) The pressure reduction modules are designed inside the tunnel, and the findings show that the double-pressure relief orifice plate has better blast attenuation effect than the one-pressure relief orifice plate, and the one-pressure relief orifice plate is better than the expansion chamber. The expansion chamber can attenuate the blast by 30%. The one-pressure relief orifice plate can attenuate the blast by 51%. The double-pressure relief orifice plate can attenuate the blast by 82%.

- (3) The overall blast attenuation trend in the numerical simulation result of the pressure reduction module matches the experimental result. The results of this study can provide a reference for future protective designs.

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Review

Blast Loaded Columns—State of the Art Review

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Abstract: The ever-present threat of terrorist attacks in recent decades gives way to research towards blast-resistant design of structures. Columns, as one of the main load-bearing elements in residential buildings and bridges, are becoming interesting targets in bombing attacks. Research of column blast load behavior leads toward increased safety by identifying shortcomings and problems of those elements and acting accordingly. Field tests and numerical simulations lead to the development of new blast load mitigation technics, either in the design process or as a retrofit and strengthening of existing elements. The article provides a state-of-the-art literature review of filed blast load tests and numerical simulations of a bridge and building columns.

Keywords: blast load; concrete columns; experimental testing; numerical modeling

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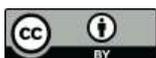
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1. Introduction

In the last five decades, terrorist attacks have become more frequent. There are different types of terrorist attacks, but according to data provided by the National Consortium for the Study of Terrorism and Response to Terrorism [1], in the last two decades, explosive attacks exceed 50% of the total number of incidents, shown in Figure 1. The attacks on The Twin Towers of the World Trade Center on 11 September 2001 and bridges in California and New York have an impact on the design of structures in the United States and also in the rest of the world [2]. In every country, the transportation system is essential for performing everyday activities, so the Blue Ribbon Panel (BRP) indicates the transportation system as one of the viable targets for a bombing attack. Due to a large number of bridges worldwide, lots of potential casualties, high repair costs, and importance in everyday life, the bridges are increasingly in the focus of terrorist attacks. This is confirmed by the fact that in the last few attacks in Nigeria in 2020, seven bridges were destroyed. It is important to identify which bridges are vulnerable due to their easy accessibility to protect against attacks. Moreover, BRP states that the columns are one of the most critical components on all types of bridges [3]. As there are many types of bridges and many ways to attack the bridge, it is difficult to predict the construction's response to the blast loadings [4]. When detonation of an explosion is under the bridge, then columns are exposed to large lateral forces, depending on standoff distance, which can result in large deformations leading to flexural or shear failures. The contact explosion can breach the column to render it incapable of supporting the dead loads. For small standoff distance, blast waves can cause a serious reduction in a concrete cross-section in terms of spalling and cratering. Since the column failure depends on the position and amount of explosives, all examined attack scenarios were observed.

The main objective of this review article is an extensive literature overview of experimental and numerical research conducted on blast-loaded columns. Both building and bridge columns are considered due to differences in their static and blast behavior. A systematic summary is given of column behavior, possible damage and failure modes, and a review of software used for blast load simulation and analysis.

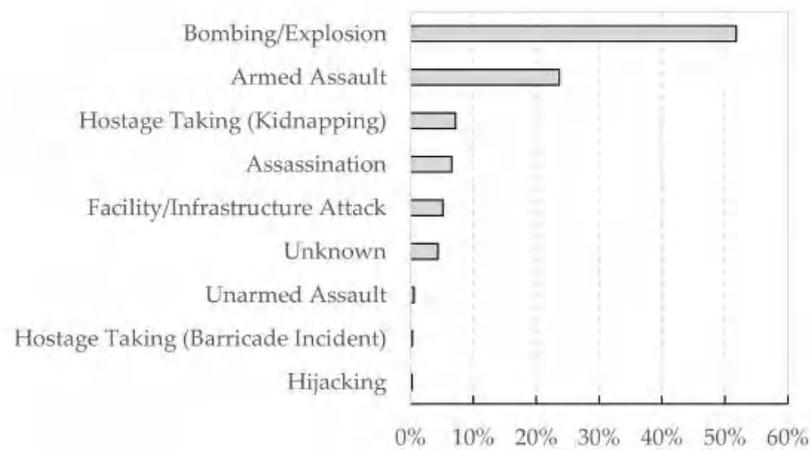


Figure 1. Percentages of terrorist attack types, based on data from [1].

2. Experimental Testing

There are no experiments on real scale specimens of bridge columns due to the experimental setup complexity and high costs, while the building columns are mainly conducted in full scale due to their maximum height of 2.5 to 3 m. Moreover, these types of experiments require special testing ground (usually military field ranges) and trained personnel for handling explosives. Even these special testing grounds have limitations regarding the maximum amount of explosives that can be used in one detonation. This also limits the scale of specimens. Research conducted in recent decades has shown that scale tests provide reliable results and the necessary knowledge to analyze the effects of blast load on full-scale structures [5].

If considering building and bridge columns, except specimen dimensions, there is a difference in their behaviors due to different levels of axial load capacity. Therefore, it is recommended to analyze bridge and building columns separately [4]. A list of conducted experimental research is provided in Table 1.

Table 1. Summary of experimental research.

Author	Year	Structural Element	Experiment Type	Material	Scale
Bruneau et al. [6]	2006	Multicolumn bents	Field	CFCSC	1:4
Fujikura et al. [7]	2008	Multicolumn bents	Field	CFST	1:4
Davis et al. [8]	2009	Bridge column	Field	RC	S. s. + 1:2
Fujikura and Bruneau [9]	2010	Multicolumn bents	Field	RC and RC SJ	1:4
Williamson et al. [4]	2011	Bridge column	Field	RC	1:2
Crawford [10]	2013	Building column	Field	RC + FRP + SJ	1:1
Burrell et al. [11]	2015	Column	Shock tube	SFRC	1:2
Zhang et al. [12]	2015	Building column	Field	CFST	1:1
Aoude et al. [13]	2015	Building column	Shock tube	UHPFRC	1:1
Codina et al. [14]	2016	Building column	Field	RC	1:1
Codina et al. [15,16]	2016	Building column	Field	RC, RC SJ, RC + polyurethane bricks	1:1
Xu et al. [17]	2016	Column	Field	UHPFRC + HSRC	1:1
Echevarria et al. [18]	2016	Bridge column	Field	CFST + RC	1:5
Fouché et al. [5]	2016	Multicolumn bents	Field	RC MSJ	1:4
Wang et al. [19]	2016	Column	Filed	RPC-FST	1:1
Zhang et al. [20]	2016	Column	Field	CFDST infilled with UHPC	1:1
Zhang et al. [21]	2017	Column	Field	CFDST	1:1
Codina et al. [22]	2017	Building column	Field	RC + reinforced resin panels	1:1
Yuan et al. [23]	2017	Bridge column	Field	RC	1:3
Wang et al. [24]	2017	Building column	Field	CFST	1:1

Table 1. Cont.

Author	Year	Structural Element	Experiment Type	Material	Scale
Li et al. [25]	2017	Building column	Field	UHPC + HSRC	1:1
Fouché et al. [26]	2017	Bridge column	Field	CFDST	1:4
Dua et al. [27]	2018	Column	Field	RC	1:1
Dua et al. [28]	2019	Column	Field	RC	1:1
Wang et al. [29]	2020	Bridge column	Field	UHPCC-FST	1:4
Kadhom et al. [30]	2020	Column	Shock tube	RC and RC + CFRP	1:2
Vapper and Lasn [31]	2020	Building column	Filed	RC, RC + GFRP	1:2

Note: CFCSC—Concrete-Filled Circular Steel Columns, CFST—Concrete-Filled Steel Tube, RC—Reinforced Concrete, SJ—Steel Jacket, MSJ—Modified Steel Jacket, FRP—Fiber-Reinforced Plastic, SFRC—Steel Fiber-Reinforced Concrete, UHPFRC—Ultra-High Performance Fiber Reinforced Concrete, HSRC—High Strength Reinforced Concrete, CFDST—Concrete-Filled Double-Skin Tubes, UHPC—Ultra-High Performance Concrete, SFRP—Steel Fiber Reinforced Polymer, UHPFRC—Ultra-High-Performance Fiber-Reinforced Concrete, CFFT—Concrete-Filled Fiber-Reinforced Polymer (FRP) tube, RPC-FST—Reactive Powder Concrete Filled Steel Tubular, HSRC—High Strength Reinforced Concrete, S. s.—small scale, GFRP—Glass Fiber Reinforced Polymer.

2.1. Bridge Columns

The experiments were carried out on standard RC columns, additionally retrofitted columns, and improved composite concrete columns. The columns are exposed to various scenarios of explosive attacks. In addition to the type of column, the scenarios also differ in the position, type, and amount of explosives.

Williamson et al. [32] provided the list of possible terrorist courses of action and indicated that the hand placed explosives on the column and large truck-bomb below the bridge superstructure can destroy columns and cause bridge collapse.

Due to similarities between the effects of the explosions and earthquakes, Bruneau et al. [6] developed a multi-hazard pier concept that they expect to provide a satisfying level of protection against failure under both loadings. All specimens were concrete-filled circular steel columns (CFCSC) with three different diameters (10.16 cm (4"), 12.7 cm (5"), 15.24 cm (6")) and a minimum steel thickness of 3.2 mm. Specimens were made in 1:4 scale of the prototype bridge columns. Due to security reasons, the actual values of charge weights and standoff distances are not provided. Experimental results showed that even a minimal increase in standoff distance and column diameter significantly reduces column deformation. The CFCSC columns showed ductile behavior and high resistance to the effects of the explosion [6]. The same scale and scenario when the explosive was located under the bridge in a car placed near to the column were examined in [7]. They assumed the charge weight similar to the blast weights predicted in FEMA (2003) [33] and FHWA (2003) [34]. Charges are set at heights of 0.25 m and 0.75 m, which correspond to the actual height of 1 m (car bomb) and the half column height, respectively. They concluded that only steel jacketing is not enough to provide adequate resistance to the large shear forces influencing the bottom of the column. Therefore, they found that a better solution is using a fully concrete-filled steel tube (CFST) continuously embedded into the footing. CFST columns provided ductile behavior and sufficient resistance to the lateral forces from earthquakes and explosions. Moreover, the advantage of CFST columns is that they do not have a breach and a spall of concrete, i.e., they do not produce flying debris [7]. Figure 2 shows the connection concept between the foundation beam and the CFST column, which provides the full moment capacity of the column. At a rotation of 3.8° of the bottom of the column, plastic deformation is visible but without cracking of the concrete. The first cracks occur at a rotation of 8.3°, while the fracture of the steel tube occurs at 17°. At the height of the explosive charge, pits and notches appeared on the steel tube, while concrete cracks occurred on the tension side at the bottom and top of the column due to the rigid boundary conditions [7]. They assumed the same blast scenario as Fujikura et al. [7] in their work at the same scale of 1:4, but there were four columns in the test specimen while the bridge prototype has three. Figure 3 shows the test setup for the same blast load scenario but on different types of columns. The RC column exhibited shear failure at the base and cracking of concrete along the column [9], RC SJ shear failure [9], and CFST column flexural failure

and buckling [7]. Fujikura and Bruneau [9], in their work, presented experimental and analytical investigations of seismically ductile RC columns and non-ductile RC columns retrofitted with steel jackets. The charge was set to a height of 0.25 m which corresponds to the actual height of 1 m (car bomb), and at this height, the column experienced the maximum deflection. All columns failed in direct shear at the base, but RC columns with steel jackets did not experience any structural damage, and the RC column experienced spalling of concrete at the bottom. Compared to the CFST columns, these columns did not exhibit a ductile behavior.

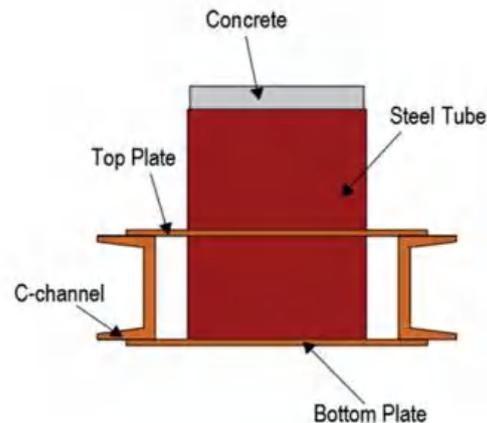


Figure 2. CFST column—details of column-to-foundation beam connection [7].

Davis et al. [8] conducted an experiment in two phases. In the first phase, they tested eight small-scale columns where they first changed the standoff distance and then the amount of explosive charge while the scaled distance was kept fixed. In the second phase, they tested 16 columns in half scale (1:2), where 10 of them were set at a small standoff to observe the mode of failure (flexure or shear) like in [4], and the remaining six were to sustain local damage (spall and breach patterns). In all samples, concrete strength, clear cover, concrete class, and reinforcement grade were unchanged. Boundary conditions for tested columns were assumed to be pinned at the top and fixed at the bottom. The test setup did not include axial load because low levels of axial loads provide greater capacity to the column, and without axial load, tests are on the conservative side. Five experimental observations and guidelines for the design of blast-loaded columns are provided in [4,8,35]:

1. Using protective fences and barriers for vehicles to increase the standoff distance;
2. The circular cross-section can maintain a lower pressure of up to 1/3 concerning a square cross-section of the same dimensions, so the second guideline is to use circular columns, and also, the pressure reducing factors on the circular column were proposed by Winget et al. [2], Marchand et al. [36], and Fujikura et al. [7], respectively, as 0.80, 0.75, and 0.45;
3. Increase in the reinforcement in the column, as this increases the shear capacity, ductility, and confinement of the concrete;
4. Use of continuous reinforcement because discrete hoops can be extracted during a blast load;
5. Placing longitudinal splices away from the charge if they cannot be completely avoided.

They also proposed three design categories (A, B, C) that depend on the scaled distance and require a different approach to designing, i.e., gravity, seismic, blast. In the C category ($Z \leq 0.6 \text{ m/kg}^{1/3}$), columns are exposed to the higher loads than columns in A ($Z > 1.2 \text{ m/kg}^{1/3}$) and B ($0.6 < Z \leq 1.2 \text{ m/kg}^{1/3}$).

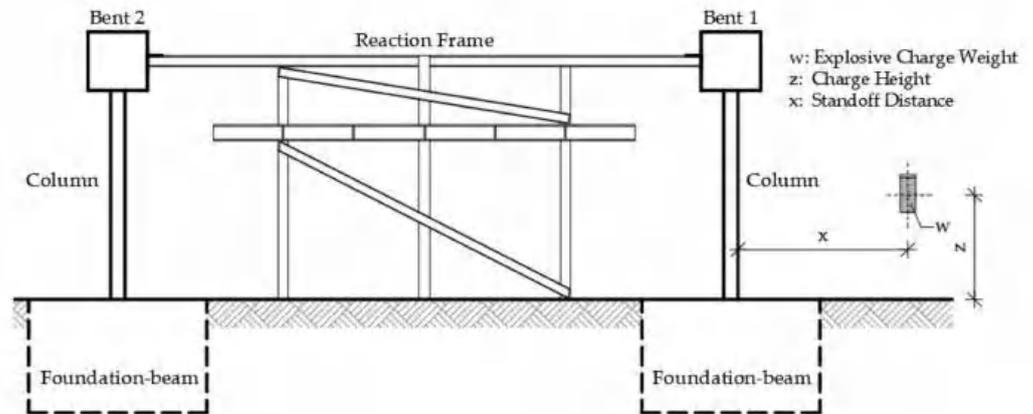


Figure 3. Test setup in [7] and [9].

Williamson et al., in Part II [35], provided a review of experiments represented in Part I [4]. Square columns experienced greater net resultant impulse than circular columns under the same blast loads and also had a larger cross-section area, so less shear occurred at the base. To increase the shear capacity of the column at the base and the ductility, it is necessary to increase the amount of transverse reinforcement. Moreover, columns with continuous spirals had a better performance than columns with discrete ties, which confirms the recommendation given in [8]. Due to changes in column design, the weight of charge, and standoff distance, several levels of damage were obtained [37,38]. The test setup is shown in Figure 4. Superficial damage means that the column performed well and has only surface damage and cracks, while minor damage means spalling of the concrete cover and cracks along the column. Deformations, flexural cracking, and spalling of concrete all together are extensive damage, while the failure of the column means that a shear occurred at the base [4].

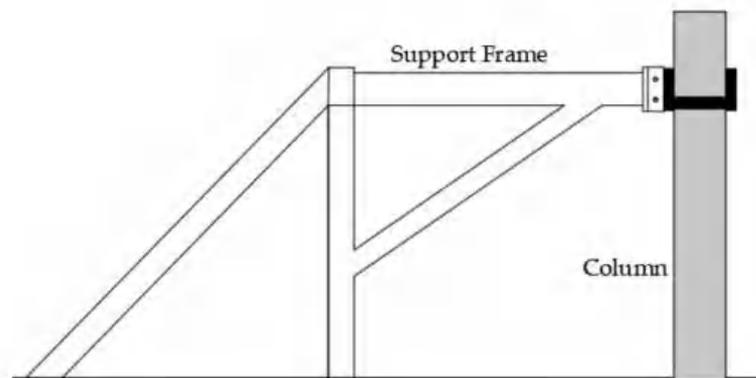


Figure 4. Test setup [4,8].

Echevarria et al. [18] tested RC and Concrete-Filled Fiber-Reinforced Polymer (FRP) tube (CFFT) bridge columns at a scale of 1:5. CFFT columns are reinforced only with longitudinal reinforcement, while RC columns have spiral hoops in addition to longitudinal reinforcement, shown in Figure 5. The quantity and distance of the explosives were not provided for safety reasons. The columns experienced minimal visual damage, but measurements showed that both concrete and steel experienced large loads and strains. In the residual test, CFFT columns showed greater strength and ductility than RC columns.

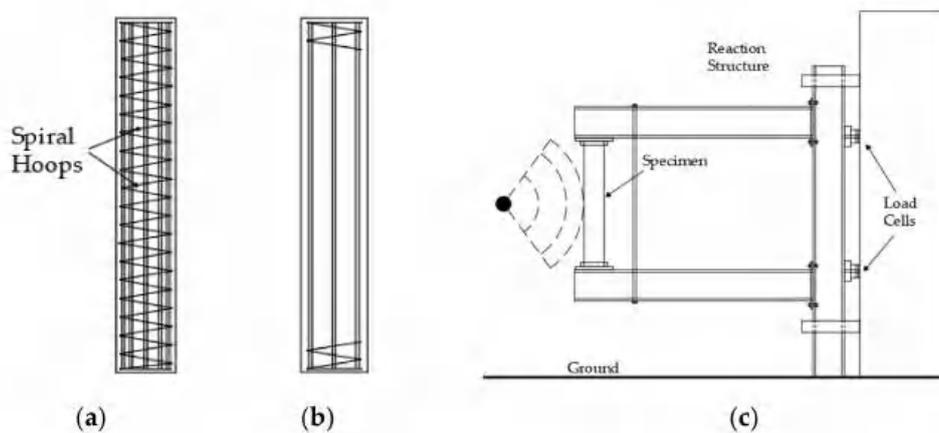


Figure 5. Reinforcement position in specimens: (a) RC column and (b) CFFT column; specimen position: (c) test setup [18].

Fouché et al. [5] made the same prototype of the bridge columns as [9] and exposed them to the same blast load scenario. To improve the behavior of the steel jacked columns to the explosion, they added structural steel collars at the top and bottom of the column. The steel collar at the base of the steel jacked RC column is shown in Figure 2 in [5]. They concluded that the modified column was effective in preventing direct shear and had increased ductility. The obtained maximum base rotation ranges from 8.6 to 10.3° , and all the columns showed satisfactory behavior.

Yuan et al. [23] experimentally tested two RC bridge columns in scale 1:3 under 1kg TNT contact explosion placed at the height of 33 cm from the ground. As a retaining structure that prevents the rotation and displacement of the column at the top, they used a wall with the opening that was placed at a distance of 1.4 m from the detonation point. Both columns experienced spallation and crushing of concrete cover, but the stirrup fracture was observed only in the square column. The damages to the front and back sides of the circular and square columns are shown in Figure 13. Therefore, the test results showed that the square column had more severe damage than the circular column.

Wang et al. [29] investigated the impact of contact explosions of 1 kg, 2 kg, and 3 kg of TNT on the mode of failure and original and residual axial capacity of Ultra-High Performance Cementitious Composite Filled Steel Tube (UHPCC-FST) bridge column. The columns were made in 1:4 scale and are tested horizontally, and the views of the test setup are shown in Figure 6. The top of the column is pinned, and the bottom of the column is fixed. The cylindrical explosive is placed at a distance of 25 cm from the lower support, which represents the actual position of the explosive in the vehicle at the height of 1 m. Quantities of explosives of 1 and 2 kg made only a crater in the column, while 3 kg fractured the tube and crushed the core, as shown in Figure 7. In the axial compression test, all columns experienced diagonal shear failure.

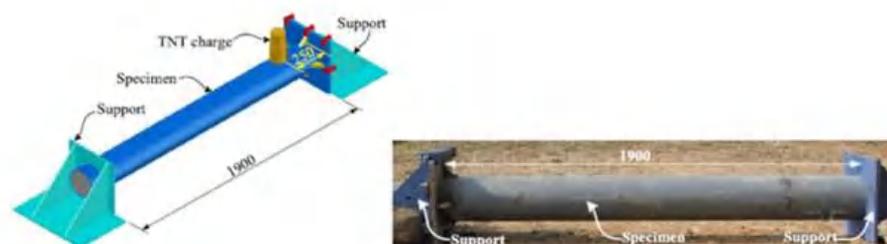


Figure 6. Schematic and field views of test setup [29]. Copyright permission obtained from authors.

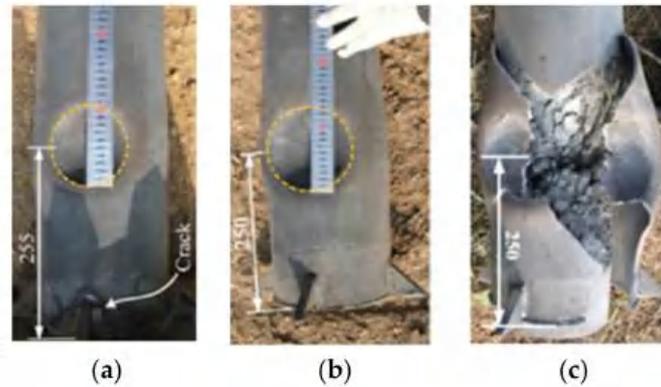


Figure 7. Damage of UHPCC-FST columns after detonation of: (a) 1 kg; (b) 2 kg and (c) 3 kg of TNT [29]. Copyright permission obtained from authors.

Based on the reviewed literature of bridge columns, maximum support rotations in field blast tests are shown in Table 2. The information can be used as a preliminary, fast damage assessment of blast-loaded columns based on the measured post-blast column rotations.

Table 2. Maximum support rotations (blast at low height).

	Top	Bottom	Crack Patterns of Concrete	Deformation
CFST [7]	1.2°	3.8°	No available	Plastic
	2.2°	8.3°	Tension side	Plastic
	4.9°	17.0°	Opening of core concrete	On-set of fracture of column
	18.7°	-	Blew away	Post-fracture of column
RC MSJ [5]	-	8.6–10.3°	Satisfactory ductile behavior	
RC [2,36,39] *	-	1.3°	Slight to moderate damage	
	-	2°	Moderate to heavy damage	
	-	3°	Lose structural integrity	
RC [40]	-	2°	Minor damage	Onset of shear failure at base
	-	4°	Collapse	Shear failure at base
RC (UFC 3-340-02) [41]		2–5°	Moderate damage	
		5–12°	Severe damage	
RC (AISC 341) [42]		2.3°	Highly ductile	

* Based on experimental testing of concrete beam elements in flexure.

2.2. Building Columns

In addition to bridges, interesting targets of terrorist attacks are buildings. By damaging the ground floor columns, the whole building loses stability, so in most of the new buildings, the ground floor columns are designed also considering the blast loads. Building columns differ from bridge columns in the magnitude of the axial loads. Moreover, the dimensions of building columns are significantly smaller, so most of the experiments conducted on building columns were in full or half-scale.

Burrell et al. [11] tested two half-scale reinforced concrete columns and six Steel Fiber-Reinforced Concrete columns (SFRC) with steel fiber content from 0 to 1.5% by volume of concrete (non-seismic and seismic detailing) at shock tube. In the experiment, the axial load equal to 30% load capacity was applied using a hydraulic jack. According to their experiments, columns designed seismically (38 mm distance between transverse reinforcement) have smaller maximum displacements and can withstand larger blast loads. Moreover, SFRC columns with non-seismic detailing (75 mm distance between transverse reinforcement) showed smaller maximum displacements and no secondary blast fragments.

Aoude et al. [13] experimentally tested nine Ultra-High-Performance Fiber Reinforced Concrete (UHPFRC) columns designed with Compact Reinforced Composite (CRC). Tests were performed in a shock tube, and a hydraulic jack was used to input the axial load. The applied blast pressures varied from 69 kPa to 689 kPa. Dimensions of the cross-section were 152 mm × 152 mm, and the free span of the column was 1980 mm. The results showed that the UHPFRC columns reduce secondary blast fragments. Increasing the proportion of fibers from 2% to 4% had a positive effect on the decrease in displacement, but a higher proportion of fibers did not result in improvements in blast behavior.

Zhang et al. [12] experimentally blast tested three square columns and one circular column. The columns were made from steel tubes filled with concrete (CFST). Specimens were placed horizontally with a simple boundary condition at both ends. A pneumatic jack was used to input the axial load, shown in Figure 8. The entire length of the specimen was 2.5 m. In the experiment, an emulsion explosive, which has a TNT-equivalent of 0.7, was used. The maximum and residual column deformations are provided for blast loads utilizing 17.5 to 35 kg TNT equivalence at a standoff distance of 1.5 m. The concrete inside the steel tube reduced local deformations, and the energy was dissipated through the global response of the element. Zhang et al. [20] tested two types of CFDST columns, circular and square, with inner and outer tubes, as shown in Figure 9. They concluded that columns filled with normal strength concrete experienced greater crushing of concrete and higher steel buckling than columns filled with UHPC. Moreover, UHPC proved to be very resistant to spalling or crushing. The tested CFDST and CFST samples have similar oscillation periods and displacements, so it is concluded that they behave similarly. Zhang et al. [21] tested six ultra-high-performance concrete-filled double skin tube columns with square hollow sections. At a standoff distance of 1.5 m, the specimen exposed to 35 kg of TNT did not experience any localized damage or steel buckling. Moreover, the axial load (25% of the maximum load) contributed to the reduction in maximum deflection in the middle of the column. They concluded that the ratio of the cavity and the section influence the overall column deflection and period of oscillation; therefore, it is recommended not to go above 0.5. Wang et al. [19] exposed four circular Reactive Powder Concrete Filled Steel Tubular columns to explosion and fire durations of 0, 60, and 105 min. Fixed supports were simulated, and on one side, an axial load was introduced. Steel tube protects columns against cracks and spalling of concrete. After the detonation of the 17.5 kg explosive, the column experienced bending and after 35 kg bending–shearing deformations. With an increase in the number of explosives and with longer exposure to fire, maximum and residual displacements increased. Wang et al. [24] studied four square and four circular CFST columns under close-range blasts. The column length was 2.5 m, and the thickness of the steel tube was 2.8 mm or 3.8 mm. The standoff distance of the emulsion explosive was 1.5 m, and the charge weight expressed through the TNT equivalence was from 25 to 50 kg. Only a 10% increase in the amount of explosive in square columns increases mid-span deflection by 200%, which is assumed to be caused by a large surface exposed to the blast load. Moreover, increasing the thickness of the steel tube by 1 mm (from 2.8 to 3.8 mm) significantly reduced the displacement (by over 50%). Global failure mode was a flexural failure, and after removing the steel, the square columns sustained spalling and crushing of concrete. Circular columns were broken into several parts. Figure 10 shows the damage to the concrete after removing the steel tube after the blast load.

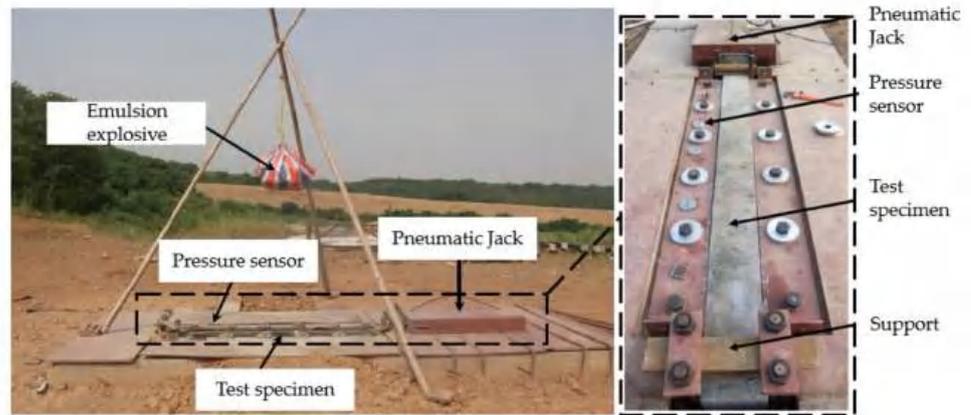


Figure 8. Test setup [24] and test pit [21]. Copyright permission obtained from authors.

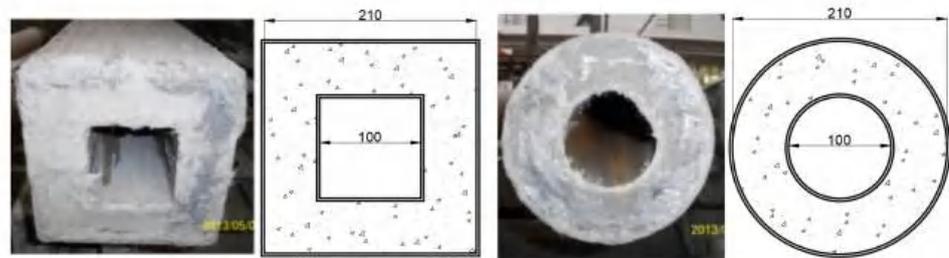


Figure 9. Cross-sections of two CFDST specimens [20]. Copyright permission obtained from authors.

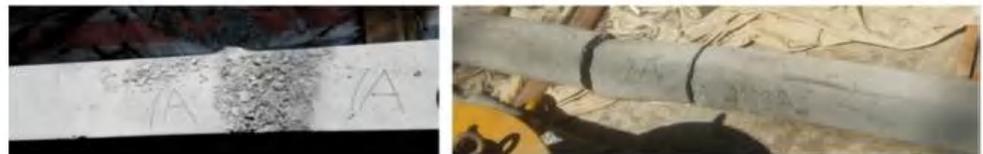


Figure 10. Columns after blast test [24]. Copyright permission obtained from authors.

Codina et al. [14] investigated the effects of close-in blast loading on the full-scale reinforced concrete column. They conducted experimental tests and numerical simulations to calibrate the numerical model. The observed column had a square cross-section of 230 mm × 230 mm and a free height of 2.44 m. The column was tested in a horizontal position, and the standoff distance from the center of the charge (8 kg of equivalent TNT—the used explosive is Gelamon VF65, which is equivalent in a mass to 65% TNT) to column 1 was 100 cm and to column 2 was 60 cm. Both columns experienced flexural damage, spallation of the concrete on the bottom side, and crushing of concrete on the exposed side, shown in Figure 11.

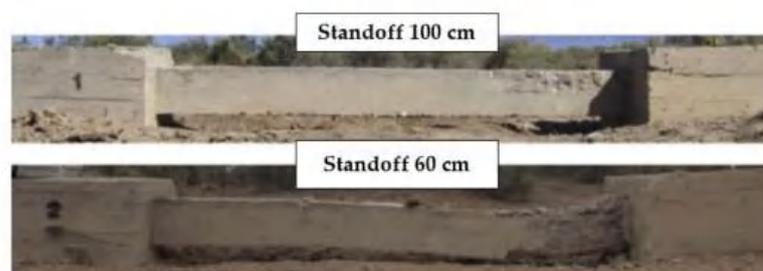


Figure 11. Damage of the RC columns after detonation at 100 cm and 60 cm standoff [14]. Copyright permission obtained from authors.

Codina et al. [15,16] investigated the behavior of RC columns and RC columns with two types of protection (steel jacket and reinforced polyurethane bricks) exposed to a near field explosion. At a standoff distance of 60 cm, 8 kg of TNT equivalent shaped into a cylinder was placed, resulting in a scaled distance $Z = 0.30 \text{ m/kg}^{1/3}$. Comparing the test results of three types of columns, the steel-jacketed column had the best results in residual capacity and in reducing final deflection. Polyurethane bricks are lighter and have cheaper protection for columns but give three times worse results than steel jacketing. It is recommended to set bricks of higher density to improve the effect in the area of blast load. In [16], they also examined reinforced resin panels with insulation layers as a possible improvement of the column. This protection system gave the best results, minimizing column damage and the greatest deflection reduction. The protection system in the field test is shown in Figure 12. Codina et al. [22], after their research on plain RC columns [14], tested RC columns covered with reinforced resin panels with an insulation layer and steel jacketing. From the obtained results, it can be concluded that a significant reduction in damage and displacement was achieved with the cladding system, but the spalling and burst of concrete cannot be prevented. Figure 14 shows a schematic representation of all specimens and their damage covered by the studies of Codina et al.



Figure 12. RC columns reinforced with: (a) polyurethane bricks [15,16] and (b) resin panels and insulation layer [22]. Copyright permission obtained from authors.

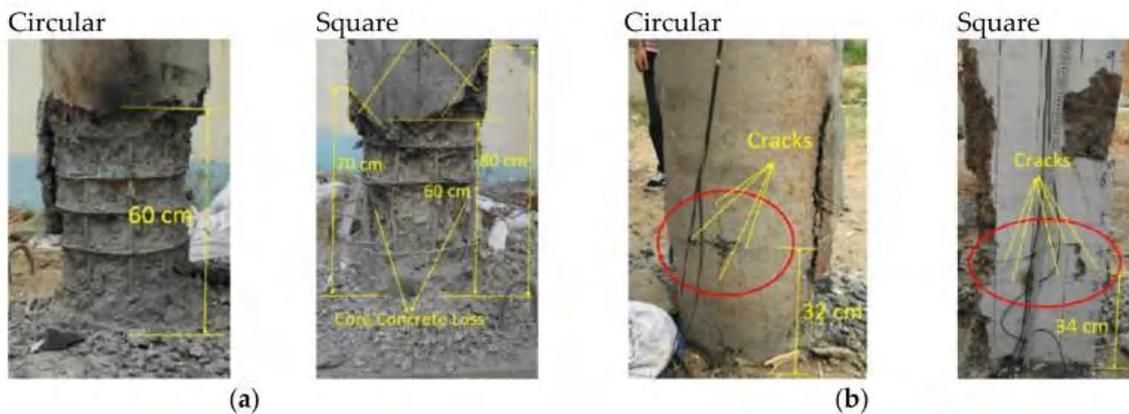


Figure 13. Comparison of damages of circular and square RC columns after the detonation of 1 kg of TNT: (a) front side and (b) back side [23]. Copyright permission obtained from authors.

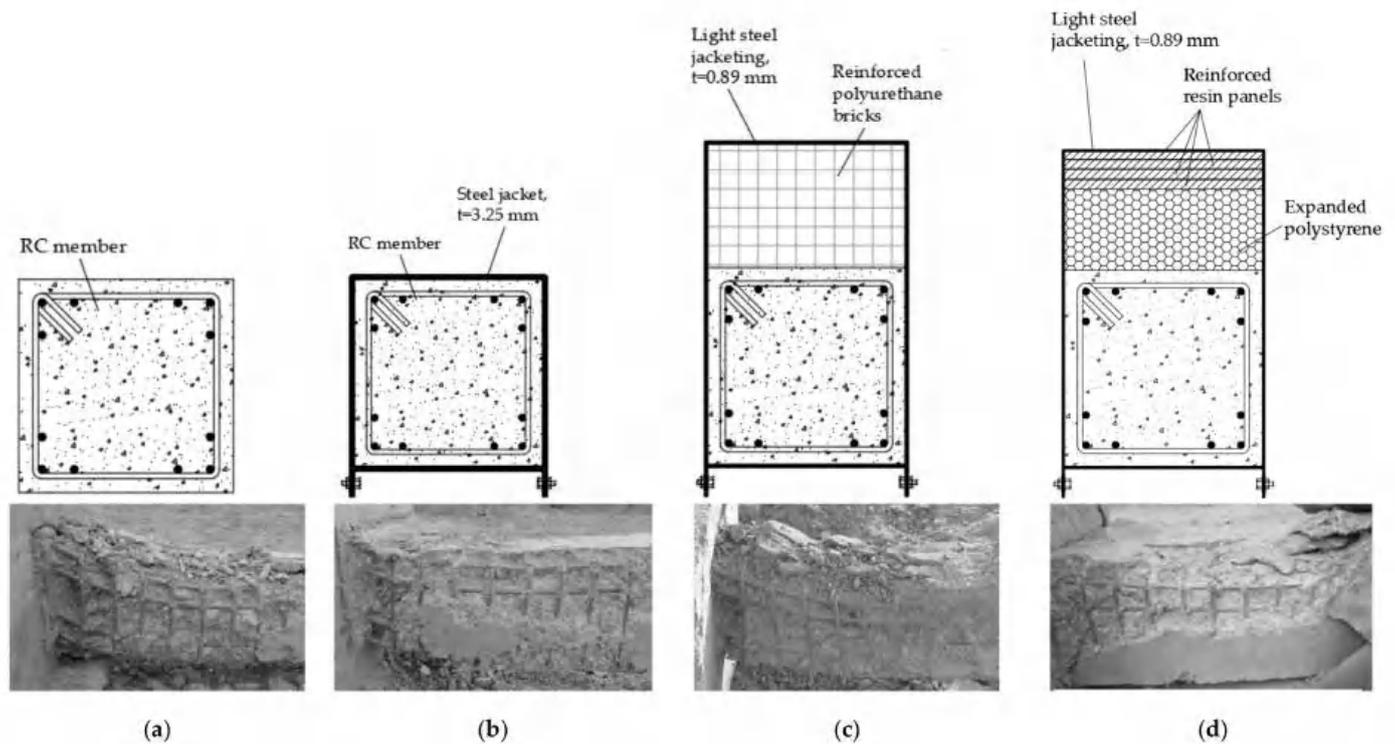


Figure 14. Damage after blast load on RC members: (a) without protection; (b) with steel jacketing; (c) with polyurethane bricks, and (d) with reinforced resin panels [15,16,22]. Copyright permission obtained from authors.

Xu et al. [17] subjected four Ultra-High-Performance Fiber Reinforced Concrete (UH-PFRC) and four High Strength Reinforced Concrete (HSRC) square columns to the effects of blast loads. Columns were tested under varied charge weights from 1.4 kg to 48 kg of emulsion explosive (TNT equivalence factor is 1.4). The standoff distance was fixed at 1.5 m in all tests. Specimens were placed horizontally, and the axial load was applied using a pneumatic jack. The results showed that UHPFRC columns could better withstand overpressure and shock waves, reducing the maximum displacements.

Li et al. [25] tested 10 Ultra-High-Performance Concrete (UHPC; six reinforced with twisted fiber and four reinforced with micro fiber) and 5 High Strength Reinforced Concrete (HSRC) columns. The length of the specimens was 2.5 m, and the cross-section was square with dimensions 0.2 m × 0.2 m. In experiments, the standoff distances of the explosive from the columns were constantly 1.5 m in all tests, but the explosive weight was changed. For the UHPC columns, weights of 17.5, 25, and 35 kg, and for the HSRC 8, 17.5 and 25 kg were used. Residual load capacity tests showed that UHPC columns after blast loads did not lose much on the axial load capacity. UHPC columns also showed much better load capacity after 35 kg TNT detonation than HSRC columns after 8 kg TNT detonation.

Fouché et al. [26] experimentally tested 12 columns at a scale of 1:4 under the blast load. They varied the void ratio, diameters, and thicknesses of the outer and inner steel tubes. The specimens' cross-section generally experienced denting, and that deformation helped to energy absorption from the overpressure from the near-contact explosions. The inner steel tube played the role of a dowel preventing direct shear failure, and this is the advantage of the CFDST columns over CFST and RC columns. On the tensile side of the column, the concrete was crushed, or horizontal flexure cracks appeared.

Dua et al. [27] tested three RC columns (30 cm × 30 cm × 375 cm) in full scale with the same material and geometrical properties on contact explosion at the bottom of the column. They used 0.1 kg of plastic explosive (PEK—TNT equivalent 1.15) and 0.5 kg and 1 kg of TNT. The TNT charge of 1 kg made a hole in the column; 0.5 kg destroyed the concrete cover, and the remaining core has no residual capacity; 115 g TNT equivalent caused the spalling

of the concrete cover. Column damage profiles are shown in Figure 15. A contact explosion causes significant local damage on at least three sides of the column, while a far-field explosion causes the worst damage on the front, exposed side. Dua et al. [28] investigated the same blast load scenario on the column concerning the increase in the cross-sectional width of the column. They experimentally tested columns of dimensions 50 cm × 30 cm, 70 cm × 30 cm, and 90 cm × 30 cm. Rectangular columns showed better behavior under contact explosion than squares. They examined the residual load-bearing capacity of the column and determined the column damage index. When the width dimension of the column subjected to the blast load is greater two and more times from depth, the damage index is lower.

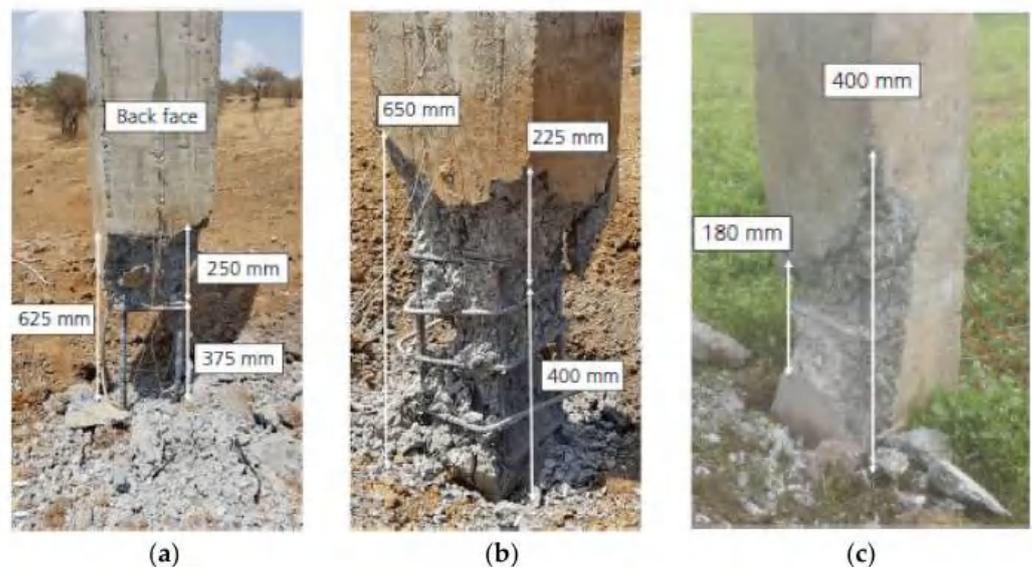


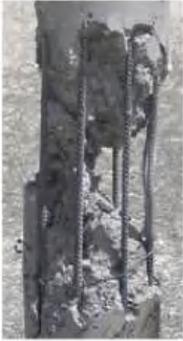
Figure 15. Column damages after blast loads: (a) 1 kg; (b) 0.5 kg and (c) 115 g TNT [27]. Copyright permission obtained from authors.

Kadhom et al. [30] examined five half-scale RC columns. Three columns are strengthened with unidirectional and woven CFRP laminates while the other two remained unprotected. They were first tested in the shock tube on the induced blast load, and thereafter, their residual axial capacity was examined. The column strengthened with CFRP laminates with $\pm 45^\circ$ woven fibers showed the best blast behavior and the most ductile response.

Fujikake and Aemlaor [43] investigated how longitudinal and shear reinforcement ratios, concrete strength, and the number of explosives affect the demolition of RC building columns. Their primary research goal is not terrorist attacks but the demolition of dilapidated concrete buildings. They used a Composite 4 (C4) explosive because of its stability and ease of shaping and placed it in the core of the column. They found that shear reinforcement plays a significant role in the residual bearing capacity after blasting; the strength of the concrete also affects the increase in residual compressive and flexural resistance capacities. The quantities of explosives and reinforcement cannot be applied to the external action of the explosion, but certainly, conclusions about the role of reinforcement and the strength of concrete are useful.

Roller et al. [44] observed two scenarios: first is a contact explosion with the amount of PETN that fits in the suitcase, and the second is the close-in scenario when the explosive is in the vehicle. They investigated the impact on RC columns and strengthened columns. Polymer concrete, SIFCON (Slurry Infiltrated Fiber Concrete), DUCON (Ductile Concrete), and Ultra-High-Performance Concrete (UHPC) were used to improve the resistance of the bridge and building columns. The appearances of column damages after contact explosion and the residual load capacities are shown in Table 3. The results showed an increase in residual bearing capacity by up to 70%.

Table 3. Damaged columns after contact detonation and the residual load capacities [44].

	RC	Polymer Concrete	SIFCON	DUCON	UHPC
Type					
Damage					
Residual load capacity	5.5%	68.6%	69.6%	49.3 (coarse)– 65.9 (fine) %	-

Xu et al. [45] tested five columns in an explosion containment vessel (ECV). The columns were exposed to an explosion of 40 g charge mass, and the distance of the explosive was changed in each test from a contact explosion to a standoff distance of 50 cm. They installed four smart aggregates (SAs) in each specimen for internal damage detection, shown in Figure 16. The propagation of the stress wave energy decreases with the formation of cracks under the blast load, and hence the amplitude of the time-domain signal recorded by piezometric smart aggregate sensors decreases with the appearance of cracks. This method of detecting internal damage has proven to be useful for completing the picture of the condition of the structure because internal cracks have a greater impact on the damage index of the structure than surface cracks.

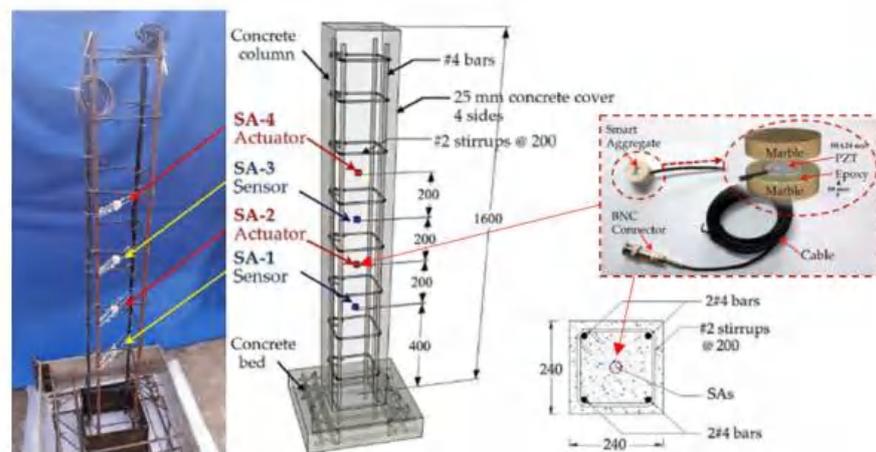


Figure 16. Position and appearance of the SA sensors [45].

Vapper and Lasn [31] examined columns measuring 100 mm × 100 mm × 1000 mm on the action of a different amount of explosion placed at a vertical standoff distance of 300 mm. They tested four types of columns: reinforced concrete columns, columns reinforced with steel fibers, and both types strengthened with Glass Fiber Reinforced Polymer (GFRP).

Plain concrete showed higher compressive strength compared to steel fiber reinforced concrete. GFRP in reinforced concrete columns did not contribute to the increase in the residual strength, while in steel fiber reinforced columns, the contribution was significant. A reduction in surface damage to GFRP-wrapped columns was also observed.

3. Numerical Modeling

Due to the increasing number of terrorist attacks, new challenges were posed to engineers. It is necessary to have a good understanding of computer programs, their capabilities, and their limitations to predict individual attack scenarios. The use of numerical simulations gives a clearer insight into the blast effects on the entire bridge and individual components. It provides the possibility of determining the most critical parts and problems that cannot be numerically simulated and need to be examined experimentally. Thus, one of the most important and difficult parts of this analysis is to properly define air blast loadings. Table 4 summarizes the software used to predict the effects of explosions on different types of columns.

Table 4. Summary of software for prediction and calculation of blast loads.

Author	Year	Structural Element	Software
Ray et al. [46]	2003	Bridge deck and column	ConWep, BlastX, SHAMRC
Marchand et al. [36]	2004	Bridge columns	BlastX, ConWep, SPAn32
Winget et al. [2]	2005	Bridge concrete girders, deck, columns	BlastX, SPAn32, Nonlin
Rutner et al. [47]	2006	Steel and composite bridge columns	MSC.Dytran
Wu et al. [48]	2009	RC and composite building columns	LS-Dyna
Hao et al. [49]	2010	RC building columns	CARLER
Elsanadedy et al. [50]	2011	RC building columns + CFRP	LS-Dyna
Williams et al. [51]	2011	RC bridge columns	LS-Dyna
Williams et al. [52]	2011	RC bridge columns	LS-Dyna
Crawford [10]	2013	RC building columns + FRP + SJ	LS-Dyna
Magali et al. [53]	2013	RC building columns	Abaqus
Eisa [54]	2014	RC building columns	Abaqus
Abladey and Braimah [55]	2014	RC building columns	Autodyn
Li and Hao [56]	2014	RC column	LS-Dyna
Shi and Stewart [57]	2015	RC building column	LS-Dyna
Liu et al. [58]	2015	RC bridge pier-bent model	LS-Dyna, ConWep
Cui et al. [59]	2015	RC column	LS-Dyna
Zhang et al. [12]	2015	CFST building columns	LS-Dyna
Codina et al. [14]	2016	RC building column	Autodyn
Zhang et al. [21]	2016	CFDST columns	LS-Dyna
Arowojolu et al. [60]	2017	RC bridge column	LS-Dyna
Eamon and Aslendi [61]	2017	RC bridge columns + SFRP	LS-Dyna
Kravchenko et al. [62]	2017	RC building columns	LS-Dyna
Kyei and Braimah [63]	2017	RC building columns	LS-Dyna
Yuan et al. [23]	2017	RC bridge columns	LS-Dyna
Abedini et al. [64]	2018	RC building columns	LS-Dyna
Li et al. [65]	2018	CFDST bridge columns	LS-Dyna
Liu et al. [66]	2018	RC bridge piers	LS-Dyna, ConWep
Li et al. [67]	2019	CFDST bridge columns	LS-Dyna
Liu et al. [68]	2019	RC building columns	Autodyn, LS-Dyna
Liu et al. [69]	2019	RC bridge column + CFRP	LS-Dyna
Thai et al. [70]	2019	RC column + SJ	LS-Dyna
Abedini et al. [71]	2019	RC column	LS-Dyna
Dua et al. [72]	2019	RC columns	LS-Dyna
Dua et al. [28]	2020	RC columns	LS-Dyna
Li et al. [73]	2020	CFDST columns	LS-Dyna
Rajkumar et al. [74]	2020	RC columns	LS-Dyna
Vavilala et al. [75]	2020	RC building columns + polymeric foam	Abaqus
Zhang et al. [76]	2020	Segmental CFST column	LS-Dyna
Yuan et al. [77]	2020	RC column	LS-Dyna
Yan et al. [78]	2020	RC columns + CFRP	LS-Dyna
Hu et al. [79]	2021	RC column + CFRP	LS-Dyna

Note: CFRP—Carbon Fiber Reinforced Polymer; FRP—Fiber-Reinforced Plastic; SJ—Steel Jacket; CFST—Concrete-Filled Steel Tube; CFDST—Concrete-Filled Double Steel Tube; SFRP—Steel Fiber Reinforced Polymer.

Conventional Weapon Effects Predictions (ConWep) [80] and BlastX [81] are programs used to calculate the effect of a blast wave from different types of detonation. ConWep is more used for air-blast calculations, including free-field and reflected blast pressure histories from the free-air, surface, and hemispherical burst explosions, and BlastX calculates internal blast pressure histories. BlastX is based on semi-empirical methods, including nonlinear addition laws for blast pressures from multiple reflecting surfaces based on computational fluid dynamics. Second-order Hydrodynamic Automatic Mesh Refinement Code (SHAMRC) [82] is also used to investigate high explosive and blast effects based on finite-difference computational fluid dynamics (CFD) code. Nonlin [83] does not make the empirical adjustments just for blast loads because it is initially designed for earthquake loads. It has similarities with SPAn32 [84] because it performs a nonlinear dynamic response history analysis taking bilinear material properties. Both programs are based on the analysis of a system with a single degree of freedom (SDOF). MSC.Dytran [85] is an explicit finite element analysis (FEA) solution for simulating blast load effects and analyzing the complex nonlinear behavior that structures undergo during detonation. Ansys Autodyn [86], LS-Dyna [87], and Abaqus [88] are programs that provide the ability to simulate detonation, wave propagation from an explosion, interaction with a structure, and nonlinear material behavior, which are known as hydrocode programs specialized for simulations in fluid dynamics.

Ray et al. [46], in their research, compared three methods with three different resolutions for air blast prediction. In the scenario of below-deck detonation, ConWep has a low resolution of air blast prediction, and the charge can be observed as hemispherical or spherical, while BlastX has a medium resolution and considers the shape of the charge and reflections of the blast pressure. The 3D bridge and blast load can be modeled in the SHAMRC because it is an advanced Eulerian-based finite difference code that has high resolution. Research shows that the highest resolution is not always necessary, as it is a mostly low resolution that provides a conservative design. The authors stated that additional analysis is needed to determine the most economical and sufficiently precise tool for a particular problem [46]. The shape of the explosive also drastically affects the resulting pressure and impulse, so it is necessary to use a program that allows the input of charge geometry.

Marchand et al. [36] determined concrete breaching using ConWep and calculated flexural response and support rotation on a reduced diameter column in SPAn32. They concluded that the strength of concrete does not significantly affect the maximum rotation of the support, but it does affect breaching, i.e., the lower strength causes greater breaching.

Winget et al. [2] use SPAn32 to calculate the flexural response of the columns and to define the equivalent SDOF stiffness and mass parameters based on the column properties. For the calculation of the blast load pressure history, the BlastX program was used. Other useful programs are AT Blast [89] and Nonlin. For the calculation of the pressure-impulse history using AT Blast, it is necessary to know the charge weight, angle of incidence, and standoff distance. However, AT Blast does not consider the effects of multiple reflections under the bridge explosions. They list four categories of bridge design concerning their importance, where category 1 represents very important bridges, and category 4, unimportant. Winget et al. did not take the real conditions of the ground and energy absorption by creating craters but the ideal reflecting surface. Footing instability, however, could also result from large ground deformations, and this aspect of behavior must also be addressed.

Wu et al. [48] numerically simulated RC and composite columns in LS-Dyna for the contact-placed TNT charges from 2.5 to 25 kg. In the simulations, they obtained a higher residual bearing capacity of the column when the explosive was placed at the height of 1.5 m from the bottom than when it was placed at the bottom.

Fujikura et al. [7], for calculation of impulse variations per unit length along the height of the column, were using the Bridge Explosive Loading (BEL) [90] program. BEL also considers the reflected pressure of the blast wave on the surface of the superstructure and on the ground.

Rutner et al. [47] studied the behavior of four types of column cross-sections in MSC.Dytran software on blast load: single-cell hollow steel section, multi-cell hollow steel section, single-cell hollow composite column, and multi-cell composite column. Compared to steel columns, the composites showed negligible deflection on blast load. The best stress distribution in the element was achieved with multi-cell composite columns, which is also visible by the displacements shown in Figure 17.

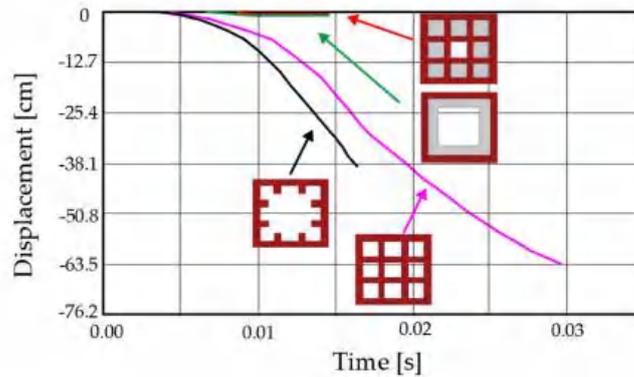


Figure 17. Time history of displacements for 4 column types [47].

Hao et al. [49] analyzed three reinforced concrete columns with the same dimensions, material strengths, and reinforcement ratios but subjected to blast loads of different scaled distances. They wanted to find the failure probability using the computer code CARLER, which is verified with Monte Carlo simulations. They defined four damage levels (D) that depend on the ratio of residual axial load carrying capacity ($N_{residual}$) when the column is damaged and the axial load of the undamaged column (N_0), shown in Table 5. Through numerical simulations, they concluded that neglecting some of the material properties of the column has minimal impact on the probability of failure, while the random changes in the blast loading have a much greater role.

Table 5. Damage levels of RC column in terms of axial load capacity.

Level of Damage [49]	$D=1 - \frac{N_{residual}}{N_0}$	Damage Limit States [57]
Low damage	0–0.2	Low damage
Medium damage	0.2–0.5	Repairable damage
High damage	0.5–0.8	Repairable damage
Collapse	0.8–1.0	Collapse

Williams and Williamson [51] emphasized the spalling of side-cover concrete because, in previous works, only the spalling of concrete off the back of reinforced concrete columns was mentioned. The aim of their research was to make and validate a numerical model with respect to the experiment explained in [38] and justified the cross-sectional response mechanisms that cause loss of side-cover concrete. For numerical simulations, they used the LS-Dyna program and the Karagozian and Case concrete (KCC) material model.

Numerical simulations in LS-Dyna showed that the shape of the column cross-section has a large influence on the resulting impulse. The authors developed expressions for calculating column shape factors for circular and square columns. The expressions are used when the R / D ratio is less than 4.5 because they provide sufficiently conservative loads but less than those experimentally determined on the walls [52].

Crawford [10] performed numerical simulations in LS-Dyna of RC columns and columns retrofitted with fiber-reinforced plastic (FRP). FRP increases the resistance of the RC columns to the blast load. For numerical modeling, the choice of concrete material model is very important, and the analysis was performed with four different concrete

models (KC, Winfrith, Continuous Smooth Cap, and RHT model). The best results were obtained with the KC model.

Magali et al. [53] performed a parametric numerical analysis in Abaqus to see which of the six varied parameters (section ratio, compressive strength of concrete, column height and thickness, charge radius, and ratio between standoff distance to the charge radius) had the greatest impact on column damage. It was shown that the column thickness, charge radius, and the ratio of standoff distance to charge radius play a significant role in the column response. They give an empirical formula based on the conducted simulations to predict the damage index of the column. Comparing the results obtained by formula and numerical simulations, the deviations are up to a maximum of 15% what is in an acceptable range.

Eisa [54] modeled RC columns in Abaqus. The position of the spherical charge remained unchanged, but the charge weight, column height, longitudinal steel reinforcement, columns aspect ratio, and transverse steel ratio were varied. Fixed boundary conditions are provided at the top and bottom of the column. In addition to column damage, they measured acceleration and displacement concerning varied parameters. Four quantities of explosives were used (45.36, 226.80, 453.59, and 1016.05 kg) and placed at a distance of 4.87 m. The increase in the lateral reinforcement in the column had the effect of reducing the displacement in the middle of the column. It is recommended to examine the influence of the axial force during the blast load and to include additional parameters such as the variation in the standoff distance of the charge.

In numerical simulations using Autodyn, Abladey and Braimah [55] tested three columns designed for different loads in accordance with the Canadian reinforced concrete design code [91]. The first type of column is designed only for gravity loads, and the distance between the transverse reinforcement is 300 mm, and the other two types of columns are designed for seismic loads, but in the second type, the distance between transverse reinforcement is 150 mm, and in the third 75 mm. Column damage is significantly less with denser reinforcement, especially at small-scaled distances. They have proven in research that regardless of the same scaled distance, in a situation where a larger amount of explosive is detonated, the column has a higher deflection.

Li and Hao [56] calibrated the numerical model for RC slab in LS-Dyna according to a previously performed experiment and then used that numerical model for RC column simulation. For good simulation of concrete spallation, the erosion criterion using principle tensile strain of 0.01 was defined. Through simulations, they concluded that denser reinforcement and greater column depth reduced spall damage, i.e., increase the confinement of concrete. The boundary conditions and column height do not play a significant role in the level of spall damage in close-in cases.

Shi and Stewart [57] analyzed a spatial and non-spatial simulation of the blast load on RC columns in LS-Dyna. They used three quantities of ANFO explosives, 50, 100, and 1000 kg, at distances from 0 to 30 m. The analysis is based on axial load-carrying capacity and concluded that the variability of the results in the spatial model is lower, and the probability of damage is significantly higher. They consider the spatial model more reliable and recommended it for future research with an additional assessment of the scale of fluctuation.

Elsanadedy et al. [50] used LS-Dyna for analyzing the behavior of the exterior building RC circular column and strengthened column with Carbon Fiber Reinforced Polymer (CFRP) sheets under blast load. Four different charge weights (100, 200, 500, and 1000 kg) of TNT at three different standoff distances (1, 4, and 15 m) and at 1 m height from the ground were analyzed. They modeled columns with different boundary conditions, first with both fixed ends and a second type with both hinged ends. For calculation of blast load parameters in all assumed scenarios, they used the software ConWep. The use of CFRP is increasing the shear capacity of the column and the strength of the column, which results in less lateral displacement, and more layers of CFRP can undergo more intense

blast loads [50]. Moreover, at the scaled distance from 0.50 to 0.68 m/kg^{1/3}, columns with CFRP showed better behavior than RC columns.

Ashalekshmi and Subha [92] modeled a bridge column in the Ansys Autodyn software to analyze the impact of concrete grade and spacing of ties under the blast load. They observed total deformation and principal stress for concrete grades M40 and M50 and ties spacings of 10 and 20 cm in the near and far-field. In the near field, the explosive is placed at the same standoff distance of 2 m, and the charge weight varies from 250 to 1500 kg. The influence of the concrete grade on the maximum deformation is visible only at the weight of explosives greater than 750 kg. With concrete M50, the principal stress is higher, but there is no big difference in it when increasing the weight of explosives. In the far-field, the explosive was placed at a standoff distance of 10 m, and the weight also varied from 250 to 1500 kg. There is a slight difference in total deformation concerning the grade of concrete. The principal stress exceeds the strength of concrete, and the difference in stress is visible for the grade of concrete. The effect of tie spacing is visible in the near field only in the increase in the maximum principal stress, while in the far-field, there is no effect on either deformation or principal stress.

Liu et al. [58] modeled the bridge column and bent it in the LS-Dyna for three design categories provided in [35]. They determined six damage mechanisms in the models, four of the column and two of bent. In all three design categories (A, B, C), spalling of concrete and crushing of the bent concrete were observed. Plastic joints in the column and shear of bent occur only in B and C categories. The shear or flexure failure of the column is most probable in category C, i.e., at the highest blast load. They found that increasing the transverse reinforcement reduces damage. In LS-Dyna, they received underestimated blast load, and therefore they used the ConWep program to calculate pressure-time diagrams.

Cui et al. [59] concluded through numerical simulations of columns in LS-Dyna that a larger cross-section and reinforcement ratio, smaller spacing between the stirrups, and a thinner concrete cover for columns exposed to close-in explosions give less damage.

Zhang et al. [12] analyzed circular and square CFST (Concrete Filled-Steel Tube) columns with tube thicknesses of 2.8 and 3.8 mm. The Emulsion explosive was used, with TNT equivalences of (0.7) 17.5, 25, and 35 kg. A numerical simulation of the columns was performed in LS-Dyna, see Figure 18, but to reduce the computation time and model congestion, an air blast model was made in the ConWep program and then imported into LS-Dyna. By comparing the obtained periods of oscillation and maximum displacements, a good match between the numerical model and the experiment was obtained. Differences are found only in residual deflections but are not considered crucial to the accuracy of the model.

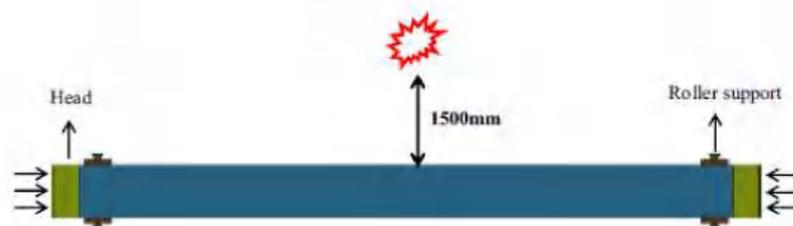


Figure 18. The numerical model of the blast test [12]. Copyright permission obtained from authors.

Zhang et al. [21] numerically modeled CFST columns with steel-fiber reinforced concrete using LS-Dyna and compared them with experimentally obtained results. For concrete, the KC model was used, and for the steel tube, the Material model 24 was used. The parameters for the concrete model were modified (f_t —tensile strength, B_1 —parameter for residual strength, w_z —strain softening, ω —confinement effect, λ , and η —damage parameters) because UHPC was used. Column erosion occurs when the maximum shear strain value reaches 0.045. Numerical research concluded that the axial load up to a certain limit has a favorable effect on the deflection in the middle of the column. It is

not recommended to use columns with a hollow section ratio greater than 0.5. Increasing the thickness of the outer steel tube affects the decrease in deflection in the middle of the column. Columns filled with UHPC have less plastic deformation than columns filled with normal strength concrete, shown in Figure 19.

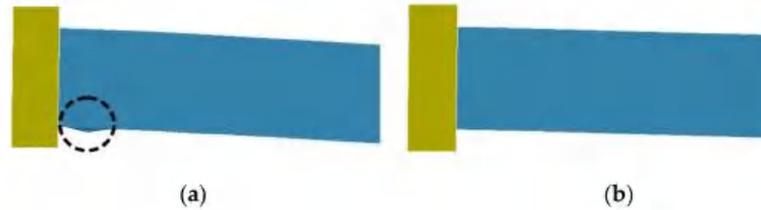


Figure 19. Failure mode of CFDST column with: (a) normal strength concrete and (b) UHPC [21]. Copyright permission obtained from authors.

Codina et al. [14] compared the numerical simulation with an already conducted experiment. The used explosive was Gelamon VF65 (65% TNT equivalence), 8 kg TNT equivalence at distances of 1 m and 0.60 m from the column, which is classified as a near-field range. Overpressure, impulse, deflection, acceleration, and visual damage to the column were measured and compared. For numerical modeling, the Ansys Autodyn program was used in which air and explosives were modeled through an Euler processor and the column through Lagrange. The optimal mesh size of concrete, steel, and the air was 10 mm. The authors made models with default RHT values and with the values proposed in [93] but concluded that the parameters that affect the strength degradation (damage factors D_1 and D_2 , and e_{min}^{fail}) and the residual strength, Y_{fric}^* (parameters B and M) should be changed in the model. For good prediction of spallation, the instantaneous geometric strain was used for erosion type with a value of 0.5. The obtained column damage with different RHT parameters is shown in Table 6. The parameters of the RHT material model (shown in Table 3) are validated for scaled distances (Z) from 0.5 to 0.3 m/kg^{1/3}.

Table 6. Column damage obtained: (a) experimentally and numerically using (b) default RHT model parameters; (c) parameters provided in [93]; (d) modified parameters by [14].

Parameters (a)	Autodyn (Default) [86] (b)	Tu and Lu [93] (c)	Codina et al. [14] (d)
B	1.6	0.7	0.35
M	0.61	0.8	0.55
RHT damage model			
D_1	0.04	0.015	0.8
D_2	1	1	1
e_{min}^{fail}	0.01	8.00×10^{-4}	0.03

Arowojolu et al. [60] studied, using LS-Dyna numerical models, the influence of axial and blast load on the RC column of the bridge. For the concrete model, they used CSCM (Continuous Surface Cap Model) and for reinforcement MAT 24. Exact quantities of explosives and distances are not given, but scaled distances from 1.77 to 0.45 m/kg^{1/3}. They concluded that when an axial load ratio of 0.25 is applied, the displacement in the middle of the column decreases but the damage of the column increases.

Eamon and Aslendi [61] made a numerical model of the column, experimentally tested in [4] using LS-Dyna software. The influence of concrete strength, reinforcement ratio, axial load, and the column wrapping with SFRP (Steel Fiber Reinforced Polymer) was

observed. The Johnson–Holmquist–Cook (JHC) model was used for concrete modeling and the elastic–plastic kinematic model for steel reinforcement. The impact of the blast load was determined in the ConWep software, and the detonation point was placed 5 cm above the ground, 40 cm from the column. SFRP proved to be an inexpensive and ductile retrofit. One layer has the largest contribution in blast capacity, while all additional layers have a small effect on increasing the capacity. They obtained a linear relationship between the concrete strength and the increase in the blast load capacity.

Kravchenko et al. [62] performed numerical simulations of the RC column in LS-Dyna. The concrete was modeled using the CSCM concrete model (type 159 material) and reinforced using the plastic–kinematic model (type 3 material). They observed the influence of detonation of 10 kg of TNT at a distance of 1.2 m from the ground and 1 m from the column. They also concluded that the reinforcement ratio has a significant impact on the behavior of columns under the blast load and that the columns in the ground floor and bases need to be better reinforced due to their easy accessibility.

Kyei and Braimah [63] modeled three RC columns in LS-Dyna, which differ in the distance between the transverse reinforcement. They designed the columns according to the instructions for the level of seismicity in the Canadian concrete design code [91]. Concrete was modeled using the Continuous Surface Cap Model (MAT_CSCM_159), and for reinforcement, they used Material Piecewise Linear Plasticity (MAT_024) model, while blast load was calculated in ConWep and then imported with Load Blast Enhanced (LBE) in LS-Dyna. They performed simulations with mesh sizes from 5 to 100 mm, and with 15 mm, they obtained a good ratio of the time spent for the calculation and the accuracy of the results compared with the experiment in [94]. The used explosive was ANFO (100, 250, 500, and 1000 kg) at scaled distances of 0.8, 1.0, and 1.5 m/kg^{1/3}. In near-field explosions, the distance between the transverse reinforcement has a significant effect on the reduction in displacement, while in far-field explosions, this effect is negligible. At a high axial load ratio (0.35), the seismically designed columns showed better behavior at scaled distances than the standard ones.

Yuan et al. [23] performed numerical simulations in LS-Dyna of the circular and square columns of the bridge, exposed to the contact explosion of 1 kg of TNT. To reduce the computation time, at the height of 1 m, in the area of the contact explosion, they placed a denser mesh (8 mm), while on the rest of the column, the mesh size was coarser (20 mm), shown in Figure 20. The principal strain of 0.5 was used as the erosion criterion. Numerical simulations well described the damage on the front sides of the column, while on the back, there are differences. They concluded that the damage was greater on the square column due to the flat surface and the higher stress concentration than on the circular column. The damage of the columns is shown in Figure 21.

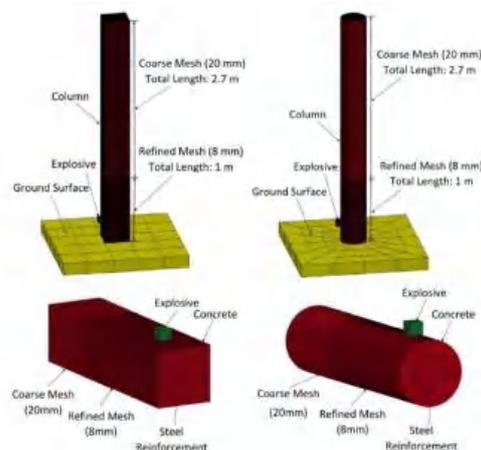


Figure 20. Detailed views of a mesh of 3D column models [23]. Copyright permission obtained from authors.

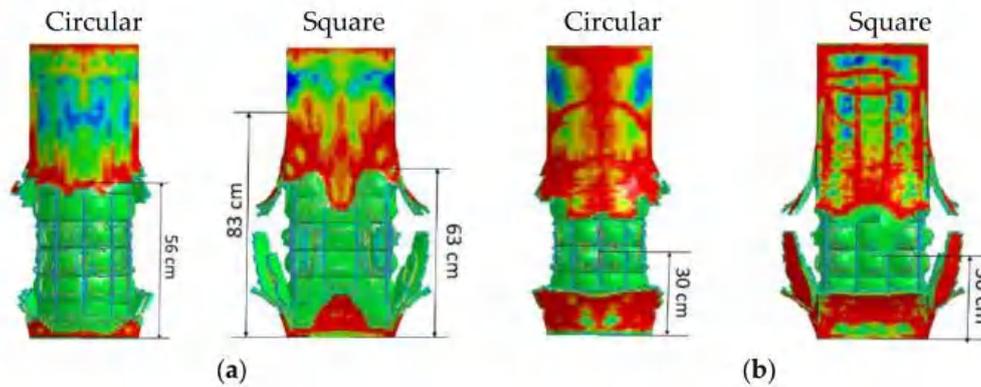


Figure 21. Comparison of damages of circular and square RC columns after the detonation of 1 kg of TNT: (a) front side and (b) back side [23]. Copyright permission obtained from authors.

Yuan et al. [77] investigated the effect of axial load on RC bridge columns subjected to far-field, close-in, and contact explosion. Columns in the far-field have mainly a flexure response, and the axial load affects the reduction in the maximum displacement in the middle of the column. In the case of a close-in detonation, a shear failure is expected, and the axial load affects the increase in the damage of the column and should not be neglected. In contact detonation, the concrete covers at the front and backside of the column spall off. The axial load reduces the damage of the concrete but increases the stress in the reinforcement and must be considered.

Li et al. [65] conducted numerical investigations on CFDST columns under contact explosion in LS-Dyna. CFDST columns have proven to be good for two reasons: the first is that the confinement of concrete by steel tube allows better energy absorption, and the second is the prevention of the spallation of the concrete cover. In work [73], the researchers concluded that increasing the cross-sectional area and the ratio of reinforcement plays a significant role in the post-blast residual capacity of CFDST columns under contact explosion. In [67], the behavior of CFDST columns subjected to close-in blast loading was studied. They concluded that the influence of the charge shape significantly affects the response and behavior of the column at scaled distances from $0.079 \text{ m/kg}^{1/3}$ to $0.175 \text{ m/kg}^{1/3}$.

Liu et al. [66] performed a dynamic and static analysis of the bridge columns in LS-Dyna. Dynamic analysis is based on the comparison of accelerations and static analysis on the determination of damage through the ratio of the residual to the ultimate axial bearing capacity. They examined the damage to the column concerning the position of the explosives, at the bottom, in the middle, and at the top, shown in Figure 22. In all three cases, bending deformation occurs at the column, but the position of the crack formation differs.

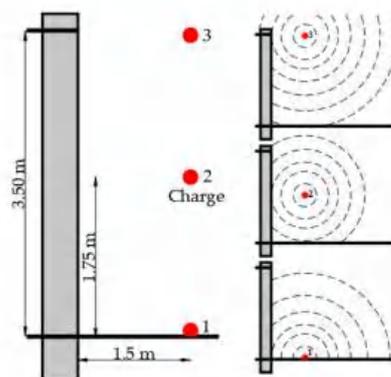


Figure 22. Shock wave propagation depending on the position of the explosive [66].

Liu et al. [68] performed a parametric analysis in LS-Dyna to determine the influence of longitudinal and transverse reinforcement ratios, longitudinal force, and boundary conditions. They concluded that at smaller-scaled distances, the increase in transverse and longitudinal reinforcement reduces the displacement in the middle of the column. It is recommended that the percentage of longitudinal reinforcement does not exceed 6% of the cross-sectional area of the column because too much reinforcement can lead to brittle failure. Analyzing the influence of the axial compressive load, they found that in an amount of up to 40%, it reduces the maximum displacement in the mid-span of the column due to the increase in moment capacity. The conclusions are based on close-in blast loading; a recommendation for future research is to conduct a parametric analysis for near and far field scenarios.

Liu et al. [69] made numerical models of RC columns strengthened with CFRP in LS-Dyna. The size of the mesh elements for all materials was 10 mm, and for air, 20 mm. The 1 kg and 2 kg TNT charges were placed in contact with the column at the height of 30 cm. The numerical results show that the dragging force of the blast load separates the CFRP from the concrete, but despite this, CFRP protects the column from contact explosion. Column damage is shown in Figure 23.

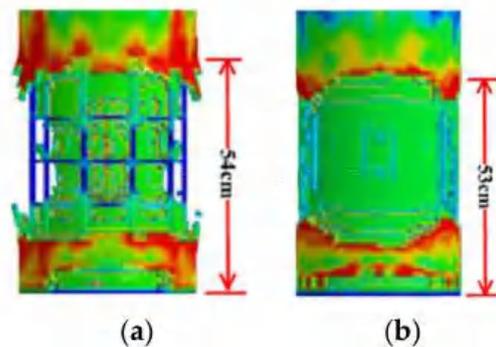


Figure 23. Column damages after detonation of 1kg TNT: (a) RC and (b) RC + CFRP column [69]. Copyright permission obtained from authors.

Thai et al. [70] modeled in LS-Dyna RC steel jacked columns (25 cm × 25 cm × 360 cm) and observed the influence of steel thickness, scaled distance, and longitudinal compressive force on the behavior of the columns under the blast load. The columns are designed according to Eurocode 2. A 10 mm mesh was used for the column, while 5 mm elements were used for the explosive. Placing 8 kg of TNT in the middle of the column causes less global damage to the column, while placing the same amount at a distance of 32 cm from the ground causes significant local damage. The scaled distance was varied from 0.10 to 0.40 m/kg^{1/3}. Increasing the steel thickness from 3 mm to 6 mm did not result in less damage.

Abedini et al. [33] investigated numerically in LS-Dyna the influence of charge and scaled distance on the level of column damage and influence of column width, concrete strength, and reinforcement ratio on the residual axial load capacity.

Dua et al. [72] performed a parametric analysis in LS-Dyna on the RC columns exposed to the contact explosion. They used from 115 g to 1000 g TNT, varied the compressive strength of concrete from 20 to 55 MPa, and reduced the distance of the transverse reinforcement from 200 to 50 mm. Increasing the transverse reinforcement reduces the damage to the concrete core, and higher compressive strength of concrete contributes to the reduction in cracks and peeling. The columns under contact explosion have local damage, while the global damage is negligible. Dua et al. [28] investigated in LS-Dyna the influence of the column cross-sectional width under the contact explosion. They concluded that a larger cross-sectional width (greater by two or more times from the depth) has a favorable effect on the behavior and damage of the column.

Rajkumar et al. [74] examined 45 numerical models of reinforced concrete columns in scale 1:4 (85 mm × 85 mm × 900 mm) in LS-Dyna. In the models, the behavior of four different cross-sections (circular, octagonal, hexagonal and square) on the blast load was examined. The circular column retains the lowest peak pressures and has the smallest deflection in the middle, while the square has the highest pressures and the largest deflection. The edges in cross-sections at small-scaled distances play a significant role in pressure retention, while with increasing scaled distance, the shape influence decreases. An increase in reinforcement in cross-section affects the improvement of the behavior of the columns during the blast load, especially in shapes that retain higher pressures.

Vavilala et al. [75] numerically simulated polymeric foam retrofitted RC columns (35.5 cm × 35.5 cm × 348 cm) in Abaqus. They used a 10 mm mesh for reinforcement and 20 mm for concrete. Columns coated with 5 mm, 8 mm, and 10 mm thick foam were exposed to 10, 25, and 50 kg of TNT. The greatest reduction in displacement in the middle of the column was obtained under 10 kg of TNT when the column was coated with 10 mm thick foam.

Zhang et al. [76] compare in LS-Dyna the behavior of a segmental CFST column with monolithic and prestressed monolithic columns. The columns were exposed to 20 and 50 kg of TNT at a standoff distance of 1.5 m. The segmental column showed a smaller residual displacement, and numerical analysis proved that a larger number of segments had a more favorable effect on the behavior of the column. Moreover, increasing the steel thickness had a beneficial effect on reducing concrete damage.

Yan et al. [78] and Hu et al. [79] used LS-Dyna for numerical simulations of RC columns retrofitted with CFRP subjected to the close-in explosion. CFRP sheets reduce the deformation and spalling of concrete. The CFRP thickness, wrapping, and dimensions ratio of the charge also had a large impact on the damage of the column and peak pressure. Debonding of CFRP is the most common form of failure, but despite this, CFRP has a role in reducing column damage during the blast load, and it may even have a role in changing the failure mode from shear to flexural deformation. The setting of the CFRP on both sides of the column needs to be further investigated because direct shear is possible due to over-reinforcement.

4. Discussion and Conclusions

4.1. Experimental Testing

Most of the experiments were conducted for the scenario of an attack by an auto-bomb located near a bridge or building column. Experimental tests on building columns are mostly full-scale, while tests on bridge columns predominate on scales of 1:3 and 1:4, due to the high cost of performing such experiments, the need for trained personnel to handle explosives, large quantities of explosives, and the field where such tests can be carried out.

Experiments showed that even a minimal increase in the cross-sectional dimensions of a column could favorably affect the behavior of the column under the blast load. Moreover, a minimal increase in the standoff distance reduces the impact and intensity of the blast load, and therefore it is necessary whenever possible to fence the column, increase visibility around the column, and reduce its accessibility. In addition to the dimensions of the column, the shape of the column plays a significant role. Circular columns retain less impulse from the blast load than square ones of the same dimensions. Squares have a larger cross-sectional area and, therefore, can better withstand shear. On the action of the contact explosion, the circular column suffered less damage than the square. However, comparing a square and a rectangular column, when the width dimension of the rectangular column is greater two and more times from depth, damage of the column is lower. The shape of the column on the blast load needs to be further investigated. In RC columns, position, quantity, anchoring, and reinforcement overlap have a great impact. Therefore, seismically designed columns have better blast load behavior than standard designed columns.

Comparing steel jacked columns, CFST and CFDST with RC columns, all showed better ductility, less cracking of the concrete, and the absence of flying debris. However,

CFDST showed the best behavior, as the inner steel tube contributes to the prevention of direct shear. With this type of column, it is important that the ratio of the cavity to the column cross-section is not greater than 0.5 and that the thickness of the steel tube is well determined (a thickness of 3.8 mm gave satisfactory results).

Columns with Ultra-High-Performance Concrete and with various Fiber-Reinforced Polymers showed better load-bearing capacity than High-Strength Reinforced Concrete.

The influence of axial load up to 30% of the total load capacity has shown a favorable effect on the reduction in the maximum displacement, but this percentage needs to be further investigated.

Recent research applies smart aggregates to measure internal cracks and internal damage because the column can be damaged and reduced load-bearing capacity without being visible on the outside. Therefore, post-blast tests are performed to determine the residual strength and ductility of the column.

4.2. Numerical Modeling

Numerical simulations make it possible to study the effect of large amounts of explosions (more than 1000 kg) on columns in full-scale. The most widely used software for analyzing the nonlinear behavior of elements on blast load is LS-Dyna. The most accurate simulation of the blast pressure requires as finer mesh as possible, which leads to a long duration of simulations and congestion of the computer processor. Due to the use of a larger mesh size than recommended, the LS-Dyna underestimates the pressures. Most researchers use the ConWep program to calculate pressures and import the resulting pressures into LS-Dyna.

Defining a model of concrete material is the most demanding because many parameters affect its behavior. Karagozian and Case (KC) concrete is mostly used in LS-Dyna. The proper definition of erosion criteria has proven to be very important in modeling column damage.

The position of the explosive plays a significant role in the behavior of the column. Placing the charge in the far-field causes a uniformly distributed load per column and global response, while a charge placed in the near-field, close-in, and contact creates local damage.

Moreover, the columns exposed to the charge placed in the lower half showed greater damage and lowered residual capacity than the columns where the charge was placed in the middle.

In all fields (far, near, close-in), changes in the quantity, shape, and position of the explosives showed a great impact on column behavior. The shape of the column, the ratio of reinforcement, and the concrete grade showed an influence only at small-scaled distances. The concrete grade does not affect the rotation of the column but does affect the reduction in concrete breach and spallation.

Until recently, the axial load on bridge columns was neglected in the calculation because it was considered to be on the safe side. However, numerical simulations showed that axial load has a large impact on increasing damage when the charge is placed near the column or in contact with the column and should not be neglected.

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Article

Mesoscale Equivalent Numerical Study of Ultra-High Performance Concrete Subjected to Projectile Impact

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Abstract: Numerical investigations on the performance of ultra-high performance concrete (UHPC) subjected to projectile impacts have attracted extensive attention, but there are still deficiencies in the accuracy and computational efficiency of related simulation methods. To make up for these deficiencies, a mesoscale equivalent model for UHPC is developed to simulate the response of UHPC under projectile impacts. In this model, an equivalent treatment is conducted on steel fibers to reduce their quantity under the premise that the interfacial shearing force between the fibers and the matrix remains equal. Based on the mesoscale equivalent model, numerical simulations of uniaxial compressive tests and projectile penetration tests on UHPC specimens are performed in LS-DYNA, and the numerical results are compared with the corresponding experimental results to verify the developed model. It is found that the mesoscale equivalent model could accurately reproduce the failure mode and stress-strain curve of UHPC specimens when the amplification factor of steel fibers is lower than 5. When the amplification factor is 5, the computational efficiency of the numerical models for penetration tests is significantly improved, and the maximum relative error between the numerical results of the crater diameter and penetration depth and experimental results is 11.7%. The successful application of the mesoscale equivalent model provides a more precise and in-depth perspective in simulating the response of UHPC with steel fibers subjected to projectile impact. Then, the influence of projectile striking velocities, UHPC compressive strengths, and volume percentages of steel fibers on the depth of penetration (DOP) are further numerically assessed. Based on the simulated data, modifications of the Young equation for predicting the DOP are conducted, and the maximum relative error of the modified equation is 13.9%. This demonstrates that the modified Young equation can accurately predict the DOP of UHPC subjected to projectile impacts.

Keywords: UHPC; mesoscale equivalent model; penetration experiments; parametric analysis; DOP prediction equation

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1. Introduction

With the enhancement of the penetration capability of earth-penetrating weapons and the frequent occurrence of local wars, studies on the resistance of new building materials under projectile impacts have aroused wide attention from engineers and researchers. Ultra-high performance concrete (UHPC) is a kind of cement-based composite material with high strength, high toughness, excellent ductility, and good energy absorption capacity, which has a promising application on protective structures that may be subjected to projectile impacts [1,2]. Experimental investigations on UHPC structures against projectile impacts have been carried out over the past few decades [3–7]. However, it is expensive and time-consuming to conduct penetration experiments, which leads to the fact that the current experimental investigations are mainly aimed at small-caliber bullets or reduced-scale projectiles. Meanwhile, it is usually difficult to obtain the expected mechanical data in

penetration tests due to the ultra-high strain rate and deceleration, which hinders the in-depth understanding of the penetration mechanism of UHPC.

Recently, efforts have been put into the numerical investigation of the dynamic response of UHPC subjected to projectile impacts, hoping to supplement the deficiencies of experimental investigations and even replace the costly penetration experiments. Prakash et al. [8] carried out numerical investigations on high-velocity projectiles penetrating steel fiber reinforced concrete (SFRC) panels with a volume percentage of steel fibers ranging from 0% to 10% using a modified RHT model. A design chart for determining the optimal panel thickness under different fiber contents and projectile kinetic energies was compiled. Blasone et al. [9] investigated the mechanical behavior of the ultra-high performance fiber-reinforced concrete (UHPFRC) under an armor-piercing projectile using the coupled plasticity-damage model DFHcoh-KST. The damage mechanisms are well simulated by the numerical tool, and it was found that the softening behavior provided by fibers had a significant influence on the damage pattern of UHPFRC targets under projectile impacts. Wan et al. [10] calibrated a set of HJC model parameters for UHPC with steel fiber according to the material test data and then simulated experiments of two different kinds of bullets penetrating UHPC targets using the modified HJC model. The numerical results showed that the modified material model could accurately simulate the depth of penetration (DOP), but it underestimated the damage, especially the tensile damage, inside UHPC targets. Liu et al. [11] presented a numerical study evaluating the performance of UHPC (90–190 MPa) under ogive-nosed projectile impacts with striking velocities from 300 m/s to 1000 m/s, where the effects of the compressive strength, projectile striking velocity, and projectile CRH on the DOP and cratering damage of targets are discussed by applying a calibrated and validated K&C material model. An empirical equation for the DOP prediction was proposed. In this work, the numerical results of projectile impacts lack sufficient experimental comparisons and verifications, and it was found that the empirical equation overestimates the penetration resistance of the UHPC. Zhou et al. [12] established a novel dynamic constitutive model for UHPC based on the KCC model, which was applied to numerically predict the resistance and damage pattern of UHPC subjected to projectile impacts and achieved good accuracy. Liu et al. [13] numerically explored the effects of steel wire mesh on the penetration resistance of reactive power concrete (RPC). An empirical equation was proposed to predict the DOP of steel wire mesh-reinforced RPC targets subjected to high-velocity projectile penetrations.

Generally, UHPC is treated as a homogeneous material when conducting simulations on UHPC targets subjected to projectile impacts, which has shown some shortcomings in reproducing and explaining the dynamic response. In fact, UHPC should be regarded as a two-phase composite material consisting of matrixes and reinforcing fibers. The addition of fibers is an important contributor to the high performance of UHPC [14]. Hence, treating UHPC as a homogeneous material is no longer appropriate, and a three-dimensional model that models the matrix and fibers discretely should be established. Zhang et al. [15], Liang et al. [16], and Peng et al. [17] built a mesoscale model to simulate the mechanical properties and failure characteristics of UHPC under static and dynamic loadings. The numerical results are in excellent agreement with corresponding experimental results, and the mesoscale model provides a mesoscopic perspective to analyze the responses of UHPC. Smith et al. [18] carried out simulations of UHPC slab penetration and perforation experiments using a mesoscale discrete particle model. It was found that the mesoscale model perfectly reproduces cratering damage, spalling damage, and the crack-bridging effect in UHPC slabs. Although the mesoscale model performs better than the homogeneous model in the penetration simulation, the computational cost of the mesoscale model is too high to be used in some large-scale penetration simulations. The main reason of the high cost of the mesoscale model is that the number of established fiber elements is too large. Therefore, some equivalent treatment must be performed on fibers to reduce their quantity in the penetration model to improve computational efficiency.

The mesoscale equivalent treatment on fibers in UHPC has rarely been studied before. In the present study, an efficient mesoscale equivalent model based on the bond-slip constitutive between steel fibers and matrixes is first developed for simulating the dynamic response of UHPC under projectile impacts and is verified by the uniaxial tests and penetration tests. Then, the validated numerical model is applied to numerically investigate the effects of the projectile striking velocity, UHPC compression strength, and volume percentage of steel fibers on the DOP of UHPC targets subjected to projectile impacts. Moreover, a modified Young equation for predicting the DOP of UHPC subjected to the projectile penetration is fitted in terms of the simulated data.

2. Penetration Experiments

2.1. UHPC Targets

In the present tests, two cuboid UHPC targets are cast for the projectile impact under different striking velocities, as shown in Figure 1. The plain sizes of the two targets are both $2.0\text{ m} \times 2.0\text{ m}$, while the thicknesses are 0.9 m and 1.5 m , respectively. Each target is wrapped by 1 cm thick steel plates except for the front and back to weaken the lateral boundary effects. Straight copper-coated steel fibers with 0.2 mm diameter and 13 mm long, as shown in Figure 2, are incorporated into the matrix at a volume content (V_f) of 2% . The yield strength of the steel fiber is 2100 MPa , and the density is 7830 kg/m^3 .



Figure 1. Cuboid UHPC target.



Figure 2. Straight copper-coated steel fibers.

Quasi-static uniaxial compression tests are carried out on prismatic specimens with dimensions of $100\text{ mm} \times 100\text{ mm} \times 300\text{ mm}$ using an electro-hydraulic servo testing

machine with a capacity of 300 tons to determine the uniaxial compressive strength of UHPC used for targets and corresponding matrixes without steel fibers, as shown in Figure 3. A total of six specimens are tested, three of which are made of UHPC with steel fibers, and the other three are made of UHPC without steel fibers (which can be named as matrix), complying with the test procedure in Chinese Standard GB/T 50081-2019 [19]. Before the compression tests, all prismatic specimens are steam cured using an automatic control system [20]. The experimental uniaxial compressive strengths of UHPC specimens with steel fibers are 144.1 MPa, 152.3 MPa, and 148.6 MPa, respectively, and the results of matrix specimens are 139.8 MPa, 134.5 MPa, and 145.3 MPa, respectively. In the subsequent discussion, the uniaxial compressive strength of UHPC and matrix are taken as the average values of the corresponding three specimens, i.e., 148.3 MPa and 139.9 MPa, respectively. At the same time, the density of UHPC specimens with steel fibers and matrix specimens is also measured, and the average values are 2430 kg/m³ and 2237 kg/m³, respectively.



Figure 3. Quasi-static uniaxial compression test on the prismatic specimen.

2.2. Projectiles

The projectile is a type of tangent ogive-nose projectile with a caliber of 8 cm ($d = 8$ cm), and detailed dimensions are shown in Figure 4. The projectiles are composed of the outer shell and inner filling, and each weighs 11.9 kg. The outer shell is made of high-strength steel; the yield strength is 1350 MPa. In order to facilitate the launch of the projectiles, the centering ring and sabot are attached to each projectile.

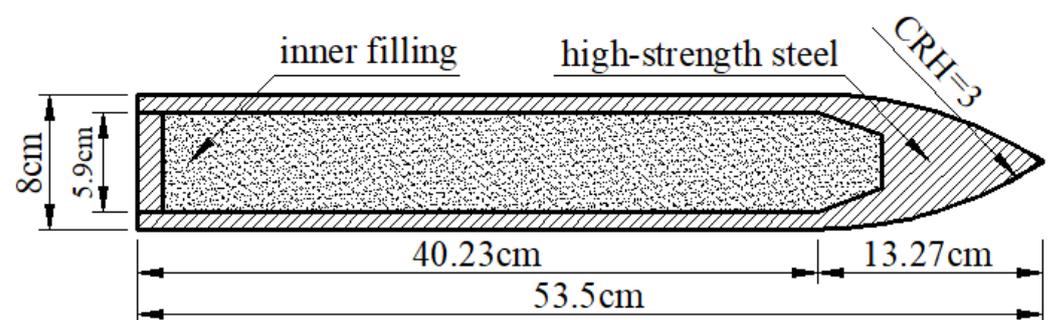


Figure 4. Ogive-nose projectile.

2.3. Experimental Setup

Figure 5 shows the test setup of the penetration tests. A 100 mm smoothbore gun is used to launch the projectiles. The targets are placed vertically on the testing bed. Some thick normal concrete blocks are placed tightly on the back of the targets to restrain the rigid body displacement in the thickness direction of the targets. The smoothbore gun barrel is adjusted to be perpendicular to the front of the targets to ensure that the projectiles can strike in the center of targets vertically. During the flight of projectiles, the centering rings and sabots would automatically separate from the projectiles, so the impact mass (m)

is 11.9 kg. Two sets of high-speed photography equipment are used to measure the actual striking velocity of the projectile (V_0) and to capture the penetration process.

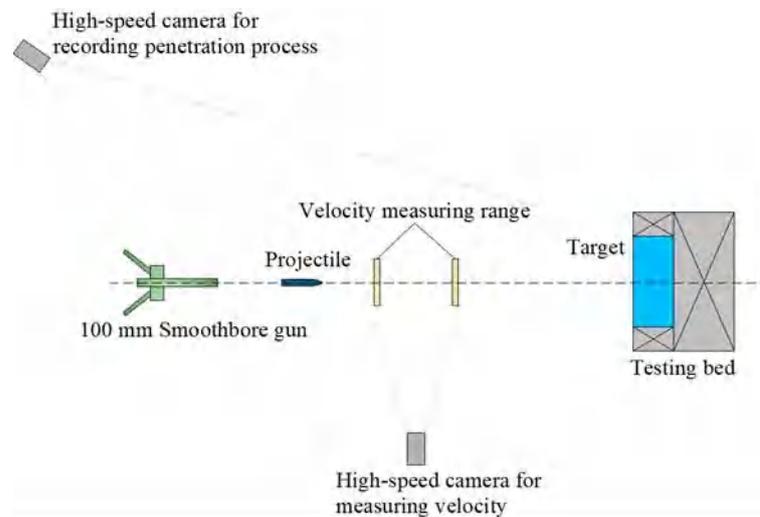


Figure 5. Experimental setup for penetration tests.

2.4. Experimental Results

Figure 6 shows a typical moment that the projectile contacts the target in the penetration process. Detailed experimental data, including the DOP and crater diameter (d_c), are listed in Table 1.



Figure 6. A certain moment in the penetration process.

Table 1. Penetration tests data.

Target	Plain Size (m)	Thickness (m)	f_c (MPa)	V_f	m (kg)	V_0 (m/s)	DOP (cm)	d_c (cm)
U-1	2.0 × 2.0	0.9	148.3	2%	11.9	410	42.1	51.3
U-2	2.0 × 2.0	1.5				664	78.6	78.2

The frontal damages of targets are shown in Figure 7. When the projectile hits the target, concrete around the contact location is ejected out and forms a circular fragment cloud. The projectile with a speed of 410 m/s bounced off the target, while the one at a striking velocity of 664 m/s was stuck in the target. The projectile impact formed an obvious funnel-shaped crater on the impact surface first and then a tunnel deep into the targets. An evident spalling phenomenon could be observed at the edge of the crater. Moreover, the concrete around the tunnel is crushed into powders by ultra-high compressive stress.

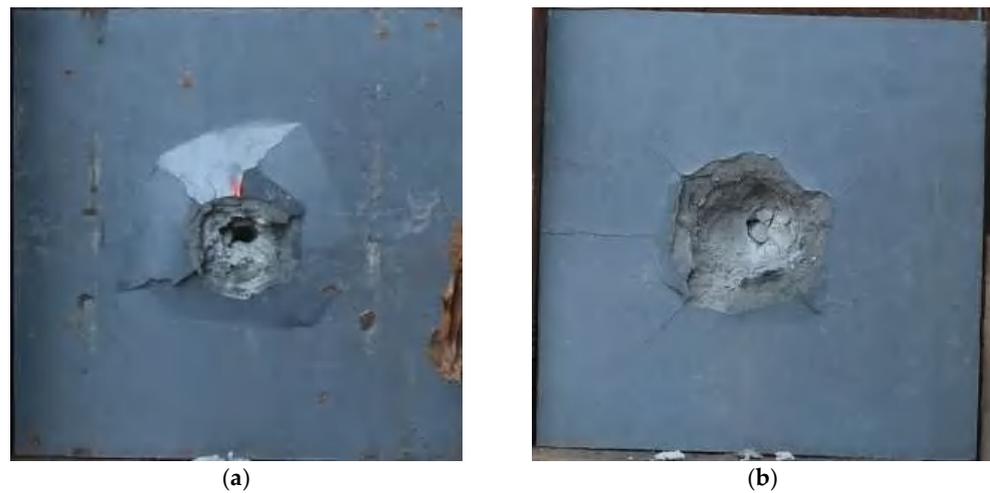


Figure 7. Frontal damages of UHPC targets: (a) U-1 target; (b) U-2 target.

As listed in Table 1, the diameter of the crater increases with the growth of the striking velocity of the projectile. In the present penetration tests, The DOP included both the depth of the crater and the depth of the tunnel. The DOP of the projectiles at striking velocities of 410 m/s and 664 m/s is 42.1 cm and 78.6 cm, respectively, which also shows an increasing tendency.

3. Mesoscale Equivalent Model for UHPC

The mesoscale model with explicit modeling of fibers is better at simulating and explaining the response of UHPC under all kinds of loadings. However, building fibers according to actual size will lead to excessive fibers for typical FE models. For example, for the 2.0 m × 2.0 m × 1.5 m cuboid target U-2 used in the above penetration tests, the number of steel fibers would be more than 70 million in a quarter model. Therefore, to reduce the computational cost of simulation, a mesoscale equivalent model is first developed in this section for UHPC with steel fibers.

3.1. Generation of Fibers

In a UHPC specimen, fibers are uniformly and randomly distributed. The generation of random straight round fibers inside a certain specimen volume V can be as follows. First, the number of fibers is calculated by $N = 4V_fV/(\pi d_f^2 L_f)$ according to the fiber diameter d_f , length L_f and volume content V_f . Random points with the number of N are uniformly generated in the specimen space and assigned to be the initial point of each fiber, labeled as (x_{1i}, y_{1i}, z_{1i}) . Then, for each fiber, the initial direction is defined by two random spatial angles φ_i and θ_i in the spherical coordinate system with the initial point as the origin. The ending point of a fiber, labeled as (x_{2i}, y_{2i}, z_{2i}) , is determined according to the initial point, random angles, and L_f , as shown in Equation (1).

$$\begin{cases} x_{2i} = x_{1i} + L_f \sin \theta_i \cos \varphi_i \\ y_{2i} = y_{1i} + L_f \sin \theta_i \sin \varphi_i \\ z_{2i} = z_{1i} + L_f \cos \theta_i \end{cases} \quad (1)$$

With the initial point and ending point determined, any fiber in the given specimen space can be identified. Equation (1) is cycled N times until all fibers are generated. All generated points and fibers will be written into node keyword file and element keyword file, respectively, which will facilitate subsequent model establishment.

3.2. Bond-Slip Constitutive Model

For building the mesoscale equivalent model for UHPC with steel fibers, the failure mode and interfacial behavior between steel fibers and matrixes must be determined. Deng

et al. [21] pointed out that the fiber pullout is the most common failure mode when UHPC is subjected to loading due to the weak fiber-matrix interface and high tensile strength of the steel fiber. Hence, the interfacial behavior between steel fibers and matrixes can be regarded as a kind of bond-slip behavior. Su et al. [22] carried out single steel fiber pullout tests and put forward a bond-slip constitutive model for UHPC with steel fibers. In this model, the interfacial shear stress τ is simplified to be constant at a given relative slip S . For different values of the relative slip, the interfacial shear stress is determined by Equation (2), where τ increases linearly with S in the bonding phase and declines exponentially with S in the debonding phase.

$$\tau = \begin{cases} G_s S & S \leq S_{max} \\ G_s S_{max} \times e^{-EXP \times D} & S > S_{max} \end{cases} \quad (2)$$

where G_s is the interfacial shear modulus, EXP is the exponent in the debonding phase, and D is the accumulated plastic displacement in all the integral time steps.

The pivotal constitutive parameters for UHPC (150 MPa) are determined according to the pullout load-slip curve, where G_s is 2393 MPa, S_{max} is 0.00125, and EXP is 0.2. Then, simulations on the static split tension test and SHPB test are performed, where the numerical results agree well with experimental results, and more details can be seen in Reference [22]. This bond-slip constitutive model is proposed, and the determined parameters are validated by static and dynamic tests and are adopted in the present study.

3.3. Equivalent Treatment on Steel Fibers

Since the basic reason for the ultra-high computational cost of the original mesoscale model is the huge number of steel fibers, the key to establishing a mesoscale equivalent model is to reduce the number through the equivalent treatment of steel fibers. Steel fibers play a bridging role in UHPC, which limits the deformation and crack development in the matrix. This kind of bridging effect is reflected by the interaction between steel fibers and matrixes; the essence is the transmission of the interfacial shear force. Therefore, the bridging effect can be equivalent as long as the interfacial shear force can be guaranteed to be unchanged when the equivalent treatment on steel fibers is performed [23,24]. The specific implementation procedure of the equivalent treatment on steel fibers is as follows.

The first step is to amplify steel fibers geometrically under the condition that the aspect ratio of fibers remains unchanged and replace the original fibers with amplified fibers.

$$d_{af} = n d_f, \quad L_{af} = n L_f \quad (3)$$

where d_{af} and L_{af} are the diameter and length of the amplified fiber, respectively, and n is the geometric amplification factor. According to the premise that the volume percentage of steel fibers V_f remains unchanged, the number of fibers N_{af} after equivalent treatment is determined as:

$$N_{af} = \frac{V_f V}{\pi (d_{af})^2 L_{af}} = \frac{V_f V}{\pi (n d_f)^2 (n L_f)} = \frac{1}{n^3} N \quad (4)$$

At the same time, due to the fibers being uniformly and randomly distributed, the number of fibers in any direction is the same, where the number is labeled as N^κ for original fibers and N_{af}^κ for amplified fibers $N_{af}^\kappa = N^\kappa / n^3$.

For a single original steel fiber embedded in the matrix, as shown in Figure 8, an increment of the axial force in an infinitesimal segment of fiber is:

$$dP = \tau_0 \cdot \pi d_f \cdot dl \quad (5)$$

where τ_0 is the actual interfacial shear stress, the interfacial shear force, i.e., the axial force of the fiber, can be computed as:

$$F = \int_0^{L_e} dP = \pi d_f \int_0^{L_e} \tau_0 dl \tag{6}$$

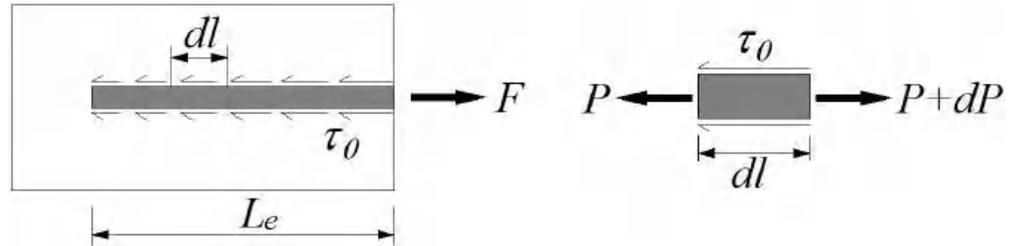


Figure 8. Free-body diagram of the infinitesimal segment of a fiber.

Based on the equivalence of interfacial shear stress distribution [22], Equation (6) is simplified to:

$$F = \tau \cdot \pi d_f \cdot L_e \tag{7}$$

Similarly, the interfacial shearing force of a single amplified fiber is:

$$F_{af} = \tau \cdot \pi (n d_f) \cdot n L_e = n^2 F \tag{8}$$

Then the resultant interfacial shearing force of the original fibers in a direction can be computed as:

$$F^k = N^k F = N^k \tau \cdot \pi d_f \cdot L_e \tag{9}$$

and the resultant force of amplified fibers in the same direction is:

$$F_{af}^k = N_{af}^k \tau \cdot \pi (n d_f) \cdot n L_e = \frac{1}{n} F^k \tag{10}$$

However, as Equation (10) shows, F_{af}^k is not equal F^k . Therefore, the second step is to modify the interfacial shear modulus G_s to be $G_{saf} = n G_s$ and then the final resultant interfacial shearing force of amplified fibers is:

$$F_{af}^k = N_{af}^k (n \tau) \cdot \pi (n d_f) \cdot n L_e = F^k \tag{11}$$

Equation (11) indicates that the interfacial shearing force of equivalent steel fibers and original steel fibers are ensured to be equal, which means the bridging effect of steel fibers is unchanged and then demonstrates that the equivalent treatment on steel fibers is theoretically reasonable. Ultimately, the mesoscale equivalent model for UHPC with steel fibers is developed based on the equivalent treatment of fibers.

4. Verification on the Mesoscale Equivalent Model

To further verify the developed model, numerical simulations on both uniaxial compression tests and penetration tests presented in Section 1 are conducted in LS-DYNA employing the mesoscale equivalent model. The numerical results are compared with corresponding experimental results, which show that the developed mesoscale equivalent model could well reproduce the behavior of UHPC under static and dynamic loadings.

4.1. Model Verification with Uniaxial Compression Tests

4.1.1. Numerical Model

The typical mesoscale finite element model for prismatic specimens under uniaxial compression is shown in Figure 9. The specimen is placed on a fixed loading plate, and the load is applied to the specimen by imposing displacement on the moving loading plate. An automatic surface-to-surface contact algorithm considering the friction effect

is adopted to simulate the contact behavior between the loading plate and the specimen. The solid element SOLID164 is used for modeling the matrix of the specimen and loading plates. The steel fibers are modeled by beam element BEAM161. The keyword *CONSTRAINED_BEAM_IN_SOLID (CBIS) [25] in LS-DYNA is adopted to simulate the bond-slip behavior between steel fibers and matrices.

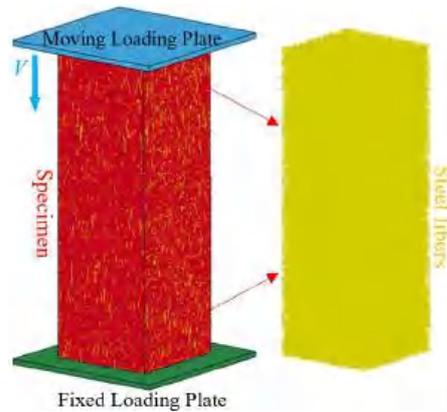


Figure 9. Typical mesoscale FE model for uniaxial compression.

For different amplification factors n , the dimensions and number of steel fibers in the mesoscale equivalent model are different, and the effect of the amplification factor on the numerical needs to be investigated. In this study, numerical simulations on uniaxial compression with different amplification factors n ($n = 1, 3, 4, 5, 6, 7$, $n = 1$ means no equivalence treatment on steel fibers) are performed. Figure 10 shows the distribution of steel fibers in a specimen with different n , where the number of steel fiber elements decreases fast with the increasing n . The critical parameters of the CBIS algorithm for the mesoscale equivalent model with different n are listed in Table 2.

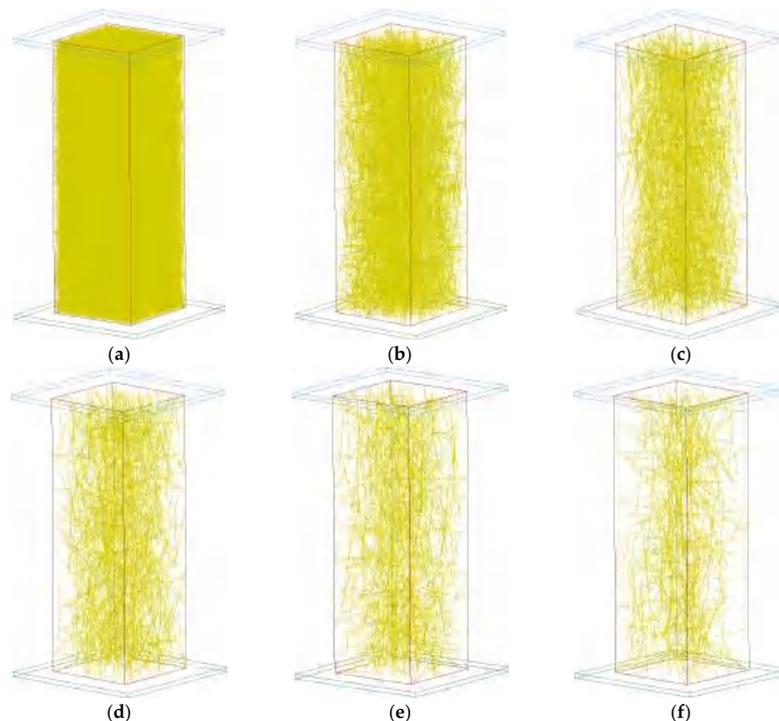


Figure 10. Distribution of steel fibers in matrix with different n : (a) $n = 1$, 146,913 beam elements; (b) $n = 3$, 5442 beam elements; (c) $n = 4$, 2296 beam elements; (d) $n = 5$, 1176 beam elements; (e) $n = 6$, 681 beam elements; (f) $n = 7$, 429 beam elements.

Table 2. Critical CBIS algorithm parameters under different n .

d	l	G_S	S_{max}	EXP
0.2·n mm	13·n mm	2393·n MPa	1.25×10^{-3}	0.2

Without adding steel fibers, the UHPC matrix could be regarded as high-strength concrete, modeled by the RHT model [26] in the present study. The critical RHT failure surface parameters A_{fail} and N_{fail} are determined by fitting the triaxial compression test data of 140 MPa HSC in Reference [27], as shown in Figure 11, where A_{fail} and N_{fail} are 1.78 and 0.35, respectively. The other critical parameters are self-defined according to Refs. [10,28,29]. Due to the stiffness of loading plates in practice being much larger than that of the specimen, the loading plates are modeled as a rigid body with the material model *MAT_RIGID. The isotropic and kinematic hardening material model *MAT_PLASTIC_KINEMATIC (*MAT_003) is chosen to build the steel fibers. Parameters of material models used in the uniaxial compression simulations are listed in Table 3.

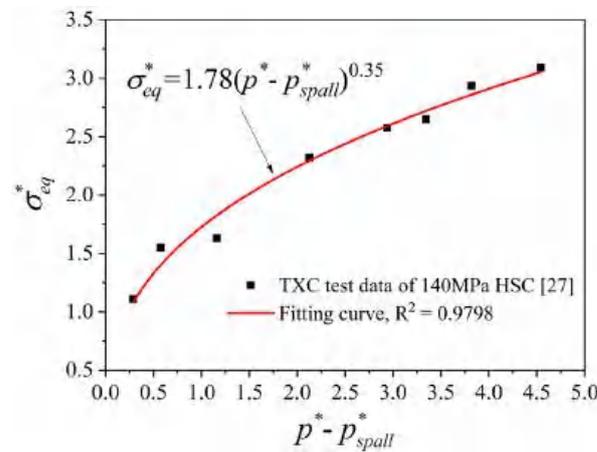


Figure 11. Determination of the parameters A_{fail} and N_{fail} .

Table 3. Parameters of material models for the UHPC matrix, loading plate, and steel fibers (units: cm-g- μ s).

Material	Material Model	Input Parameter	Value
UHPC matrix	*MAT_RHT	Shear modulus G	0.19
		Mass density ρ_0	2.273
		Compressive strength f_c	0.0014
		Failure surface constant A_{fail}	1.78
		Failure surface constant N_{fail}	0.35
		Residual Surface constant B	0.8
		Residual Surface constant m	0.3
		Damage constant D_1	0.045
		Damage constant D_2	1.0
		Minimum strain at fracture $EFMIN$	0.011
Loading plate	*MAT_RIGID	Mass density RO	7.83
		Young's modulus E	2.0
		Poisson's ratio PR	0.25
Steel fibers	*MAT_003	Mass density RO	7.83
		Young's modulus E	2.0
		Poisson's ratio PR	0.30
		Yield strength $SIGY$	0.021
		Tangent modulus $ETAN$	0.0021
		Failure strain FS	0.2

4.1.2. Results and Discussion

The numerical failure modes of specimens under different amplification factors n are shown in Figure 12. For the cases of $n = 1, 3, 4, 5$, the specimens exhibit ductile shearing failure modes, where a main oblique crack develops from the end of the specimen to the middle in the direction of about 45 degrees, which is the same as the typical experimental failure mode shown in Figure 12g. However, for the case of $n = 6$ and $n = 7$, the failure mode is quite different from the other cases and the experimental result, where the damage occurs firstly in the middle of the specimen and is approximately concentrated in the middle, which displays as brittle splitting failure mode to a certain content. Shearing failure is a type of ductile failure, while splitting failure belongs to brittle failure. Although the volume percentage of steel fibers is the same, the failure mode tends to change from ductile failure to brittle failure with the increase of the amplification factor. The difference between failure modes under different amplification factors indicates that steel fibers greatly influence the ductility of UHPC.

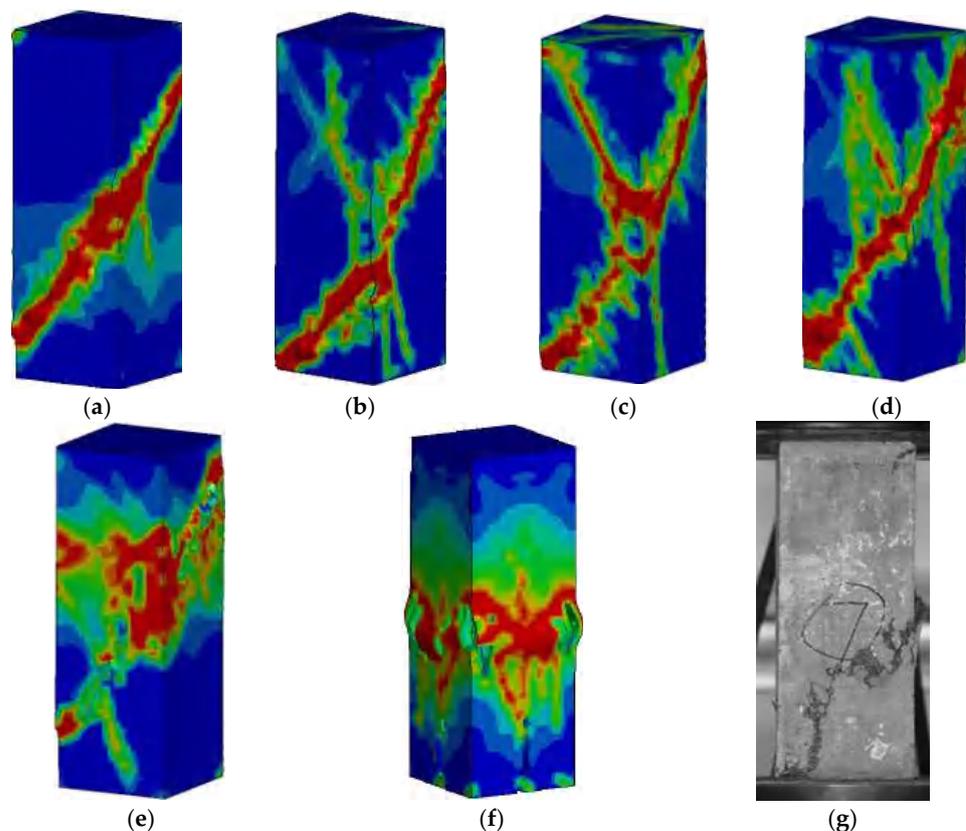


Figure 12. Failure mode of specimens: (a) $n = 1$; (b) $n = 3$; (c) $n = 4$; (d) $n = 5$; (e) $n = 6$; (f) $n = 7$; (g) $n = 8$.

Figure 13 shows the uniaxial compressive curves with different amplification factors n and the average full stress-strain curve obtained from uniaxial compression tests. There is little difference in the ascending branches of stress-strain curves obtained from the numerical simulations under different amplification factors, and all the ascending branches are in good agreement with the test data. As for the descending branches of the stress-strain curves, the numerical results of $n = 1, 3, 4$, and 5 are in good accord with the test data. Nevertheless, for the cases of $n = 6$ and $n = 7$, the numerical descending branches drop more sharply than the other cases and have much lower residual strength, which presents a great difference from the experimental result. If taking the area under the full stress-strain curve of uniaxial compression as an index to characterize the toughness of specimens, Figure 13 shows that the toughness decreases with the increase of the amplification factor.

The variation of the toughness index caused by the change in amplification factor shows that steel fiber influences the toughness of UHPC significantly.

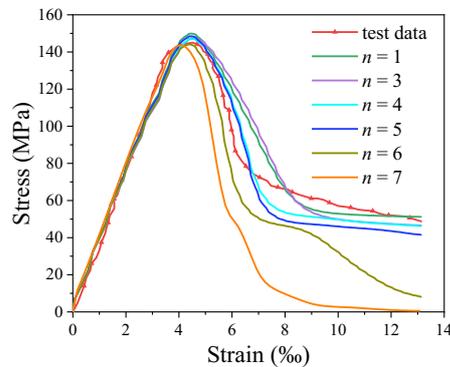


Figure 13. Uniaxial compressive curves of specimens.

Steel fibers improve the ductility and toughness of UHPC by the bridging effect in the matrix. According to the above comparison and discussion, the bridging effect gradually weakens with the increase in the amplification factor. In the present study, the mesoscale equivalent model can accurately simulate the uniaxial compression behavior of UHPC when the amplification factor is lower than 5. However, with further increases in the amplification factor, the mesoscale equivalent model loses its validity. This phenomenon may be because the equivalence of the resultant interfacial shearing force of steel fibers, as shown in Equation (11), is destroyed due to the excessive amplification in the fiber dimension. The equivalent treatment on fibers is based on the assumption that there are enough original fibers in any direction to synthesize a certain number of amplified fibers. When the amplification factor is too large, such as $n = 6$ or $n = 7$, the number of original fibers in some directions may not be enough to synthesize sufficient amplified fibers, leading to the uneven distribution of amplified fibers and causing Equation (11) to fail. Therefore, to ensure the accuracy of the simulation, it is recommended that the amplification factor should not be greater than 5.

4.2. Model Verification with Penetration Tests

Taking the target U-2 as an example, the estimated computation time of penetration simulations under $n = 1-5$ is listed in Table 4. The computation time falls quickly as the amplification factor increases. Consequently, the amplification factor adopted for penetration simulations is determined to be 5 for balancing the accuracy and computation time of numerical calculation.

Table 4. The estimated computation time of penetration simulations on target U-2 under different amplification factors.

n	1	3	4	5
computation time	1400 h 16 min	51 h 54 min	23 h 18 min	11 h 35 min

4.2.1. Numerical Model

Since no obvious yaw phenomenon is observed in the penetration process, quarter finite element models based on the equivalence method are established, as shown in Figure 14. The keyword *CONSTRAINED_GLOBAL is used to simulate the symmetric boundary condition by defining two global boundary constraint planes. As for the outer surfaces of targets, the fixed boundary condition is imposed according to the experimental setup. Matrix elements (SOLID164) within five times the projectile diameter are refined, and the minimum matrix element size is 0.8 cm. There are 352,512 steel fiber elements (BEAM161) uniformly distributed in the target with 90 cm thickness and 587,571 steel

fiber elements in the target with 150 cm thickness. The outer shell and inner filling of the projectile are modeled by the solid element SOLID164 and are connected by joint nodes. The eroding surface-to-surface contact algorithm is used to simulate the contact behavior between the projectile and matrix.

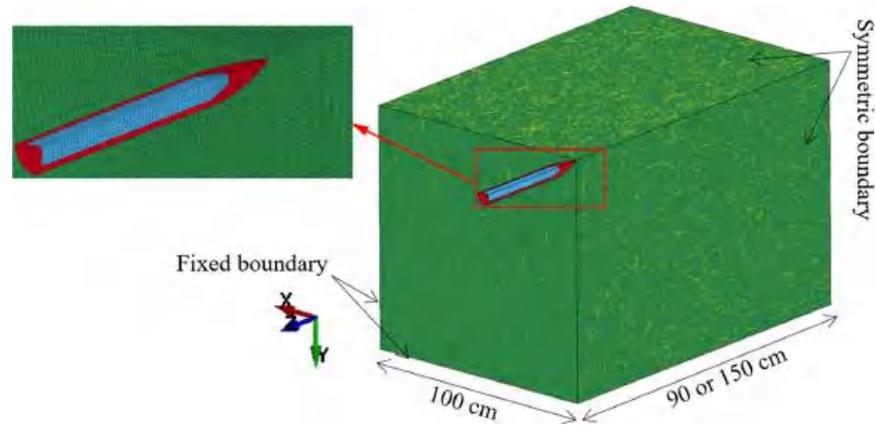


Figure 14. FE model for projectile penetration.

In the process of penetrating, the outer shell of the projectile withstands high temperatures, which will soften the material. Thus, the Johnson–Cook model [30,31] (*MAT_015) applied for metals subjected to large strains, high strain rates, and high temperatures, coupled with *EOS_GRUNEISEN, is adopted to build the outer shell. Material model *MAT_003 is used to build the inner filling of the projectile. The material models and parameters for the projectile are listed in Table 5.

Table 5. Parameters of material models for projectile [3] (units: cm-g- μ s).

Material	Material Model	Input Parameter	Value
Outer shell of projectile	*MAT_015	Shear modulus	0.84
		Mass density	7.83
		Poisson’s ratio	0.25
		$a/b/n/c/m$	0.01350/0.00477/0.18/0.012/1.0
		Failure stress	−2
		D_1	0.15
		D_2	0.72
		D_3	1.66
		$C/S_1/\gamma/A$	0.4596/1.357/1.71/0.43
Inner filling of projectile	*MAT_003	Mass density RO	1.63
		Young’s modulus E	0.1
		Poisson’s ratio PR	0.45
		Yield strength $SIGY$	0.0006
		Tangent modulus $ETAN$	0.001
		Failure strain FS	3.0

4.2.2. Results and Discussion

Figure 15 shows the numerical results of the penetration process for U-1, where the striking velocity is 410 m/s and the target thickness is 90 cm. The “History Variable #4” refers to the damage parameter D in the RHT model, where $D = 0$ means no damage and $D = 1$ means fracture of the material. It can be seen that severe damage mainly occurs around the ballistic trajectory, and the region of damage distribution expands as penetration depth increases.

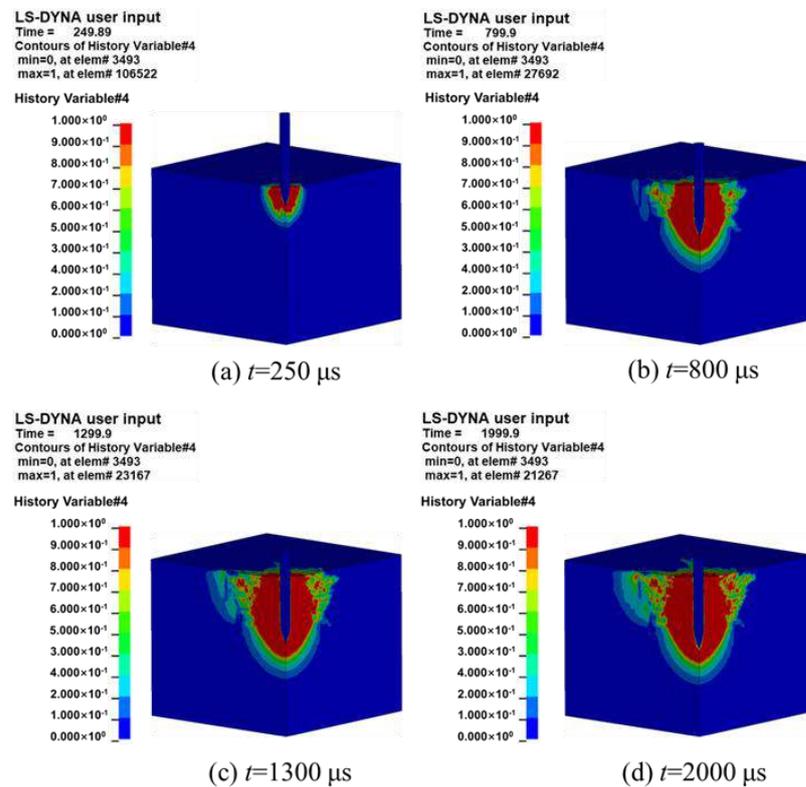


Figure 15. Numerical results of the penetration process for U-1, where the striking velocity is 410 m/s, and the target thickness is 90 cm.

Frontal damage contours of targets U-1 and U-2 are shown in Figure 16. As the striking velocity of the projectile increases, the impact surface of the target tends to be subjected to more severe local damage. The numerical results of the crater diameter on targets U-1 and U-2 are 53.8 cm and 77.3 cm, respectively. Comparing the numerical and experimental results, the maximum relative error in the crater diameter is just 5.9%, showing perfect consistency. In addition to the crater, the projectile impact also forms cracks on the impact surface. The numerical results of cracks show that there are four main cracks on the impact face of target U-1 and eight main cracks on the impact face of target U-2, and these cracks develop radially and orthogonally. The width, number, and distribution range of cracks expand with increased striking velocity. For target U-1, the number and distribution of the simulated cracks are in good agreement with the experimental results. While for target U-2, the difference between the numerical results of the cracks and the experimental results is a little large, which may be caused by the slight yaw of the projectile penetrating into U-2 in the penetration tests.

The time-displacement curves and time-velocity curves of the projectiles are shown in Figure 17. The numerical values of DOPs at striking velocities of 410 m/s and 664 m/s are 40.4 cm and 87.1 cm, respectively. The numerical results and experimental results of DOPs are compared in Figure 18, where the relative errors are -3.8% and 11.7% , respectively. The reason for the slightly larger error in the case of 664 m/s may be that the projectile has a slight yaw, as shown in Figure 7b. Although the yaw is not obvious, it could increase the resistance of the projectile penetration into the target and then reduce the DOP. The maximum relative error of the DOP is lower than 15%, which indicates that the numerical method could reasonably predict the DOP when projectiles penetrate UHPC targets. Figure 17 shows that the velocities of the three projectiles gradually decrease from positive values to negative values with the increase of the DOP. The negative values of the velocities mean that the projectiles bounce in the opposite direction, which is consistent with the experimental phenomenon. The rebound velocities of projectiles at striking velocities of 410 m/s and 664 m/s are 16.7 m/s and 22.6 m/s, respectively.

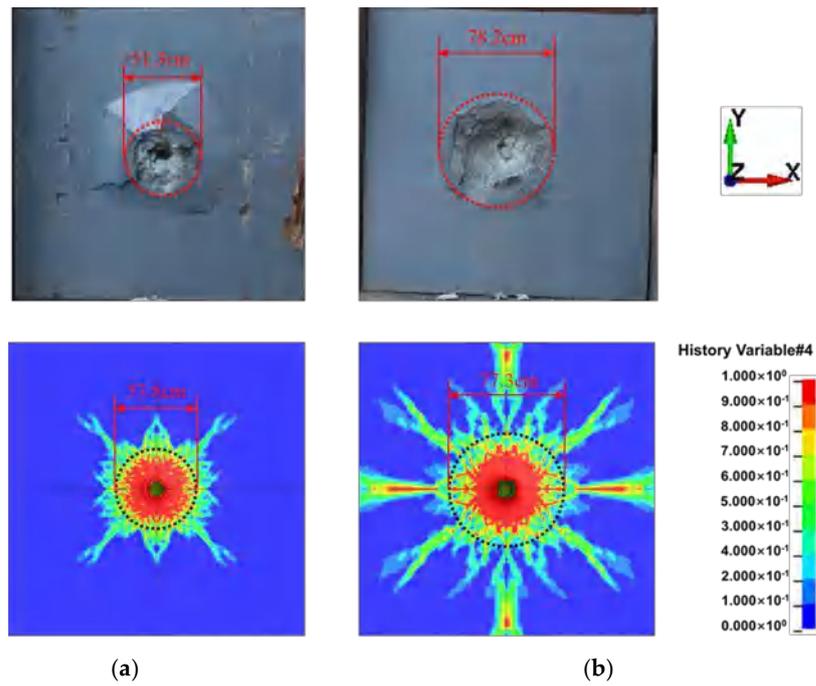


Figure 16. Comparison of frontal damage between the experimental and numerical results; (a) U-1, 410 m/s; (b) U-2, 664 m/s.

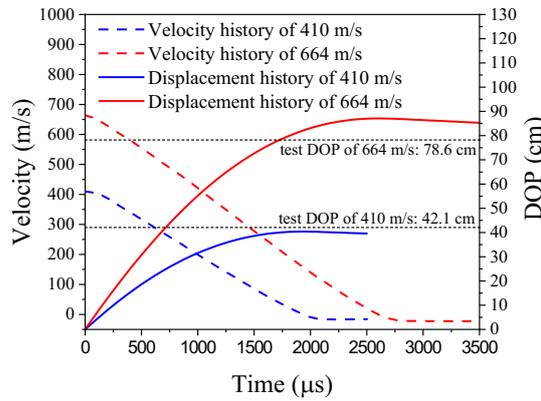


Figure 17. Velocity and displacement histories of projectiles at striking velocities of 410 m/s and 664 m/s.

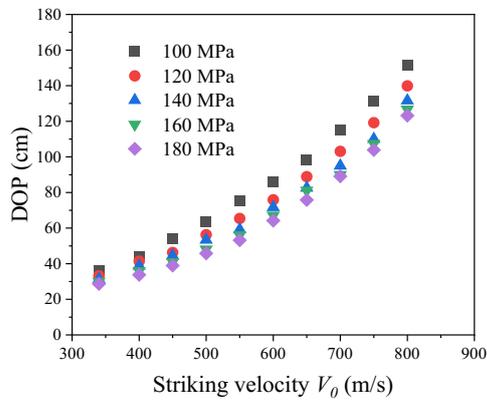


Figure 18. DOP versus striking velocity V_0 for 100 MPa, 120 MPa, 140 MPa, and 180 MPa UHPC targets, where the volume content of the steel fiber is 2%.

Comparing the frontal damage and the DOP between experiments and numerical simulations indicates that the mesoscale equivalent model can reasonably simulate the dynamic performance of UHPC under projectile penetration when the amplification factor is 5.

5. Numerical Investigation on DOP

The DOP is the most important and noteworthy index, which reflects the resistance of concrete defensive structure under the projectile impact, and it is greatly influenced by the projectile's striking velocity and the concrete's compression strength [5]. As for UHPC, steel fibers play an important role in improving tensile strength and toughness, which makes it attractive to study the influence of steel fibers on the DOP. Therefore, based on the validated mesoscale equivalent model, 62 penetration scenarios considering different striking velocities, compression strength, and volume percentage of steel fibers are simulated to investigate the specific effect of these parameters on the DOP. All the simulated results of the DOP are listed in Table 6.

Table 6. Numerical results of the DOP.

Simulation No.	V_0 (m/s)	f_c (MPa)	V_f	Simulated DOP (cm)
1	340	100	2%	36.4
2	400			43.8
3	450			54.0
4	500			63.8
5	550			75.2
6	600			85.7
7	650			98.0
8	700			115.2
9	750			131.3
10	800			151.6
11	340	120	2%	33.2
12	400			41.4
13	450			46.3
14	500			56.3
15	550			65.4
16	600			75.9
17	650			88.9
18	700			103.1
19	750			119.2
20	800			139.9
21	340	140	2%	31.3
22	450			44.5
23	500			53.4
24	550			59.2
25	600			71.7
26	650			82.6
27	700			95.0
28	750			110.0
29	800		131.6	
30	400		1%	44.3
31			1.5%	40.9
32			2%	38.8
33			2.5%	37.6
34			3%	36.8
35			3.5%	36.4
36			4%	36.2

Table 6. Cont.

Simulation No.	V_0 (m/s)	f_c (MPa)	V_f	Simulated DOP (cm)
37	340	160	2%	30.0
38	400			35.9
39	450			41.0
40	500			48.1
41	550			56.1
42	600			67.1
43	650			81.1
44	700			89.9
45	750			107.4
46	800			126.7
47	340	180	2%	28.7
48	450			39.0
49	500			45.8
50	550			53.2
51	600			64.2
52	650			75.8
53	700			89.1
54	750			103.9
55	800			123.2
56				1%
57		1.5%	36.8	
58		2%	35.8	
59	400	2.5%	34.5	
60		3%	34.3	
61		3.5%	34.0	
62		4%	33.8	

5.1. Effect of Striking Velocity

Figure 18 shows the DOPs of UHPC targets with 2% steel fiber volume content at striking velocities of 340 m/s, 400m/s, 450 m/s, 500m/s, 550 m/s, 600 m/s, 650m/s, 700 m/s, 750 m/s, and 800 m/s. The numerical results manifest that for the UHPC targets of 100 MPa, 120 MPa, 140 MPa, 160 MPa, and 180 MPa, the DOP increases with the increasing striking velocity. Taking the scenario of the 140 MPa UHPC target as an example, the DOPs at striking velocities of 400 m/s, 500 m/s, 600 m/s, 700 m/s, and 800 m/s are 38.8 cm, 53.4 cm, 71.7 cm, 95.0 cm, and 131.6 cm, respectively. The increments between adjacent striking velocities are 14.6 cm, 18.3 cm, 24.3 cm, and 39.0 cm, respectively, and present an exponentially growing trend. For UHPC targets with other compression strengths, the growing trend is similar. The above findings show an exponential relationship between the DOP and striking velocity for UHPC.

5.2. Effect of Compression Strength

As shown in Figure 19, the DOP decreases with increasing compression strengths of UHPC targets when the striking velocity is fixed. Comparing the numerical results of UHPC targets under projectile penetration with the striking velocity at 550 m/s, where DOPs of 100 MPa, 120 MPa, 140 MPa, 160 MPa, and 180 MPa targets are 75.2 cm, 65.4 cm, 59.2 cm, 56.1 cm, and 53.2 cm, respectively, it can be observed that the increments between the adjacent compression strength are 9.8 cm, 6.2 cm, 3.1 cm, and 2.9 cm, respectively, showing a decreasing trend. Hence, the enhancement of compression strength contributes to improving the resistance of the UHPC target against projectile penetration. However, it should also be noted that the improvement effect diminishes with increasing compression strengths.

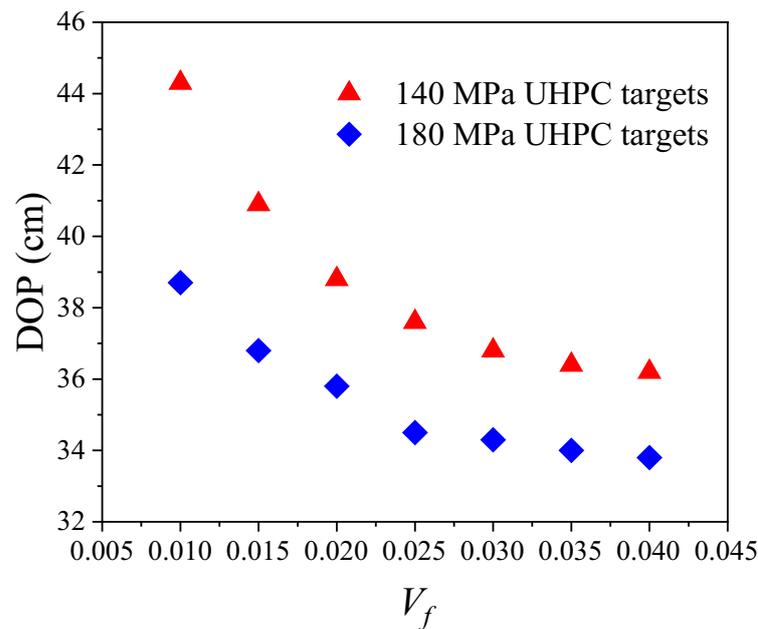


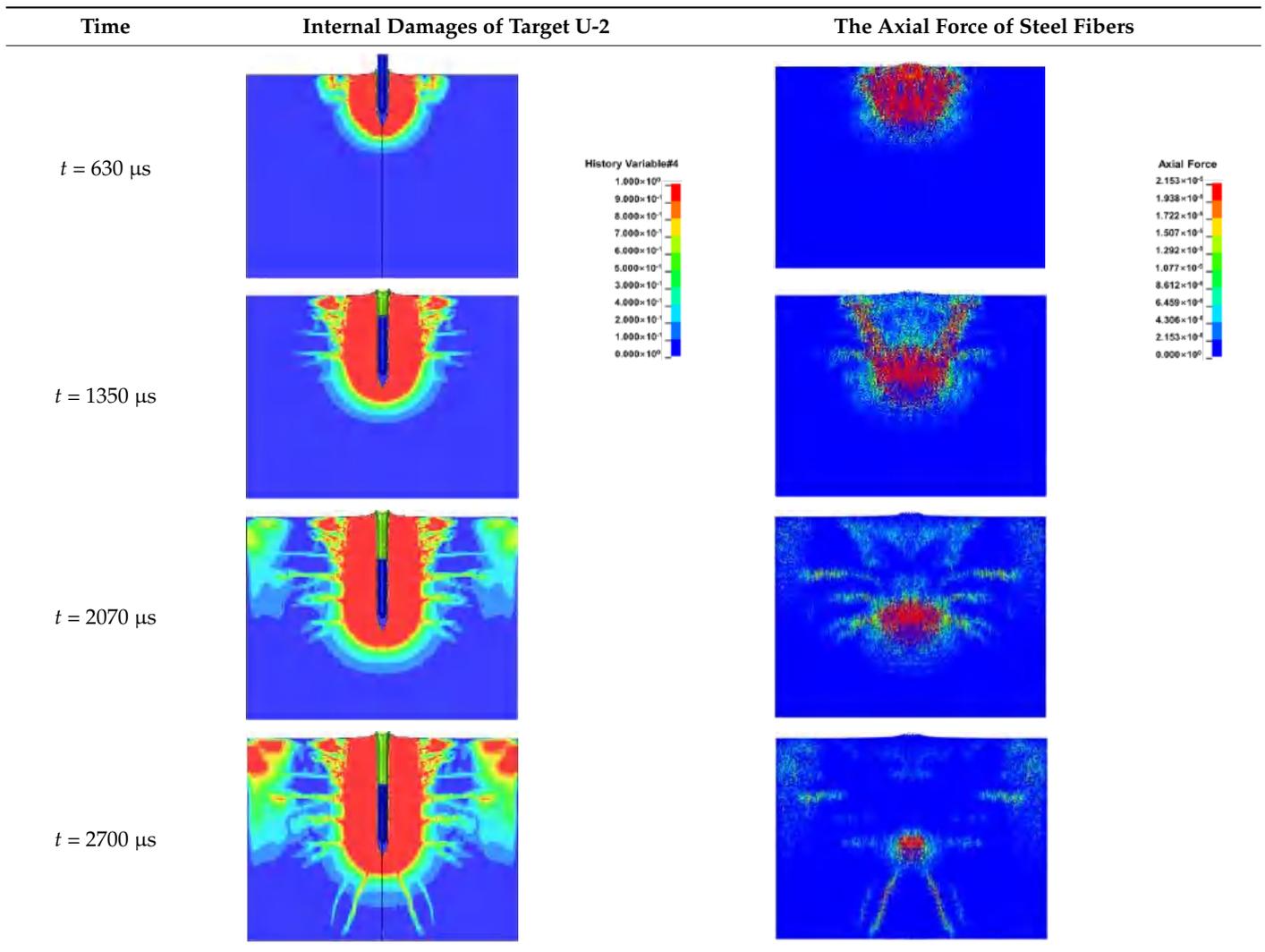
Figure 19. DOP versus volume content of the steel fiber V_f for 140 MPa and 180 MPa UHPC targets, where the striking velocity is 400 m/s.

5.3. Effect of Steel Fiber

Axial force contours of steel fibers for target U-2 in four different moments, along with internal damages and cracks in the matrix, are listed in Table 7. From $t = 630 \mu\text{s}$ to $t = 2700 \mu\text{s}$, the axial force in steel fibers develops synchronously with the expansion of internal damages and cracks. Since damage is accumulated by the plastic strain, the damage parameter can reflect the deformation of matrix elements. This deformation will cause a relative slip between the matrix and steel fibers embedded in the matrix, resulting in the generation of axial force in the steel fibers. The axial force will increase until the maximum relative slip S_{max} is reached and then decline until total debonding happens. Therefore, taking the regions around the ballistic trajectory at $t = 630 \mu\text{s}$ and regions where the cracks develop at $t = 2070 \mu\text{s}$ as examples, the axial force of steel fibers in the regions with damage is larger than that in the undamaged regions. The above findings show that the steel fibers can exert a bridging effect by limiting the deformation of matrix elements and the development of cracks, improving the resistance of UHPC targets under projectile penetration.

Figure 19 shows the DOPs of 140 MPa and 180 MPa UHPC targets with volume percentages of steel fibers of 1.0%, 1.5%, 2.0%, 2.5%, 3.0%, 3.5%, and 4.0%, where the striking velocity is 400 m/s. No matter the UHPC target of 140 MPa or 180 MPa, the DOP decreases with the increasing volume percentage of steel fibers. DOPs into 140 MPa UHPC targets with 1.0% and 4.0% V_f is 44.3 cm and 36.2 cm, respectively, where the 3% increase in the volume percentage of steel fibers results in an 18.3% decrease in the DOP. For the 180 MPa UHPC targets, the corresponding DOPs are 38.7 cm and 33.8 cm, respectively, and the decrease in the DOP is 12.7%. It can be seen from Figure 19 that when V_f increases from 3% to 4%, the reduction in the DOP is not obvious, which demonstrates that there is a limit to reducing the DOP by adding more steel fibers.

Table 7. Development of internal damages and steel fibers axial force in target U-2.



5.4. Modification of Young Equations for DOP Prediction

Based on an extensive experimental database of projectile impact, Sandia National Laboratories proposed the empirical Young penetration equations to predict the DOP into natural earth materials and concrete. With the expansion of the experimental database, Young equations have been constantly updated, and the latest version for concrete is as follows (in SI units):

$$DOP = 0.00000153 N K_e K_h (t_c T_c)^{-0.06} (11 - P)(35/f_c)^{0.3} (m/A)^{0.7} (V_0 - 30.5), \quad V_0 \geq 61 \text{ m/s} \quad (12)$$

$$N = 0.18(CRH - 0.25)^{0.5} + 0.56, \text{ for tangent ogive nose shapes} \quad (13)$$

$$K_e = (F/W_1)^{0.3} \quad (14)$$

$$K_h = 0.46 (m)^{0.15}, \text{ when } m < 182 \text{ kg}; \text{ else, } K_h = 1.0 \quad (15)$$

where N is the nose performance coefficient of the projectile, K_e is the correction coefficient for edge effects in concrete targets, and K_h is the correction coefficient for the lightweight projectile. More details about the three coefficients are presented in Reference [32]. t_c is the cure time of concrete, T_c is the thickness of the target, P is the volumetric percentage of rebars in concrete targets, f_c is the compression strength of concrete, m is the mass of the

projectile, A is the cross-sectional area of the projectile, V_0 is the striking velocity of the projectile.

The Young equations comprehensively consider the parameters that affect the DOP, which makes them applicable in many penetration scenarios. For normal concrete and high-strength concrete with compression strength under 100 MPa, the accuracy of the Young equations is well validated. However, according to the above parametric analysis, the functional relationship between the DOP and the striking velocity of the projectile for UHPC is different from the linear relationship described in the original Young equations. At the same time, the original Young equations do not consider the influence of steel fibers on the penetration depth. Consequently, the original Young equations are not applicable to predicting the DOP into UHPC with ultra-high strength and steel fibers and should be modified. The parametric analysis shows that the functional relationships between the DOP and striking velocity V_0 , compression strength f_c , and volume percentage of steel fibers V_f can be described by an exponential function, power function, and quadratic polynomial, respectively. Hence, the modified Young equation for UHPC without rebar is expressed as:

$$DOP = \alpha \cdot N K_e K_h \cdot (t_c T_c)^{-0.06} (f_c^\beta) (k_0 + k_1 V_f + k_2 V_f^2) (m/A)^{0.7} (e^{\gamma V_0}) \tag{16}$$

where N , K_e , and K_h are the same as the original equations, and α , β , k_0 , k_1 , k_2 , and γ are the undetermined coefficients.

According to the numerical results of the DOP listed in Table 6 and the equation form expressed as Equation (13), multivariate nonlinear fitting is performed in MATLAB to determine the pending parameters. The values of α , β , k_0 , k_1 , k_2 , and γ are determined to be 0.00344, -0.43 , 292.8, -5048 , 77946, and 0.003, respectively. In the end, the modified Young equation is as follows:

$$DOP = 0.00344 N K_e K_h (t_c T_c)^{-0.06} (f_c^{-0.43}) (292.8 - 5048 V_f + 77946 V_f^2) (m/A)^{0.7} (e^{0.003 V_0}) \tag{17}$$

where f_c is in the unit of "MPa", V_f is a unitless percentage, m is in the unit of "kg", A is in the unit of "m²", V_0 is in the unit of "m/s", and eventually the DOP is in the unit of "cm". The correlation index R^2 of the fitted function is 0.9977, indicating that the modified equation for the DOP prediction has high goodness of fit.

The DOPs calculated by the modified Young equation and the original Young equation are compared with the experimental data presented in this study and References [3,5,33], as listed in Table 8. It can be seen that the original Young equation greatly underestimates the penetration resistance of UHPC, and the maximum relative error is 26.9%. However, the relative error between the calculated DOPs by the modified Young equation and experimental data is within 14%, which indicates that the modified Young equation proposed in the present study can accurately predict the DOP of UHPC targets subjected to projectile impact.

Table 8. Comparison of the DOPs between results of equation calculation and experimental results for different UHPC targets.

Specimen	f_c (MPa)	V_f (%)	m (kg)	d (cm)	V_0 (m/s)	DOP in Tests (cm)	DOP of Modified Equation (cm)	Relative Error	DOP of Original Equation (cm)	Relative Error
U-1	148	2	11.9	8	410	42.1	39.2	-6.9%	52.7	25.2%
U-2	148	2	11.9	8	664	78.6	84.0	6.8%	92.7	17.9%
UHPC-SF-1 [3]	140	3	0.329	2.53	553	12.9	12.9	-0.2%	16.4	26.9%
UHPC-SF-2 [3]	140	3	0.329	2.53	683	16.6	18.7	12.7%	20.4	23.1%
UHPC-SF-3 [3]	140	3	0.329	2.53	808	20.8	23.7	13.9%	25.8	24.0%
A-5-1 [5]	114	3	0.341	2.53	510	13.4	12.7	-4.9%	16.9	26.1%
1:3 [33]	153	1.5	6.3	7.5	622	51.0	52.2	2.4%	59.7	17.1%

6. Conclusions

In this work, a mesoscale equivalent model is first developed to numerically investigate the dynamic response of UHPC subjected to projectile impacts in a more refined and efficient way. Experiments on UHPC subjected to uniaxial compression and projectile impacts are conducted and are used to validate the developed model. Relying on the mesoscale equivalent model, the influence of projectile striking velocities, UHPC compression strengths, and volume percentages of steel fibers on the DOP is numerically investigated, and a modified Young equation for predicting the DOP is proposed. The main conclusions can be drawn as follows:

- (1) The equivalent treatment on steel fibers is to amplify the size of the fibers and the interfacial shearing modulus between fibers and the matrix by n times synchronously. The interfacial shearing force is analytically proven to be equal to that before the equivalent treatment conducted on steel fibers, demonstrating that the equivalent treatment on steel fibers is theoretically reasonable. The mesoscale equivalent model for UHPC with steel fibers is successfully developed based on the equivalent treatment on fibers. When the amplification factor of steel fiber is lower than 5, the proposed model can accurately simulate the uniaxial compression behavior of UHPC specimens. However, when the amplification factor is greater than 5, the model cannot well characterize the ductility and toughness of UHPC.
- (2) When the amplification factor of steel fibers is lower than 5, the mesoscale equivalent model can accurately reproduce the failure mode and stress-strain curve of the UHPC specimens under the uniaxial compression. The computational cost of the numerical simulations of penetration experiments decreases rapidly with the increase of the amplification factor. With an amplification factor of 5, the maximum relative error between the numerical results of the cater diameter and penetration depth and experimental results is 11.7%, indicating that the mesoscale equivalent model has high accuracy.
- (3) The mesoscale equivalent model provides a more refined and in-depth perspective into numerically investigating the response of UHPC subjected to projectile impacts. The numerical investigation of the DOP shows that the DOP increases exponentially with the increase of the projectile striking velocity. The decreasing relationships between the DOP and the compression strength and volume percentage of steel fibers can be described by the power function and quadratic polynomial, respectively. Steel fibers exert a bridging effect by limiting the deformation of matrix elements to improve the penetration resistance of UHPC, but there is a limit to reducing the DOP by adding more steel fibers.
- (4) Based on the simulated data of the DOP, a modified Young equation is proposed for predicting the DOP of UHPC targets subjected to projectile impacts. The maximum relative error between the modified equation and experimental data is 13.9%, showing the proposed equation has high accuracy.

Future studies will focus on establishing a mesoscale equivalent model for UHPC with fibers in different shapes and materials or numerically investigating the dynamic response of UHPC under blast loadings based on the mesoscale equivalent model.

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Article

A Machine Learning Model for the Prediction of Concrete Penetration by the Ogive Nose Rigid Projectile

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Abstract: In recent years, research interest has been revolutionized to predict the rigid projectile penetration depth in concrete. The concrete penetration predictions persist, unsettled, due to the complexity of phenomena and the continuous development of revolutionized statistical techniques, such as machine learning, neural networks, and deep learning. This research aims to develop a new model to predict the penetration depth of the ogive nose rigid projectile into concrete blocks using machine learning. Genetic coding is used in Python programming to discover the underlying mathematical relationship from the experimental data in its non-dimensional form. A populace of erratic formulations signifies the rapport amid dependent parameters, such as the impact factor (I), the geometry function of the projectile (N), the empirical constant for concrete strength (S), the slenderness of the projectile (λ), and their independent objective variable, X/d , where X is the penetration depth of the projectile and d is the diameter of the projectile. Four genetic operations were used, including the crossover, sub-tree transfiguration, hoist transfiguration, and point transfiguration operations on supervised test datasets, which were divided into three categories, namely, narrow penetration ($X/d < 0.5$), intermediate penetration ($0.5 \leq X/d < 5.0$), and deep penetration ($X/d \geq 5.0$). The proposed model shows a significant relationship with all data in the category for medium penetration, where $R^2 = 0.88$, and $R^2 = 0.96$ for deep penetration. Furthermore, the proposed model predictions are also compared with the most commonly used NDRC and Li and Chen models. The outcome of this research shows that the proposed model predicts the penetration depth precisely, compared to the NDRC and Li and Chen models.

Keywords: penetration; machine learning; concrete; rigid projectile; symbolic regression

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1. Introduction

1.1. Research Background

In the 20th century worldwide, various studies were conducted to produce innovative concrete variations [1]. However, since the 19th century, ordinary concrete is still the foremost distinctive practical material and is commonly used to build structures against accidentally-occurring impact loads [1]. These accidentally-occurring impact loads, such as vehicle crashes, plane crashes, tsunami, tornadoes, and flying objects are the main sources of penetration in concrete [2]. Figure 1 shows the illustration of penetration that occurs due to ogive nose rigid projectile impacts on concrete structures [2].

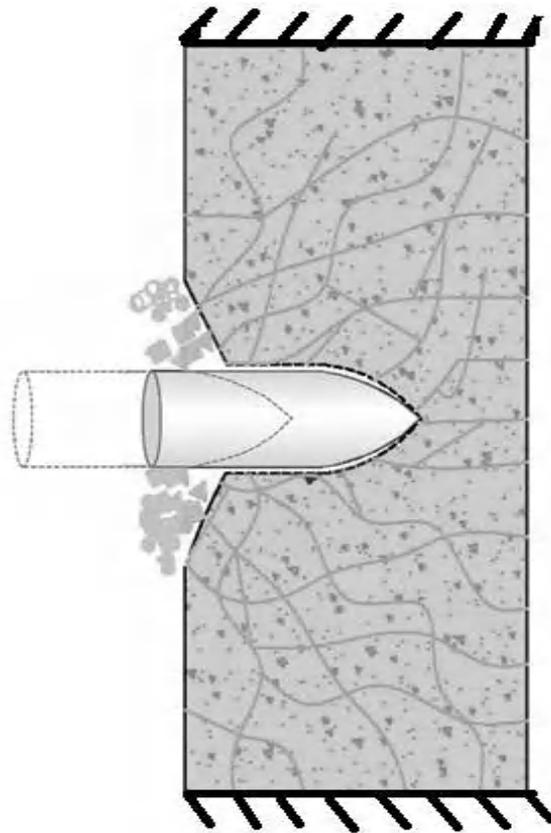


Figure 1. Penetration of concrete caused by ogive nose rigid projectile.

The historical penetration depth prediction model, that was developed based on experimental data, showed that the USA most frequently used the Petry model/the modified Petry model to predict penetration depth into a concrete block [1,3]. The Petry model was initially developed in 1910 and is considered the oldest available empirical model in literature [1,3]. Later, the Petry model was modified by Q.M. Li in the S.I unit [1,3]. In 1941, the Ballistic Research Laboratory (BRL) proposed the penetration prediction model [1,3]. Commencing in 1943, the Army Corp of Engineers also established the ACE model [1,3]. In 1946, the National Defense Research Committee (NDRC) proposed a model founded on the ACE model [1,3,4] The limitation of the NDRC hypothesis was that the NDRC model was used for rear penetration depths [1,3]. The NDRC also suggested a nose shape factor N^* for projectiles, as mentioned in Table A1. Later, Ammann and Whitney's model was anticipated to foretell the penetration of concrete traceable to the effect of strenuously precipitated particles at comparatively sophisticated velocities [1,3]. According to Kennedy [5], the Ammann and Whitney model can be used for velocities over 1000 ft/s [1,3]. Whiffen further continued research in the British Road Research Laboratory in the United Kingdom, using experimental data acquired from World War II, and developed a prediction model for penetration depth [1,3]. Kar altered the NDRC model, employing reversion regarding the modulus of elasticity (E), where E is the modulus of elasticity of the projectile, and E_s is the modulus of elasticity of steel [1,3]. In the United Kingdom, Barr recommended a UKAEA model by reforming the NDRC model derived from wide-ranging research on nuclear power plant structures [1,3]. The Halder–Hamieh model [1,3,6] of penetration depth relies on the dimensionless impact factor (I_a), where and N^* is the nose shape factor [1,3,6]. Adeli and Amin improved the impact factor (I_a), familiarized by Halder and Hamieh, using regression on Sliter's experimental data [1,3]. Hughes revised the NDRC hypothesis and suggested a model where I_h is a non-dimensional impact factor [1,3] and N_h is a nose shape factor [1,3]. Hughes proposed that the tensile and compressive strength of the concrete (f_t/f_c) ratio is constant [1,3,7]. Hughes also showed the importance of the strain rate

and the dynamic increase factor (DIF), represented by 'S' [1,3,7]. Healy and Weissman introduced a model for penetration depth by revising the NDRC and Kar models [1,3]. The CRIEPI Model, mentioned below in Table A1, assumes the thickness of the concrete barrier $H_r = 20$ cm (0.2 m) [1,3]. In 1985, the United Kingdom Nuclear Electronics (UKNE) began intrinsic research on the behavior of concrete structures resisting hard projectile penetration by creating the R3 Concrete Impact Working Party [1,3]. The UMIST model for penetration depth (X) improved the form with the reflection of the nose shape [1,3]. Li and Chen [1,3,8] further advanced Forrestal et al.'s [1,3,9] model and proposed a semi-empirical or semi-analytical model for penetration depth (X). The model is in a non-dimensional homogenous form, and these models are valid for an extensive scope of penetration depth, where I is the impact function, and N is the geometry function. $S = 72f_c^{-0.5}$ is an empirical function of f_c (MPa). The Li and Chen model is applicable for $X/d \geq 5.0$ [2,4]. Li and Chen [1,3,8,10] recommended $X/d < 5.0$ for small-to-medium penetration depths, where h is the length of the nose of the projectile. In the case of narrow penetrations when $X/d < 0.5$, the penetration depth is given by [1,3,8,10]. Chen and Li [1,3,8,10] recommended a simplified model of $X/d = 0.5(I)$ to predict the penetration depth for deep penetration. The details of the discussed models are given in Appendix A Table A1.

Since 2005, researchers turned their attention to developing models in terms of the required critical impact energy [1,2]. Furthermore, research also turned to the advancement of concrete, creating high-performance concrete (HPC), ultra-high-performance concrete (UHPC), and high-performance fiber-reinforced concrete (HPFRC). However, until today, normal concrete and normal reinforced concrete are still the most commonly used materials for the construction of structures. The revised interest of researchers on concrete penetration depth emerged in 2015, when Hussein and Dalvand implemented an evolution of the statistical model developing technique, the neural network, to develop a model for the prediction of the rigid projectile penetration depth in concrete. Furthermore, in 2021, research was conducted using gradient tree boosting machine learning to predict RC panel failure modes under impact loading. These two kinds of research emerged with the need for the redevelopment of the model, using advanced machine learning tools to develop a prediction model based on an extensive collection of experimental data. Therefore, this research is focused on developing a model based on an extensive collection of 257 experimental test data, using machine learning symbolic regression. Furthermore, the proposed prediction model is validated with the prediction of the NDRC and Li and Chen models, because, in the literature, the most prominent model used to predict penetration depth is the Li and Chen dimensionless model [1].

1.2. Research Motivation

Based on the literature, there is a need to modernize the penetration depth prediction model in recent years. This is mainly due to the complexity of penetration phenomena, where there is a continuous evolution of techniques, such as neural networks and machine learning, especially since concrete is still commonly used in structures [11,12].

In 2015, Hussein and Dalvand used a neural network on 70 experimental datasets taken to estimate the penetration depth in concrete targets using an ogive nose rigid projectile [8,10,13–16]. The neural network models have a very low rate of error and high correlation coefficients compared to the regression-based models [17]. In 2021, Thai et al. used gradient tree boosting machine learning to predict RC panel failure modes under impact loading. The accuracy of the prediction result was not as high as expected, due to the lack of data and the unbalance experimental output features. However, this new approach was recommended for further investigation [18].

1.3. Research Objectives

The main objective of this research is to develop a modernized model for predicting penetration depth using machine learning symbolic regression genetic programming in Python. The model can predict the penetration depth with greater accuracy on a wide range of 257 tests, compared with the NDRC and Li and Chen models. The total of 257 test datasets from [2,8,10,13–16,19] is divided into three categories of narrow penetration depths, where $X/d < 0.5$; intermediate penetration depth, where $0.5 < X/d < 5.0$; and deep penetration, where $X/d > 5.0$. X is the penetration depth and d is the diameter of the projectile. The significance of this research is that a reasonable number of 257 test datasets are trained and tested using machine learning symbolic regression with a genetic programming crossover, sub-tree transfiguration, hoist transfiguration, and point transfiguration to develop a model with better accuracy. The proposed model further compares with the NDRC and Li and Chen equations.

2. Machine Learning Symbolic Regression Genetic Programming Using Python

An interpretable supervised machine learning symbolic regression was used to discover the fundamental scientific equation to describe a Python correlation. Symbolic regression discovers mathematical equations using genetic programming [20]. A populace of erratic formulations is used to signify rapport amid dependent parameters, such as the impact factor (I), the geometry function of the projectile (N), the empirical constant for concrete strength (S), the slenderness of projectile (λ) and their independent objective variable, X/d , where X is the penetration depth of the projectile and d is the diameter of the projectile. In the iterative process, the consecutive propagation of a series of events is progressed from its predecessor's generation by choosing the populace's adequate entities to undertake genetic maneuvers. Genetic encoding yields a sequence of entirely unplanned programs (or models) and further predicts scientific equations to express the relationship between the reliant and sovereign parameters [21]. Four genetic operations used in genetic programming are the crossover, sub-tree transfiguration, hoist transfiguration, and point transfiguration operations, as shown in Figure 2. A sub-tree is randomly selected from the front-runner of an event in the crossover method and is replaced with a randomly selected sub-tree, since it is the front-runner of one more event (Figure 2a). A sub-tree is randomly chosen from the front-runner of an event in the sub-tree mutation method, and it is replaced with a sub-tree that is generated randomly (Figure 2b). The hoist mutation method selects a sub-tree of a randomly selected sub-tree from the winner of a tournament, which replaces the previously chosen sub-tree (Figure 2c). The point mutation method randomly selects some nodes from the tournament winner and replaces them with other building blocks (Figure 2d).

Symbolic regression generates a populace of unplanned scientific formulations with independent data factors as variants [22]. This scientific formulation genus transfigures and emerges as innovative models via genetic encoding [22]. The resulting formulations endeavor to predict trial data by appraising the specified metric (e.g., mean absolute error, root mean-squared error, or mean-squared error) between the predicted and actual values as shown in Figure 3.

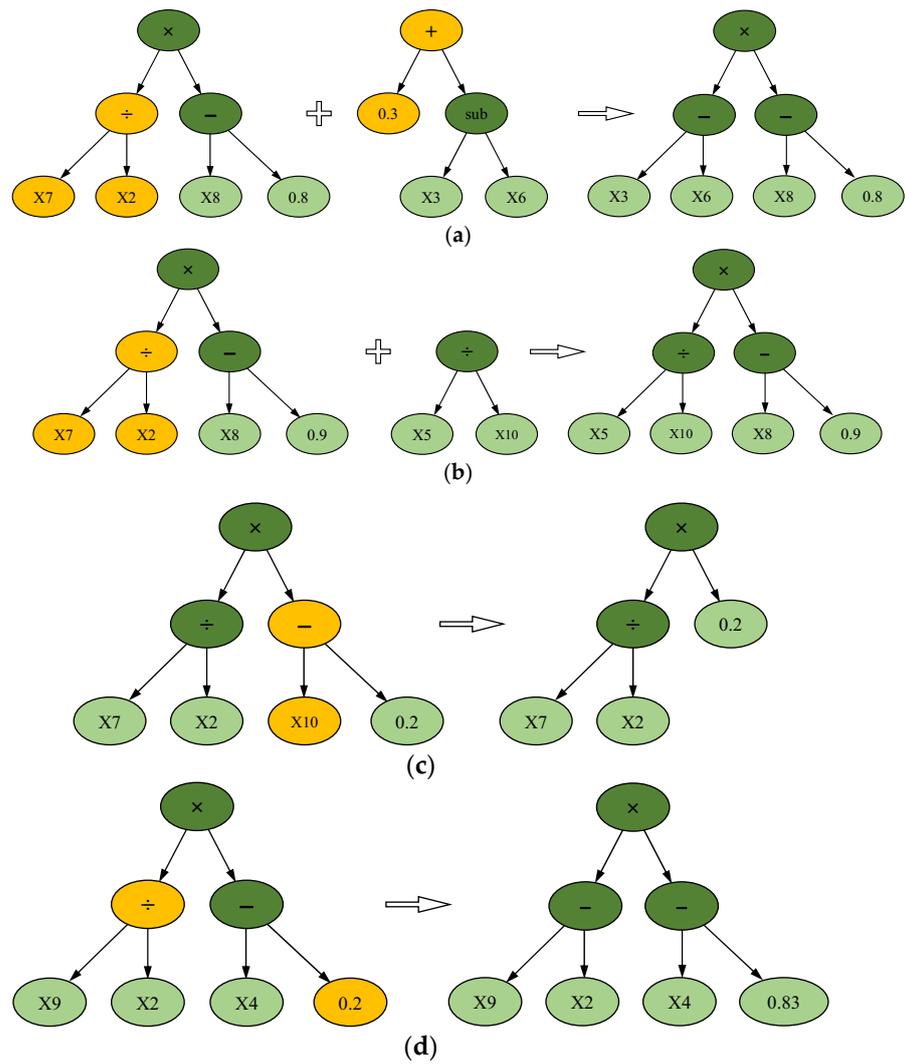


Figure 2. (a) The schematic diagrams of crossover method genetic operations. (b) The schematic diagrams of sub-tree transfiguration genetic operations. (c) The schematic diagrams of hoist transfiguration, genetic operations. (d) The schematic diagrams of point transfiguration, genetic operations.

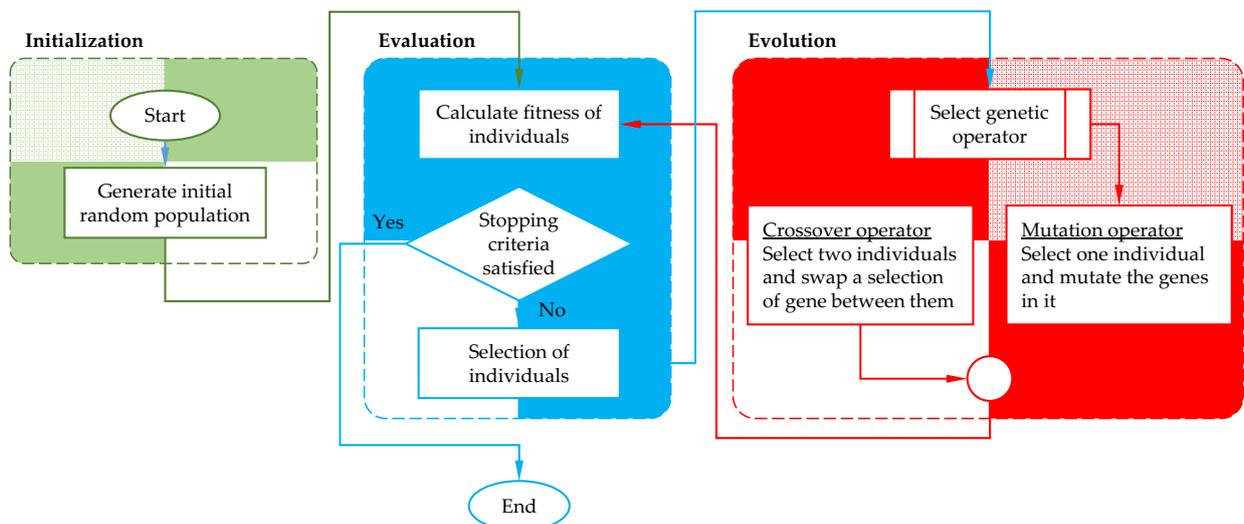


Figure 3. Flowchart of a genetic algorithm.

3. Data Analysis

The penetration of the projectile can be divided into three categories, namely, deep penetration ($X/d \geq 5.0$), intermediate penetration ($0.5 \leq X/d < 5.0$), and narrow penetration ($X/d < 0.5$), where X is the penetration depth and d is the projectile’s diameter [10]. Experimental data in dimensionless form, shown in Table 1, is used for the symbolic regression analysis to obtain a mathematical model [2,8,10,13–16,19]. The data’s dimensionless parameters include the impact factor $I = \left(\frac{MV_0^2}{f_c d^3 S}\right)$, where M is mass of the rigid projectile, V_0 is impacting projectile velocity, f_c is the compressive strength of the concrete, d is the diameter of the rigid projectile, and $S = 72f_c^{-0.5}$ is the empirical constant for concrete strength. The geometry function of the projectile is $N = \left(\frac{M}{l_c d^3 N^*}\right)$, where M is the mass of the rigid projectile, N^* is nose shape factor projectile, l_c is the density of concrete, and d is the diameter of the rigid projectile. The slenderness of the projectile is $\lambda = \left(\frac{M}{l_c d^3}\right)$, where M is the mass of the rigid projectile, l_c is the density of concrete, and d is the diameter of the rigid projectile. The empirical constant for concrete strength is $S = 72f_c^{-0.5}$, and X/d refers to the dimensionless penetration depth ratio-to-projectile diameter. The dimensionless penetration X/d is taken as a dependent target variable, whereas I , N , λ , and S are independent variables or predictors.

Table 1. Dimensionless data with penetration depths for regression analysis.

λ	S	N	I	$(X/d)_{\text{test}}$	λ	S	N	I	$(X/d)_{\text{test}}$
15.2	21	143.4	14.45	9.83	5.31	11.81	5.31	0.66	0.86
15.2	21	143.4	36.55	24.15	5.31	11.04	5.31	0.78	0.91
15.2	21	143.4	47.28	27.79	5.31	12.41	5.31	1.33	1.22
15.2	21	143.4	54.73	32.04	5.31	12.07	5.31	1.25	1.29
15.2	21	143.4	93.77	49.54	5.31	11.81	5.31	1.36	1.31
15.2	21	143.4	133.13	65.79	5.31	11.04	5.31	1.16	1.29
15.2	21	200	12.5	8.59	10.75	12.41	10.75	1.47	1.51
15.2	21	200	35.94	24.15	10.75	12.07	10.75	1.43	1.73
15.2	21	200	54.73	33.98	10.75	11.81	10.75	1.45	1.58
15.2	21	200	85.07	51.32	10.75	11.04	10.75	1.37	1.58
15.2	21	200	118.67	66.56	10.75	12.41	10.75	2.77	2.36
19.64	12	125.9	8.45	6.43	10.75	12.07	10.75	2.71	2.89
19.73	12	126.5	17.33	11.52	10.75	11.81	10.75	2.42	2.27
19.66	12	126	18.93	15.28	10.75	11.04	10.75	2.29	2.36
19.77	12	126.7	29.02	17.84	6.12	10.81	6.12	2.41	1.45
19.73	12	126.5	32.62	19.52	6.12	11.39	6.12	1.65	1.47
19.62	12	125.8	36.55	27.1	6.12	11.86	6.12	0.71	1
19.53	12	125.2	33.6	19.07	6.12	10.8	6.12	1.7	1.5
19.57	12	125.5	43.48	22.57	3.14	11.39	3.14	0.35	0.65
19.62	12	125.8	46.02	23.05	3.14	11.17	3.14	0.44	0.77
19.53	12	125.2	64.02	32.19	3.14	11.17	3.14	1.39	1.1
19.6	12	125.6	76.45	35.61	5.85	10.77	5.85	1.91	1.25
19.91	7	127.6	23.15	13.12	6.34	10.77	6.34	1.16	1.2
19.72	7	126.4	24.7	14.28	6.34	11.19	6.34	2.54	1.65
19.93	7	127.8	25.85	16.25					
19.87	7	127.4	25.34	15.69	14.45	15	144.51	1.66	3.15
19.91	7	127.6	39.83	23.42	14.4	15	144.03	2.29	4.07
19.76	7	126.7	38.35	22.49	12.98	11.26	121.91	3.77	3.94
					14.44	15	135.65	3.42	5.51
15.2	15.2	143.4	21.95	13.16	14.31	15	134.43	3.42	5.91
15.2	15.2	143.4	34.62	19.35	12.96	11.26	121.69	5.05	4.99
15.2	15.2	143.4	56.3	34.83	12.93	11.26	237.71	6.47	8.01
15.2	15.2	143.4	75.08	42.57	14.47	15	266.01	4.86	7.61
15.2	15.2	143.4	96	58.05	12.97	11.26	121.79	6.55	5.91

Table 1. Cont.

λ	S	N	I	$(X/d)_{test}$	λ	S	N	I	$(X/d)_{test}$
15.2	15.2	143.4	118.24	65.79	14.5	15	136.15	5.36	8.14
15.2	15.2	200	20.28	13.16	14.28	15	134.12	6.62	9.58
15.2	15.2	200	39.47	20.9	14.47	15	135.89	6.66	9.06
15.2	15.2	200	54.45	31.73	19.64	12	125.69	8.46	6.43
15.2	15.2	200	76.9	44.12	12.97	11.26	121.83	9.08	6.96
15.2	15.2	200	99.95	58.82	14.58	15	136.91	6.94	9.97
15.2	15.2	200	128.38	68.11	14.45	15	135.68	9.41	12.6
24.84	8.6	234.3	21.43	14.78	14.49	15	136.09	9.61	12.34
24.84	8.6	234.3	39.63	23.65	14.49	15	136.09	9.71	12.21
24.84	8.6	234.3	71.31	37.44	14.46	15	135.78	9.71	13.39
24.84	8.6	234.3	90.72	46.8	12.97	11.26	238.38	13.37	12.99
24.84	8.6	234.3	103.09	45.32	13.02	11.26	122.23	13.89	12.34
24.84	8.6	234.3	110.97	46.31	19.91	7	127.44	23.15	13.12
24.52	10.5	231.3	17.28	12.13	19.72	7	126.18	24.7	14.28
24.52	10.5	231.3	20.95	13.77	19.87	7	127.16	25.34	16.25
24.52	10.5	231.3	31.29	18.36	19.94	7	127.58	25.93	15.69
24.52	10.5	231.3	44.64	25.57	24.64	10.5	231.41	17.36	12.13
24.52	10.5	231.3	68.09	34.43	14.47	15	266.07	12.26	16.4
24.52	10.5	231.3	70.99	40.33	24.84	8.6	233.31	21.42	14.78
24.52	10.5	231.3	85.31	46.23	14.54	15	136.5	12.32	15.49
24.52	10.5	231.3	107.23	57.38	24.63	8.7	231.3	21.97	14.14
24.52	10.5	231.3	120.29	64.26	24.61	7.9	231.13	24.51	15.08
24.52	10.5	231.3	151.92	66.56	19.73	12	126.25	17.33	11.52
					24.52	10.5	230.26	20.95	13.77
24.63	8.7	232.4	21.97	14.14	19.66	12	125.83	18.93	15.28
24.63	8.7	232.4	41.92	24.19	14.84	20.2	194.91	12.95	8.51
24.63	8.7	232.4	74.7	41.38	15.19	21	199.51	12.5	8.59
24.63	8.7	232.4	114.5	64.04	19.76	7	126.46	38.35	22.49
24.63	8.7	232.4	151.85	78.33	19.91	7	127.44	39.83	23.42
24.63	8.7	232.4	70.37	35.96	14.84	20.2	139.34	14.98	10.06
24.63	8.7	232.4	111.12	57.14	15.19	21	142.63	14.45	9.83
24.63	8.7	232.4	152.64	71.92	14.84	15.2	194.91	20.22	13.16
24.61	7.9	232.2	24.5	15.08	24.52	10.5	230.26	31.28	18.36
24.61	7.9	232.2	42.21	25.9	14.84	15.2	139.34	21.88	13.16
24.61	7.9	232.2	77.91	40.33	24.61	7.9	231.13	42.21	25.9
24.61	7.9	232.2	118.85	63.93	24.84	8.6	233.31	39.63	23.65
24.61	7.9	232.2	121.78	64.26	19.77	12	126.52	29.02	17.84
24.61	7.9	232.2	171.15	87.54	24.63	8.7	231.3	41.85	24.19
					19.73	12	126.25	32.62	19.52
39.69	11.02	39.69	0.18	0.71	19.53	12	125	33.6	19.07
3.48	11.1	3.48	0.15	0.56	19.62	12	125.55	36.55	27.1
3.48	11.1	3.48	0.15	0.66	24.56	10.5	230.69	44.71	25.57
2.45	11.1	2.45	0.15	0.41	19.57	12	125.27	43.48	22.57
3.48	11.1	3.48	0.17	0.61	14.84	15.2	139.34	34.52	19.35
2.45	11.1	2.45	0.15	0.34	19.62	12	125.55	46.02	23.05
2.45	12.22	2.45	0.24	1.31	14.84	15.2	194.91	39.37	20.9
2.93	12.22	2.93	0.15	0.32	24.63	8.7	231.3	70.37	35.96
2.45	12.22	2.45	0.17	0.41	24.84	8.6	233.31	71.31	37.44
2.94	12.22	2.94	0.2	0.78	24.61	7.9	231.13	78.42	40.33
3.48	12.22	3.48	0.27	0.73	24.63	9.04	231.3	68.77	49.75
3.48	12.22	3.48	0.22	0.49	24.63	9.04	231.3	69.81	60.39
2.45	12.22	2.45	0.3	0.57	24.63	8.7	231.3	74.7	41.38
2.45	12.22	2.45	0.08	0.07	24.58	10.5	230.84	68.24	34.43
2.94	12.22	2.94	0.3	0.54	24.52	10.5	230.26	70.98	40.33
3.48	11.34	3.48	0.3	1.47	14.84	20.2	194.91	37.24	23.99

Table 1. Cont.

λ	S	N	I	$(X/d)_{test}$	λ	S	N	I	$(X/d)_{test}$
2.45	11.34	2.45	0.33	0.95	15.19	21	199.51	35.93	24.15
2.94	11.34	2.94	0.1	0.1	14.84	20.2	139.34	37.88	23.99
2.94	11.34	2.94	0.22	0.44	15.19	21	142.63	36.55	24.15
2.45	11.76	2.45	0.4	0.75	19.53	12	125	64.02	32.19
13.04	11.76	13.04	0.38	0.5	24.84	8.6	233.31	90.72	46.8
4.6	11.76	4.6	0.25	0.66	14.84	15.2	194.91	54.29	31.73
2.45	11.76	2.45	0.27	0.26	14.84	15.2	139.34	56.12	34.83
13.04	11.76	13.04	0.37	2.15	24.84	8.6	233.31	103.07	45.32
2.45	11.76	2.45	0.1	0.21	24.56	10.5	230.69	85.46	46.23
4.6	11.76	4.6	0.11	0.36	19.6	12	125.41	76.45	35.61
0.83	10.6	0.83	0.09	0.32	24.61	7.9	231.13	116.93	64.26
1.63	11.67	1.63	0.2	0.95	24.61	7.9	231.13	118.86	63.93
0.83	12.13	0.83	0.16	1.04	24.84	8.6	233.31	110.94	46.31
0.69	11.25	0.69	0.08	0.54	24.61	7.9	231.13	121.79	64.26
0.69	11.41	0.69	0.04	0.13	24.63	8.7	231.3	111.12	57.14
1.05	11.1	1.05	0.05	0.34	14.84	20.2	139.34	48.85	27.86
0.53	10.6	0.53	0.06	0.23	15.19	21	142.63	47.14	27.79
0.95	11.41	0.95	0.28	0.34	24.63	8.7	231.3	114.5	64.04
0.95	10.74	0.95	0.35	0.55	24.5	10.5	230.12	107.15	57.38
0.7	11.67	0.7	0.24	0.5	14.84	15.2	139.34	74.85	42.57
0.7	11.25	0.7	0.35	1	14.84	20.2	139.34	56.73	31.73
0.56	11.03	0.56	0.14	0.5	14.84	20.2	194.91	56.73	34.06
96.15	14.28	96.15	5.25	3.6	15.19	21	142.63	54.74	32.04
96.15	14.51	96.15	10.97	5.8	15.19	21	199.51	54.74	33.98
96.63	12.29	96.63	1.14	1.2	14.84	15.2	194.91	76.67	44.12
96.63	10.99	96.63	2.09	1.6	24.66	10.5	231.56	121.02	64.26
96.63	10.96	96.63	2.23	2	24.63	8.7	231.3	151.85	78.33
96.63	11.14	96.63	1.92	1.7	24.63	8.7	231.3	152.64	71.92
5.02	11.13	5.02	0.04	0.09	24.61	7.9	231.13	171.16	87.54
5	12.67	5	0.07	0.25	14.84	15.2	139.34	95.7	58.05
5	12.17	5	0.3	1.13	24.61	7.9	231.13	185.72	92.79
5	12.34	5	0.23	0.38	14.84	15.2	194.91	99.65	58.82
5.05	12.74	5.05	0.39	0.56	24.56	10.5	230.69	152.18	66.56
5.56	15.31	5.56	0.03	0.06	14.84	20.2	194.91	88.14	51.08
5.65	13.93	5.65	0.03	0.05	15.19	21	199.51	85.05	51.32
5.65	13.93	5.65	0.06	0.04	14.84	15.2	139.34	117.88	65.79
5.61	12.9	5.61	0.1	0.12	14.84	15.2	194.91	127.99	68.11
5.65	13.93	5.65	0.14	0.14	14.84	20.2	139.34	97.18	49.54
10.46	13.93	10.46	0.06	0.06	15.19	21	142.63	93.77	49.54
10.46	13.93	10.46	0.1	0.06	14.84	20.2	194.91	122.96	66.56
5.31	12.41	5.31	0.83	0.95	14.84	20.2	139.34	137.97	65.79
5.31	12.07	5.31	0.79	1.02					

The correlation among predictors is shown in Figure 4 as a correlation matrix. The correlations are calculated separately for narrow, medium, and deep penetration. N and λ values are identical in the narrow penetration data. Hence, the data have a perfect correlation of 1.0. S and N are also highly correlated. In the medium penetration data, N is highly correlated with I and λ . In the deep penetration data, the correlation among predictors is weak. A summary of the statistics of predictors for different penetration types is shown in Table 2.

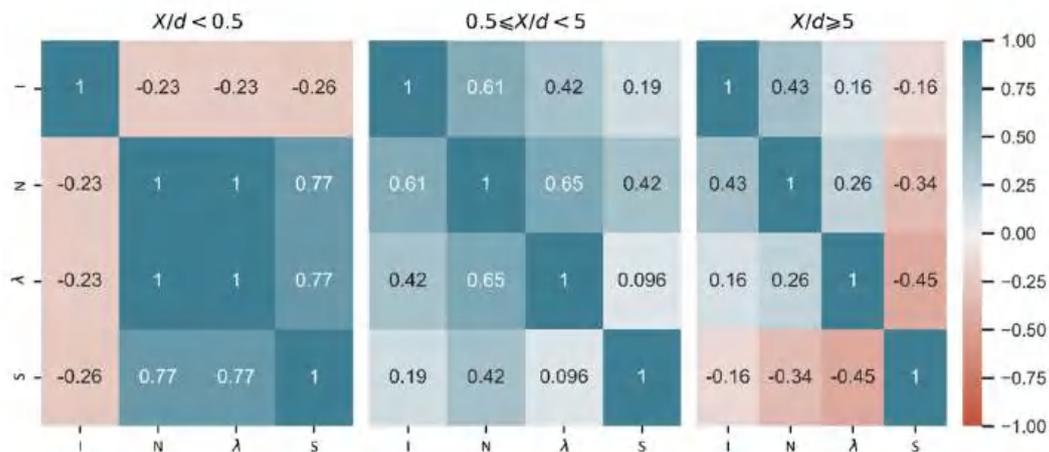


Figure 4. Correlation among N, I, S, λ , and X/d using Python.

Table 2. Descriptive statistics of N, I, S , and λ in narrow, intermediate, and deep penetrations.

Statistics	Independent Variables											
	I			N			λ			S		
	Narrow	Intermediate	Deep	Narrow	Intermediate	Deep	Narrow	Intermediate	Deep	Narrow	Intermediate	Deep
Count	26	59	174	26	59	174	26	59	174	26	59	174
Mean	0.12	1.19	60.05	3.83	22.29	182.81	3.83	14.20	19.96	12.24	11.74	12.75
Std	0.074	1.15	44.29	2.60	39.85	47.24	2.60	25.93	7.30	1.24	0.89	4.55
Cov	0.62	0.97	0.74	0.68	1.79	0.26	0.68	1.83	0.37	0.10	0.08	0.36
Min	0.03	0.08	3.42	0.53	0.56	96.15	0.53	0.56	12.93	10.60	10.74	7.00
25%	0.06	0.30	23.15	2.45	3.14	135.94	2.45	3.14	15.19	11.34	11.10	8.70
50%	0.10	0.79	45.37	2.94	5.31	194.91	2.94	5.31	19.65	11.99	11.67	12.00
75%	0.15	1.68	90.72	5.43	10.75	231.30	5.43	10.75	24.61	12.84	12.15	15.20
Max	0.28	5.25	185.72	10.46	144.51	266.07	10.46	96.63	96.15	15.31	15.00	21.00

Std: standard deviation; Cov: coefficient of variation; 25%: 25th percentile (i.e., 25 percent of data is below this value); 50%: 50th percentile; 75%: 75th percentile.

In order to explore the distribution of predictors, a boxplot (or a box-and-whisker plot) of each predictor for the different penetration ranges is shown in Figure 5. The median is shown by the vertical line inside the box, whereas the left and right sides of the box are the first and third quartiles, respectively. Most of the data lie in the first and third quartiles, and the lines that are referred to as whiskers extend on the right and left sides of the box to indicate the range. There is a clear separation of I, N , and λ in the different penetration ranges. The median of S is the same for different penetration ranges, while there is more deep penetration data. More outliers in all predictors in the medium penetration are observed, as seen by the points plotted beyond the whiskers.

Figures 6–8 show pair plots of the predictors for narrow, medium, and dense penetration ranges, respectively. The diagonals are density plots calculated from the data through a kernel density estimate (KDE) [20,21]. The KDE plots can be considered as smoothed histograms. The plots above the diagonal are scattered plots with bivariate KDE contours overlapping the scatter points. The bivariate KDE plot estimates the probability density of two variables. The shaded contours represent different density levels. The plots below the diagonal are scattered plots with linear regressions among predictors. In the narrow penetration range, the distribution of I, N , and S is skewed, with a high correlation between N and S . In the medium penetration range, the distribution of all predictors is skewed with a high correlation of N with I and λ , respectively. The I and S distributions are skewed in the deep penetration range, whereas N and λ have two modes. Furthermore, all predictors are weakly correlated in the deep penetration range.

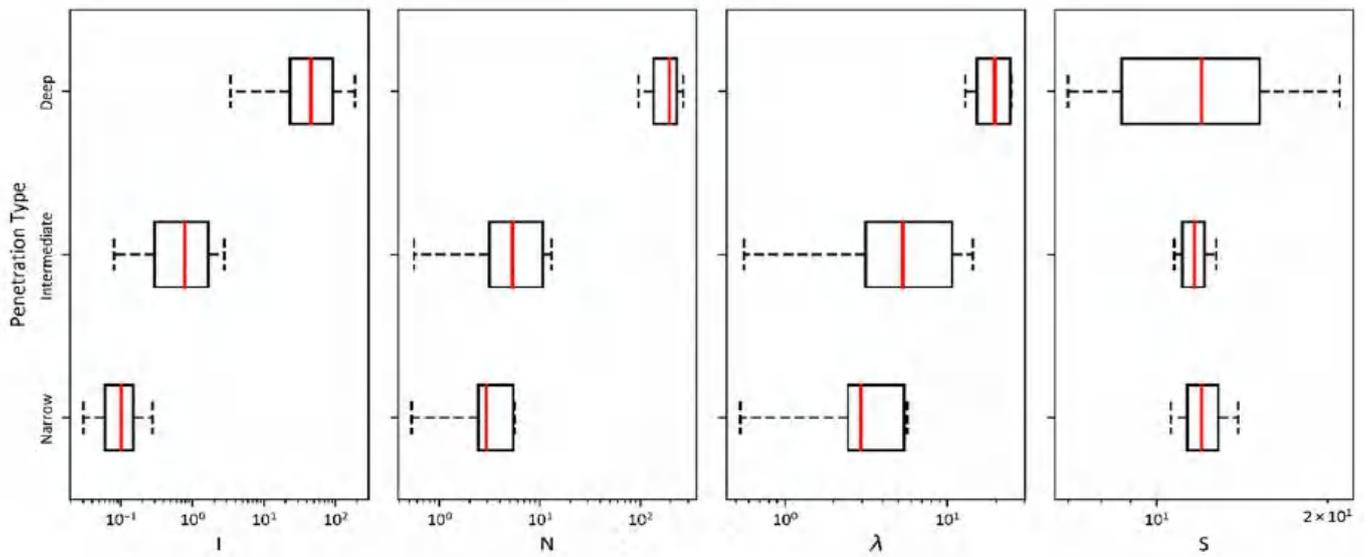


Figure 5. Box-and-whisker plot of N , I , S , and λ , in narrow, intermediate, and deep penetration using Python.

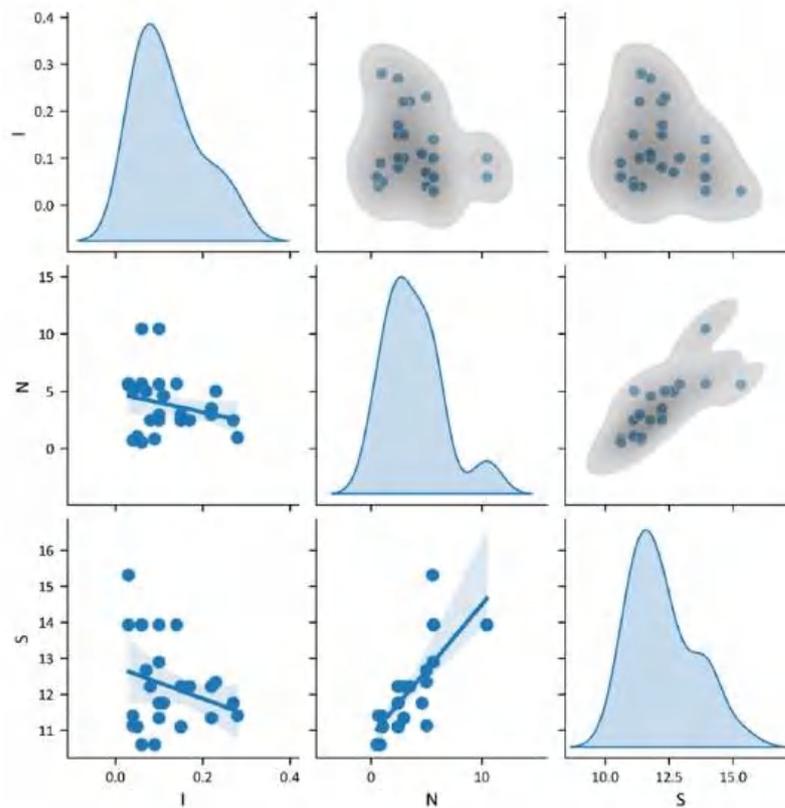


Figure 6. Pair plot of N , I , and S , for narrow penetration data using Python.

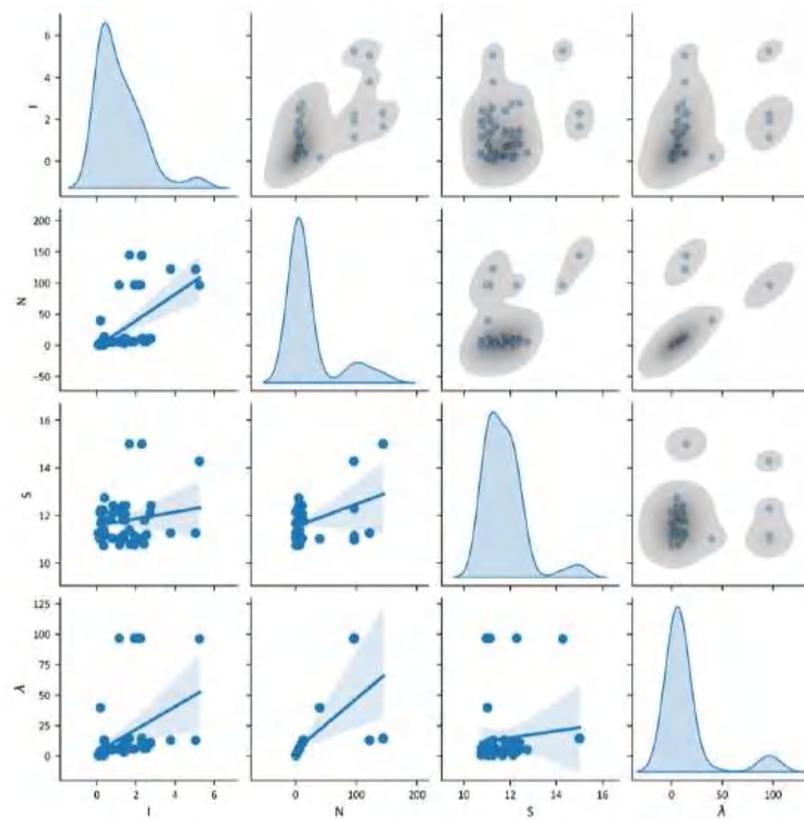


Figure 7. Pair plot of N , I , and S , for intermediate penetration data using Python.

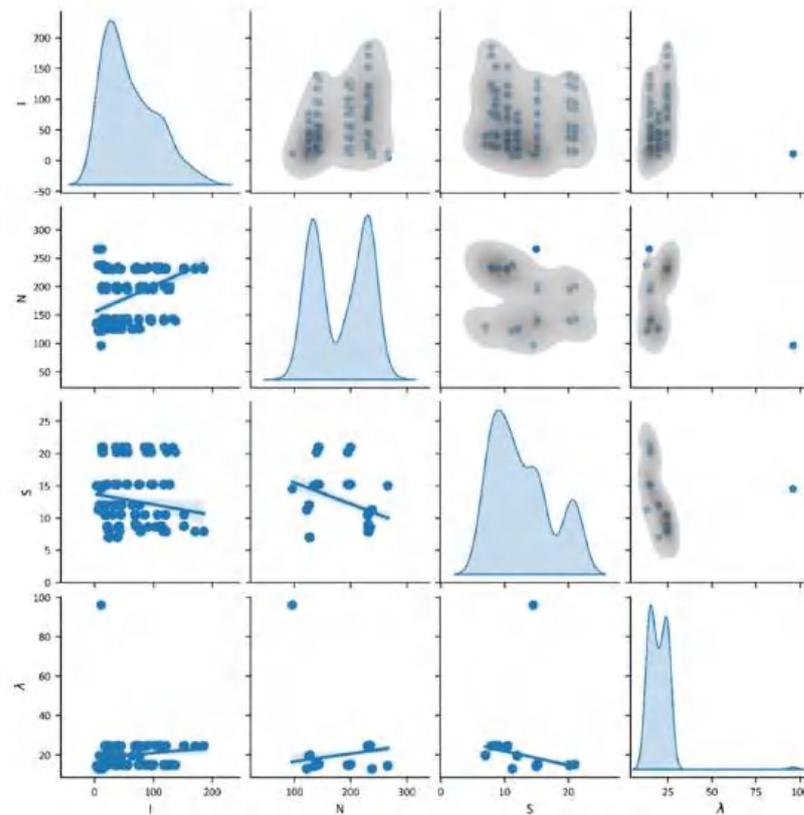


Figure 8. Pair plot of N , I , and S , for deep penetration data using Python.

4. Proposed Model Using Symbolic Regression in Python

Symbolic regression is performed using gplearn [22], which executes genetic encoding in Python through a scikit-learn stimulated and reconcilable application programming interface (API). The hyperparameters used for symbolic regression in gplearn are listed in Table 3. The experimental dataset of 257 observations [2,8,10,13–16,19] as shown in Table 1, is further separated according to the penetration type. Datasets belonging to narrow, intermediate, and deep penetrations consist of 26, 59, and 174 data observations, respectively. Symbolic regression is performed separately for each dataset to obtain the underlying mathematical expressions to describe the best relationship. For intermediate and deep penetration datasets, 70% of data is used to construct the mathematical model (i.e., train the model), and 30% of data is used to test the mathematical model’s performance (i.e., test the model). Since data observations in the narrow penetration are small and consist of 26 data observations only, all data is used to construct the mathematical model. The hyperparameters used for symbolic regression in gplearn are shown in Table 3.

Table 3. The hyperparameters for symbolic regression in gplearn.

Parameter	Value		
	EQ 1	EQ 2	EQ 3
population size	5000	5000	5000
generations	60	60	60
stopping_criteria	0.01	0.01	0.01
p_crossover	0.9	0.7	0.7
p_subtree_mutation	0.01	0.01	0.1
p_hoist_mutation	0.01	0.05	0.05
p_point_mutation	0.01	0.1	0.1
function_set	+, −, ×, ÷	+, −, ×, ÷, √	+, −, ×, ÷
tournament size	25	25	25
parsimony_coefficient	0.0003	0.002	0.003
metric	MAE	MAE	MAE
const_range	(−5, 5)	(−5, 5)	(−5, 5)

The following is an explanation of the hyperparameters: population size: number of mathematical formulas in each generation; generations: maximum number of generations; stopping_criteria: MAE value that program stops; p_crossover: crossover probability; p_subtree_mutation: subtree mutation probability; p_hoist_mutation: hoist mutation probability; p_point_mutation: point mutation probability; function_set: building blocks containing mathematical operators; parsimony_coefficient: a constant that penalizes large individuals by adjusting their MAE to make them less favorable for selection; metric: measures how well an individual fits; const_range: the range of constants included in the model.

The mathematical model obtained for narrow, medium, and deep penetration datasets from symbolic regression is shown in the Figures 9–11 as expression trees (ETs).

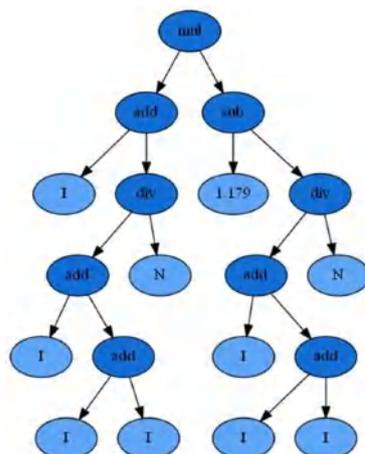


Figure 9. Model equation for narrow penetration, in expression tree form, identified by symbolic regression.

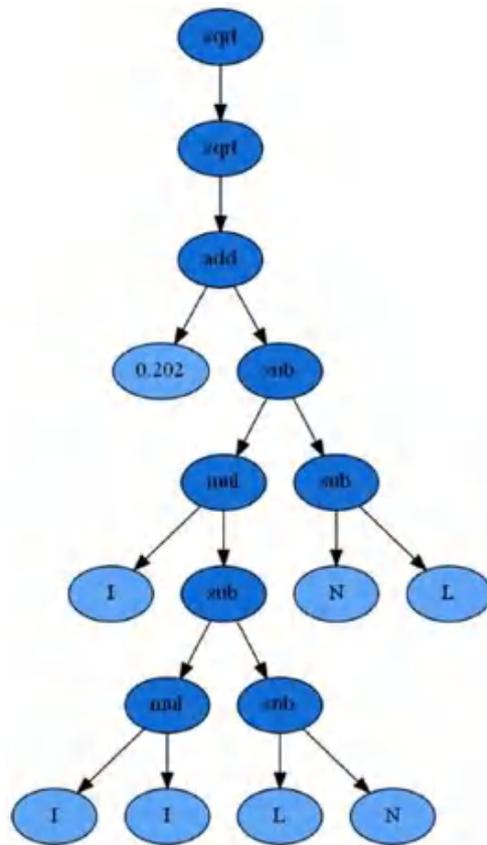


Figure 10. Model equation for intermediate penetration, in expression tree form, identified by symbolic regression.

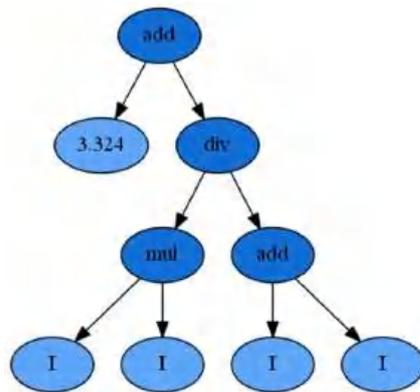


Figure 11. Model equation for deep penetration, in expression tree form, identified by symbolic regression.

The proposed equations, obtained from symbolic regression, can be re-written in mathematical form, as follows.

$$\frac{x}{d} = \left(I - \frac{3I}{N}\right) \left(1.179 - \frac{3I}{N}\right) \text{ for } \frac{x}{d} < 0.5 \tag{1}$$

$$\frac{x}{d} = I^3 + (I - 1.0)(N - \lambda) + 0.202 \text{ for } 0.5 < \frac{x}{d} < 5.0 \tag{2}$$

$$\frac{x}{d} = 0.5I + 3.324 \text{ for } \frac{x}{d} > 0.5 \tag{3}$$

5. Results and Discussion

The mathematical model performances are shown in Figures 12–14 and Table 4 for narrow, intermediate, and deep penetration, respectively. Table 4 shows the R^2 , MSE, MAE of proposed model in comparison with NDRC, and Li and Chen model. Figure 12a shows the comparison of the model-predicted values with the actual values of narrow penetration. The coefficient of determination, R^2 , is 0.59. Figure 12b illustrates the residual plot of predicted values. The residuals show a constant variance and are evenly spread out. The LOWESS (locally weighted scatterplot smoothing) fit is close to zero but it shows some divergence, as the variance is high at a high predicted value, which is, possibly, an outlier. The frequency distribution of residuals is shown in Figure 12c.

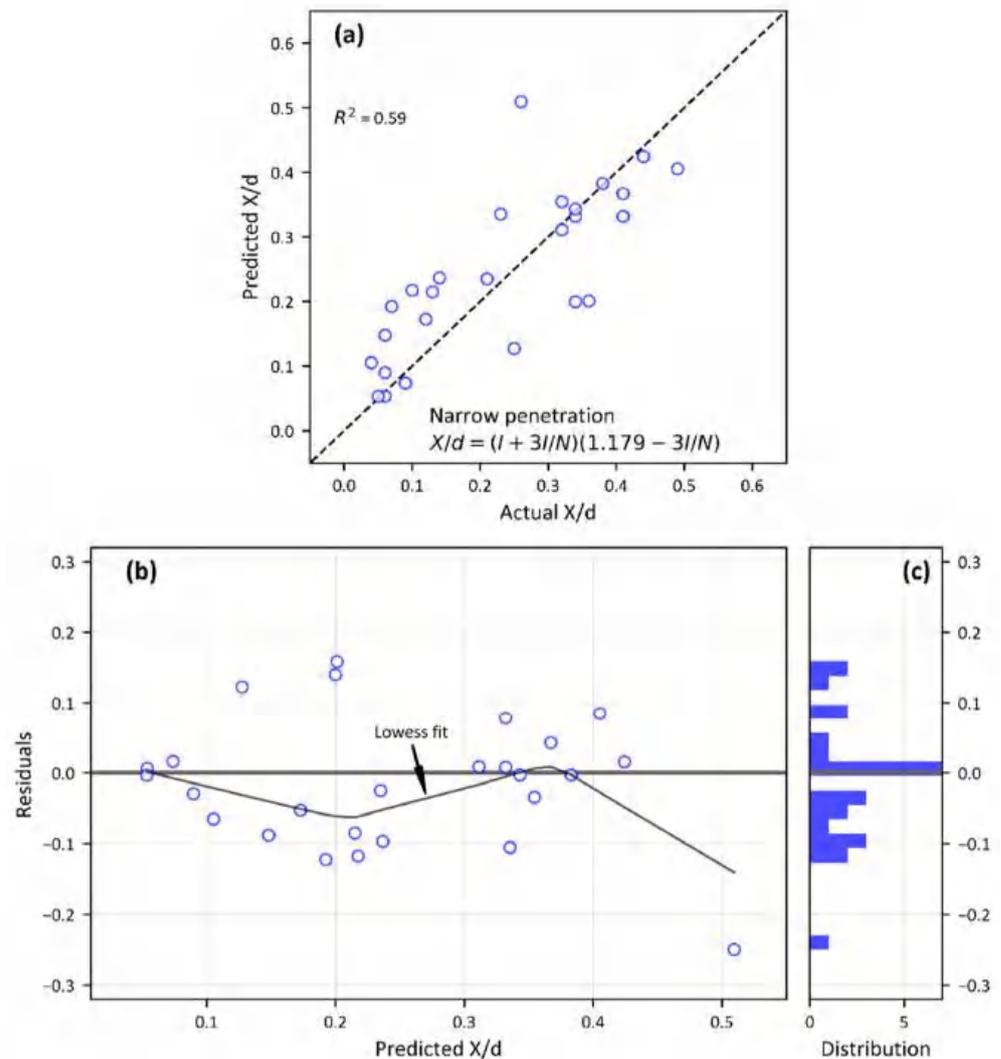


Figure 12. Symbolic regression model for narrow penetration. (a) Comparison between actual and predicted values for all data, (b) residual plot of predicted values, and (c) frequency distribution of residuals.

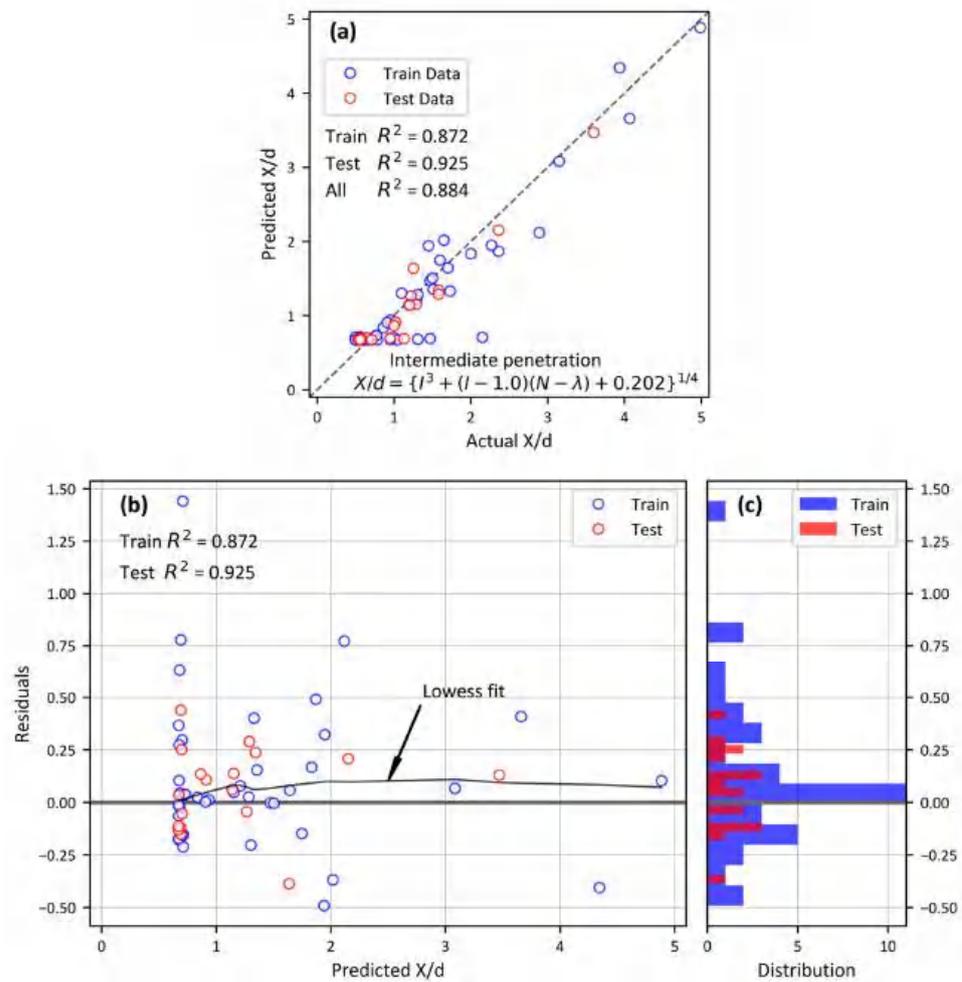


Figure 13. Symbolic regression model for intermediate penetration. (a) Comparison between actual and predicted values for train and test datasets, (b) residual plot of predicted values, and (c) frequency distribution of residuals.

Table 4. Statistics of previous studies compared to the proposed study.

Metric	Comparison of Performance between Different Equations			
	NDRC	Barr	Li and Chen	Sym. Reg. ^a (Present Study)
Narrow penetration ($X/d < 0.5$)				
R ²	−3.262	0.145	−0.031	0.590
MSE	0.085	0.017	0.021	0.008
MAE	0.270	0.097	0.118	0.068
Intermediate penetration ($0.5 \leq X/d < 5$)				
R ²	0.746	0.650	0.746	0.884
MSE	0.103	0.142	0.231	0.106
MAE	0.222	0.261	0.330	0.216
Deep penetration ($X/d \geq 5$)				
R ²	0.565	−	0.963	0.967
MSE	199.369	−	16.877	15.087
MAE	11.295	−	2.799	2.470

^a Equation (1) for narrow penetration, Equation (2) for intermediate penetration, and Equation (3) for deep penetration.

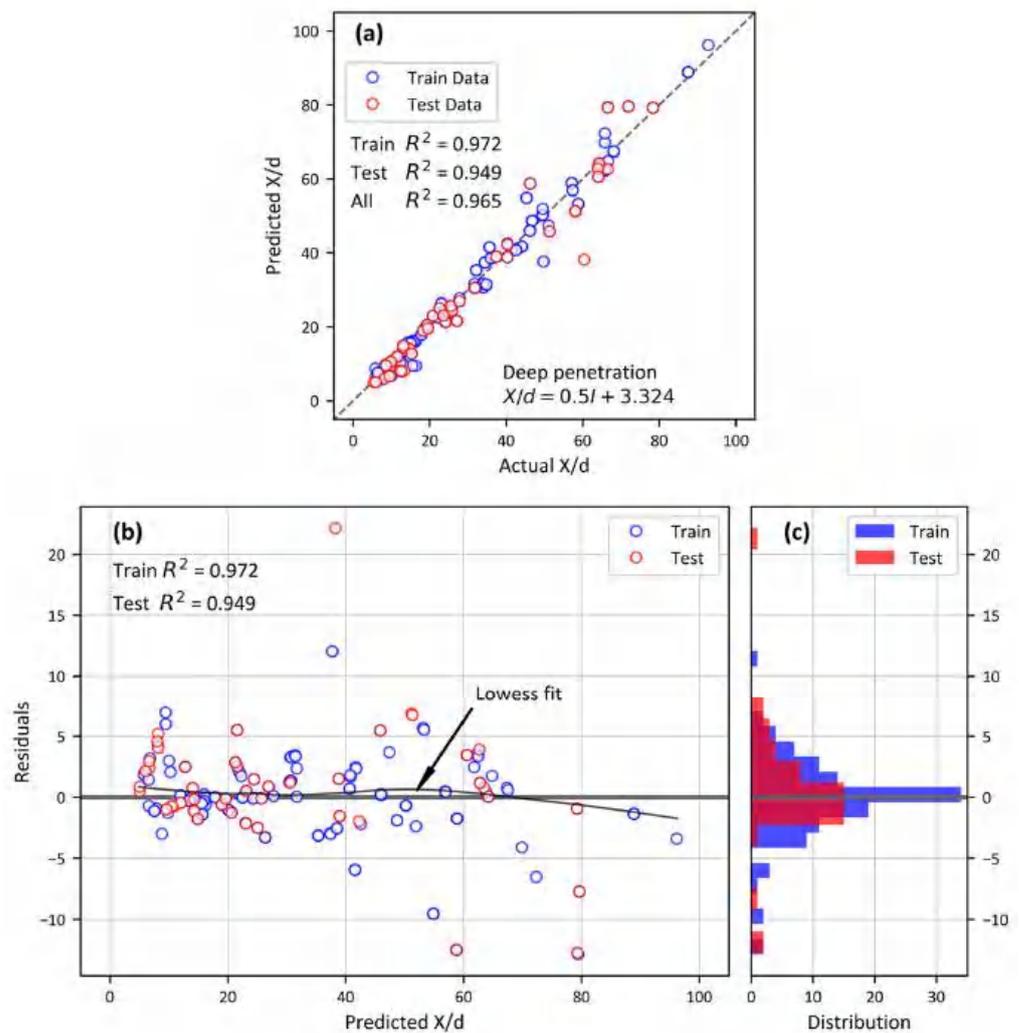


Figure 14. Symbolic regression model for deep penetration. (a) Comparison between actual and predicted values for train and test datasets, (b) residual plot of predicted values, and (c) frequency distribution of residuals.

For the intermediate penetration data, Figure 13a compares model-predicted values with actual values. The prediction of the train and test datasets show that the model performs well on unseen test data. The coefficient of determination, R^2 , for all data is 0.88. The residuals of predicted values are evenly distributed, as seen in the Figure 13b,c, which shows constant variance. The LOWESS fit is near zero and does not display divergence at low or high predicted values, as seen in Figure 13b.

For deep penetration data, Figure 14a compares model-predicted values with actual values. The model performs well on unseen test datasets. The coefficient of determination, R^2 , for all data is 0.97. The residual of predicted values is evenly distributed at low or high predicted values, exhibiting a constant variance (Figure 14b,c). The LOWESS fit is close to zero, showing an absence of divergence at low or high predicted values.

Furthermore, Figure 15 shows the prediction of the proposed equation for narrow, medium, and deep penetrations compared to the Barr, NDRC, Li and Chen, and L equations. Figure 15a shows that the NDRC and Li and Chen models are almost not applicable in the narrow penetration depths. However, the present model can be useful, compared to other models with less accuracy, due to the complexity of penetration phenomena. Figure 15b,c shows that the proposed model for the prediction of X/d is more accurate, as compared to the NDRC and Li and Chen models within the range of $0.5 \leq X/d < 5.0$ and $X/d \geq 5.0$.

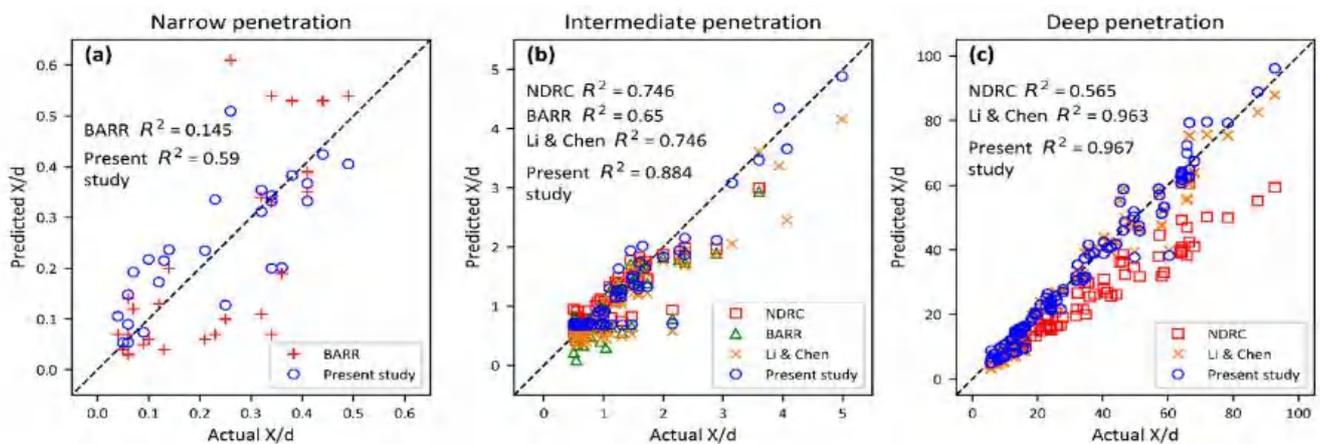


Figure 15. Comparison of performance with different equations for (a) narrow penetration, (b) medium penetration, and (c) deep penetration.

6. Conclusions

The concrete penetration of rigid projectiles is a complex phenomenon that depends on several concrete strength parameters and projectiles. For centuries, continuous research has been conducted to predict penetration with respect to advanced tools and technology. In recent years, machine learning has evolved as an advanced statistical tool that is capable of solving complex phenomena, such as penetration, with acceptable accuracy. This research developed a new model that considers four genetic operations (crossover, sub-tree transfiguration, hoist transfiguration, and point transfiguration operations) using symbolic regression machine learning tools in Python to predict penetration and to compare with the well-established NDRC and Li and Chen models. The three equations are proposed for predicting $X/d < 0.5$, $0.5 \leq X/d < 5.0$, and $X/d \geq 5.0$, respectively. The proposed equations show good relationships between test data and predicted X/d , with $R^2 = 0.88$ for $0.5 \leq X/d < 5.0$, and $R^2 = 0.96$ for $X/d \geq 5.0$. Furthermore, the proposed model is also compared with the predictions of the NDRC and Li and Chen equations. The significance of this research shows that proposed equation predictions are more accurate than the NDRC and Li and Chen models within $0.5 \leq X/d < 5.0$ and $X/d \geq 5.0$. In conclusion, it is recommended to use machine learning tools to achieve great accuracy in complex studies such as penetration, scabbing, and perforation.

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Appendix A

Table A1. Previous studies equations for prediction of penetration depth with limitations.

References	Equation in S.I. Unit	
Petry model [1,3,5]	$\frac{x}{d} = k \frac{M}{d^3} \log_{10} \left(1 + \frac{V_o^2}{19,974} \right)$	
Ballistic Research Laboratory (BRL) Model [1,3]	$\frac{x}{d} = \frac{1.33 \times 10^{-3}}{\sqrt{f_c}} \left(\frac{M}{d^3} \right) d^{0.2} V_o^{1.33}$	
Army Corp of Engineers (ACE model) [1,3,23]	$\frac{x}{d} = \frac{3.5 \times 10^{-4}}{\sqrt{f_c}} \left(\frac{M}{d^3} \right) d^{0.2} V_o^{1.5} + 0.5$	
National Defense Research Committee (NDRC) Model [1,3,4,24]	$G = 3.8 \times 10^{-5} \frac{N^* M}{d \sqrt{f_c}} \left(\frac{V_o}{d} \right)^{1.8}$	$G = \left(\frac{x}{2d} \right)^2$ $\frac{x}{d} \leq 2$
		$G = \frac{x}{d} - 1$ $\frac{x}{d} > 2$
		$\frac{x}{d} = 2G^{0.5}$ $G \geq 1$
		$\frac{x}{d} = G + 1$ $G < 1$
Ammann and Whitney model [1,3]	$\frac{x}{d} = \frac{6 \times 10^{-4}}{\sqrt{f_c}} N^* \left(\frac{M}{d^3} \right) d^{0.2} V_o^{1.8}$	
Whiffen model [1,3,25]	$\frac{x}{d} = \left(\frac{2.61}{f_c^{0.5}} \right) \left(\frac{M}{d^3} \right) \left(\frac{d}{a} \right)^{0.1} \left(\frac{V_o}{533.4} \right)^n$	$n = \frac{97.51}{f_c^{0.25}}$
Kar Model [1,3,26,27]	$G = 3.8 \times 10^{-5} \frac{N^* M}{d \sqrt{f_c}} \left(\frac{E}{E_s} \right)^{1.25} \left(\frac{V_o}{d} \right)^{1.8}$	$\frac{x}{d} = 2G^{0.5}$ $G \geq 1$
		$\frac{x}{d} = G + 1$ $G < 1$
UKAEA model [1,3,28]	$G = 3.8 \times 10^{-5} \frac{N^* M}{d \sqrt{f_c}} \left(\frac{V_o}{d} \right)^{1.8}$	$\frac{x}{d} = 0.275 - [0.0756 - G]^{0.5}$ $G \leq 0.0726$
		$\frac{x}{d} = [4G - 0.242]^{0.5}$ $0.0726 \leq G \leq 1.06$
		$\frac{x}{d} = G + 0.9395$ $G \geq 1.06$
		$G = 0.55 \left(\frac{x}{d} \right) - \left(\frac{x}{d} \right)^2$ $\frac{x}{d} < 0.22$
		$G = \left(\frac{x}{2d} \right)^2 + 0.0605$ $0.22 \leq \frac{x}{d} \leq 2.0$
		$G = \frac{x}{d} - 0.9395$ $\frac{x}{d} \geq 2.0$
Haldar and Hamieh model [1,3,6]	$\frac{x}{d} = 0.2251 I_a + 0.0308$ $\frac{x}{d} = 0.0567 I_a + 0.6740$ $\frac{x}{d} = 0.0299 I_a + 1.1875$	$I_a = \frac{MN^* V_o^2}{f_c d^3}$ $0.3 \leq I_a \leq 4.0$
		$I_a = \frac{MN^* V_o^2}{f_c d^3}$ $4.0 \leq I_a \leq 21$
		$I_a = \frac{MN^* V_o^2}{f_c d^3}$ $21 \leq I_a \leq 455$
Adeli and Amin Model [1,3]	$\frac{x}{d} = 0.0416 + 0.1698 I_a - 0.0045 I_a^2$ $0.0123 + 0.196 I_a - 0.008 I_a^2 + 0.0001 I_a^3$	$I_a = \frac{MN^* V_o^2}{f_c d^3}$ for $0.3 \leq I_a \leq 4$
		$I_a = \frac{MN^* V_o^2}{f_c d^3}$ $4 \leq I_a \leq 21$
Hughes Model [1,3,7]	$\frac{x}{d} = 0.19 \frac{N_h L_h}{S}$	$I_h = \frac{MV_o^2}{f_c d^3}$ $S = 1.0 + 12.3 I_h (1.0 + 0.03 I_h)$ $I_h < 3500$
Healy and Weissman Model [1,3]	$G = 4.36 \times 10^{-5} \left(\frac{E}{E_s} \right) \frac{N^* M}{d \sqrt{f_c}} \left(\frac{V_o}{d} \right)^{1.8}$	$\frac{x}{d} = 2G^{0.5}$ $G \geq 1$
		$\frac{x}{d} = G + 1$ $G < 1$
CREIPI Model [1,3,29]	$\frac{x}{d} = \frac{0.0265 N^* M d^{0.2} V_o^2 \left[114 - 6.83 \times 10^{-4} f_c^{\frac{2}{3}} \right]}{f_c^{\frac{2}{3}}} \left[\frac{(d+1.25H_r)H_r}{(d+1.25H_o)H_o} \right]$	
UMIST model [1,3]	$\frac{x}{d} = \left(\frac{2}{\pi} \right) \frac{N^*}{0.72} \frac{MV_o^2}{\sigma_r d^3}$ $\sigma_t(Pa) = 4.2 f_c(Pa) + 135 \times 10^6 + [0.014 f_c(Pa) + 0.45 \times 10^6] V_o$	

Table A1. Cont.

References	Equation in S.I. Unit		
Li and Chen Model [1,3,8,10]	$\frac{x}{d} = \sqrt{\frac{(1+(\frac{k\pi}{4N}))}{1+(\frac{I}{N})} \frac{4kI}{\pi}}$	$I = \frac{I^*}{S} = \frac{1}{S} \left(\frac{MV_0^2}{f_c d^3} \right)$	$\frac{x}{d} \leq 5$
	$\frac{x}{d} = \frac{2}{\pi} N \ln \left[\frac{1+(\frac{I}{N})}{1+(\frac{k\pi}{4N})} \right] + k$	$N = \frac{\lambda}{N^*} = \frac{1}{N^*} \left(\frac{M}{I_c d^3} \right)$ $S = 72 f_c^{-0.5}$	$\frac{x}{d} > 5$
	$\frac{x}{d} = 1.628 \left(\frac{(1+(\frac{k\pi}{4N}))}{1+(\frac{I}{N})} \frac{4kI}{\pi} \right)^{1.395}$	$k = \left(0.707 + \frac{h}{d} \right)$	$x/d < 0.5$
	$\frac{x}{d} = \sqrt{\frac{\frac{4kI}{\pi}}{1+(\frac{I}{N})}}$		for $\frac{x}{d} \leq k$
	$\frac{x}{d} = \frac{2}{\pi} N \ln \left(1 + \frac{I}{N} \right) + \frac{k}{2}$	If $N \gg 1$	for $\frac{x}{d} > k$
	$\frac{x}{d} = 1.628 \left(\frac{\frac{4kI}{\pi}}{1+(\frac{I}{N})} \right)^{1.395}$		$x/d < 0.5$
	$\frac{x}{d} = \sqrt{\frac{4kI}{\pi}}$		$\frac{x}{d} \leq k$
	$\frac{x}{d} = \frac{k}{2} + \frac{2I}{\pi}$	When $I/N \ll 1$	$\frac{x}{d} > k$
	$\frac{x}{d} = 1.628 \left(\frac{4k}{\pi} I \right)^{1.395}$		$x/d < 0.5$

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Article

Analytical Model Formulation of Steel Plate Reinforced Concrete Walls against Hard Projectile Impact

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Abstract: Steel plate reinforced concrete (SC) walls can effectively resist projectile impact by preventing the rear concrete fragments flying away, thus attracting much attention in defence technology. This work numerically and analytically investigated the hard projectile perforation of steel plate reinforced concrete walls. Impact resistance theories, including cavity expansion analysis as well as the petaling theory of thin steel plates were used to describe the cratering, tunneling and plugging phases of SC walls perforation. Numerical modeling of SC walls perforation was performed to estimate projectile residual velocity and target destructive form, which were validated against the test results. An analytical model for SC wall perforation was established to describe the penetration resistance featuring five stages, i.e., cratering, tunneling and plugging, petaling with plugging and solely petaling. Analytical model predictions matched numerical results well with respect to projectile deceleration evolution as well as residual velocity. From a structural absorbed energy perspective, the effect of front concrete panel and rear steel plate thickness combinations was also studied and analyzed. Finally, equivalent concrete slab thickness was derived with respect to the ballistic limit of SC walls, which may be helpful in the design of a protective strategy.

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Keywords: steel plate reinforced concrete walls; FE simulation; perforation analytical model; cavity expansion analysis; thin plates petaling

1. Introduction

Characterized with easy shaping, efficient fabrication and construction, concrete material structures are widely used for most civilian and military infrastructure, e.g., nuclear power plants, liquefied natural gas storage tanks and civil air defence, which are designed to withstand extreme loading, such as aircraft engine impact as well as internal and external missile impact [1–3]. Under projectile penetration, attaching a relative thin steel plate onto the concrete wall rear (protected) face would better protect the inner inhabitants and vulnerable instruments by preventing rear face ejected fragmentation, which might occur even if the concrete panel is not breached. In practice, steel plate reinforced concrete (SC) walls have been effectively used as primary and secondary shelters in protective structures. It is important to develop a simple but robust analytical model for projectile perforation on SC walls.

SC walls have superior performance in terms of resisting impact loading, since the rear steel plate induces a considerable effect on limiting crater development and preventing the pulverized pieces from flying away. Concerning the impact resistance of the SC walls, experimental studies have compared and analyzed projectile impact tests on concrete slabs with and without the rear steel plate. Remennikov et al. [4,5] investigated the static and impact performance of SC walls in which no shear connectors were utilized to connect the steel faceplates and the concrete core. Owing to the specially designed connection details, the tested panels exhibited tensile membrane resistance at large deformations.

Armour-piercing projectiles impacting high-strength concrete backed by armoured steel were studied by Feng et al. [6], showing spaced composite targets had larger residual penetration depth than segmented ones. Kojima [7] concluded that the rear attached steel plate has little effect on enhancing the impact resistance of SC walls, while it can efficiently restrain scabbing and spalling fragments. Aiming at identifying the influence of the steel plate on local damage of SC walls, Tsubota et al. [8] performed a series of impact tests in which the steel plate was placed on the rear, front and both faces of a concrete panel, respectively. The rear steel plate attached to reinforced concrete (RC) plate could prevent local damage caused by perforation, while the front steel plate attached to the impacted face had a relatively slight effect. Abdel-Kader and Fouda [9] showed that the steel plate placed on the rear face had better impact performance than the front one, which further validated the above conclusion. Hashimoto et al. [10] also showed that a thicker steel plate would result in less local damage under low-velocity impact. Mizuno et al. [11] found that SC panels have better impact resistant performance than RC panels, which enabled reduction of panel thickness by almost 30%.

Bruhl et al. [12] developed an analytical and empirical model of SC walls under rigid projectile impact and proposed a three-step method for designing SC walls against missile impact which can be used to evaluate the ballistic limit of SC walls. For rigid projectile perforation, Grisaro and Dancygier [13] investigated the thickness of the SC composite barrier with respect to energy absorption. Wu et al. [14] performed projectile perforation tests on monolithic and segmented RC panels with a rear steel plate. The ballistic performances of layered RC and SC targets were analyzed quantitatively, which further validated the equivalent approach in Ref. [13]. To assess the core concrete thickness and steel ratios effect on failure modes, Lee and Kim [15] numerically evaluated the impact resistance of SC and RC panels via a LS-DYNA solver, suggesting SC walls can better resist impacting loads than RC panels.

Although extensive studies on SC structures have been conducted both experimentally and numerically, analytical models of hard projectile perforation on SC walls need to be further explored. Cavity expansion analysis and petaling theory were combined to describe the cratering, tunneling and plugging phases of SC wall perforation. Front concrete wall perforation resistance was analyzed by perforation modeling via non-linear transient dynamic solver LS-DYNA. The thin plate petaling theory was utilized to analyze the process of rear face plate damage by projectile impact. With the same ballistic limit, a semiempirical analytical model, converting SC walls to equivalent thickness concrete panels, was developed and validated. This work may shed some light on SC wall ballistic performance related to protective structure design.

2. Impact Resistance Theories

Since the composite structure of SC walls consists of a front concrete plate attached to a rear steel plate, the ballistic performance of SC walls should be roughly separated into three parts: penetration of the front concrete, perforation of the steel plate and interaction with both concrete and steel. When a projectile perforates a concrete panel with a certain thickness it goes through three response stages: front crater scabbing, stable tunneling, and rear crater plugging (spalling) [16]. The impact perforation response of thin steel plates has been successfully analyzed by the petaling theory proposed by Wierzbicki [17].

This work developed a semiempirical analytical model of projectile penetration of steel plate reinforced concrete walls based on five penetration stages, as plotted in Figure 1. Stage 1 is projectile cratering on the front concrete impact face. Stage 2 represents stable tunneling inside the concrete block. Afterwards, rear face plugging occurs in Stage 3 until the projectile starts to hit the rear steel plate. Stage 4 occurs when the projectile interacts with the steel plate and fragments the concrete rear face. Finally, the projectile head has interaction only the steel plate in stage 5. Hence, the penetration resistance consists of concrete resistance and steel plate resistance. Classical resistance equations are introduced below.

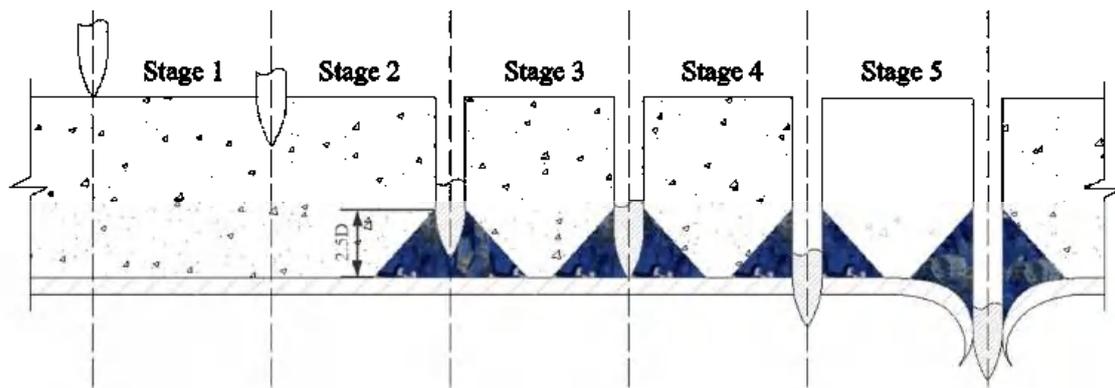


Figure 1. Diagram of a projectile penetrating a steel plate reinforced concrete wall.

2.1. Cavity Expansion Analysis for Concrete

Introduced by Bishop et al. [18], cavity expansion analysis was applied to solve the governing equations of the spherical cavity or cylindrical cavity expansion process in an elasto-plastic incompressible medium. Then Forrestal and Luk et al. [19] extended this to compressible material penetration problems. This classical model has been successfully applied to metal and concrete material penetration analyses.

Projectile penetration inside a solid medium can be regarded as a spherically symmetric cavity expansion process. Embedded in an infinite and isotropic medium, the zero initial radius cavity expands at a constant velocity. Due to the instantaneous rise of dynamic pressure, spherical stress waves are generated to form different response regions corresponding to the concrete constitutive law. According to continuum mechanics, equations of mass conservation and momentum conservation [8] in Euler coordinate for compressible spherical cavity expansion analysis at time t are:

$$\text{div } \sigma = \rho a \tag{1}$$

$$\frac{\partial \rho}{\partial t} + \text{div}(\rho v) = 0 \tag{2}$$

where σ is the stress tensor, a is acceleration, and v is velocity. In a spherical coordinate system, the governing equations [9] can be expressed as:

$$\frac{\partial \sigma_r}{\partial r} + \frac{2(\sigma_r - \sigma_\theta)}{r} = -\rho \left(\frac{\partial v}{\partial t} + v \frac{\partial v}{\partial r} \right) \tag{3}$$

$$\rho \left(\frac{\partial v}{\partial r} + \frac{2v}{r} \right) = - \left(\frac{\partial \rho}{\partial t} + v \frac{\partial \rho}{\partial r} \right) \tag{4}$$

where r is the radial coordinate, and σ_r and σ_θ are radial and circumferential Cauchy stress.

With the corresponding constitutive law for concrete materials, attempts have been made to numerically solve the governing equations with the Runge-Kutta method [20]. The normal stress σ_n , referred as penetration resistance, acting on the projectile nose is usually expressed as:

$$\sigma_n = \rho v^2 + R \tag{5}$$

where ρv^2 and R represent the inertial dynamic resistance and static resistance. Forrestal and his coworkers [21–23] deemed the target strength parameter as $R = S f'_c$ where S is a dimensionless constant and f'_c is the unconfined compressive strength of concrete. After validation with extensive penetration data [21,22,24–26], the semiempirical penetration model gives:

$$S = 82.6 \left(f'_c / 10^6 \right)^{-0.544} \tag{6}$$

Recently, researchers have pointed out that the above penetration resistance equation only applies to medium caliber projectile cases (shank diameter ranging from 12.9 mm to 30.5 mm) [27]. To avoid the concrete size effect in penetration, this work focuses on a 1-inch (25.4 mm) diameter projectile meeting the application range.

2.2. Perforation with Shear Plugging

After cratering and tunneling, the interaction between the projectile and shear plugging fragments can be treated as a collision problem. In Figure 2, the fragments with velocity V_{rf} are pushed away by the projectile, whereby the shear plugging zone is a frustum-of-cone with cone slope angle φ . The projectile residual velocity V_r can be derived from:

$$\frac{1}{2}mV_{r0}^2 = \frac{1}{2}mV_r^2 + \frac{1}{2}\rho_c\Omega V_{rf}^2 \tag{7}$$

where V_{r0} is the initial projectile velocity of the plugging stage, fragments velocity can be estimated by $V_{rf} = \eta V_r$ and $\eta = 0.2$ according to [14], the cone slope angle $\varphi = 62.5^\circ$ is given by Peng et al. [28], and Ω is the volume of the ejected frustum-of-cone fragment, which can be expressed as:

$$\Omega = \frac{\pi}{12}H_p(4 \tan^2 \varphi H_p^2 + 6d_p \tan \varphi H_p + 3d_p^2) \tag{8}$$

where H_p is the shear plugging thickness, and d_p is the projectile diameter.

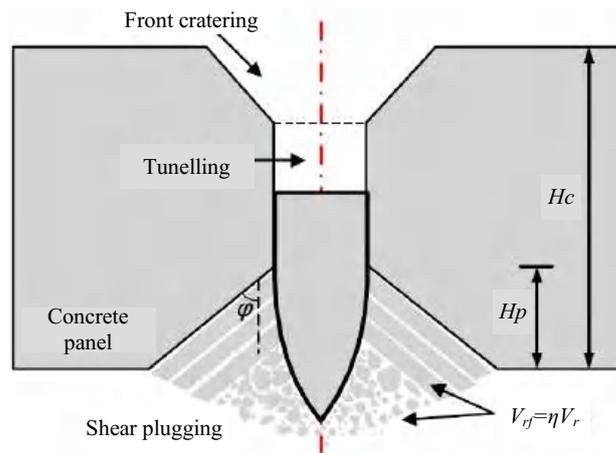


Figure 2. Concrete plate plugging due to perforation [16].

2.3. Thin Plate Petaling

Characterized by multiple symmetric petals forming, petaling damage is a common failure mode of thin metal plates when subjected to localized high intensity loadings, e.g., projectile perforation as shown in Figure 3. The shock wave pressure and the cavity tear the thin shell to generate a much larger radius. The tearing fracture energy is related to the bending energy through the petal local radial curvature and the circumferential curvature. From an energy perspective, the energy consumption of projectile perforation on the thin metal plate results from petaling energy and plastic deformation:

$$E_c = W_L + W_G \tag{9}$$

where W_L denotes the petaling energy, and W_G represents the plastic energy during plate deformation.

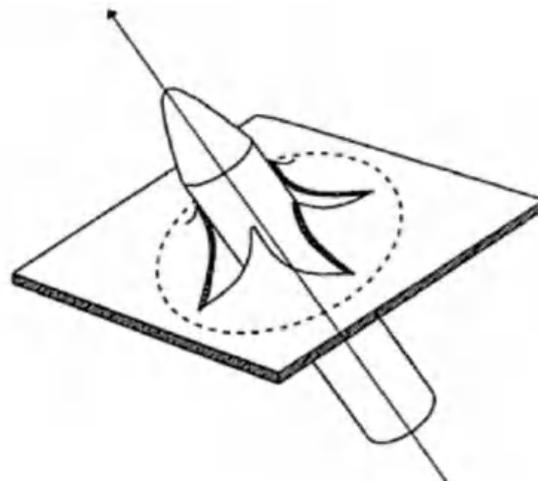


Figure 3. Petaling diagram [17].

According to Wierzbicki [17], the petaling energy due to projectile perforation can be expressed as:

$$W_L = 3.37\sigma_0 \bar{\delta}^{0.2} h_t^{1.6} d_p^{1.4} \tag{10}$$

where $\bar{\delta} = \delta_t/h_t$, $\delta_t = R_t/(\sqrt{3}\sigma_u)$, R_t denotes the impact toughness, δ_t is the crack tip opening displacement parameter (CTOD), h_t is the plate thickness, σ_u is the ultimate strength and σ_0 is the nominal strength.

Landkof and Goldsmith [29] measured the plastic deformation energy according to the ballistic tests, hence thin plate perforation energy consumption E_c can be written in the dimensionless form:

$$\frac{E_c}{\sigma_0 d_p^3} = 3.37\bar{\delta}^{0.2} \left(\frac{h_t}{d_p}\right)^{1.6} + 2.8\left(\frac{h_t}{d_p}\right)^{1.7} \tag{11}$$

3. Perforation Model Validation

Recently, Wu et al. [14] conducted penetration experiments to study the ballistic performance of reinforced concrete panels with a rear steel plate which provided valuable data for SC wall perforation analyses. For the sake of analytical model formulation, hypotheses needed to be verified via extensive test data both from experiments and simulations. This section aims to develop the FE numerical model for SC walls perforation which was validated against test data.

3.1. Perforation Test and FE Model

With an ogival-shaped nose, hard projectiles were used for penetration tests in which no apparent erosion occurs. The total projectile mass was 428 g, the shank diameter was 25.3 mm and the caliber-radius-head (CRH) of the ogival nose was 3.0, as shown in Figure 4.

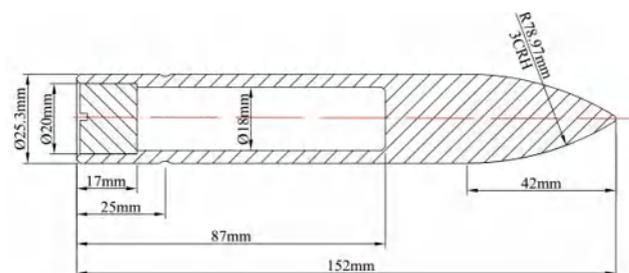


Figure 4. Projectile dimension [14].

Simulations of the SC walls perforation tests were conducted with 200-mm thick concrete panels. Figure 5 depicts the target dimension together with its reinforced mesh. The projectile impact location point is denoted by ‘×’. For the investigated concrete, the unconfined compressive strength of cylinder sample (f'_c) was 41 MPa.

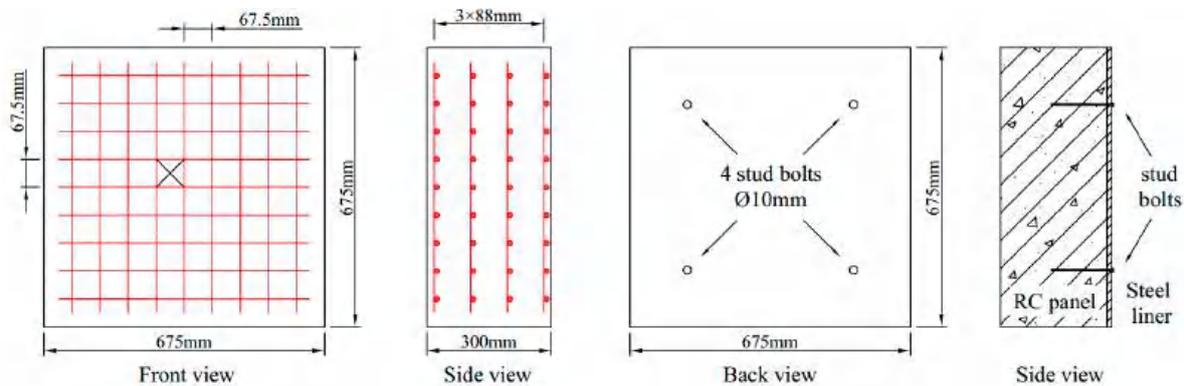


Figure 5. Geometric dimensioning of SC walls target.

For the numerical study, LS-DYNA, the extensively applied explicit solver, was adopted for impact simulations. Since the projectile material was of high strength and great hardness, the projectile was modeled as MAT_RIGID in the simulation. Successfully applied for concrete penetration simulations [30], the Holmquist-Johnson-Cook (HJC) model [31] was chosen to model the concrete material. Originally presented by Holmquist, Johnson and Cook [32], the HJC concrete model was developed for the purpose of impact computations where the material experiences large strains, high strain rates and high pressures. Coupled with isotropic damage, the HJC concrete model is an elastic-viscoplastic model [33] where the deviatoric response is determined by the following constitutive law:

$$\sigma_Y^* = [A(1 - D) + Bp^{*N}][1 + C \ln \epsilon^*] \tag{12}$$

in which $\sigma_Y^* = \sigma_Y / f'_c$ is the normalized equivalent stress, and $p^* = p / f'_c$ is the normalized pressure.

The equation of state in HJC is characterized by three stages: plate elastic (for $p < p_c$) $p = K\mu$; pore collapse (for $p_c \leq p \leq p_1$) $p = p_c + (p_1 - p_c)(\mu - \mu_1) / (\mu_1 - \mu_c)$; compaction (for $p > p_1$) $p = k_1\bar{\mu} + k_2\bar{\mu}^2 + k_3\bar{\mu}^3$ with $\bar{\mu} = (\mu - \mu_1) / (1 + \mu_1)$; k_1, k_2 , and k_3 are constants.

The HJC model includes a scalar damage formulation, where the damage evolution is accumulated from both the equivalent plastic strain increment $\Delta\epsilon_{eq}^P$ and the equivalent plastic volumetric strain increment $\Delta\mu_{eq}^P$. The damage evolution is expressed as:

$$\Delta D = \frac{\Delta\epsilon_{eq}^P + \Delta\mu_{eq}^P}{\epsilon_p^f + \mu_p^f} \tag{13}$$

where, $\epsilon_p^f + \mu_p^f = D_1(p^* + T^*)^{D_2}$, ϵ_p^f and μ_p^f are plastic strain and plastic volumetric strain corresponding to fracture, T^* is the normalized tensile strength, and D_1 and D_2 are damage constants.

The steel plate was modeled by the Johnson-Cook (JC) model [34] for its wide adoption in the metal impact engineering domain. JC is a strain rate and temperature-dependent (adiabatic assumption) visco-plastic material model [35,36]. This model is suitable for problems in which strain rates vary over a large range. The JC model expresses the flow stress with the form:

$$\sigma_Y = [A + B\epsilon_p^N][1 + C \ln \dot{\epsilon}^*] \tag{14}$$

where σ_Y is the effective stress, ϵ_p is the effective plastic strain, ϵ^* is the normalized effective plastic strain rate (typically normalized to a strain rate of 1.0 s^{-1}), N is the work hardening exponent, and A, B, C are constants determined by calibration.

With reference to [37], the steel rebar was modeled using MAT_PLASTIC_KINEMATIC. For the concrete HJC model and the steel 1006 JC model, the main parameters are listed in Tables 1 and 2 where those model parameters have been validated against available penetration tests [38,39]. In the test set up, all the top and bottom surfaces of the target were constrained. Element sizes were strictly controlled to guarantee that the steel mesh nodes coincide with concrete element nodes. Figure 6 shows the finite element model developed for SC wall penetration. Table 3 lists the element numbers of projectile, concrete panel, steel mesh as well as steel plate. The refined concrete mesh of the impact area was 3 mm, with 6 mm mesh for the outer region, which have been proven as converged meshes for penetration simulation [40]. The projectile was meshed with 1 to 3 mm size hexahedrons. The meshing sizes of the reinforced rebars were 3 mm and 6 mm. The rear steel plate was modeled with $1 \text{ mm} \times 1 \text{ mm} \times 1 \text{ mm}$ for the center area and $1 \text{ mm} \times 2 \text{ mm} \times 2 \text{ mm}$ for the outer region.

Table 1. Material parameters of concrete (units: cm-g- μs).

RO 2.24	G 0.1486	A 0.79	B 1.6	C 0.007	N 0.61	FC 4.1×10^{-4}	T 4.1×10^{-5}	EPSO 1×10^{-6}	EFMIN 0.01
SFMAX 7.0	PC 1.6×10^{-4}	UC 0.001	PL 0.008	UL 0.1	D₁ 0.04	D₂ 1.0	K₁ 0.85	K₂ −1.71	K₃ 2.08

Table 2. Material parameters of steel plate (units: cm-g- μs).

RO 7.896	G 0.818	A 3.5×10^{-3}	B 2.75×10^{-3}	N 0.36	C 0.022	M 1	TM 1793	TR 293	EPSO 1×10^{-6}
CP 0.452×10^{-5}	PC 0	SPALL 2	IT 0	D₁ −0.8	D₂ 2.1	D₃ −0.5	D₄ 0.0002	D₅ 0.61	C2/P 1

3.2. Numerical Results, Validation and Discussion

In this work, five penetration simulations with different striking velocities (V_s), were carried out to validate the numerical model. After simulation, numerical predictions of projectile residual velocities ($V_{r,n}$) were compared against test data ($V_{r,e}$) as shown in Figure 7a. As listed in Table 4, the numerical predictions agreed well with the test data. It is also suggested that with a 436 m/s striking velocity, the projectile perforated the SC wall with quite a small residual velocity which was overestimated by the numerical model. Figure 7b shows the projectile velocity history during perforation, implying that it takes less time to perforate the SC wall at higher striking velocity.

Figure 8 compares the post-test target and numerical results from different views. Figure 8a shows the constraints of the target, which was fixed in the steel frame. The actual destructive forms of the rear target are also shown. A three-stage perforation model [16,40] consisting of front impact crater, ballistic tunnel, and a nearly frustum-of-cone shaped rear crater is shown in Figure 8b. In Figure 8c, the steel plate deformation is shown and it was notable that the neighboring area of the plate around the projectile suffers severe deformation. From a cross-section view of the SC walls, Figure 8d illustrates the damage mode of the target in which the foregoing three-stage thick plate model was numerically verified. From the validation results in terms of damage mode and residual velocity, the numerical predictions match well with SC wall perforation tests.

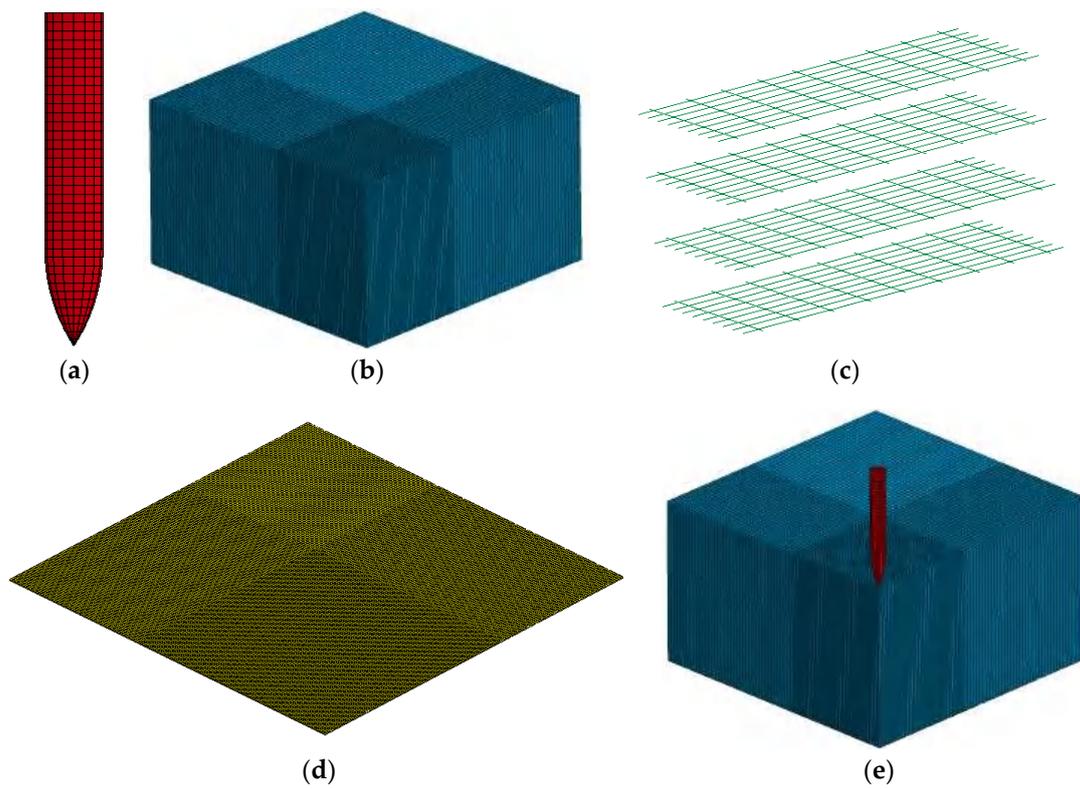


Figure 6. Finite element model. (a) Projectile; (b) concrete slab; (c) reinforcement mesh; (d) rear steel plate (e) projectile impact SC walls.

Table 3. Element numbers for each component of the SC wall perforation model.

Part	Projectile	Concrete Slab	Steel Mesh	Steel Plate
Number of elements	800	1,822,500	8712	18,496

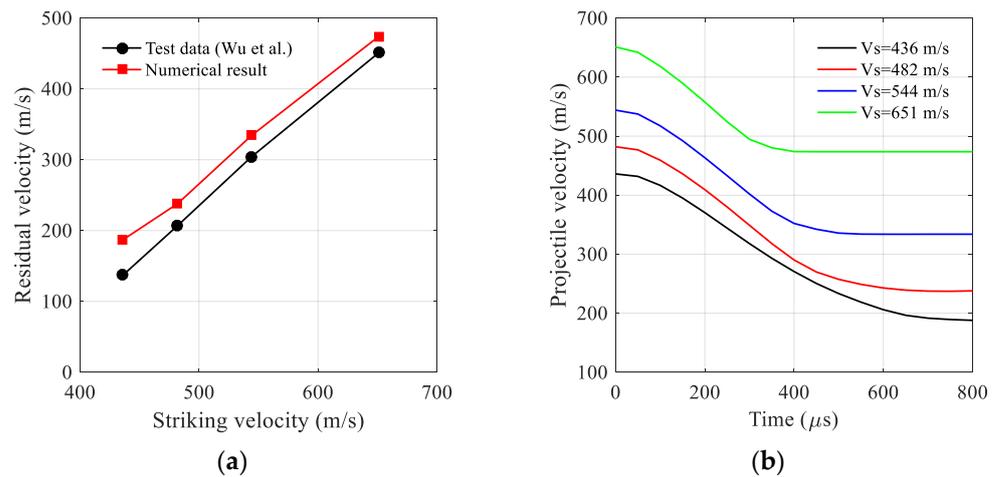


Figure 7. Numerical results of projectile perforation on SC walls. (a) Residual velocity comparison; (b) projectile velocity evolutions.

Table 4. Comparison between experiment and simulation.

V_s (m/s)	$V_{r,e}$ (m/s)	$V_{r,n}$ (m/s)
436	137	187
482	207	238
544	304	334
651	451	474

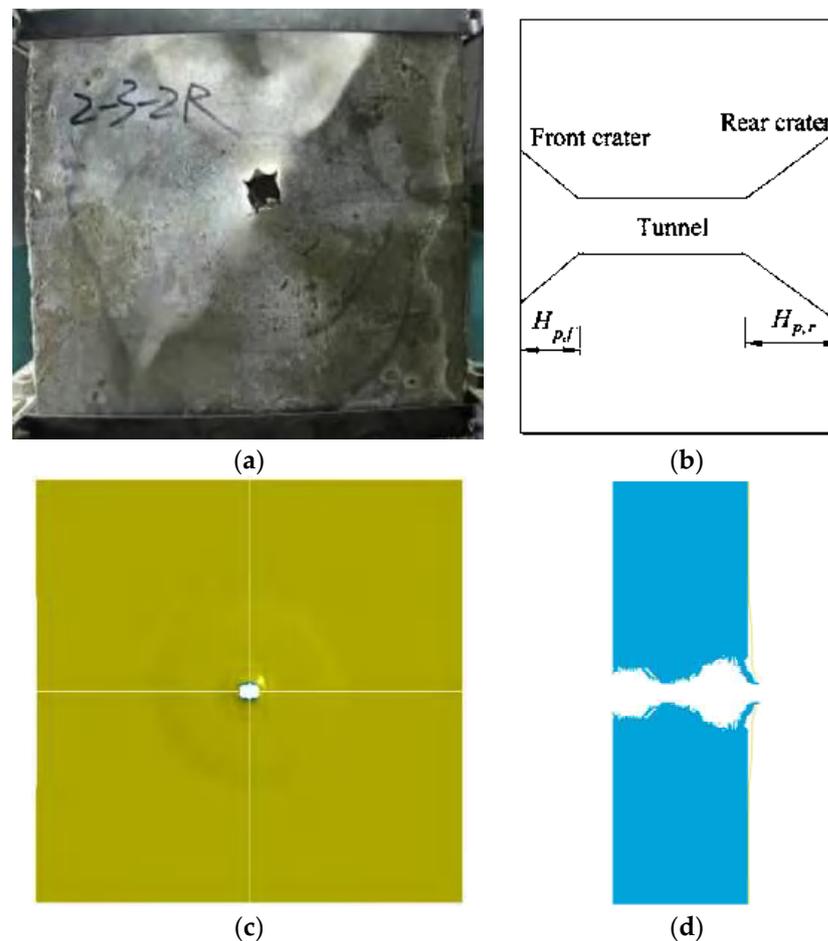


Figure 8. Damaged contour of SC walls. (a) Rear view of SC walls; (b) section view of concrete panel; (c) numerical results of damaged steel plate; (d) numerical results of damaged concrete panel.

4. Numerical Study of SC Walls Perforation

This section describes the extensive numerical investigations on SC wall perforations with different concrete panels and steel plates. SC walls with different front concrete panel and rear steel plate thickness combinations are investigated in this section considering protective structure analysis.

4.1. Model Setting

The previously described ogival nose projectile was used as the penetrator. With an 800 mm length and width, the investigated concrete panels were attached with 3 mm to 11 mm thicknesses of the rear steel plate. With the same cross section, the concrete panel thicknesses were selected as 150 mm, 200 mm and 250 mm, respectively. For SC walls with 200-mm thick concrete panel, the striking velocities of the projectile were set at 550, 600, 650, 700 and 800 m/s. Furthermore, a striking velocity of 600 m/s was set for perforation simulations with 150 mm and 250 mm thick concrete panels.

For the sake of computational cost, the SC walls perforation models were developed as quarterly symmetric bodies with symmetric boundaries. Figure 9 shows the FE model in which the grids near the impact region were refined. For the 1/4 model of SC walls with a combination of 11 mm thickness steel plate and 250-mm thick concrete panel, the element numbers for the three parts are given in Table 5. These material models and their parameters are the same as in the validation model. An eroding algorithm was applied to all the interactions between the contact components.

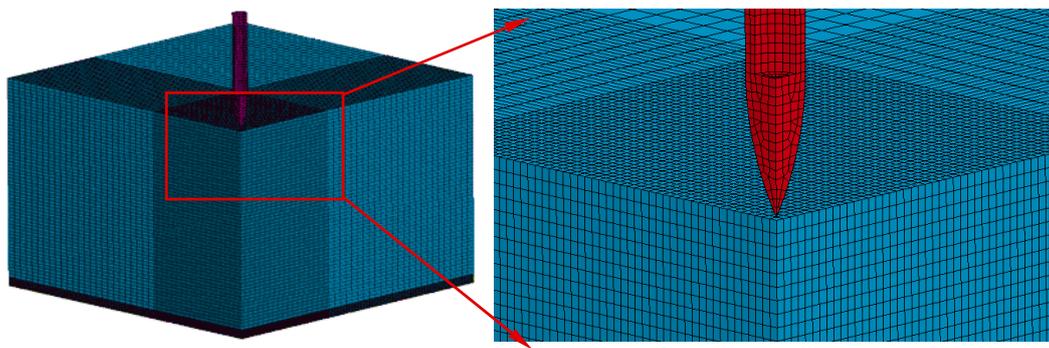


Figure 9. Overview of the meshed FE model.

Table 5. Element numbers for each part.

Part	Projectile	Concrete	Rear Steel Plate
Number of elements	696	376,875	50,000

4.2. Results Discussion of SC Walls Perforation

Seven kinds of thickness combinations of concrete panel and steel plate were used for the SC wall perforation analysis. Figure 10 shows the numerical results for an SC wall with a 250 mm thick concrete panel and an 11 mm thick steel plate. The projectile striking velocity was 600 m/s and the SC wall was perforated as expected. The destructive area of concrete rear surface was larger than its front impact surface due to conical plugging occurring in the back area. In Figure 10b, the von-Mises stress distribution contour exhibits a circular character.

To examine the penetration responses of SC walls under various striking velocities, a 200-mm thick concrete panel supported by rear steel plates with different thickness was numerically studied. Backing steel plates with different thickness were assumed to have various effects on the penetration resistance. Figure 11a illustrates that projectile residual velocity decreased with increasing thickness of the rear steel plate when the striking velocity was close to the ballistic limit [4]. Figure 11b shows the residual velocities for 150, 200 and 250 mm thick concrete panels subjected to a 600 m/s striking velocity impact. With the residual velocity increasing, the curve of velocity over time shows a slightly oscillating character. Under 600 m/s striking velocity impact, the projectile velocity history during SC wall perforation with a 200-mm thick concrete panel is shown in Figure 11c. The early penetration responses were almost the same, due to the fact that the backing steel plate had no influence on impact resistance of the front concrete target. Concerning the later perforation process, results imply that the rear steel plate has a significant effect on penetration resistance. Figure 11d shows the striking velocities and the residual velocities after perforation of SC walls with 200 mm thick concrete panels. With increasing striking velocity, it seems that the rear steel plate had a less pronounced effect on penetration resistance and thus the residual velocities tend to converge.

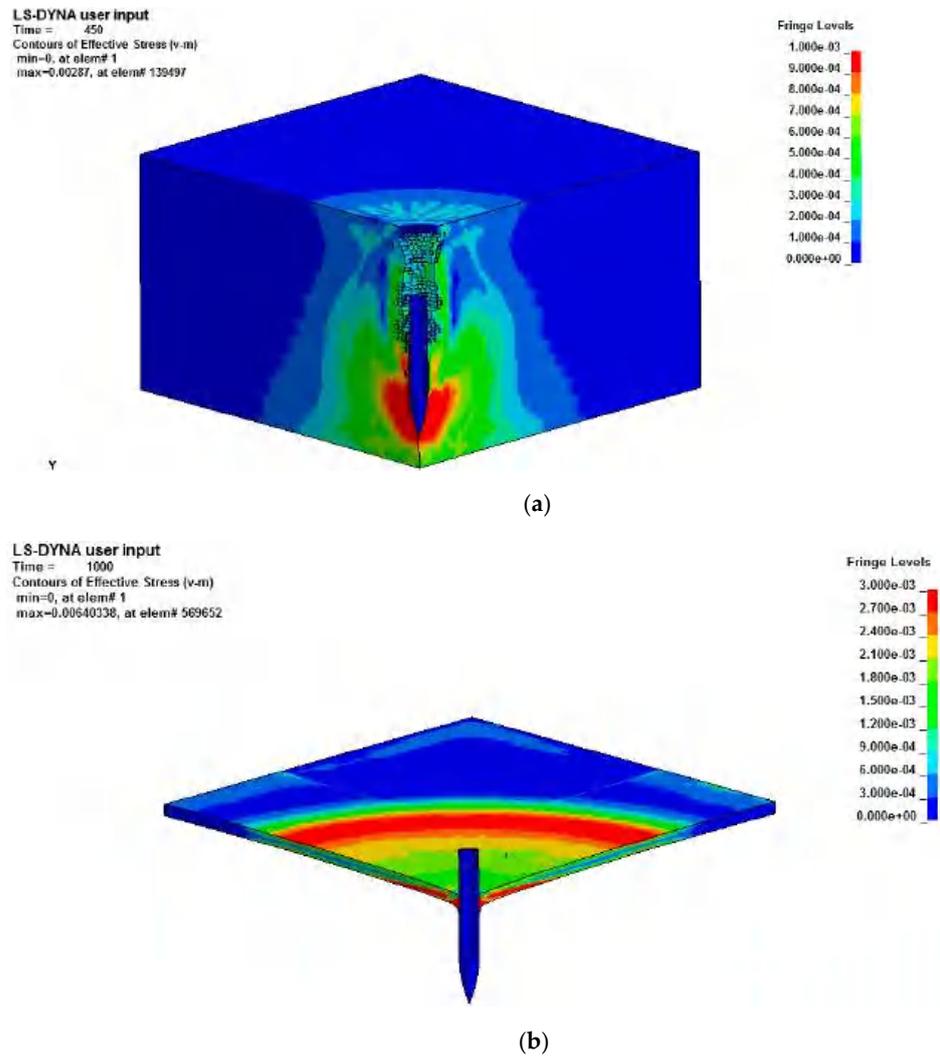


Figure 10. Impact response of an SC wall. (a) von-Mises stress distribution in front concrete panel; (b) von-Mises stress distribution in rear steel plate.

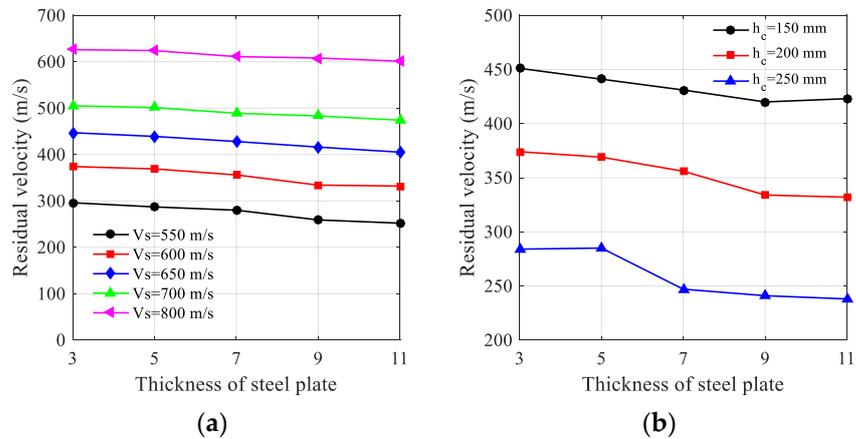


Figure 11. Cont.

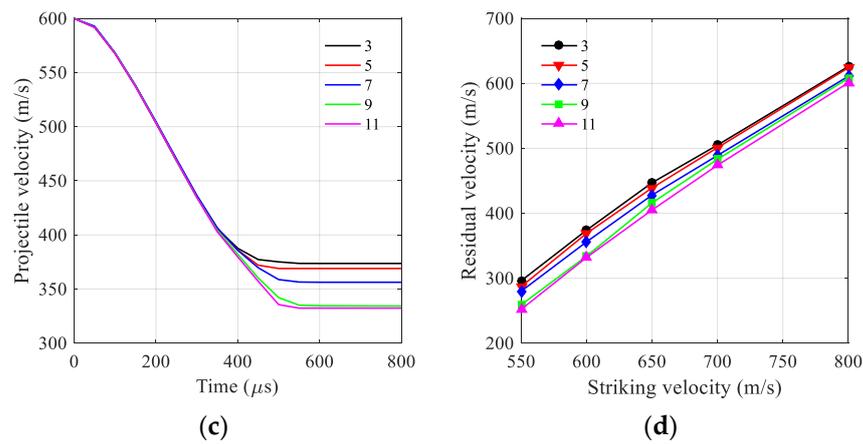


Figure 11. Numerical simulation results. (a) Effects of steel plate; (b) effects of concrete panel; (c) projectile velocity history; (d) residual velocity corresponding to striking velocity.

4.3. Free Surface Boundary Effect

Although cavity expansion analysis was successfully applied to projectile deep penetration in concrete, the typical penetration resistance equation proposed by Forrestal et al. [23] was not suitable for a projectile perforation scenario with a concrete panel of limited thickness. The front and back free surfaces might degrade the material strength thus reducing the penetration resistance during cratering and shear plugging, as shown in Figure 12. Therefore, the penetration resistance prior to shear plugging should be revised in the case of a concrete panel with limited thickness.

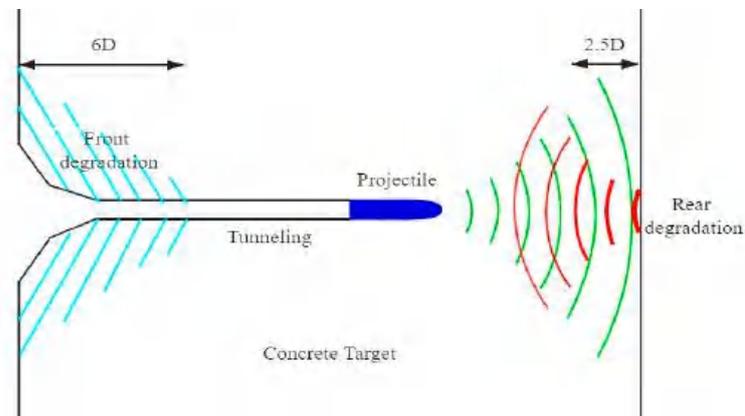


Figure 12. Front and rear free boundary effect on projectile penetration responses.

To develop a penetration resistance equation with respect to penetration velocity, numerical simulations with constant projectile velocity were performed to derive the penetration resistance acting on the projectile nose. Since the front and rear free surface might negatively affect the penetration resistance, concrete panels with thicknesses of 150, 200, 250, 300, 400, 500, 600 and 700 mm were selected for simulation. According to the results plotted in Figure 13, it is interesting that penetration resistant force increased to a plateau. This can be explained by the fact that the front surface degraded the penetration resistance until reaching about $6d_p$. The rear free surface effect was estimated by penetration simulation of panels with 250 mm and 300 mm thickness. Both had a stable plateau during tunneling and started to drop at a position about 68 mm to 70 mm away from the rear surface. The shear plugging height was about $2.5d_p$ which matches well with experimental data in Ref. [34]. For 150 mm and 200 mm thickness concrete panels, both the front and rear free surface affected the penetration resistance, implying no stable tunneling.

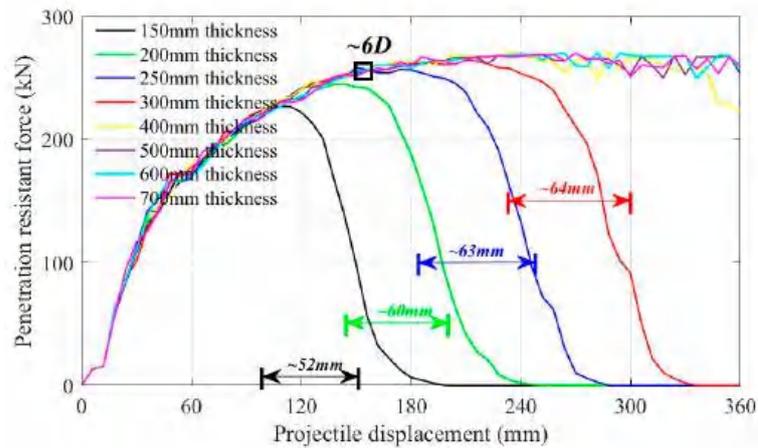


Figure 13. Penetration resistance for panels of different thickness.

5. Theoretical Analyses of Hard Projectile Perforation on SC Walls

According to the foregoing spherical cavity expansion theory, as well as thin steel plate petaling destruction, a semiempirical analytical model was proposed derived from numerical results. The SC wall composite target was composed of concrete and steel; therefore, the penetration resistance acting on the projectile nose was attributed to the concrete and steel plates. The concrete resistance at different penetration stages is related to cavity expansion analysis. The thin steel plate destruction mode due to perforation is generally petaling, hence the penetration resistance may be derived with a petaling model.

5.1. Penetration Resistance Force

The typical ogival-shaped nose projectile is illustrated in Figure 14, where the projectile nose length and shank diameter are denoted as h and d_p . Assuming a projectile with a striking velocity V , the normal expansion velocity perpendicular to the nose curve is $v = V \sin \theta$. Take the micro segment length dx to study the stress distribution over the infinite projectile nose surface ds . The resistant force df can be treated as the normal stress σ_n projection along the projectile axial direction:

$$df = \sigma_n \sin \theta ds \tag{15}$$

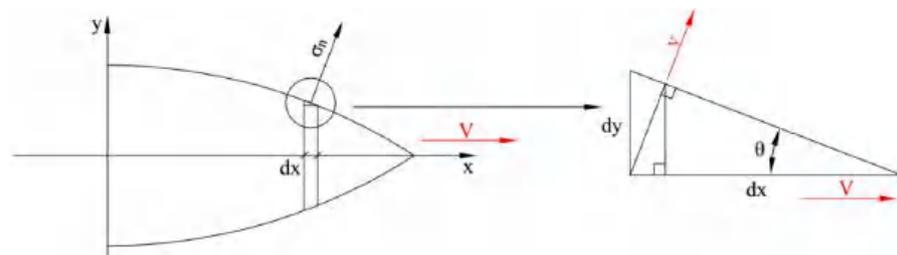


Figure 14. Penetration resistance on the projectile nose.

Integrating the normal stress σ_n over the projectile nose to achieve the axial penetration resistant force:

$$F = \int \sigma_n \sin \theta ds \tag{16}$$

$$ds = 2\pi |y| \sqrt{1 + |k|} dx \tag{17}$$

For thick concrete targets, the Forrestal model [41] derived from the empirical formula $S = 82.6(f'_c/10^6)^{-0.544}$, was applied to describe the static penetration resistance. Hence,

the normal resistant stress can be expressed as: $\sigma_n = Sf'_c + \rho v^2$, where f'_c is the unconfined cylinder compressive strength of concrete and ρ is the density of concrete target.

When the ratio of the metal plate thickness h_t and the shank diameter d_p is no larger than 0.5, petaling is generally the damage form of the thin metal target subjected to vertical impact. During the rear steel plate petaling process, the perforation energy E_c in Equation (11) can be regarded as mean resistant force accumulation over the deformation. Therefore, the mean resistant force can be estimated as $F_{mean} = E_c/d$, whereas d is the actual petaling displacement.

5.2. Stages of SC Walls Perforation

Based on the SC wall penetration resistance mechanism, a five-stage semiempirical analytical model was developed, i.e., projectile nose part penetration in concrete (cratering), stable penetration (tunneling), shear plugging, plate petaling with concrete plugging and plate petaling only, as depicted in Figure 1.

For stage 1 and 2, the projectile penetrates the front concrete panel with different contact areas. Hence, a deep penetration model [32,33] is utilized to analyze the first and second perforation stages. For penetration in the concrete panel, the normal stress $\sigma_{n,c}$ acting on the projectile nose can be expressed as $\sigma_{n,c} = Sf'_c + \rho v^2$ where $f'_c = 41$ MPa, $\rho = 2240$ kg/m³. The static resistance term in the normal direction $Sf'_c = 82.6 \times (f'_c/10^6)^{-0.544} \times f'_c = 449.17$ MPa and the dynamic term ρv^2 change with actual projectile velocity.

During stage 3, the projectile passes through the concrete fragments until it hits the rear steel plate. For thicker rear plates, the pulverized concrete pieces in the rear crater have more support and provide more penetration resistance to the projectile. Related to its thickness, a certain deformation happens to the rear steel plate. According to previous literature [14,28,42,43], it is assumed that the rear crater depth of the pulverized concrete near rear surface follows the relationship $H_{p,r} = 2.5d_p$. With reference to [44,45], the penetration model with spherical cavity expansion analysis can be modified to describe the penetration response in pulverized concrete. Due to the decreasing static resistance (fragile material), the normal penetration resistant stress $\sigma'_{n,c}$ can be expressed as:

$$\sigma'_{n,c} = (0.1 + 0.3 \times \frac{h_t}{d_p})Sf'_c + \rho v^2 \quad (18)$$

For stages 4 and 5, the projectile penetrates both the pulverized concrete as well as the rear steel plate. Considerable deformation occurs to the rear steel plate with destructive form. The process of a projectile perforating a thin steel plate is complex; therefore, the energy method is used to analyze rear steel plate perforation. The mean penetration resistant force is noted as F_{mean} , and the length is assumed to be $d = h_t + 1.5h$ according to the simulation results. For steel 1006 material, $\sigma_y = 350$ MPa, $\sigma_u = 500$ MPa, $p = 0.36$ and $R = 35$ J/cm².

As the projectile perforates the SC wall with a 200-mm thick concrete panel and a 3-mm thick steel plate, the resistant force acting on the projectile nose with a 700 m/s striking velocity is given in Figure 15, where five different stages of SC walls perforation are depicted in different colors for better visualization.

5.3. Analytical Model Validation against FE Simulation

SC walls with a 200-mm thick concrete panel and a 3-mm thick steel plate were studied to validate the analytical model with respect to numerical results in which the projectile striking velocity was 700 m/s.

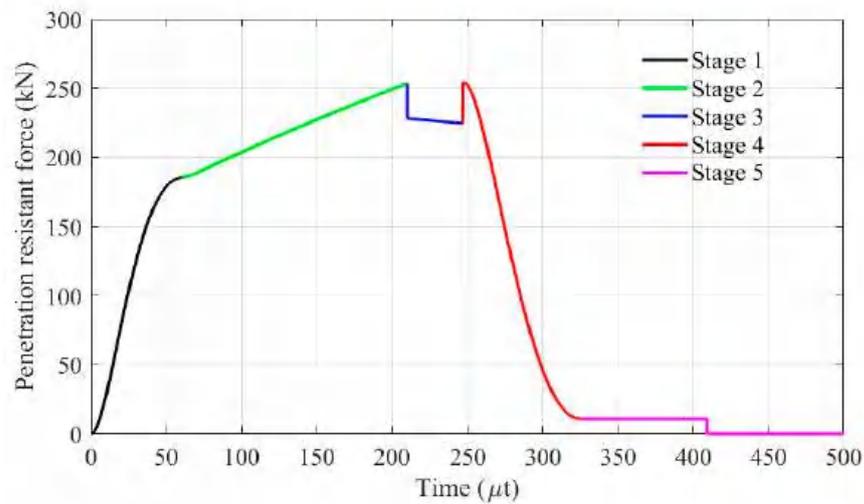


Figure 15. Penetration resistant force for five stages.

The projectile movement process, projectile velocity history, as well as projectile deceleration evolution are plotted in Figure 16. The analytical model has good consistency and regularity with simulation results concerning projectile residual velocity and deceleration history.

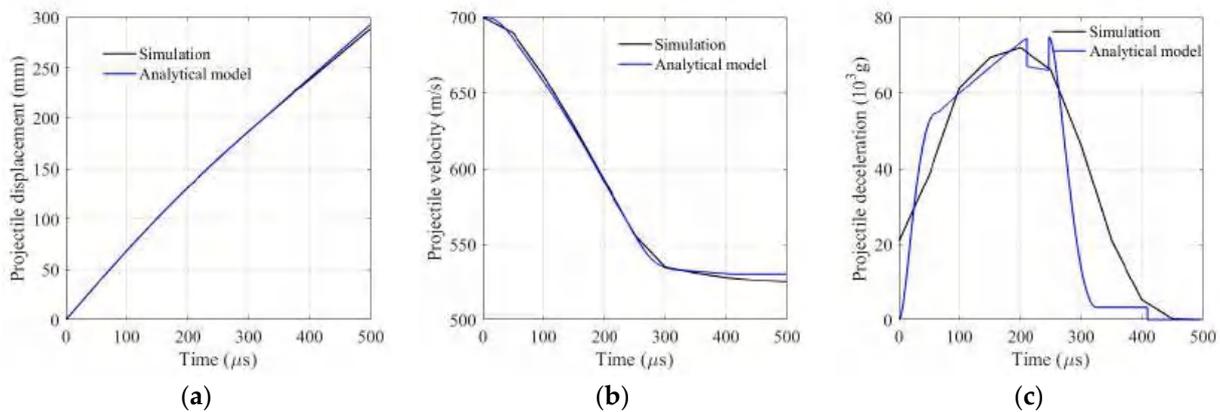


Figure 16. Comparison of analytical model and simulation. (a) Projectile displacement history; (b) projectile velocities history; (c) projectile deceleration history.

It is important to validate the analytical model with different boundary conditions, e.g., various thickness combinations of concrete panel and steel plate, and different projectile striking velocities. SC wall perforation of SC walls of 150, 200 and 300 thickness concrete and 3, 5, 7, 9 and 11 mm thicknesses of steel plate were studied both analytically and numerically. Striking velocities ranging from 650 m/s to 800 m/s were calculated and compared with simulation data. A good match is shown in Figure 17, suggesting the analytical model is validated with FE simulation and thus can be applied to subsequent discussion.

5.4. Rear Steel Plate Effect

For the SC walls with 200 mm thick concrete panels, the penetration responses, i.e., energy consumption and rear steel plate contribution, were explored analytically by taking into account steel plates with different thickness. In Figure 18a, all five curves have the tendency to converge at increasing striking velocity. Under different striking velocities, the absorbed energy of SC walls with a 200-mm thick concrete panel and different thickness steel plates is shown in Figure 18b, where SC wall consumption energy increases with increasing striking velocity. For the SC walls with a 200 mm thick concrete panel, energy

consumed by the rear steel plates is depicted in Figure 18c. The absorbed energy increases with increasing rear steel plate thickness. Under penetration impact of different projectile striking velocities, their energy absorption shows a tendency of linear increase.

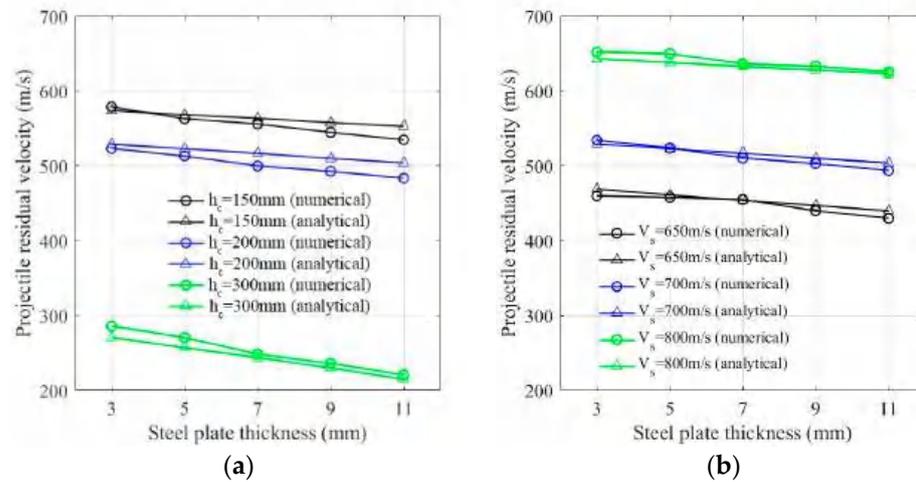


Figure 17. Model validation in terms of residual velocity. (a) Different thickness combinations; (b) different projectile impact velocity.

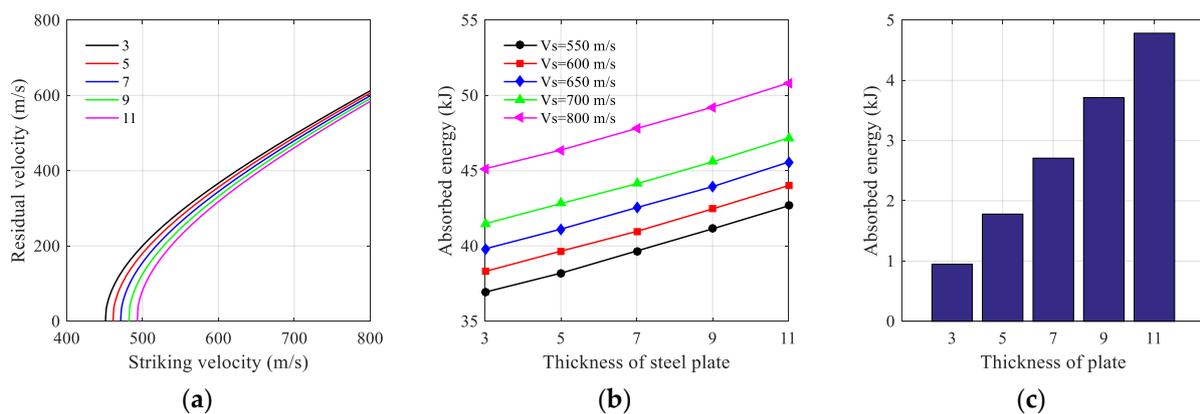


Figure 18. Results of analytical model for perforation on SC walls with $h_c = 200$ mm. (a) Residual velocity; (b) absorbed energy by SC walls; (c) absorbed energy by steel plate.

The effect of concrete panel thickness on energy consumption during perforation was studied. Figure 19a shows the absorbed energy of SC walls with different concrete panels. Figure 19b shows the equivalent concrete panel thickness determined for various SC wall perforation with the same ballistic limit.

$$h_{eq} = h_c + 3.06h_t \tag{19}$$

To evaluate the penetration resistance of SC walls with front concrete panels and rear steel plates, the concept of equivalent thickness of a concrete panel was considered. It was assumed that the equivalent concrete panel thickness had the same ballistic limit as the corresponding SC walls. After data fitting, Equation (19) was derived for the description of equivalent concrete panel thickness in terms of h_c and h_t of SC walls. This may help researchers to estimate the ballistic limit of SC walls.

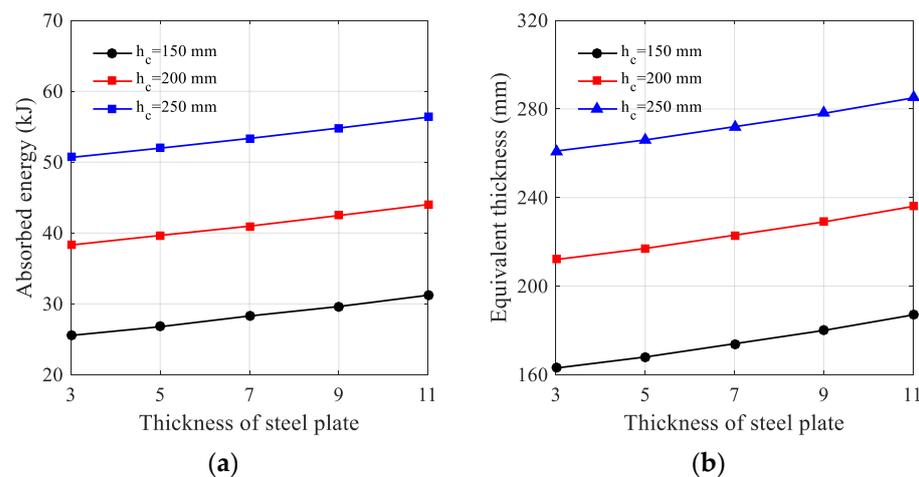


Figure 19. Perforation behavior of SC walls with different thicknesses of concrete panels. (a) Absorbed energy for SC walls; (b) equivalent concrete thickness.

6. Conclusions

The penetration resistance of SC walls with various thickness combinations of concrete panels and steel plates was explored both numerically and analytically. Through validation against experimental data, the numerical model and semi-empirical analytical model were further investigated to derive the following conclusions. (1) The attached rear steel plate has a positive influence on impact resistance by preventing the pulverized concrete pieces from flying away. (2) With increasing striking velocity, the rear steel plate has a less pronounced effect on the penetration resistance and thus the residual velocities tend to converge (3) Combining spherical cavity expansion analysis and the thin plate petaling theory, an analytical model with five stages of SC wall perforation was proposed and validated against a numerical simulation. (4) With the same ballistic limit, the equivalent concrete panel thickness can be derived with respect to the SC walls as $h_{eq} = h_c + 3.06h_t$.

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Article

Dynamic Response of Reinforced Recycled Aggregate Concrete Pavement under Impact Loading

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Abstract: Airport runway pavements often undergo the direct impact of aircraft landings. For the purposes of designing the structure, it is of great importance to know about the dynamic response of the pavement and its behavior under impact loading. However, the dynamics and failure mechanisms of reinforced recycled aggregate concrete pavements subjected to impact loading are seldom explored in the literature. For this purpose, four reinforced recycled aggregate concrete pavements with different thickness and ratios of reinforcement, and one reinforced normal concrete pavement, were manufactured and tested under impact loading using the drop-weight impact frame system. The impact force characteristics, crack patterns, deformation responses, and strain developments of reinforced concrete pavements subjected to impact loading were evaluated and compared. The above-mentioned study revealed that with an increase in the reinforcement ratio, both the deformation and the steel strain were reduced. Increasing the thickness would reduce the degree of damage and the impact force of reinforced concrete pavement (RCP) but increase the deformation. The results show that under the same compressive strength, the dynamic performance of the reinforced recycled aggregate concrete pavement was worse than that of the reinforced normal concrete pavement because of its lower elastic modulus and weaker interfacial transition zone. The dynamic performance of reinforced recycled aggregate concrete pavement could be improved by increasing the thickness and reinforcement ratio. The use of recycled aggregate concrete (RAC) in RCP is a technically feasible application of the material within the scope of this experimental study.

Keywords: RCP; recycled aggregate concrete; impact loading; impact force; cracking; peak displacement

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1. Introduction

With the development of national defense construction and civil aviation and the use of a large number of high-speed heavy aircraft, the safety and reliability of airport runway structures are highly sought after. The airport runway system does not only bear the direct impact caused by aircraft landing, but also may encounter large impact loading due to the hard landings of aircraft crashes [1]. Currently, the design specifications of airport pavement structure take the structure under static load as the research object, while impact effects due to hard landings have not been taken into consideration in the design of airport runway pavements [2,3]. It should be emphasized that airport pavements are constructed to provide adequate support for the loads imposed by airplanes, and produce a firm, stable, and smooth surface, and it should be strictly required that there will be no debris or other particles caused by landing, or they could be sucked into the engine and cause serious

engineering accidents. In order to satisfactorily meet these requirements, the pavement must be of sufficiently good quality to ensure not failing under the applied load.

Reinforced concrete pavement (RCP) refers to a pavement with embedded steel reinforcing bars in the concrete for crack control. The bars keep the cracks tightly closed, thus allowing longer joint spacing, resulting in an intact and smooth surface that ensures structural integrity and improves the performance of the pavement [4,5]. Reinforced concrete pavement is widely used in airport runway landing areas due to its good performance and low maintenance needs.

The performance of RCP depends on critical stresses and deflections imposed by repeated traffic and environmental loading, and fatigue fracture caused by these stresses is considered to be the limit state in the design of pavement structure. Therefore, in the current design code provisions on pavements, the main task is to determine the thickness of each component of the pavement structure to ensure that it can provide a satisfactory structural life at design fatigue limits. At present, the widely available analysis method of wheel load stress is based on Hertz's elastic thin plate theory [6], and Westergaard [7] has further proposed the solution of Winkler foundation under different load conditions. With the emergence of finite element software, the mechanical stresses are evaluated by a three-dimensional analysis and thermal stresses by two-dimensional analysis [8–11]. However, neither the Federal Aeronautics Administration (FAA) nor the Civil Aviation Administration of China (CAAC) considers impact effects due to hard landings of heavy aircraft in the design of airport runway pavements. The design code recommends that the dynamic effect be taken into account by multiplying the dynamic amplification factor.

A large amount of abandoned concrete is produced during the reconstruction and extension of airport runways, which has a substantial effect on the environment. Sustainability concerns are at the forefront of our society; unfortunately, the abandoned concrete is a non-renewable resource. The use of recycled waste concretes in construction application and pavement construction is one way to promote sustainable development. Consequently, many researchers have investigated the use of recycled concrete aggregate (RCA) in the production of new concrete, which is named recycled aggregate concrete (RAC). Most findings have indicated that the compressive strength, splitting tensile strength, flexural strength, and modulus of elasticity for RAC decrease with an increase in the content of RCA [12–15]. Furthermore, some pieces in the literature have studied RAC structural elements, such as columns, beams, slabs, and pavements; the results show that the incorporation of RAC has negative effects on the performance of these elements [16–21]. In general, the desired reduction caused by using RAC is limited, and further engineering research is encouraged, since satisfactory performance can still be achieved.

Dynamic response of reinforced concrete structural elements under impact loading has been investigated through experiments by many researchers [22–29]. Zineddin et al. [23,24] investigated the effects of different types of slab reinforcements and impact energy on the dynamic response and behavior of reinforced concrete slabs. The addition of steel reinforcement provided substantial strength enhancement to the slab, promoted crack formation on the top surface, and increased the stresses and strains that the concrete and steel materials could safely undergo, especially under higher impact energy. Othman et al. [25] conducted an experiment to investigate the effect of steel reinforcement distribution on the dynamic response of high strength concrete (HSC) plates, taking into account the effects of the main bottom steel reinforcement ratio (1.0, 2.0, and 3.0%) and the steel reinforcement arrangement (single or doubly reinforced plates). The results showed that the change of reinforcement ratio and/or reinforcement arrangement has no significant effect on impulse and absorbed energy values for same impact loading condition, while the impact duration decreased with the increase in reinforcement ratio. The reinforcement arrangement could affect the crack pattern; the HSC plates with single reinforcement typically failed by localized sudden punching, and the HSC plates doubly reinforced typically failed in a ductile punching mode.

Xiao et al. [26] studied the effects of loading rates on the performance of reinforced concrete (RC) slabs. From test results, the damage process, failure mode, strain rate, and energy absorption capacity of RC slabs were similar between the high-loading-rate test and the low-velocity impact test. Therefore, it was suggested that the high load rate test results could be used to analyze the performance of the RC slabs under low-velocity impacts. Both longitudinal and transverse reinforcements were effective in enhancing the maximum strength of specimens. However, the damage to the slab under both high-rate and impact loadings can be more efficiently reduced by adding shear stirrups. In another experimental program [27], five 1200 mm square RC slabs were tested with different nose shapes, diameters of impacted area, drop weights and drop heights in another experimental program; the punching shear failure mode was observed for all the specimens that failed during the test. The damage to the slabs increased with the increase in impact energy, and more impact energy was required to fail RC slabs when the diameter of the impacted area increased.

In order to better understand the effects of supporting conditions on the behavior of RC slabs subjected to impact load, some research has been undertaken. Özgür et al. [28] found that the number of drops until failure was lower for the specimens with four hinge supports than those for the specimens with four fixed supports, but higher than those for the specimens with two opposite hinge supports. The authors also reported that the acceleration, velocity, and displacement decreased due to an increase in the support stiffness. Chiaia et al. [29] studied two-way reinforced concrete slabs over different kinds of yielding supports and concluded that reducing the support rigidity could decrease the displacement and stress of the whole structure. Husem et al. [30] found that the energy-absorbing capacity was decreased by an increase of span size in both fixed and free supported RC slabs, and that the maximum midspan displacement values increased only in free supported RC slabs; however the span size has no considerable effect in fixed supported RC slabs.

Furthermore, some studies were made to improve the impact behavior of RC slabs by blending in other materials, such as steel fibers, carbon fabric, polypropylene fiber, etc. An experimental program by Hrynyk et al. [31] revealed that the increased addition of the steel fibers was effective in increasing slab capacity, reducing crack widths and spacings, and mitigating local damage under impact. The research by Beckmann et al. [32] investigated blending steel fibers; carbon fabric showed substantial advantages in the resistance to the impact load and to the penetration of the impactor but had only a minor influence on concrete strain. AlRousan et al. [33] studied the impact resistance of RC slabs blending with polypropylene fiber; the result showed that the proper quantity of polypropylene fiber could significantly improve the impact resistance of RC slabs, and that a suitable content of polypropylene fiber was 0.90%. Ong et al. [34] studied the impact resistance of concrete slabs blending with four substances (polyolefin, polyvinyl, alcohol, and steel); the result showed that hooked-end steel fiber concrete slabs had the best cracking and energy absorption characteristics compared to other slabs.

However, there are a limited number of studies that comprehensively explore the impact dynamics and failure mechanisms of reinforced recycled aggregate concrete pavement. Wu [35] studied the damage and failure model of rigid concrete pavement under drop weight impact, and obtained the extent of damage, failure mode, deformation, and acceleration. The test results show that the concrete pavement without steel reinforcement was fragmented into three segments and showed brittle failure mode. Cai et al. [36] analyzed the dynamic deflection and velocity response of airport concrete pavement under impact loading and drew the conclusion that the velocity response amplitude decreased with the increase of slab thickness, and that the deflection at the center of the slab decreased with the decrease of pavement slab size.

Most of the studies conducted under impact loading were focused on natural aggregate concrete (NAC) structural elements. Recently, a few studies have been performed on the dynamic response of RAC structural elements under drop weight impact loading [37–40]. Vali et al. [38] studied the behavior of RAC slabs under impact loading with different

replacement ratios of RCA and found that the stiffness of RAC slabs decreased with an increasing replacement ratio of RCA, which led to decreases in the punching shear strength at first crack stage and in the ultimate punching shear strength.

In order to design the airport pavement scientifically and to ensure its service life and safe operation, it is necessary to conduct experimental research on the performance of reinforced recycled aggregate concrete pavement under likely impacts. In fact, the excessive pavement damage due to the impact load of hard landings is extremely difficult to measure. Therefore, in order to better understand the behavior of reinforced recycled aggregate concrete pavement subjected to impact loading, an experimental program has been designed and conducted in this paper. The structural dynamic responses are measured during the drop-weight impact tests, and together with the cracking mechanisms, can provide a basis for investigating the impact behavior of reinforced recycled aggregate concrete pavement with different thickness and reinforcement ratios. In the meantime, the drop-weight impact test is also done on the reinforced normal concrete pavement, and the impact force characteristics, crack patterns, deformation responses, and strain developments are compared with those of reinforced recycled aggregate concrete pavement.

This paper presents the details of a well-organized and well-equipped experimental investigation with two main research objectives:

- (1) To investigate the effect of steel reinforcement distribution and slab thickness on the impact force characteristics and impact behaviors of RCP;
- (2) To evaluate the applicability of reinforced recycled aggregate concrete pavement under impact loads compared to the reinforced natural aggregate concrete pavement.

2. Experimental Methods

2.1. Materials

(1) Cement and Water

The cement used throughout this study was Portland cement (P.O 42.5) conforming to standard GB175-2020 and obtained from Conch Cement Group. The detailed properties of the cement are shown in Table 1. Clean and fresh water was used for casting and curing of the samples and RCP specimens.

Table 1. Properties of cement.

Loss on Ignition (%)	Initial Setting Time (min)	Final Setting Time (min)	Specific Surface Area (m ² /kg)	Compressive Strength (Mpa)		Flexural Strength (Mpa)	
				7 Days	28 Days	7 Days	28 Days
2.35	170	290	337	27.3	45.6	5.6	8.2

(2) Fine Aggregate

The fine aggregate used in the mixes was locally-sourced river sand with a maximum particle size of 4.75 mm and a fineness modulus of 2.60, which met the requirements for a Class II gradation medium sand.

(3) Coarse Aggregate

The coarse aggregate involved two types of natural coarse aggregate (NCA) and recycled coarse aggregate (RCA). The NCA was crushed calcareous limestone from the stone quarries which had continuous gradation with a particle size of 5~20 mm. The RAC was supplied by the local plant from the demolition of twenty- to thirty-year-old concrete pavements, the original strength grade of which was unknown. Prior to the experiment, the aggregates larger than 20 mm were screened, and impurities such as bricks and wood were removed.

Tests were conducted on the NCA and RCA according to the Code of GB/T 25177-2010 and GB/T 14685-2011; the physical properties of the coarse aggregates are shown in Table 2,

and Figure 1 shows the sieve analysis of the coarse aggregates. Both NCA and RCA met the specification that ensured the appropriate properties of the fresh and hardened concrete.

Table 2. Physical properties of coarse aggregate.

Type of Coarse Aggregate	Apparent Density (g/m ³)	Clay Content (%)	Water Absorption (%)	Crushing Value Index (%)
NCA	2644	0.6	0.9	8.2
RCA	2567	1.2	3.4	14.5

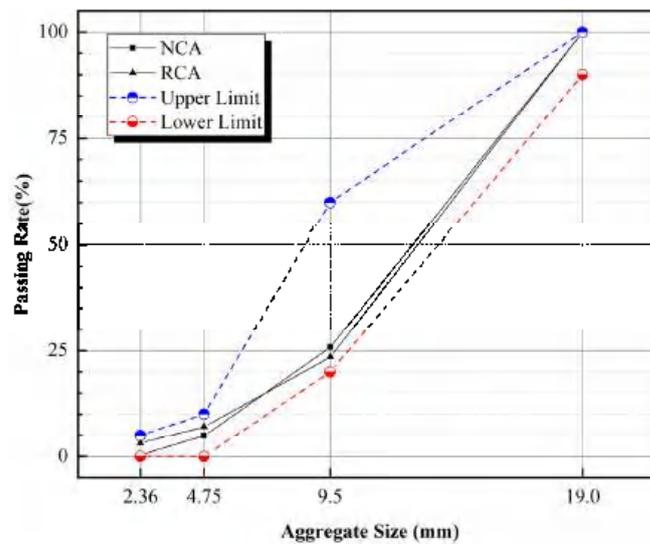


Figure 1. Grading of coarse aggregates.

(4) Steel reinforcement

The diameter of the longitudinal reinforcing bars in the RCP specimens was 8 mm, and the relevant properties were tested by using a tensile test machine. The bar response showed an obvious yield plateau, and the measured yield strength (f_{yk}), ultimate strength (f_{uk}) and elongation after fracture were 426 Mpa, 600 Mpa, and 18.2%, respectively.

2.2. Concrete Mix Design

Two types of concrete were designed in this experiment: Natural Aggregate Concrete (NAC) with natural fine and coarse aggregate, and Recycled Aggregate Concrete (RAC) using natural fine aggregate and RCA of 100% mass replacement. The mix proportion of NAC and RAC were designed to have a similar compressive strength of 45 Mpa, and the concrete mixture proportions for the NAC and RAC are listed in Table 3.

Table 3. Matrix proportions of different mixes.

Type of Concrete	Cement (kg/m ³)	Fine Aggregate (kg/m ³)	Coarse Aggregate(kg/m ³)	Water (kg/m ³)	Water-Cement Ratio	Sand Rate (%)
NAC	18.57	22.10	46.96	7.80	0.42	32
RAC	22.29	21.03	44.69	7.80	0.35	32

Three cube samples of 100 mm and 150 mm size were cast for compressive strength and tensile splitting strength tests, respectively, and three prism samples with 450 mm × 150 mm × 150 mm length were made for the purpose of testing the flexural strength. The mechanical properties of NAC and RAC were measured at 28 days under the standard curing condition according to GB/T 50081-2002, as shown in Table 4 based on average values for three tested samples.

Table 4. Mechanical properties of concrete.

Type of Concrete	Compressive Strength, 28 Days		Split Tensile Strength, 28 Days		Flexural Strength, 28 Days	
	Mean (Mpa)	Standard Deviation	Mean (Mpa)	Standard Deviation	Mean (Mpa)	Standard Deviation
NAC	46.80	1.694	3.55	0.141	5.64	0.303
RAC	48.28	1.236	3.21	0.172	5.44	0.376

2.3. Description of RCP Specimens

Five types of 1000 mm square RCP slab specimens with different longitudinal reinforcement spacing (100 and 150 mm), thickness (60, 70, and 80 mm), and RAC replacement ratio (0 and 100%) were designed for the experimental program. The RCP specimens were named by thickness (cm), type of concrete, the location of the impacting load, and additional information (1 meant the longitudinal reinforcement spacing was 100). For example, two types of concrete were used in the experimental program: NAC was named N, and RAC was named R. M represents that the RCP specimen was subjected to impact load at its mid-point. The details of all RCP specimens are summarized in Table 5, and their reinforcement layouts are shown in Figure 2.

Table 5. Summary of RCP specimen.

RCP Specimen	Thickness (mm)	Type of Concrete	Bar Spacing (mm)	Average Compressive Strength (Mpa)	Maturing Age
6MR	60	RAC	150	48.67	1 year, 11 days
7MR	70	RAC	150	48.73	1 year, 8 days
8MR	80	RAC	150	47.78	1 year, 9 days
7MN	70	NAC	150	47.21	1 year, 11 days
7M1R	70	RAC	100	49.43	1 year, 11 days

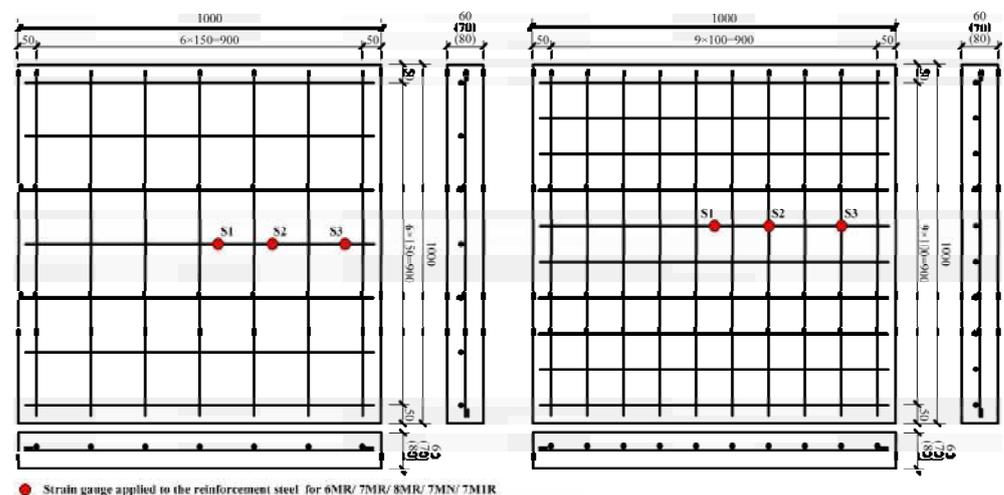


Figure 2. Size and reinforcement layout of RCP specimen (mm).

All RCP specimens adopted the single-layer reinforcement scheme with equal amounts of reinforcement in both planar directions, resulting in two layers of bars. The diameter of reinforcement bars was 8 mm, and the rebar spacing within the RCP specimens ranged from 100 mm to 150 mm. For slab thicknesses of 60 mm, 70 mm, and 80 mm, the thicknesses of concrete protective cover from the bottom surface were 25 mm, 30 mm, and 35 mm to ensure that the steel bars were located in the middle of the slab. According to MT/T 5004-2010, the single-layer steel bars should be located in the lower 1/3~1/2 thickness of the slab [2]. The transverse and longitudinal steel pieces were bundled by steel wires

to create a continuous mesh, and those steel meshes were fixed on the cement cushion block during vibration to achieve an accurate positioning in the slab. Short bars of 20 mm length and 8mm diameter were welded at both ends of all reinforcement bars to enhance anchoring capacity and ensure sufficient reinforcement (Figure 3).

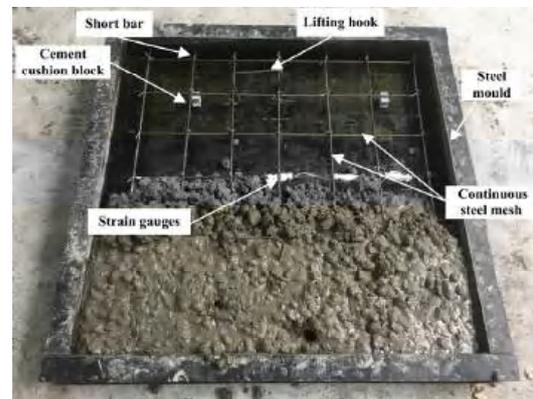


Figure 3. Casting of RCP specimen.

To ensure consistency and good quality of each RCP specimen, the concrete was mixed and cured in-house using a 60 L single horizontal-axis forced mixer in the Concrete Materials Laboratory. In addition to the RCP specimens, three 100 mm × 100 mm × 100 mm cube samples were cast from the same batch to characterize the compressive strength of the concrete material. The cube samples and RCP specimens were cured in natural environment conservation, and watered in the first 7 days to ensure the strength of concrete and prevent cracks. The average compressive strength of cube specimens was measured synchronously after the impact experiment, which is listed in Table 5.

2.4. Instrumentation

To fully document the dynamic response of the RCP specimens, various kinds of sensors were installed to monitor specimen displacements, accelerations, concrete strains, and reinforcing bar strains during the test.

Two laser-type displacement sensors were installed to capture the vertical displacement distribution of the RCP specimens; this kind of sensor is generally capable of capturing higher response frequencies than is the linear variable differential transformer, making it more suitable for impact testing applications. Two accelerometers were mounted on the top surface to capture the acceleration distribution of the RCP specimens, and one accelerometer was attached to the drop-weight to estimate the impact force. These accelerometers had the capacity of measuring accelerations within the range of ± 5000 g and were used to measure accelerations along the vertical axis of motion. Moreover, three strain gauges with 5 mm gauge lengths were glued to the bottom surface of reinforcement bar prior to concrete pouring, and a total of five strain gauges with 50 mm gauge lengths were arranged on the top and bottom surfaces of the RCP specimens. The range of the magnitude and rate of strain were detected by eight strain gauges applied to the concrete and the reinforcement bar. The arrangement and designation of these sensors are given in Figure 4.

2.5. Test Program

The RCP specimens were placed on the top of a compacted sand-and-gravel layer in a steel strongbox. The net internal size of the steel box was 1020 mm × 1020 mm × 650 mm, which was slightly larger than the size of the RCP specimens; all four edges of the RCP specimens were free edges without constraints. The sand-and-gravel layer was composed of sand and gravel in a mixing ratio of 1:2, and with no clods, roots, or other sundries inside. The maximum size of gravel aggregate was limited to 26 mm, and particle size distributions of the sand-and-gravel layer conformed to the requirements of continuous

gradation [41]. The sand-and-gravel layer was compacted to a degree of 0.97, in order to enhance its strength and provide a high-quality subbase.

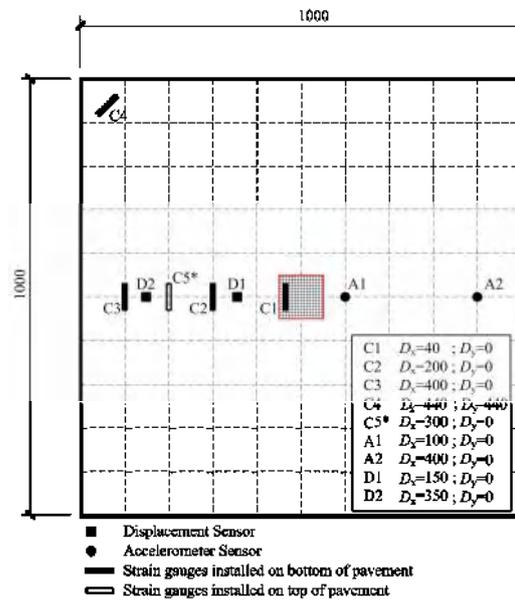


Figure 4. Casting of RCP specimen.

The drop-weight test frame consisted of two columns, the drop-weight, and two vertical guide rails; the rails were used to guide the drop-weight during the fall. The drop-weight in the experimental program was comprised of a cross-beam with a span of approximately 2.5 m and an impactor with a striking surface of a 20 cm diameter hemispherical nose. The total mass of the drop-weight was 200 kg, and the drop-height of the drop-weight was set at 1.0 m above the top surface. A rubber pad with dimensions of 100 mm × 100 mm × 10 mm was placed on the RCP specimen exactly in the contact zone, which simulated the impact cushioning effect of the landing gear. Prior to performing a test, the drop-weight was lifted up along the guide rail to the desired height and secured to an electric clamping style release mechanism. After debugging all related devices and instrumentation, the drop-weight was released in a free-fall condition to generate the impact loads, and all digital data were recorded synchronously. The schematic diagram of the setup and the test configuration is illustrated in Figure 5.

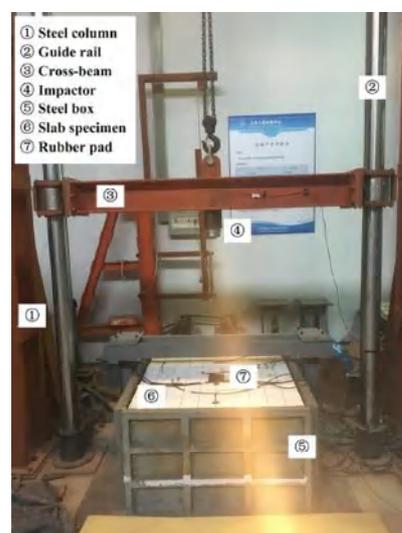


Figure 5. Experimental setup.

3. Experimental Results and Discussion

3.1. Impact Force Characteristics

According to the test program described above, the acceleration-time history of the drop-weight was obtained from the measurement data of the A3 accelerometer, as shown in Figure 6. The acceleration-time history was used to calculate the impact force-time history. The formula is $F(t) = ma(t)$, where m is the mass of the drop-weight that remained at 200 kg consistently in all tests. The impact force-time histories of different tests were similar in shape, showing a high magnitude peak followed by few small magnitude shocks, which were caused by the rebounding of the drop-weight after impacting the RCP specimens. Compared with other subsequent peaks, the magnitude of first impact was very high, and therefore, the impact response under first impact is the most central issue for this research [22,23].

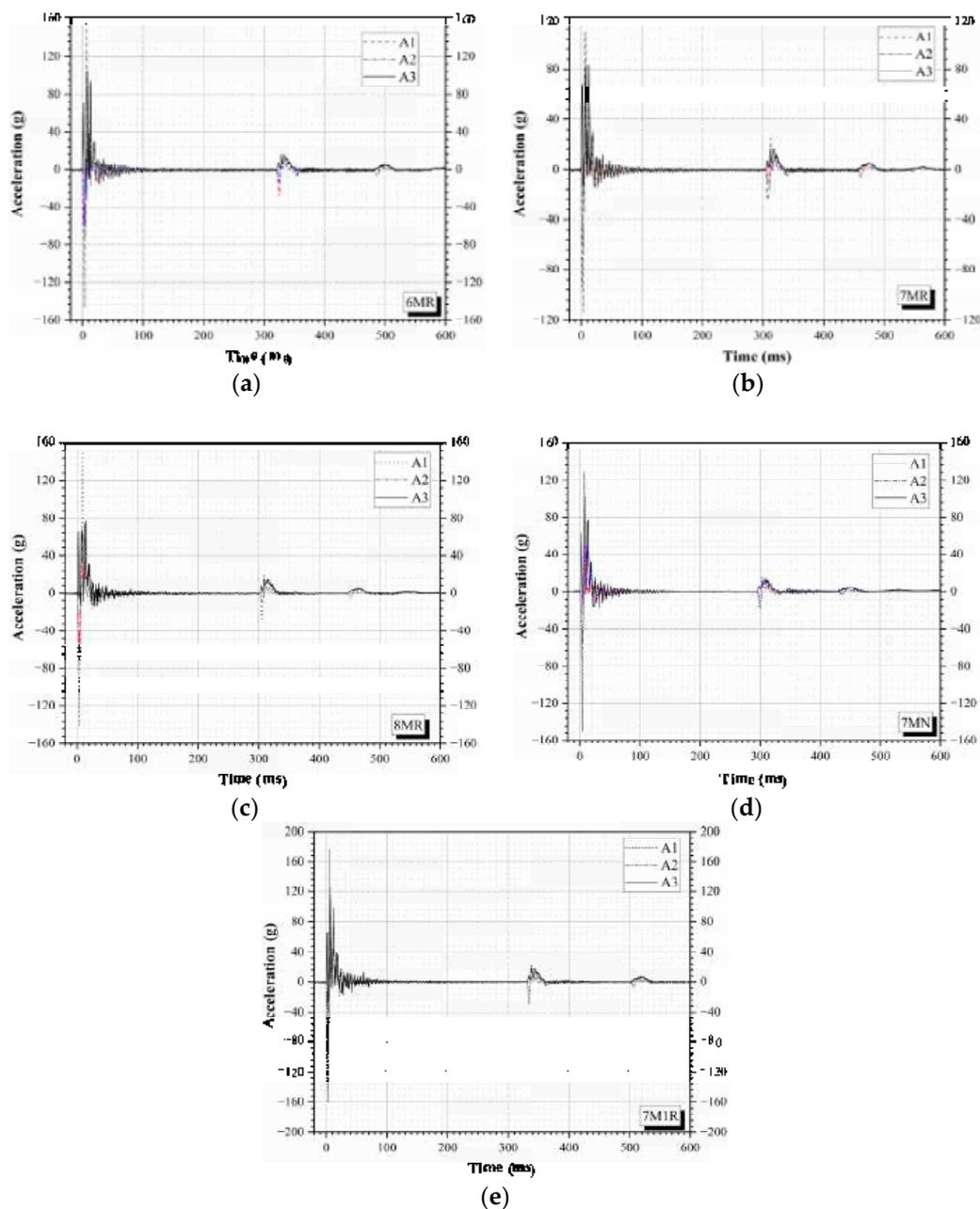


Figure 6. Acceleration-time histories of the (a) 6MR specimen, (b) 7MR specimen, (c) 8MR specimen, (d) 7MN specimen, and (e) 7M1R specimen.

To assess the inertial force during the impact test, a simple integration approach is proposed; that is, to use the recorded A1 and A2 acceleration data to estimate the inertial force of the RCP specimen [26]. The acceleration-time histories of the A1 and A2 accelerometer are shown in Figure 6. The accelerometer A2 was placed further away from the impact area than was the accelerometer A1, and it was thought that the phase diversity of the acceleration between A1 and A2 reflected the lag in response, owing to the force propagating from the point of impact to the edge [42]. The RCP specimen gained significant downward acceleration immediately after impact and upward inertial force was induced. The RCP specimen was divided into three tributary integration areas according to the positions of accelerometers, as shown in Figure 7. For Area 1, the acceleration was assumed to be uniformly distributed and its value was equal to the value of the A1 sensor. The acceleration was assumed to have a linear distribution for both Area 2 and Area 3. For Area 2, the values of the inner and outer boundaries of acceleration were the values of the A1 and A2 sensors, respectively. For Area 3, since the size of the RCP specimen was much larger than that of the impact area, it could be considered that the acceleration at the far edge of the specimen was sufficiently weak to be ignored. Accordingly, the values of the inner and outer boundaries of the acceleration were the value of the A1 sensor and zero, respectively. The inertial force of the RCP specimen could be calculated by summing up the inertial forces of the three tributary areas.

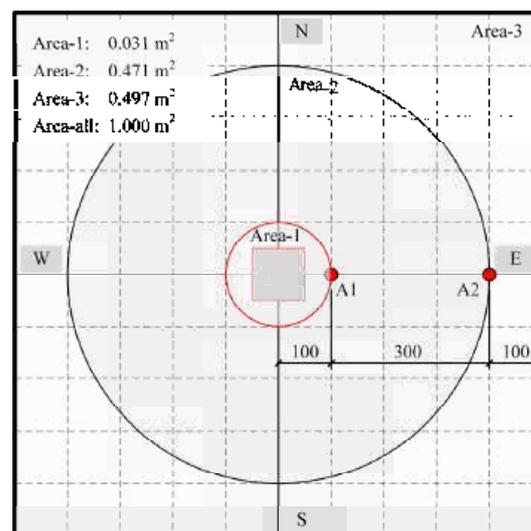


Figure 7. Integration areas of the inertial force.

For all the specimens, impact force and inertial force rose immediately and reached their peak values shortly after the impact began. The impact force suddenly dropped due to the deformation and cracking of the RCP specimen. As the impact force decreased, the inertia force decreased and dissipated. The peak impact force was detected after circa 7–8.5 ms following the first contact between the drop-weight and the RCP specimen, and the result is summarized in Table 6. There was a time lag of circa 1–3 ms between the peak impact force and the peak inertia force, which was due to the stress wave propagation travelling gradually from the impact area to the far edge [25,31,43]. This stress wave traveled at varying speeds within the speed of sound, depending on the mass, the density and the elastic modulus of the concrete type used [44]. Comparing the impact force with the inertia force, it is obvious that the peak amplitude of impact force is greater than that of the inertia force. The reason is that most of the impact force was converted into inertia force, while a portion of impact force was balanced by ground reaction force during impact.

Table 6. Impact force characteristics of RCP specimen.

RCP Specimen	$F_{im,p}$ (kN)	$F_{in,p}$ (kN)	I_p (kN.s)	V_r (m/s)	E_{im} (J)	E_{ab} (J)	E_{ab}/E_{im} (%)
6MR	206.1	85.9	1143.4	1.292	1960.0	1793.0	91.5
7MR	175.9	110.7	1148.5	1.315	1960.0	1787.0	91.2
8MR	150.9	116.1	1153.0	1.338	1960.0	1781.0	90.9
7MN	200.2	105.1	1151.1	1.328	1960.0	1783.5	91.0
7M1R	220.5	111.7	1198.5	1.565	1960.0	1715.0	87.5

$F_{im,p}$: peak impact force; $F_{in,p}$: peak inertial force; I_p : impulse; v_r : rebound velocity; E_{im} : impact energy; E_{ab} : absorbed energy.

All RCP specimens were impacted by a free fall of 200 kg drop-weight from a constant height of 1.0 m, so in the case of ignoring friction and air resistance, when the drop-weight impacted the RCP specimen, the instantaneous impact velocity v_{im} was about 4.4 m/s and the maximum impact energy E_{im} was 1.96 kJ. The reported impulse I_p is the time integration of impact force. The impulse–momentum theorem states that the impulse is equal to the change of momentum [23–25]. Thus, it is expected that the rebound velocity v_r of the drop-weight can be calculated by the formula $I_p = m \times v_{im} - m \times v_r$. Once the rebound velocity v_r is known, the residue kinetic energy of the drop-weight can be calculated. In calculating the energies for the impact test, the current study neglected the energy dissipated in the following mechanisms: the free vibration of the RCP specimen and steel strongbox, and the energy losses due to heat and noise [37,45]. Thus, the energy absorbed by the RCP specimen can be calculated via subtracting the residue kinetic energy of drop-weight from the impact energy [27]. The calculated rebound velocities v_r , absorbed energies E_{ab} , and its ratio over impact energies of different tests are listed in Table 6.

It can be seen from Table 6 that the characteristic value of impact force varies in accordance with the longitudinal reinforcement spacing, the concrete type, and the thickness. Comparing 7MR with 7MN, the peak impact force of 7MR is 13.8% smaller than that of 7MN. According to the contact theory proposed by Hertz [46], the force between two objects in contact is proportional to the relative elastic modulus. The 7MR specimen showed a lower impact force, which can be attributed to a lower modulus of elasticity of the RCA mix compared with the NAC mix. The peak inertial force of 7MN is very similar to that of 7MR, and the difference between the two is less than 5%. This is likely due to the fact that the steel reinforcement contributed more to the stiffness of the RCP specimen in this state, thus reducing the relative influence of concrete on the overall stiffness.

A tendency is observed that the peak impact force increased with the increase in reinforcement ratio. As can be seen from Figure 8, the peak impact force of 7M1R is 25.3% higher than that of 7MR, which is the maximum value among all RCP specimens. Therefore, increasing the reinforcement ratio could improve the stiffness of the RCP specimen and have a significant effect on the impact force [22,23]. In addition, it was observed that the peak inertial force slightly increased as the reinforcement ratio increased.

Compared with 7MR, the peak impact force of 6MR increased by 17.1%, while the peak impact force of 8MR decreased by 14.2%. According to the research results of Xiao [26], the thickness could increase the impact resistance and stiffness of the RCP specimens; therefore, an increase in the peak impact load should be also observed. The reason for this is that, although 6MR, 7MR and 8MR specimens had the same reinforcement layout scheme of D8@150, the reinforcement ratio decreased with the increase in thickness. When considering the peak impact load, the influence of the thicknesses was relatively lower compared with the effect of the reinforcement ratio, which has paramount relevance. In addition, it was observed that the impact force duration slightly decreased as the reinforcement ratio increased in the 8MR, 7MR, 6MR and 7M1R specimens. As can be seen from Figure 8, the peak inertia force variation rules of the RCP specimen were different. The peak inertia force of the 8MR specimen with the largest thickness was maximum, while the peak inertia force of the 6MR specimen with the lowest thickness was minimum.

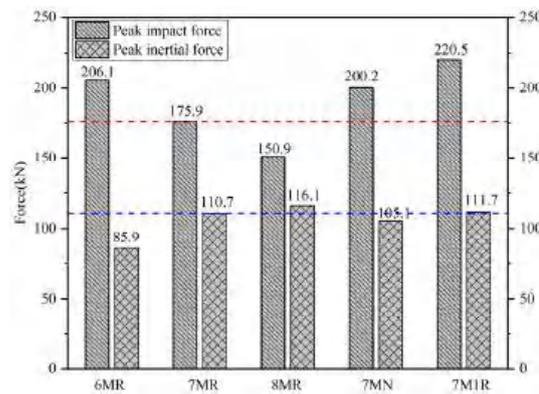
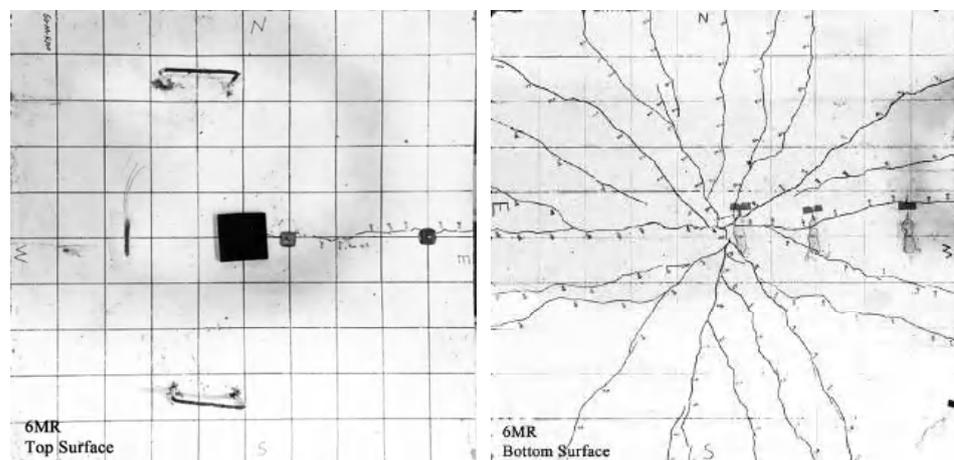


Figure 8. Peak impact load and peak inertial force.

Except for 7M1R, the energy dissipation ratio E_{ab}/E_{im} of all RCP specimens exceeds 90%, implying that the RCP specimens dissipate most impact energies through deformation and cracking. The energy dissipation ratio E_{ab}/E_{im} of 7MR specimen was similar to that of 7MN specimen. For 7M1R specimen, as shown in Figure 9, there was slight damage on the surface after impact, and approximately 87.4% of the impact energy was imparted to the specimen. For the less damaged specimen, more impact energy could be stored through the temporary elastic deformation of the specimen [27]. This stored energy would return to the drop-weight when the elastic deformation recovered, resulting in greater rebound speed v_r . On the contrary, severely damaged specimens had already entered their plastic stage and more impact energy was dissipated in the form of permanent deformation or crack damage. The 6MR specimen was severely damaged and the energy consumption ratio E_{ab}/E_{im} reached 91.5%. As seen in Figure 9, the damage characteristics and crack patterns after the impact also confirmed this phenomenon.

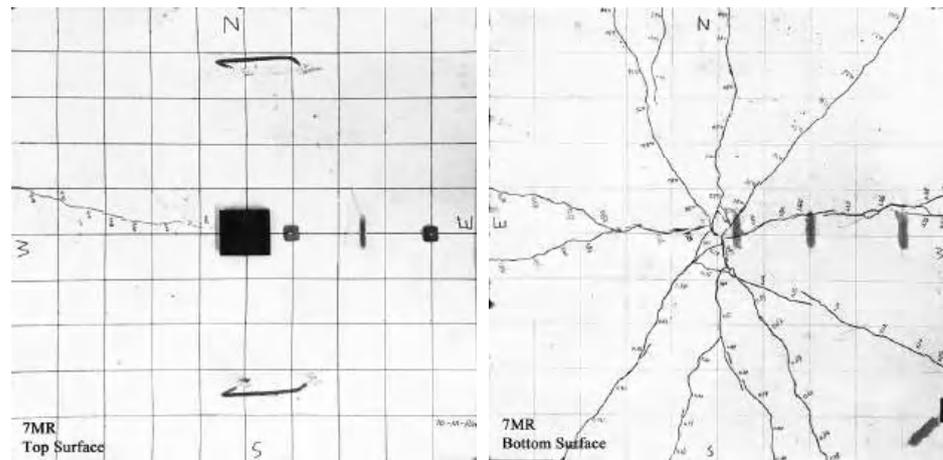
3.2. Damage Characteristics and Crack Patterns

Prior to the test, the surfaces of the PCP specimen were painted white and then meshed with spacing of 100 mm grids in order to observe damage characteristics and crack patterns. The “E/S/W/N” symbols were marked at top, bottom and side surfaces of the RCP specimen, and the RCP specimen was divided into four regions according to direction. The cracks that developed after each test were marked, and the crack widths were measured manually by HC-CK101 Concrete Crack Width Meter. For impacting at the mid-point, the sketched cracks profiles of the RCP specimen are shown in Figure 9.

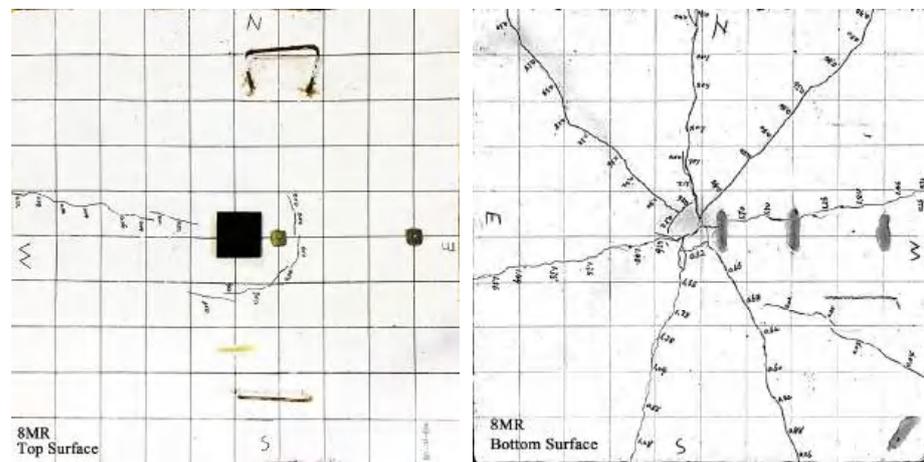


(a)

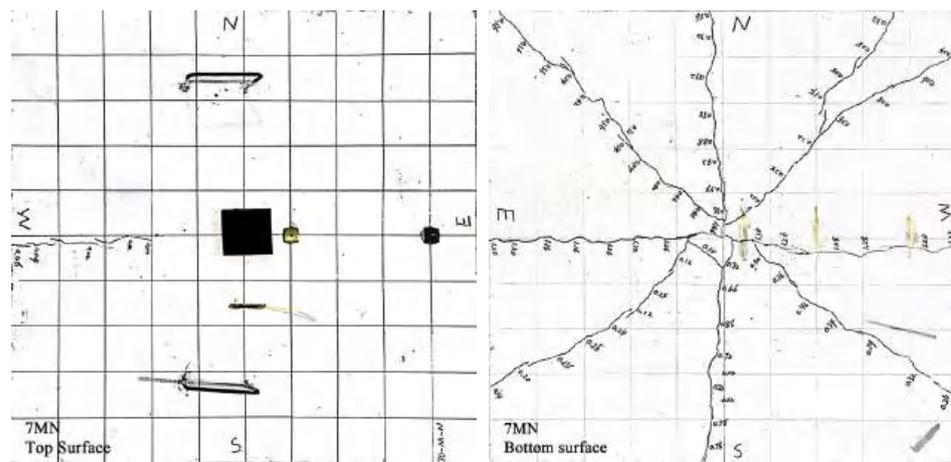
Figure 9. Cont.



(b)



(c)



(d)

Figure 9. Cont.

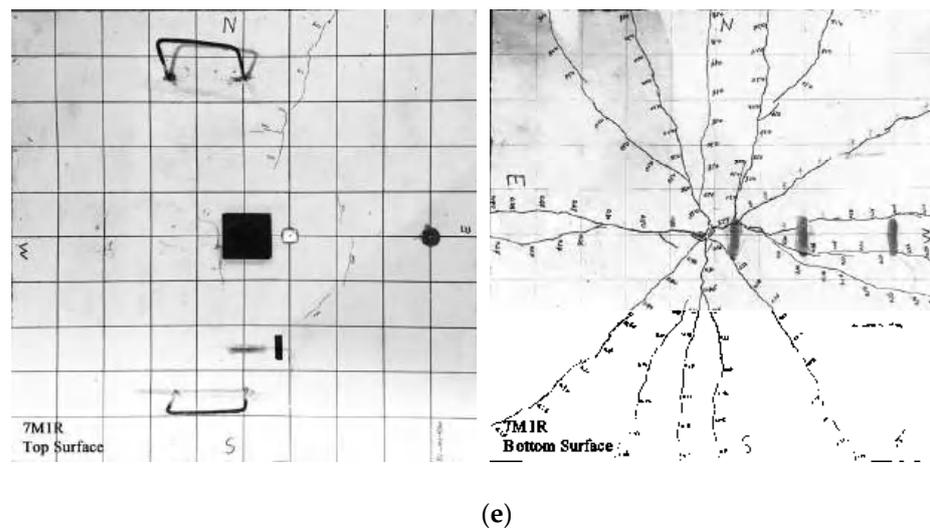


Figure 9. Sketched cracks profiles of the (a) 6MR specimen, (b) 7MR specimen, (c) 8MR specimen, (d) 7MN specimen, and (e) 7M1R specimen.

The type of damage and crack development mode on the bottom surface of all RCP specimens are similar. The crack patterns mainly appeared as the radial crack and diagonal crack, indicating that the deformation of the specimen was a global flexural deformation. The major radial crossing cracks and the failure took place simultaneously. The concrete scabbing was quite limited and mainly centralized in the region of 200 mm × 200 mm beneath the impacting point. The maximum residual width of the crack was also found in this region, which reached up to 1.46 mm~1.8 mm. For all RCP specimens, the widths of radial cracks were larger than those of diagonal cracks. This is because the radial crack developed prior to the diagonal crack, and the radial crack could dissipate more impact energy, thus reducing the crack width [27]. There were two different crack patterns on the top surface of the RCP specimen: one was the radial crack propagating from the bottom surface towards the top surface, and the other was the circumferential crack with the impact point as the center. No obvious penetration was observed on the surface of the RCP specimen that was found in References [47,48]. The final crack properties of all tested specimens are presented in Table 7.

Table 7. Final crack properties of all RCP specimens.

RCP Specimen	Bottom Surface			Top Surface		
	Crack Pattern	Num of Crack	Maximum Crack Widths	Crack Pattern	Num of Cack	Maximum Crack Widths
6MR	radial crack, diagonal crack	21	1.60	radial crack	1	0.08
7MR	radial crack, diagonal crack	9	1.80	radial crack	1	0.12
8MR	radial crack, diagonal crack	7	1.80	circumferential crack radial crack	2	0.08
7MN	radial crack, diagonal crack	8	1.46	radial crack	2	0.06
7M1R	radial crack, diagonal crack	14	1.50	circumferential crack radial crack	2	0.04

The final damage status of the 7MR specimen is similar to that of the 7MN specimen. In the E-W direction, radial cracks were fully developed, and their widths were in the range of 1.2–1.4 mm for both 7MR and 7MN. While, in the N-S direction, the cracks generated in 7MR are slightly more than those in 7MN, this could be attributed to the character of RCA, whose adhesive mortar and cracks caused by procession have an adverse effect on the behavior of the concrete matrix. Furthermore, due to the high brittleness of RAC, the radial crack propagation was normally unstable [49]. As shown in Figure 10, the radial cracks extending from the bottom to the top run along the W-E direction toward the impact point, simultaneously with the crack widths being gradually reduced. The radial crack

widths of 7MR on the top surface were larger than those of 7MN. Furthermore, the radial crack of 7MR extended to the impact point, while the radial crack of 7MN extended only a quarter of the slab span. With the same thickness, reinforcement ratio and concrete grade, RAC has little influence on damage characteristics and crack patterns of RCP specimens; however, the crack resistance of 7MR is slightly lower than that of 7MN.

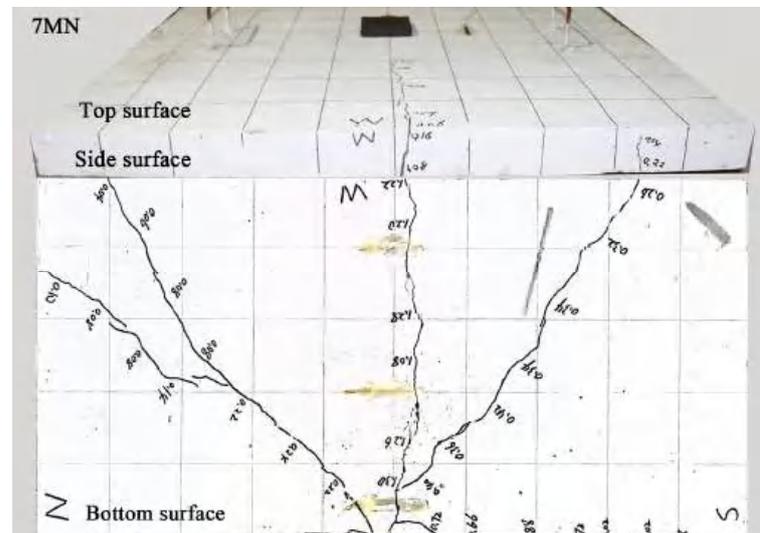


Figure 10. Radial cracks extending from the bottom to the top.

The number of cracks on the bottom surface decreased with the increase in the thickness of the slab. With the reduction of the thickness from 70 mm to 60 mm, multiple tightly spaced hairline cracks formed on the bottom surface, while as the thickness increased from 70 mm to 80 mm, the development of cracks was strongly limited. The crack width widened as the number of cracks decreased. These results suggest that there is an association between crack resistance and slab thickness. The crack patterns on the top surface varied with the change of the thickness of the slab. The radial cracks extending from the bottom to the top were found in the 6MR, 7MR, and 8MR specimens, and the circumferential cracks around the impact area were detected only in the 8MR specimen. These circumferential cracks with a hairline width less than 0.06 mm did not close and developed in the range of a half-circle. The circumferential cracks indicated that localized damage in the form of limited concrete penetration on the impact surface had occurred in the 8MR specimen. The change in crack patterns was due to the stiffness of the specimen increasing as a result of the increase in the slab's thickness, and partial impact energy needing to be dissipated through local damage deformation during the impact [48,50]. Except for the cracks mentioned above, the remaining area on top surface of 8MR specimen was nearly undamaged.

Both circumferential cracks similar to those in 8MR and radial cracks similar to those in 7MR were found on the top surface of 7M1R specimen. In comparison with 8MR, the circumferential cracks of 7M1R had further distributed distance, thinner width, and smaller range. The development of the radial cracks on the top surface of 7M1R was limited when compared to that of 7MR. Based on the observed damage and crack development in tested specimens, it was found that the crack pattern was more affected by the thickness than by the reinforcement ratio. More steel reinforcement would induce a localized failure of concrete [23].

3.3. Displacement Response

The displacement-time histories of D1 and D2 are shown in Figure 11, and it is found that the displacement-time history shapes of all RCP specimens are similar in terms of magnitude, time response, and residual displacement. With each impact event performed, the RCP specimen exhibited progressively increasing peak displacements, and

then decreased to a stable residual displacement, followed by few small displacements due to rebounding of the drop-weight after impacting the RCP specimen. It should be recalled that the magnitude of first impact was very high compared with other subsequent peaks; therefore, the peak displacement and residual displacement due to the first impact were recorded in Table 8. The final cumulated residual displacement would affect the performance of the RCP specimen; it was also recorded in Table 8. It can be seen that the displacement at D1 point was larger than that at D2 point for all RCP specimens from Figure 11. This is because when the drop-weight impacted the top surface, due to the limited impact area, sufficient impulse should be provided in this area to prevent the drop-weight from falling until it stopped. Therefore, compared with other areas, the stress around the impact area was greater, the damage was more serious, and the deformation was more obvious.

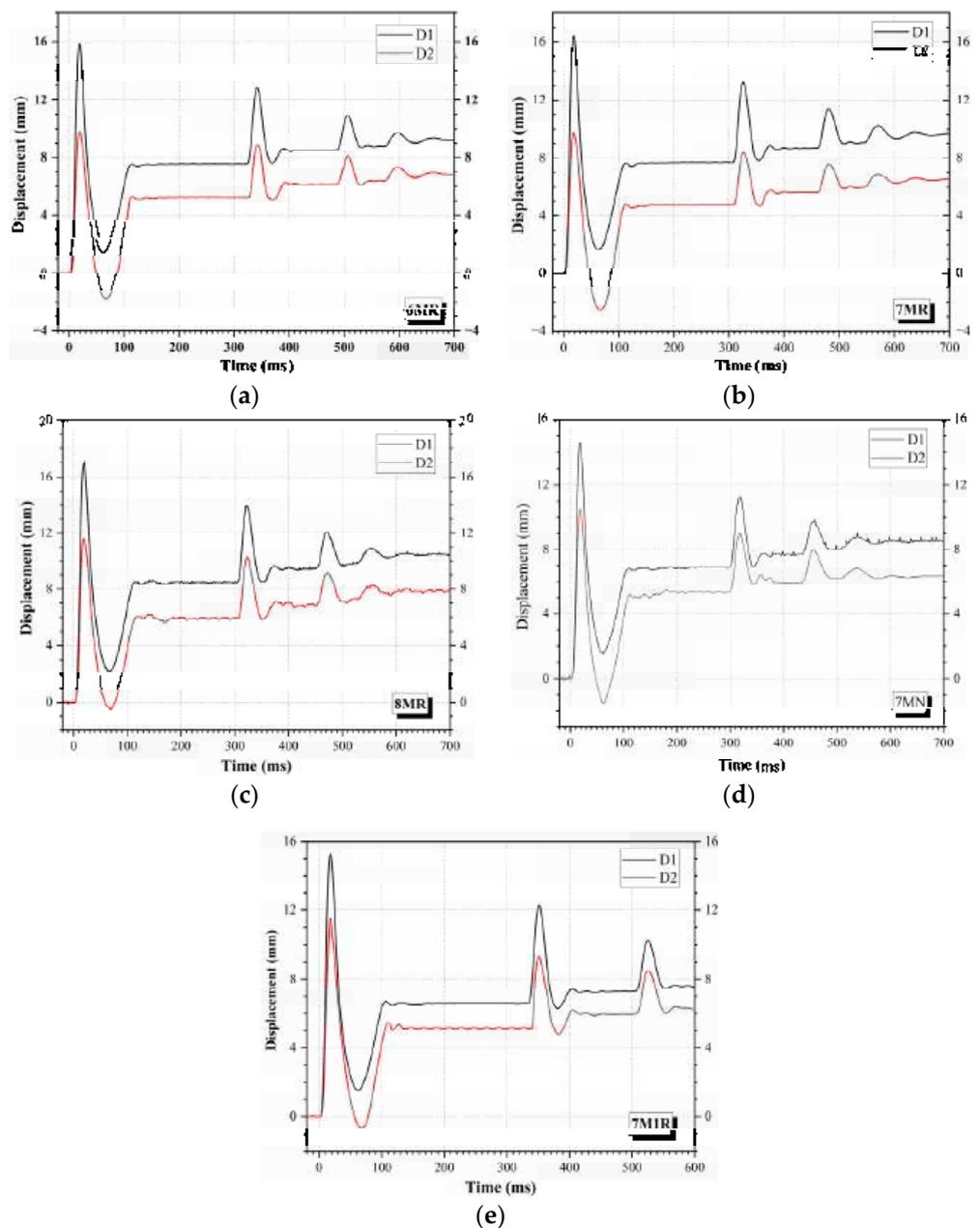


Figure 11. Displacement-time histories of the (a) 6MR specimen, (b) 7MR specimen, (c) 8MR specimen, (d) 7MN specimen, and (e) 7M1R specimen.

Table 8. Displacement response of all tested specimens.

RCP Specimen	D1			D2		
	ω_{p1}	ω_{r1}	ω_{fr1}	ω_{p2}	ω_{r2}	ω_{fr2}
6MR	15.87	7.46	9.24	9.80	5.04	6.83
7MR	16.40	7.54	9.67	9.75	4.67	5.83
8MR	17.10	8.47	10.42	11.61	5.91	7.89
7MN	14.60	6.84	8.50	10.55	5.27	6.23
7M1R	15.25	6.63	8.16	11.53	5.34	6.66

ω_{p1} and ω_{p2} : the peak displacement at D1 point and D2 point, respectively; ω_{r1} and ω_{r2} : the residual displacement under first impact at D1 point and D2 point, respectively; ω_{fr1} and ω_{fr2} : the final cumulated residual displacement at D1 point and D2 point, respectively.

As can be seen from Table 8, the peak and residual displacements at D1 point of 7MN specimen were lower than that of 7MR specimen, reduced by 11.0% and 5.3%, respectively. According to the literature [51,52], the elasticity modulus of concrete decreased with the increase in RCA replacement ratio. In addition, the micro cracks in adhesive mortar of RCA had a detrimental effect on crack development, which would reduce the stiffness of the specimen. With increased distance from the impact area, both the peak displacement and the residual displacement showed an opposite trend to that of before. The peak and residual displacements at D2 point of 7MN specimen were higher than that of 7MR specimen, increased by 8.2% and 12.9%, respectively. This meant that the difference between D1 and D2 was decreasing, indicating that the deformation on the front surface of 7MN specimen became gentle, and showed more flexural response.

The reinforcement ratio plays an important role in peak deflection and residual displacement [47]. As the reinforcement ratio was increased from 0.48% to 0.72%, the peak and residual displacements at D1 point decreased by 7.0% and 12.2%, respectively. The reason for such behavior may be attributed to the fact that more steel bars could effectively arrest the propagation of cracks inside the concrete, thus improving the stiffness of RCP specimens. Compared with the 7MR specimen, the 7M1R specimen exhibited smaller displacement amplitudes under same impacts and was expected to be able to undergo larger displacement amplitudes before failure. The variation trend of the displacement at D2 point of 7M1R specimen was similar to that of 7MN specimen, and the peak value and residual displacement at D2 point of 7M1R specimen are 18.3% and 12.1% higher than that of 7MR specimen, respectively.

At D1 point, compared with the peak and residual displacements of the 7MR specimen, those of the 6MR specimen had undergone approximately 3.2% and 1.11% decrease, respectively, while those of the 8MR specimen had undergone approximately 4.3% and 12.3% increase, respectively. The peak and residual displacements of 6MR specimen and 7MR specimen at D2 point were not significantly different. In this case, it is thought that before the overall deformation occurred, the impact energy of 6MR specimen would have been dissipated through the development of the dense radial cracks on the bottom surface. The 8MR specimen always maintained a large displacement value at D1 and D2, indicating that when the drop-weight impacted against the 8MR specimen, almost all the impact energy was dissipated through global deformations. As described in Section 3.2, the 8MR specimen had the fewest number of radial cracks on the bottom surface among all RCP specimens. At the same time, as the velocity of drop-weight progressively slowed down with the increase of displacement, the circumferential cracks were formed on the top surface near the impact area.

As can be seen from Figure 11, under first impact, the RCP specimen reached the peak downward displacement and then rebounded upward. The amplitude of D1 and D2 rebound displacement changed differently between specimens. For 7MR, 6MR, and 7MN, the peak upward displacement at D2 point were -2.53 mm, -1.79 mm, and -1.49 mm respectively (downward is positive), while the peak upward displacement at D1 point remained positive. The specimen showed a trend of reverse bending deformation, and the

stress wave bounced back from the base to the surface to form tensile stress, which could further explain the radial cracks which appeared on the top surface of these specimens. For 8MR and 7M1R specimens, the upward displacements of D2 point were relatively small, and the radial crack development was limited due to the higher stiffness.

The 8MR specimen with the minimum reinforcement ratio had the maximum final cumulative residual displacement, while the 7M1R specimen with the maximum reinforcement ratio had the minimum final cumulative residual displacement. The final cumulative residual displacement of the RCP specimen was found to correlate with the reinforcement ratio more than with other factors.

Research addressing the displacement shapes of the RCP specimens could provide more information regarding the impact response, which was difficult to directly observe in the displacement analysis at D1 and D2 points. Therefore, the displacement shapes of the RCP specimens were addressed in this paper, which provides a quantitative index for comparing the global impact responses of the RCP specimens. Accelerometers A1 and A2 were arranged along the same axis as displacement sensors D1 and D2, which were 50 mm away from the left and right sides of D1 and D2, respectively, as shown in Figure 4. As described in Section 3.1, the acceleration-time history $a(t)$ at A1 point and A2 point of the RCP specimens were recorded by A1 and A2 accelerometers, respectively. The corresponding velocity $v(t)$ and displacement $d(t)$ responses can be calculated by numerical integration of the acceleration time histories using the Newmark Beta method [22,53]:

$$v_{i+1}(t + \Delta t) = v_i(t) + [(1 - \alpha)a_i(t) + \alpha a_{i+1}(t + \Delta t)]\Delta t$$

$$d_{i+1}(t + \Delta t) = d_i(t) + v_i(t)\Delta t + [(1/2 - \beta)a_i(t) + \beta a_{i+1}(t + \Delta t)]\Delta t^2$$

The acceleration was assumed to vary linearly between two instants of time in this study; α and β were chosen as 1/2 and 1/6, respectively [22]. Before impact, the initial velocity and initial displacement of the surface were considered to be zero. By assuming symmetric displacement response of the RCP specimen, the deflected shape along the midline of the top surface was plotted by linking the measured displacement data at D1 and D2 points and the calculated displacement data at A1 and A2 points. Uniform displacement with the value calculated by A1 was assumed in the impact area. The deflected shape of all the specimens is plotted at an interval of 2.0 ms and shown in Figure 12.

The value of deformation of the 7MR specimen was small in the initial 4 ms, and then increased rapidly. A global deformation on the top surface could be observed during the impact process, showing elastic flexural behavior. The deflection of the impacted area increased more rapidly than did the deflection of unloaded area, as shown in Figure 12. After reaching its peak displacement at about 14 ms, the impacted area began to rebound, while the unloaded area continued its downward movement for another few millimeters, and then rebounded at 18.5 ms. In the end, the displacement shape of the 7MR specimen flattened out again. In this case, the previously discussion of development of radial cracks observed on the bottom surface are believed to be attributable to the flexural displacements developed in the 7MR specimen. The similar behaviors were also observed in the displacement shapes of other specimens.

The punching shear behavior was observed in the 8MR specimen, indicating that the deflection of the impacted area increased much more rapidly than did the deflection of the unloaded area during 10ms to 16ms. Under the impact events, a slight development of localized displacements was observed to occur on one side of the impact region; however, no significant punching region was observed, and few instances of mass penetration had occurred. By comparing all displacement shapes shown in Figure 12, it can be seen that the displacement shapes of all RCP specimens were uniformly distributed, indicating that the failure of the specimen was mainly caused by the flexural deformation.

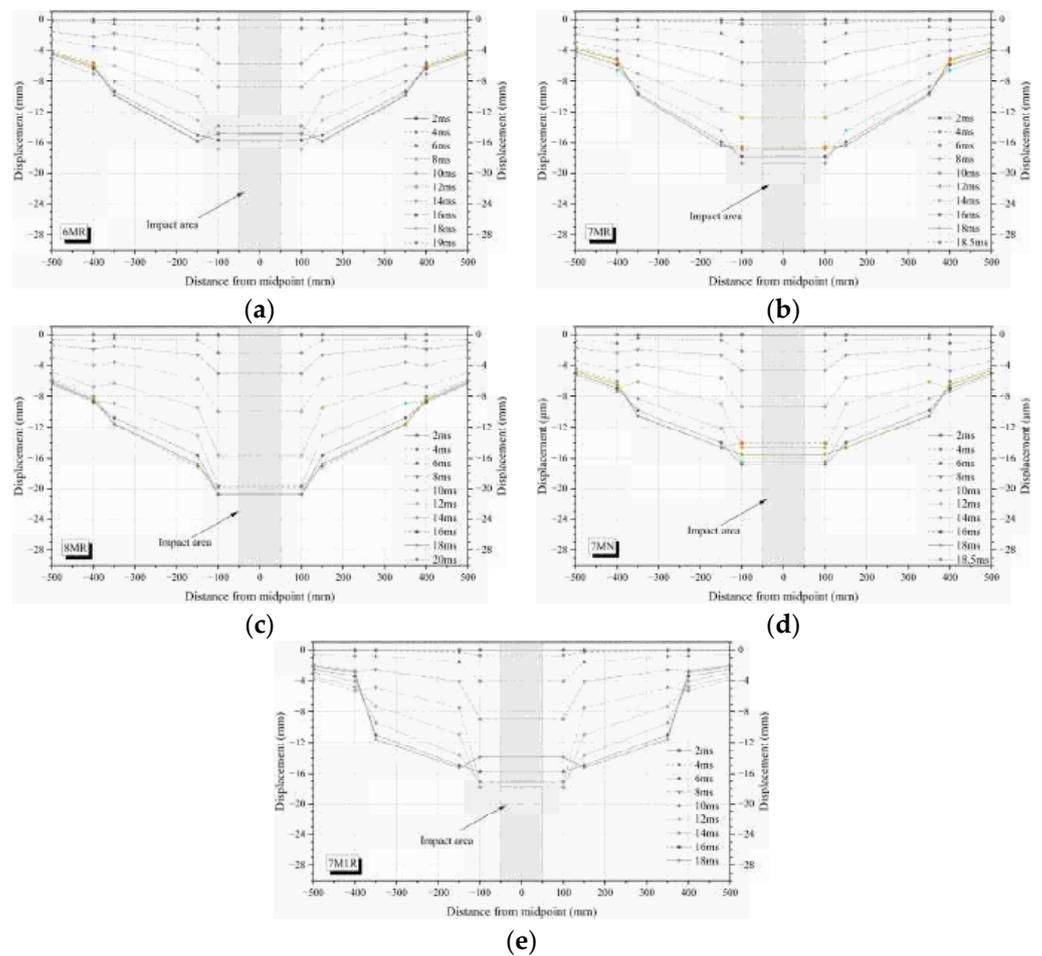


Figure 12. Deflected shapes of the (a) 6MR specimen, (b) 7MR specimen, (c) 8MR specimen, (d) 7MN specimen, and (e) 7M1R specimen.

3.4. Strain Due to Impact Load

The material strain was detected by five strain gauges applied to the concrete and by three strain gauges applied directly to the reinforcement steel. Figure 13a shows the strain evolution of the 7MR specimen, used as the reference specimen in comparison with all other specimens. Figure 13b shows a zoomed detail of the graphs under first impact. After the impact, a compressive strain of the concrete on the top can be seen, while a tensile strain of the concrete on the bottom can be observed. The C1 strain gauge was placed in central impact region and was disrupted about 2.0 ms after the first contact of the drop-weight. The failure of the C2 strain gauge could be determined from the horizontal plateau of strain-time history in Figure 13a. The C4 strain on the bottom was very small, indicating that, for the specimen with four free edges, the strain in the corner area of the RCP specimen could be ignored during the impact process, something which could be confirmed by the sketched cracks profiles in Figure 9. The compression strain of C5 on the top lasted for about 30 ms. At 14 ms, the maximum compressive strain of C5 reached about -2248μ . Corresponding to the compression strain of the concrete mentioned above, the peak value of the tensile strain of S1 was measured $10,292 \mu$. Regarding the strain of the reinforcement steel, the tensile strain decreased with the increase in distance from the impact area.

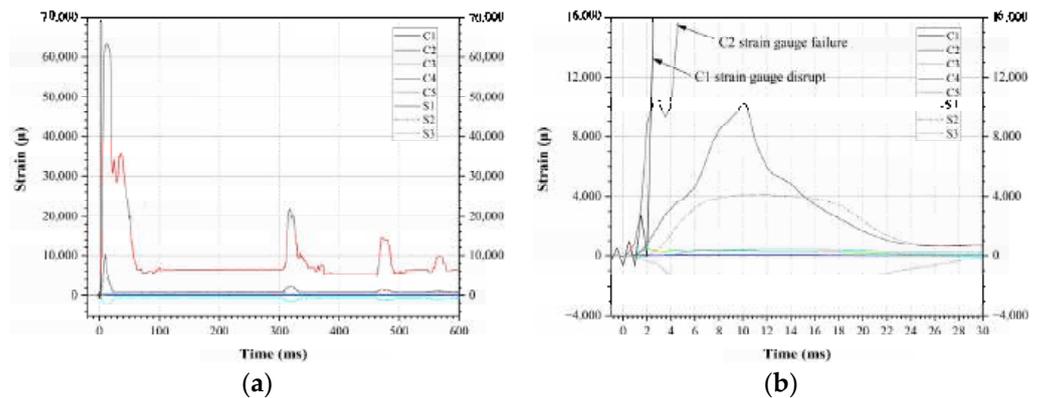


Figure 13. Concrete strain and steel strain of 7MR specimen (a) overall, and (b) in detail.

In this paper the steel strain of different RCP specimens was compared by means of the strain of the S1, as shown in Figure 14. For 6MR, 7MR and 8MR specimens, the strain values of the S1 strain showed a similar behavior, a rapidly increasing tensile strain followed by a very sharp drop. It can be observed that, with the increase in thickness, the peak S1 strain decreased, and the duration of the tensile strain shortened. For 7MN, the strain-time history showed a slower and smoother strain evolution than did the other specimens. In terms of peak strain values, the difference between the 7MR and the 7MN was 41.5%, while the difference between the 8MR and the 7MN was fairly small. It can be seen from Figure 14 that the steel strain of the 7M1R specimen was a compressive strain in the first 2.5 ms after the impact. This effect is indicative of the local material behaviors due to the impact, which is also described in Refences [32]. After a short duration of 4.0 ms, the S1 strain changes from compressive strain into tensile strain, indicating the transition from local deformation to global flexural behaviors. The more longitudinal reinforcement, the greater the decrease of tensile strain and the smoother the curve shape.

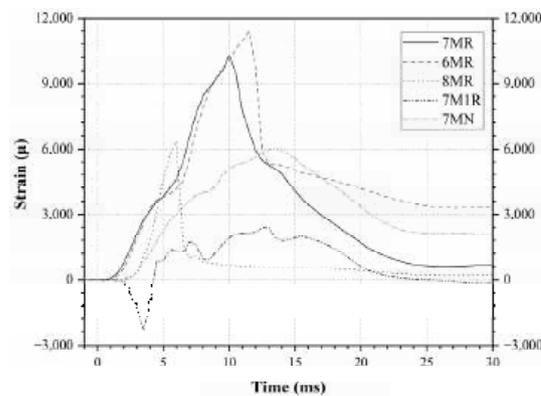


Figure 14. S1 steel strain under first impact.

4. Conclusions

The experimental investigation of five RCP specimens under impact load is presented in this paper. The acceleration, the displacement, and the strain time histories were recorded under constant impact energy in order to determine dynamic response of RCP under impact loading. The following conclusions can be drawn from the experimental study that was conducted:

- (1) The peak impact force increased with the increase in reinforcement ratio. The impact force reached its peak value immediately after the impact, but the displacement, concrete strain and steel strain reached their peak value a few microseconds later. Therefore, the peak impact force cannot be directly considered the true impact resistance capacity.

- (2) The increase in slab thickness resulted in an increase in the peak inertia force, but it decreased the peak impact force. Moreover, the energy consumption ratio reached 91.5% in 6MR specimen, which been severely damaged.
- (3) All RCP specimens had similar crack patterns on the bottom surface, and the number of cracks decreased with the increase in the slab thickness. The reinforcement arrangement could affect the crack pattern; circumferential cracks on the top surface appeared in the 7M1R slab with 100 mm reinforced spacing, and similar cracks were not found in the 7MR slab with 150 mm reinforced spacing.
- (4) The reinforcement ratio played an important role in peak deflection and residual displacement. As the reinforcement ratio increased from 0.48% to 0.72%, the peak and residual displacements at D1 point decreased by 7.0% and 12.2%, respectively. The global flexural response could be observed in the RCP specimens. Microscopic punching shear failure modes were observed only in the 8MR and 7M1R specimens.
- (5) The 7MN specimen showed lower peak and residual displacement and higher peak impact force compared to the 7MR specimen, but no significant difference was observed between damage characteristics and crack patterns in the 7MR and 7MN specimens.
- (6) The influence of using RAC in RCP was relatively small, even at 100% RCA replacement ratio, and the impact of using RCA was diminished for RCP made with 100 mm longitudinal reinforcement spacing.

However, due to the limited investigation conducted here, further research is being recommended to increase the database of test results for the RCP. This study was designed so that the RCP impact occurred at the center of the slab; however, in the real world, the impact can take place at other locations as well, and the response of the slabs under such conditions may be significantly different.

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Article

Study on the Penetration Characteristics of Water Entry Rod Projectile into Liquid Cabin at an Attack Angle

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Abstract: The penetration of a projectile into a warship broadside liquid cabin is usually a non-ideal penetration process. To explore the protective effects of the broadside liquid cabin of a large warship against the non-ideal penetration of rod projectiles and to provide reference for the design of new liquid cabin structures, ballistic impact tests of rod projectiles penetrating the liquid cabin at different attack angles were carried out. Combined with numerical calculation, the impact of the attack angle on the water entry and penetration characteristics of the projectile into the liquid cabin as well as their failure modes were studied. The overturning and yawing of the projectile in water were analyzed. The pressure load characteristics in the liquid cabin and the deformation/failure modes of the projectile and the liquid cabin were identified. The results showed that: multiple overturning and yawing occur in the projectile with an initial attack angle during penetration into liquid; the yaw direction is mainly affected by the initial attack angle and projectile attitude; the projectile mainly undergoes four basic failure modes, namely, asymmetric mushrooming at the projectile nose, side erosion, overall plastic bending and fracture; the actual failure of the projectile is a combination of the basic failure modes; the overall plastic bending and fracture are mainly related to the length to diameter ratio, initial attack angle and initial projectile velocity; the front plate of the liquid cabin may undergo tearing along the central plastic hinge line of the plate: at a small attack angle, the tear is “I” shaped, and at a large attack angle, it is “X”-shaped.

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Keywords: rod projectile; angle of attack; penetration; liquid cabin; overturn and yaw; trajectory

1. Introduction

To improve the penetration and destructive capabilities of projectiles, anti-armor weapons are widely equipped with rod projectiles with large length to diameter ratio (L/D ratio), strong anti-interference ability, good flight stability, high energy density and strong armor-piercing ability [1], whereas underwater weapons mostly adopt explosively formed projectiles (EFPs) [2,3]. As the warship broadside can be easily attacked by anti-ship weapons due to its large area, liquid cabin structures are often set on the broadside to protect the inner structure against penetration or armor-piercing damage caused by fragments and debris generated by the warhead shell and the outer plate of the cabin [4]. However, before penetrating into the liquid cabin, the projectile inevitably interacts with the stiffened plate and other obstacles in the empty cabin, changing the ballistic characteristics such as attitude angle of the projectile or causing asymmetric deformation of the projectile, and thus affecting its penetration capability [5]; when attacking underwater targets, torpedoes and projectiles usually enter the water at an attack angle; thus, it is of great significance to study the non-ideal water entry of rod projectiles for warship protection.

Generally, the water entry of projectiles can be divided into four stages: (1) Impact stage. As a projectile impacts the water at a high speed, causing a large impact force on the projectile, the projectile nose is prone to mushrooming deformation, and meanwhile the impact by the projectile leads to the formation of high-speed shock waves in the water, which propagates in a semicircle [6]. (2) Flow formation stage. After being impacted, the

water will be separated by the projectile and flow away from its sides as the projectile moves forward, resulting in cavitation. Whip can be easily observed in this stage, where the pressure on the projectile is far less than that in the impact stage. (3) Open cavity stage. The cavity generated after the water entry of the projectile does not disappear immediately. It is still connected with the air at the water surface. As the projectile moves, the cavity gradually expands. When the projectile moves forward in the cavity, the uneven contact between the projectile nose and the water cause projectile swinging, resulting in ballistic instability. At the same time, due to tail swing, the tail contacts the cavity, causing tail flap, which further increases the uneven force on the projectile, leading to greater yaw [7,8]. (4) Closed cavity stage. After formation for some time, the cavity closes at a point on or below the water surface. Cavity closure may generate jet. When the jet strikes the cavity wall and causes its deformation, or hits the projectile and changes its trajectory, the projectile will detach from the cavity, leading to a significant change in the direction of projectile movement [9,10].

At present, the water entry problem of projectiles is generally studied from two perspectives: a projectile penetrating the free water, such as the water entry of aerial torpedo and spacecraft; a projectile penetrating the liquid tank, such as the penetration of a projectile into an aircraft fuel tank, and fragments penetrating the broadside liquid cabin. Researchers have conducted extensive research on the penetration of rod projectiles under ideal conditions.

In terms of the ideal penetration of projectiles into free water, Karman [11] proposed the added mass method to calculate the water entry impact load, and derived the formula of the water entry impact load using the conservation of momentum; Wagner [12] considered the lifting of water surface based on Karmans work, and introduced the wave influence factor to optimize the theory; Cointe [13] established a two-dimensional water entry impact model through progressive matching; Takagi [14] employed the potential method to accurately calculate the added mass, water entry velocity and penetration depth, which were in good agreement with the test results; Mojtaba [15] considered the transient model of cavity shape and established a complete model of the water entry of cylindrical projectile. Alekseevskii [16] and Tate [17] built a theoretical analysis model of the penetration of long-rod projectiles into semi-fluid medium. However, the water entry of projectile is a three-dimensional phenomenon. Both the fluid flow and projectile movement are asymmetric, and the forces involved are quite complex [18]. Chen [19] conducted experiments and found that projectile tumbling and yaw occurred in the cases of both vertical and oblique water entry of the projectile. Projectile tumbling occurs due to the change of the pitch angular velocity when the hydrodynamic force acting on the projectile nose does not pass through the center of gravity of the projectile, resulting in the so-called “whip” phenomenon [20]. Li Tianxiong [21] verified by numerical simulation that symmetrical projectiles also undergo ballistic yawing during vertical water entry.

With respect to the ideal penetration of a projectile into a liquid-filled structure, this process involves not only the interaction between the projectile and water, but also the interaction between water and the liquid tank. As the projectile penetrates the liquid tank, its energy is transferred to the water and tank structure, causing high pressure on them. This phenomenon is called hydrodynamic ram (HRAM) [22]. Researchers have studied the HRAM effect caused by impact on liquid-filled structures from various aspects: (1) Pressure load characteristics. Shi [23] verified that the rise time of initial shock waves under the HRAM effect is in microseconds; Gao [24] experimentally demonstrated that cavitation load is the dominant factor of liquid-filled cell failure; Li [25–27] summarized the load characteristics of rod projectiles penetrating the liquid-filled structure through penetration test and numerical calculation. Then, according to the load characteristics, he divided different areas to establish a simplified calculation model, and compared the structure protection performance; Disimile [28] adopted high-speed photography to show the generation of pressure waves and how the cavitation region expands and collapses. (2) Remaining projectile characteristics. Deletombe [29] proposed that the

projectile will undergo overturning, deformation or even fracture when interacting with the water medium; Shen [30–32] analyzed the rule of the mushrooming deformation of the projectile and its influence on penetration resistance by tests and numerical calculations of fragments penetrating the liquid cabin, and proposed the formulas for calculating penetration resistance and velocity considering the influence of projectile deformation. (3) Structural deformation and failure. Artero and Nishida [33,34] found through tests and numerical calculations that water will cause secondary damage to the fluid-filled tubes, thus reducing the structural strength, and that the damage to the rear plate is more serious; David [35] conducted high-velocity impact test at 1000–3000 m/s with fluid-filled aluminum alloy containers, and the wall plate exhibited petal cracking.

In terms of water entry under non-ideal penetration conditions, which includes oblique water entry, water entry with an attack angle and water entry of asymmetric projectiles, currently, a large number of studies are focused on the water entry of asymmetric projectiles and oblique water entry. Takashi Isobe [36] performed oblique water entry tests using projectiles of various shapes, and had the following findings: when the projectile nose is moving in the water, the fluid exerts a lift force on it because of the cavitation caused by projectile motion; due to the pressure difference between the upper and lower surfaces of the projectile, cavitation occurs on the two surfaces is not symmetrical, and then the amount of water displaced by the projectile on the free surface side is less than the other side; thus, the pressure on this side is lower, resulting in asymmetric pressure on the upper and lower surfaces, thus exerting a lift force on the projectile and causing projectile ricochet; the nose shape of the projectile greatly affects the ballistic stability in water: the flat-nosed projectile has better ballistic stability underwater, the sharp-nosed projectile is prone to ballistic instability, and the ogive-nosed projectile tend to overturn in the early stage of water entry [37–40]; compared with symmetric nose, after the asymmetric nose enters the water, a non-axial component exists in the hydrodynamic force on the nose [41], and an overturning moment acts on the projectile, changing its attitude angle. This further affects the hydrodynamic force on the projectile, leading to the nonlinear increase in the overturn and yaw of the projectile [42,43]. The more asymmetric the projectile nose is, the worse the attitude and trajectory stability of the projectile is, and the projectile is prone to instability and inclination, which leads to yaw.

There are much less studies on the water entry by a projectile at an attack angle. Li [44] used numerical calculation to simulate the vertical water entry by a projectile at a small attack angle, and found that under certain attack angles, the projectile tail contacts with the cavity, thus reducing the ballistic stability, and that the spinning of the projectile in water has little impact on ballistic stability. However, Truscott [45,46] found that the spinning motion of the projectile induces a lateral force on the projectile, resulting in a curved trajectory. Yao [47] studied the water entry by the underwater vehicle through numerical simulation and revealed that positive attack angles suppress projectile whipping, whereas negative attack angles aggravate this phenomenon; therefore, ballistic change easily occurs under negative attack angles. Liang Jingqi [48] conducted a study using the LS-DYNA program and found that greater attack angle leads to faster axial velocity attenuation of the projectile, and thus greater overturning angle and projectile velocity. Wang Zhen [49] studied the oblique water entry of projectile at small attack angles with LS-DYNA, and discovered that the attack angle determines the direction and magnitude of the moment on the projectile nose.

To sum up, detailed investigations on the residual characteristics, pressure load characteristics and structural deformation of projectiles have been conducted through the ideal penetration test of the water-entry projectile. During the oblique water entry, entry with an angle of attack and water entry of an asymmetric projectile, ballistic yaw will occur. Among these cases, the penetration at an attack angle has been less investigated, and numerical calculation is the main method adopted. Few experimental studies have been carried out on the water entry by high-velocity projectiles with attack angles because the initial attack angle is difficult to realize in ballistic tests and is not conducive to ballistic control.

Therefore, in the present study, the tests of high-velocity rod-shaped projectiles penetrating the liquid cabin at various attack angles are performed, and the overturning and yawing of the projectile in water, the pressure load characteristics and the deformation/failure modes of the projectile and the liquid cabin are analyzed by combining experimental tests with numerical calculations.

2. Experimental and Numerical Methods

2.1. Ballistic Experiment Design

2.1.1. Water Tank Model

Two types of cylindrical rod projectiles with a diameter of 14.5 mm and lengths of 23.5 mm and 29 mm are employed in the test. The projectile material is 45# steel.

As shown in Figure 1, the water tank is designed with a volume of $300 \times 400 \times 600$ mm, with no cover on the upper side to simulate the free surface of the liquid cabin of an actual warship. The front and rear plates of the water tank have flange structures to facilitate the connection with the target. The front and rear plates and the pressplate are connected with the water tank using bolts with a diameter of 15 mm. A rubber pad is added between the target plate and the water tank to keep water tightness. One side of the water tank is made of 30 mm thick polymethyl methacrylate (organic glass) for the convenience of high-speed photography. On the other side, 50 mm gridlines are drawn for the calculation of the position and velocity of the projectile in water. The front plate is 2 mm thick, and the rear plate 4 mm thick. The material of the water tank and front/rear plates is Q235 steel.

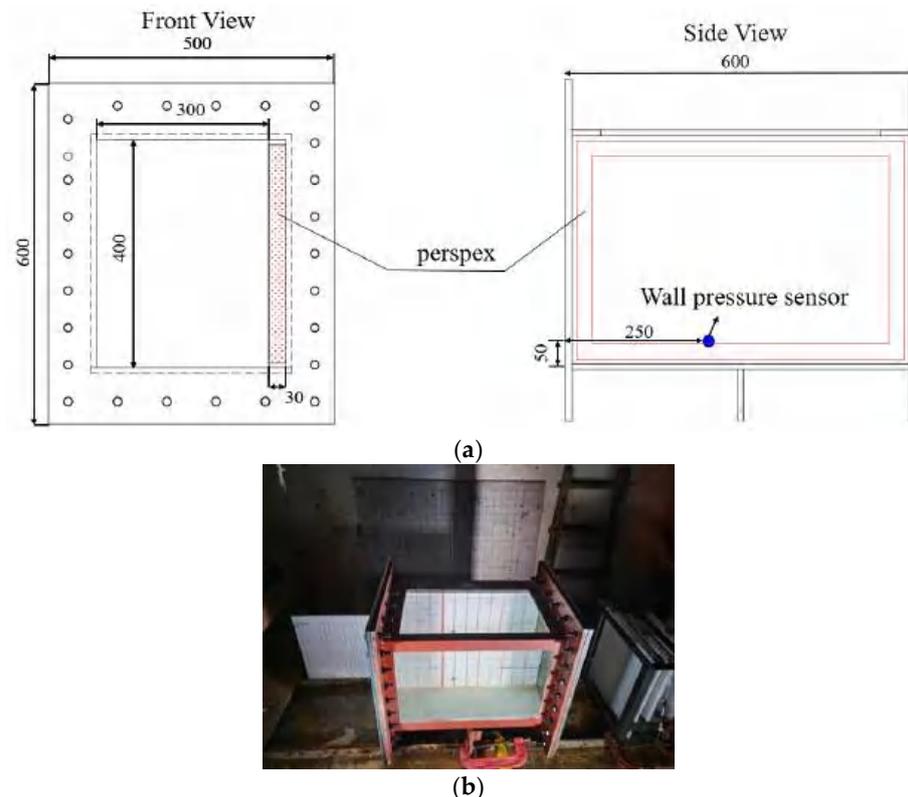


Figure 1. Diagram and photo of the water tank model. (a) Diagram of water tank model. (b) Water tank model.

2.1.2. Experimental Equipment and Setup

A 14.5 mm ballistic gun is used to fire the projectiles (bullets). The sabot is designed to be not closely fit with the projectile, so that initial disturbances can be generated after the projectile exits the bore, leading to the formation of the initial attack angle as the projectile impacts the target. The initial attack angle is obtained by the combination of high-speed

photography and analysis of the breach morphology of the front wall of the water tank. A sabot collector is set to avoid the influence of sabot on the tests.

A laser speed measuring system is installed at the front side of the water tank (liquid cabin) to measure the initial velocity of the projectile (Figure 2a), and a target-net speed measuring system and a projectile recovery box are set at the rear side to measure the residual velocity of the projectile and recover the remaining projectile, respectively. The laser speed measurement system uses Chengdu Test TST6260 transient signal tester, the maximum sampling can reach 20 Msp/s.



Figure 2. Test equipment and recording devices. (a) Laser velocimeter; (b) high-speed camera, pressure sensor and target-net velocimeter.

The X213 high-speed camera made by Revealer Company in Hefei, China, is used to observe the projectile attitude in water and the cavitation process (Figure 2b). It shoots at 10,000 frames per second, i.e., one picture is taken every 100 μ s.

To measure the pressure load in water, a wall pressure sensor with a measuring range of 0–68 MPa and a sampling frequency of 1 MHz is set at a position 5 mm from its center to the bottom of the water tank (Figure 1a).

The experimental setup is shown in Figure 3.

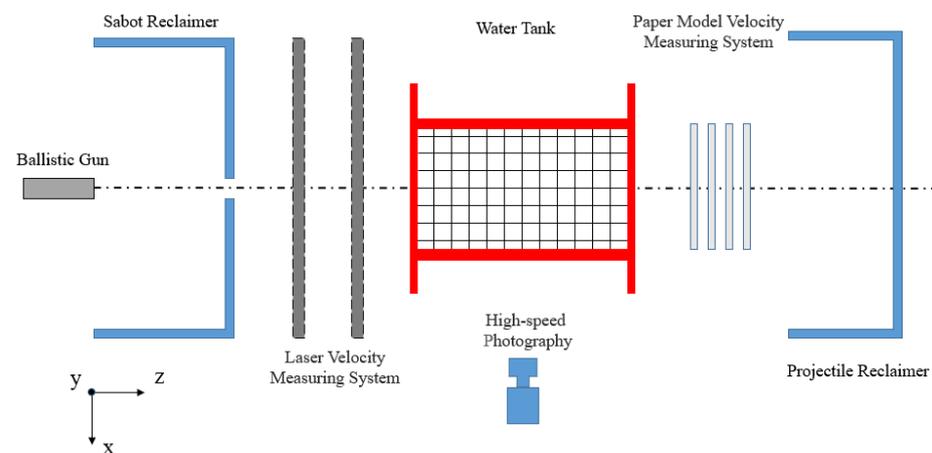


Figure 3. Experiment setup.

2.2. Brief Experimental Results

Three penetration tests are conducted with the water tank, and the initial physical state and initial projectile velocity are recorded. Since the rear plate of the water tank have been perforated, the residual velocity of the projectile is not measured. After the tests, the residual projectile mass and its displacement in all directions are obtained through measurements. The test conditions and results are shown in Table 1.

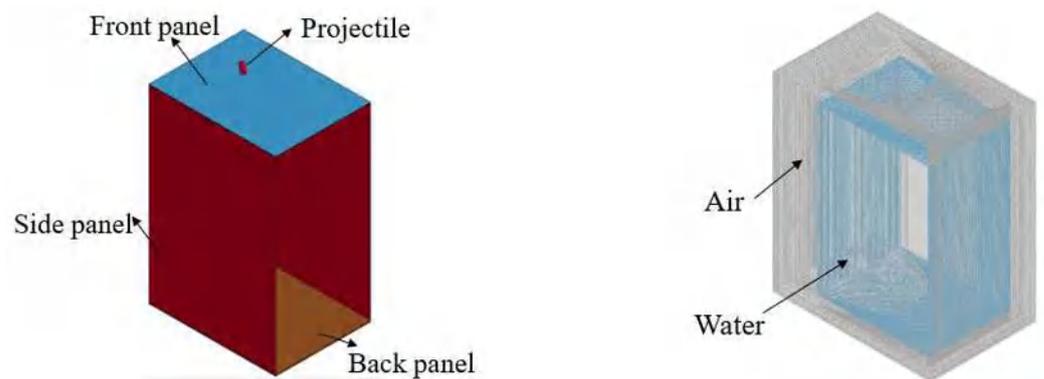
Table 1. Test conditions and results.

Test No.	Y-Axis Attack Angle (°)	Z-Axis Attack Angle (°)	Total Attack Angle (°)	Projectile Length	L/D Ratio	Projectile Weight M (g)	Initial Velocity V_0 (m/s)	Residual Weight of Projectile M (g)	Displacement cm		
									X	Y	Total
1	−30	−40	−50	23.5	1.6	29.9	1016.0	29.08	3.3	−1.9	3.8
2	−15	66	67	23.5	1.6	29.3	991.1	29.56	−0.6	6.8	6.83
3	25	38	45	29	2	37.3	1004.9	36.54	3.5	5.6	6.6

2.3. Numerical Calculation Model and Effective Verification

2.3.1. Numerical Model

The water tank model is simplified as shown in Figure 4. The models are built with SolidWorks, and Hypermesh is used to generate grids. The solid structures such as the projectile and target plate use Lagrangian solid elements; the side walls of the water tank do not involve damage and deformation; therefore, Lagrangian shell elements are used for them. The water domain is constructed in the water tank, the air domain is established outside the water tank; Eulerian elements are adopted for the water and air domains, and the common node approach is applied, so as to realize the flow of water and air.

**Figure 4.** Numerical model of water tank and Eulerian domains.

The projectile is divided into 14 equal parts in the radial direction, 16 equal parts in the circumferential direction, and 1.2 mm grids along the length direction. The grid of the water and air domains is 4 mm in size, and the grids are refined in the central area with a size of 2 mm. The grids of the front and rear plates of the water tank are divided into 1 mm cube elements. Because side walls are thick and can be regarded as rigid walls, their grid size is 5 mm.

2.3.2. Numerical Calculation Method

As the numerical calculation involves the fluid flow and the interaction between fluid and structure, the fluid-solid coupling algorithm in the LS-DYNA R10 software is used. The LAGRANGE_IN_SOLID card is used to realize the structure-fluid coupling. Since gravity cannot be ignored during the projectile's water entry, the LOAD_BODY command is employed to set the gravity field in the Eulerian domain, and the INITIAL_HYDROSTATIC_ALE command is set to define the hydrostatic pressure in the water domain. The contact between the projectile, the steel plate and water tank are set to ERODING_SURFACE_TO_SURFACE.

2.3.3. Material Model

(a) Q235 steel

The water tank and steel plate are made of Q235 steel. The Cowper-Symonds constitutive model is adopted, and its dynamic yield strength σ_d is:

$$\sigma_d = \left(\sigma_0 + \frac{EE_h}{E - E_h} \varepsilon_p \right) \left[1 + \left(\frac{\dot{\varepsilon}}{D} \right)^{1/n} \right] \tag{1}$$

where σ_0 is the static yield strength, E is the elastic modulus, E_h is the strain hardening modulus, $\dot{\varepsilon}$ is the equivalent strain rate, and D and n are the strain rate parameters. Table 2 lists the material parameters.

Table 2. Material parameters of Q235 steel.

Parameter	Value	Parameter	Value	Parameter	Value
Yield strength σ_0 /MPa	235	n	5	Failure strain ε_f	0.28
Strain hardening modulus E_h /MPa	250	D (s ⁻¹)	40.4		

(b) 45# steel

The material of the projectile is 45# steel, and Johnson–Cook constitutive model is selected:

$$\sigma = (A + B\varepsilon_p^n) \left(1 + C \ln \frac{\dot{\varepsilon}_p}{\dot{\varepsilon}_{p0}} \right) \left[1 - \left(\frac{T - T_0}{T_m - T_0} \right)^m \right] \tag{2}$$

where σ is the dynamic yield strength of steel, ε_p is the plastic strain, A is the static yield limit, B is the strain hardening modulus, n is the strain hardening index C is the strain rate coefficient, $\dot{\varepsilon}_{p0}$ is the critical strain rate, m is the thermal softening index, T is the temperature, T_m is the melting point of the material, and T_0 is the room temperature.

The J-C failure model is adopted to describe the failure of the materials:

$$\varepsilon_f = \left\{ D_1 + D_2 \exp \left[D_3 \frac{\sigma_h}{\sigma_{eff}} \right] \right\} \left[1 + D_4 \ln \dot{\varepsilon}^* \right] (1 + D_5 T^*) \tag{3}$$

where D_1 – D_5 are material parameters, σ_{eff} is the von Mises stress, σ_h is the hydrostatic pressure of the material under the triaxial stress, $T^* = (T - T_r)/(T_m - T_r)$ is the dimensionless temperature, T_r is the room temperature, and T_m is the melting point of the material.

The material parameters of the projectile are presented in Table 3.

Table 3. Mechanical parameters of the projectile.

Parameter	Value	Parameter	Value	Parameter	Value
shear modulus G /GPa	80.8	C	0.0483	T_m /K	1793
A /MPa	335	m	0.804	T_0 /K	300
B /MPa	350	D_1	0.8	D_2	0.76
n	0.782	D_3	1.57	D_4	0.005
C_v /J·(kg·K) ⁻¹	477	D_5	-0.84		

(c) Water

The Grüneisen equation of state (EOS) is adopted for water:

$$P = \frac{\rho_0 C^2 \mu [1 + (1 - \frac{\gamma_0}{2})\mu - \frac{a}{2}\mu^2]}{\left[1 - (S_1 - 1)\mu - S_2 \frac{\mu^2}{\mu+1} - S_3 \frac{\mu^3}{(\mu+1)^2}\right]^2} + (\gamma_0 + a\mu)E \tag{4}$$

where ρ_0 is the density, C is the speed of sound, γ_0 is the Grüneisen parameter, $\mu = \rho/\rho_0 - 1$, a is volume correction, and S_1, S_2 and S_3 are curve fitting parameters.

(d) Air

When air is an ideal gas without viscosity, its EOS is a linear polynomial:

$$P = C_0 + C_1\mu + C_2\mu^2 + C_3\mu^3 + (C_4 + C_5\mu + C_6\mu^2)E \tag{5}$$

where C_0-C_6 are the parameters and E is the internal energy. Let $C_0 = C_1 = C_2 = C_3 = C_6 = 0$, $C_4 = C_5 = \gamma - 1$, so that it has ideal gas characteristics, where γ is the adiabatic exponent. The material parameters of water and air are shown in Table 4.

Table 4. Material parameters of fluid.

Parameter	ρ_0 (kg/m ³)	ν_d	C (m/s)	S_1	S_2	S_3	γ_0	a	C_4	C_5	E_0 (J/m ³)
water	1000	0.89	1448	1.98	0	0	0.11	3			0
air	1.22								0.4	0.4	2.53×10^5

2.3.4. Verification of Calculation Results

Numerical calculation is performed according to the attack angle and initial projectile velocity measured in the tests. The calculation results are compared with the experimental data. With the measured projectile positions in water at different times, the corresponding velocities can be calculated and fitted. The velocity comparison is shown in Figure 5, and the maximum error is within 15%. It can be considered that the velocity attenuation trend of the numerical calculation is in good agreement with the experimental results.

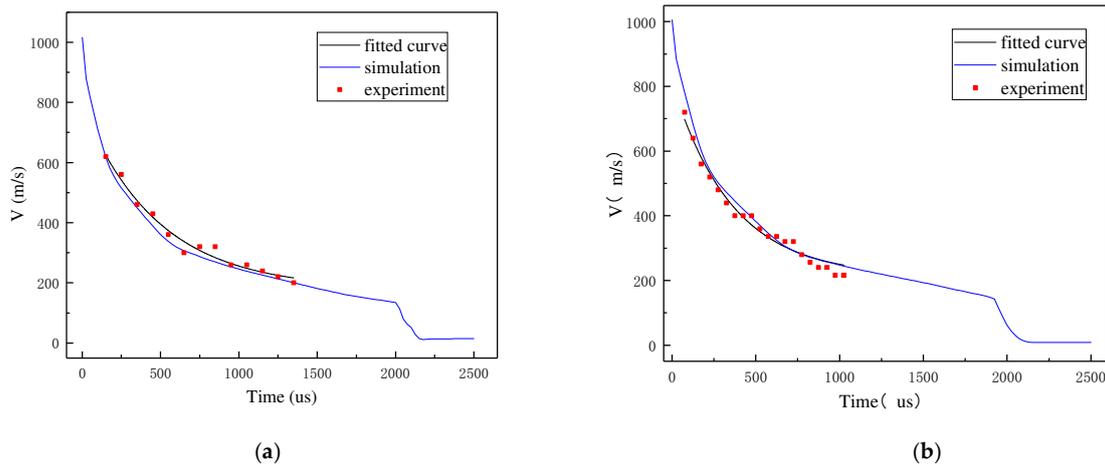


Figure 5. Comparison of tested and simulated velocities. (a) Test 1 velocity curve; (b) test 3 velocity curve.

Figure 6 gives a comparison between the measured load curve in the projectile water entry test and numerically calculated loads. The shock wave peak overpressure measured in Test 1 is 19.1 MPa, and the initial shock wave pressure in the simulation is 15.6 MPa, with an error of 18%, whereas the error between simulation and experimental results in Test 3 is 22%. The reason for this difference is: the grids in the water domain are slightly larger, so the shock wave attenuation speed is faster than the actual situation; however, reducing the

grid size will increase the computing time at a geometric rate. Therefore, it is considered that the numerical calculation gives reasonable results on the premise of ensuring accuracy, and the error is within the acceptable range.

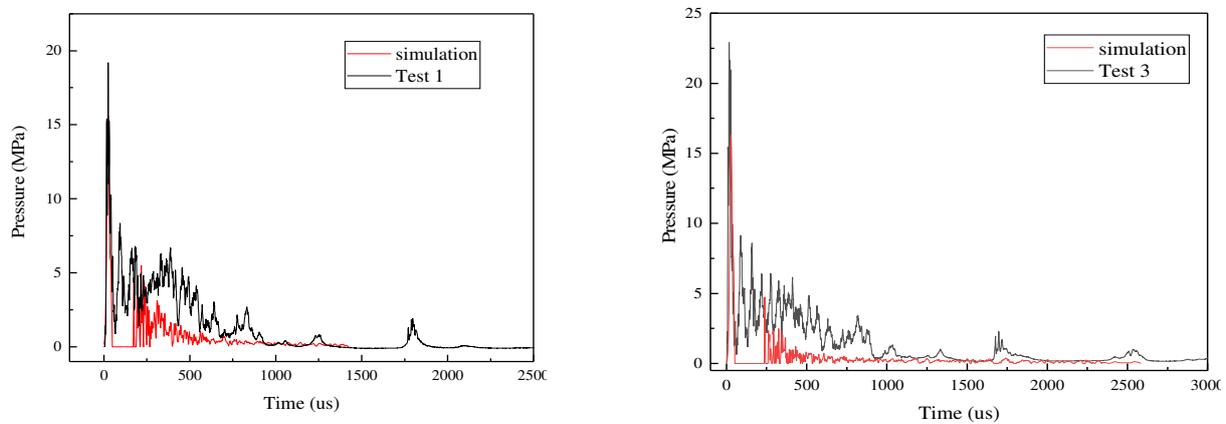


Figure 6. Comparison of experimental and simulated loads.

Figure 7 and Table 5 show a comparison of the remaining projectile after the test and numerical simulation. The side erosion unique to the water entry of the projectile at an attack angle and the asymmetric shape of the nose are successfully simulated, and the error between the calculated and experimental residual mass is very small, demonstrating that the failure modes of the projectile obtained by numerical calculation agrees well with the experimental data.



Figure 7. Comparison of projectile failure morphology between the test and numerical simulation.

Table 5. Comparison between experimental and calculated residual mass of the projectile.

Test No.	Projectile Mass M (g)	Residual Mass of Projectile in Test M (g)	Residual Mass of Projectile in Simulation M (g)	Error Value (%)
1	29.9	29.08	28.24	2.84
2	29.3	29.56	29.03	1.81
3	37.3	36.54	36.52	0.05

To sum up, the proposed numerical calculation methods and numerical models can well simulate the water entry of the projectile, and check with the experimental data, striking a good balance between accuracy and efficiency.

2.4. Test Conditions

To study the influence of the initial attack angle and initial velocity on the deformation morphology of the liquid cabin and the projectile after the penetration of the projectile into the liquid cabin, 30 working conditions for the projectile with a diameter of 14.5 mm and a length of 69.6 mm (L/D ratio = 4.8) penetrating the liquid cabin are investigated using the

numerical calculation method. TEST/FEM indicates that ballistic impact test is used in this working condition, supplemented by numerical calculation method, whereas FEM uses the finite element algorithm verified in Section 2.3 for calculation. The specific working conditions are listed in Table 6.

Table 6. Test conditions.

Condition No.	Initial Velocity V_0 (m/s)	Attack Angle ($^\circ$)	Test Method	Condition No.	Initial Velocity V_0 (m/s)	Attack Angle ($^\circ$)	Test Method
No. 1	1016	−50	TEST/FEM	No. 18	1200	60	FEM
No. 2	991	67	TEST/FEM	No. 19	1600	0	FEM
No. 3	1004	45	TEST/FEM	No. 20	1600	15	FEM
No. 4	400	0	FEM	No. 21	1600	30	FEM
No. 5	400	15	FEM	No. 22	1600	45	FEM
No. 6	400	30	FEM	No. 23	1600	60	FEM
No. 7	400	45	FEM	No. 24	2000	0	FEM
No. 8	400	60	FEM	No. 25	2000	15	FEM
No. 9	800	0	FEM	No. 26	2000	30	FEM
No. 10	800	15	FEM	No. 27	2000	45	FEM
No. 11	800	30	FEM	No. 28	2000	60	FEM
No. 12	800	45	FEM	No. 29	2400	0	FEM
No. 13	800	60	FEM	No. 30	2400	15	FEM
No. 14	1200	0	FEM	No. 31	2400	30	FEM
No. 15	1200	15	FEM	No. 32	2400	45	FEM
No. 16	1200	30	FEM	No. 33	2400	60	FEM
No. 17	1200	45	FEM	No. 34	1200	8	FEM

3. Results and Analysis

3.1. *Overturning and Yawing during Underwater Penetration of Projectile at an Attack Angle*

Figure 8 shows the photos of the three groups of tests taken by the high-speed camera. After the projectile enters the water with an attack angle, the cavitation region generated by its high-velocity motion is asymmetric and curved. This is because the penetration attitude and trajectory of the projectile with an attack angle are not stable in water, and the penetration direction of the projectile at the moment it enters the water is related to the initial attack angle of the projectile. In the case of positive initial attack angles, the projectile tends to yaw upward such as in Tests 2 and 3; in the case of negative initial attack angles, the projectile tends to yaw downward, such as in Test 1 (the anticlockwise rotation of the attack angle about the axis is defined as positive attack angle and the clockwise rotation as negative one).

The attitude angle of the projectile in water is an important factor affecting the yaw of the projectile. Taking Test 3 as an example, it can be clearly observed from the photos that the attitude of the projectile keeps changing, overturning anticlockwise first and then clockwise. The attitude angle of the projectile on the XY plane, its Y-axis velocity and displacement are read from numerical calculations (Figures 9–11). When the projectile enters the water with its nose up, the nose and lower side of the projectile are the main positions contacting with water. Because the lower side of the projectile is subjected to the dynamic pressure of the water, which is perpendicular to the contact surface, a lift force is exerted on the projectile, generating a Y-direction velocity, leading to the upward yawing of the projectile. At this time, the force exerted by water on projectile nose is the largest. Due to the existence of the attitude angle of the projectile, the force does not pass through the center of mass of the projectile, and the resultant force is on the upper side of the center of mass. This gives an anticlockwise overturning moment to the projectile, leading to the anticlockwise overturning of the projectile. At $t = 130 \mu\text{s}$, the projectile overturns to the maximum incident flow area (around 90°), and the resultant force on the projectile basically passes through its center of mass. However, as the projectile continues

to overturn anticlockwise due to inertia, the original “tail” of the projectile turns over to the “nose” position, and the direction of the resultant force on the projectile becomes downward. The Y-axis velocity of the projectile starts to decrease, and the resultant force is below the projectile’s center of mass; thus, the projectile is subjected to a clockwise overturning moment, which generates an angular acceleration in the opposite direction to the overturning direction, and the anticlockwise overturning velocity decreases gradually. At $t = 342 \mu\text{s}$, the angular velocity of the projectile declines to 0 rad/s , the projectile starts to overturn clockwise, and the Y-axis velocity of the projectile is increasing smaller. When the projectile overturns clockwise to the maximum incident flow area, the projectile starts to be subjected to an anticlockwise moment; thus, the angular velocity starts to decrease, the resultant force becomes upward, and the Y-axis velocity increases; finally, as the projectile contacts the back plate, the penetration ceases.

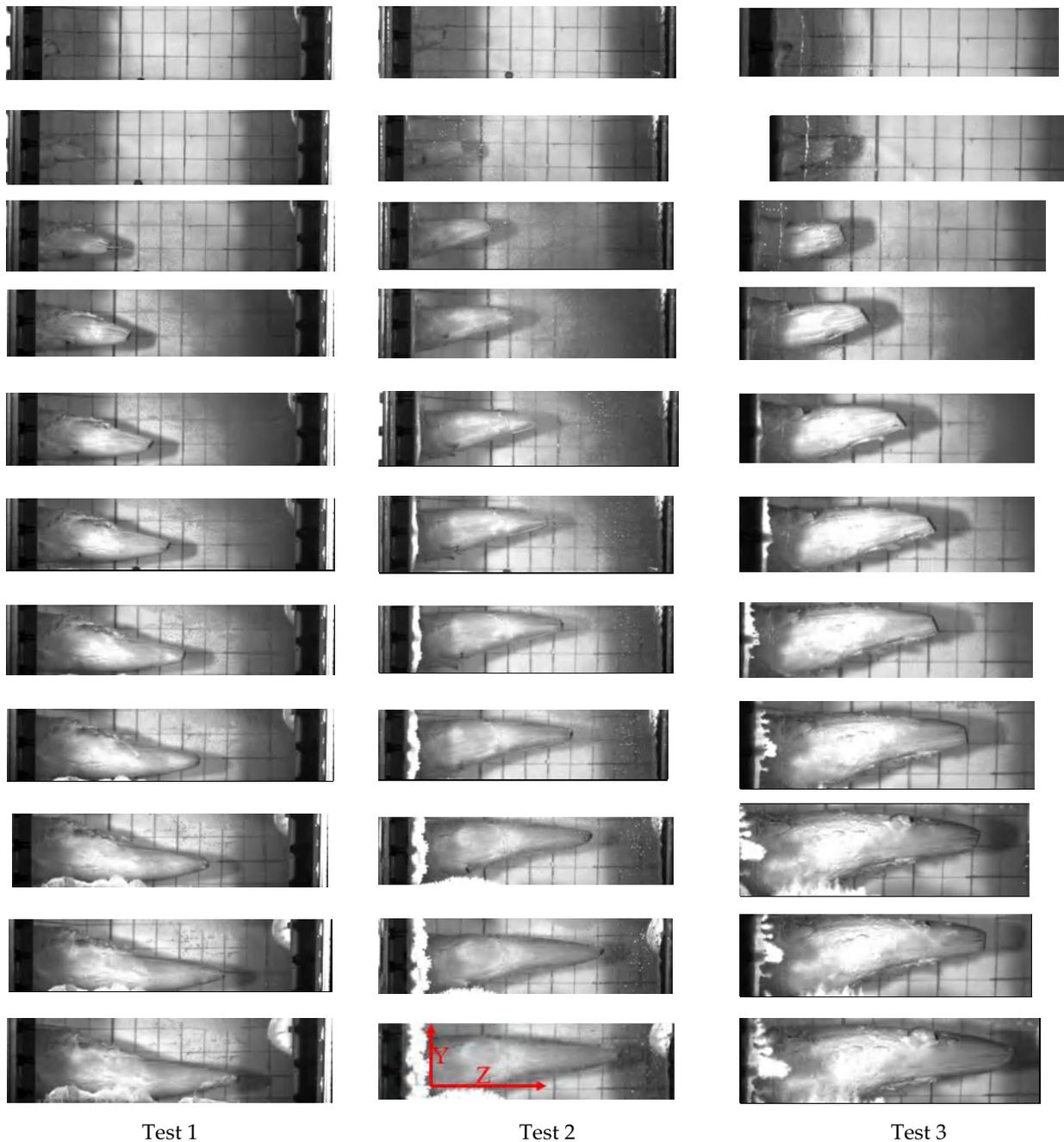


Figure 8. Photographs of underwater penetration process.

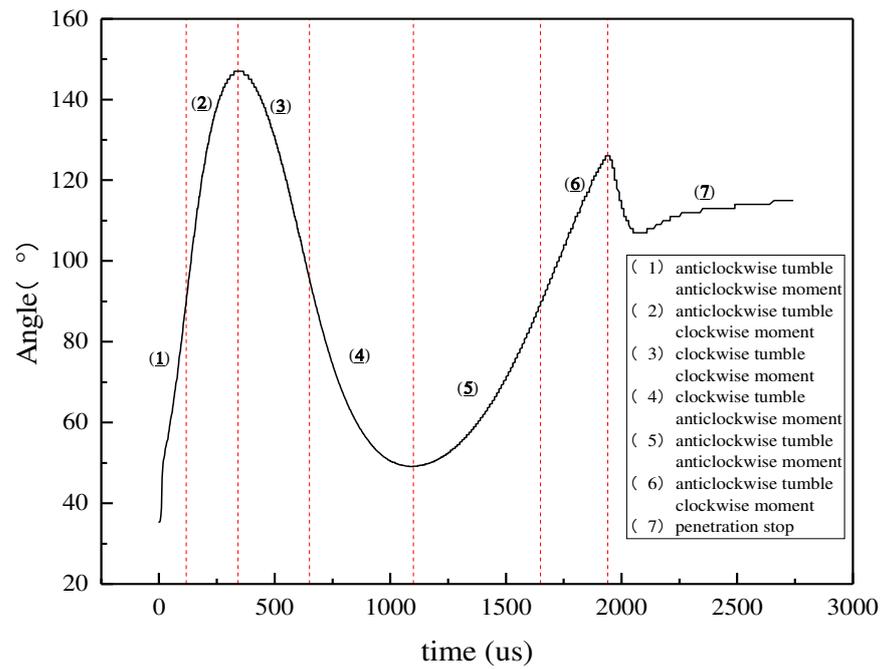


Figure 9. Attitude angle change of projectile in Test 3. (The red dash line is the dividing line of different stages).

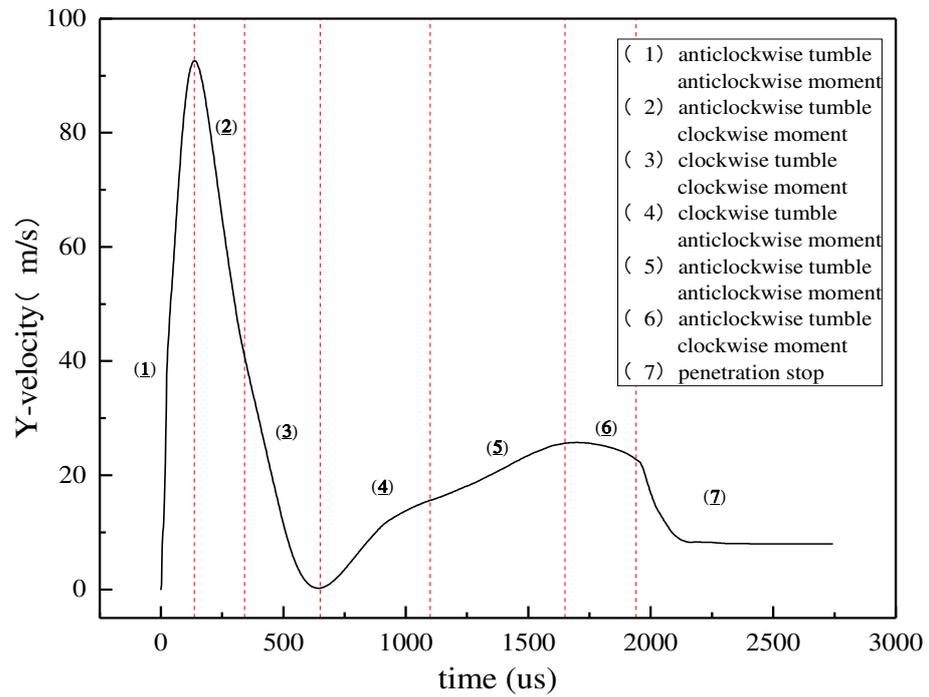


Figure 10. Y-axis velocity curve of projectile. (The red dash line is the dividing line of different stages).

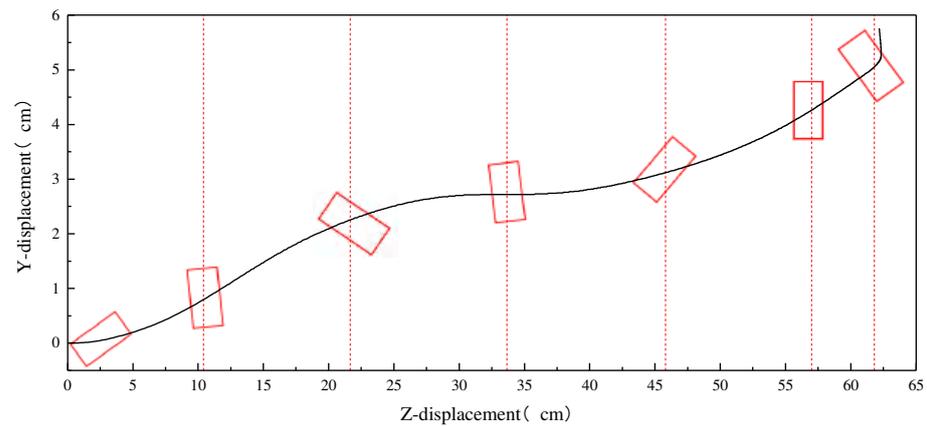


Figure 11. Trajectory of projectile in Test 3. (The red dash line is the dividing line of different stages).

In summary, during projectile penetration into liquid at an attack angle, the projectile is always in the overturning state. Due to the change of the projectile position contacting with water, the projectile will undergo overturning many times. The yaw of the projectile is affected by the attack angle and attitude angle as the two angles determine the direction of the force on the projectile during penetration. As the initial velocity during water entry is large, greater dynamic pressure on the projectile leads to greater Y-direction component. Therefore, the yaw velocity of the projectile is fast at the initial stage of water entry; when the second overturn occurs, the projectile velocity becomes smaller, and the Y-direction component of the projectile is also smaller; thus, the trajectory tends to be stable with a relatively small yaw velocity at the later stage of water entry.

3.2. Analysis of Residual Characteristics of Projectile Penetrating into Liquid Cabin at an Attack Angle

Figure 12 shows the deformation and failure morphology of the projectiles after the penetration tests under various attack angles. Mass abrasion occurs in both radial and axial directions of the projectile (the residual mass increases due to high-temperature fusion of the projectile and front plate fragments in Test 2). Compared with water entry under normal penetration (Figure 4d), the water entry projectile with an attack angle undergoes obvious asymmetric deformation. In the radial direction, one side of the projectile presents overall wavy erosion, whereas the other side shows no deformation; in the axial direction, one side of the projectile undergoes mushrooming deformation, and the other side suffers slight mass loss.

Adiabatic shear failure occurs during the high-velocity projectile impact on the target plate, generating a large amount of heat. As the heat is transferred to the projectile, the yield strength of the projectile is reduced, causing mushrooming and erosion to the projectile nose; the sides of the projectile with an attack angle also impacts the front plate, causing “strip” erosion. The impacted part of the front plate drives the nearby region to move backward, and the velocity of the plate exceeds that of the projectile. Then, a gap is produced between the projectile and the front plate after the initial impact, and as the projectile moves forward, its side impacts the front plate again, causing “strip” erosion to this side again; after multiple impacts, the side shows wavy erosion. After the projectile enters the water, one side and the nose of the projectile are impacted by the water, leading to erosion and mushrooming deformation of the projectile; due to the reduced velocity of the projectile after water entry, its deformation in water is relatively small; cavitation occurs on the other side of the projectile due to its high-velocity motion; the projectile on this side is always in the cavitation region and does not contact with the water; thus, deformation does not occur to this side; when the projectile impacts the rear plate, a small deformation occurs as the velocity at this time has been completely decayed.

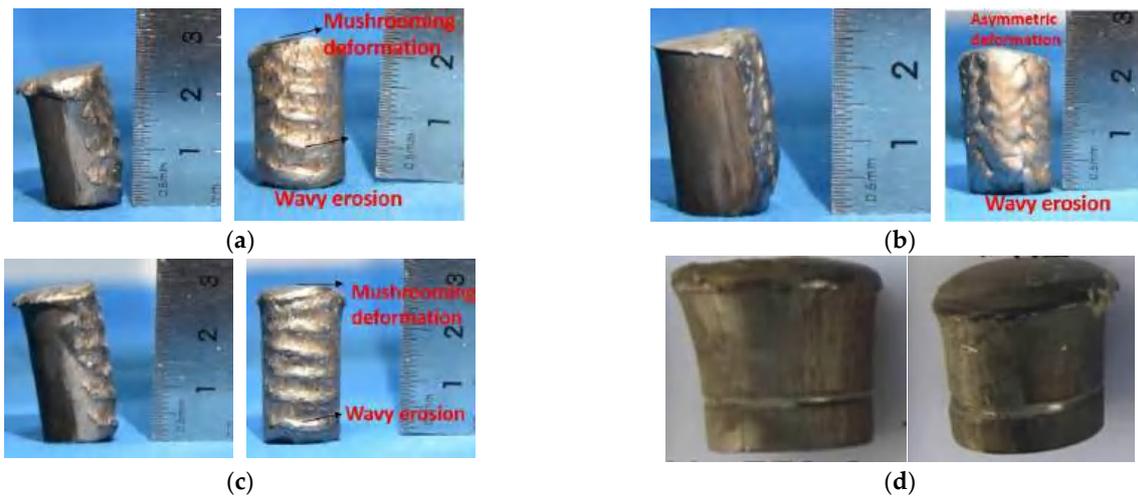


Figure 12. The remaining projectile after the tests. (a) Remaining projectile in Test 1; (b) remaining projectile in Test 2; (c) remaining projectile in Test 3; (d) remaining projectile under normal penetration [24].

In conclusion, when the initial velocity of the projectile is 1000 m/s, the penetration into the front plate takes the shortest time but is the main stage when projectile failure occurs, featuring a small deformation and mass abrasion of the projectile.

The failure morphology of the high-velocity rod projectile is studied by numerical calculation. The specific working conditions are shown in Figure 13.

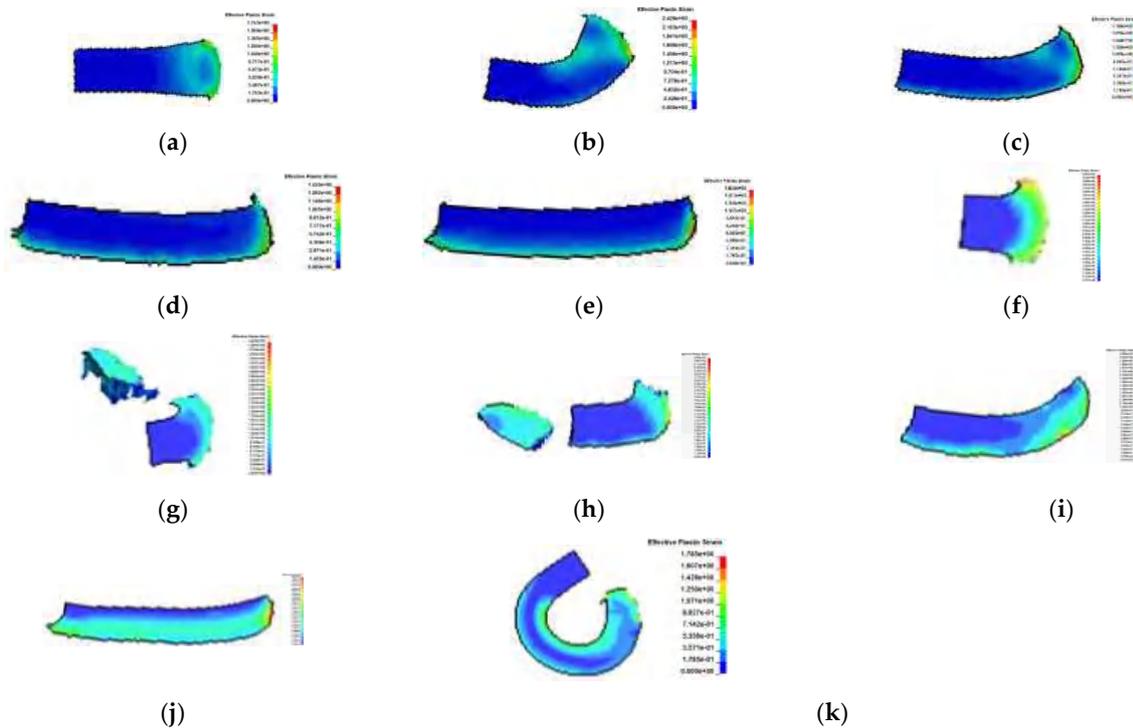


Figure 13. Projectile failure morphology under different working conditions (a) No. 19 erosion-mushrooming; (b) no. 20 overall plastic bending; (c) no. 21 overall plastic bending; (d) no. 22 side erosion; (e) no. 23 side erosion (f) no. 29 erosion—mushrooming; (g) no. 30 fracture; (g) no. 31 fracture; (i) no. 32 overall plastic bending; (j) no. 33 side erosion; (k) no. 34 bending.

As the front plate is thin, the penetration of the high-velocity projectile into liquid is the main stage of projectile failure. As can be seen from Figure 13, the projectile enters the

water at a high velocity, which causes great deformation of the projectile. When projectile water entry is under normal penetration, the projectile undergoes erosion-mushrooming deformation, and the greater the velocity, the more serious the erosion; when projectile water entry is under penetration at an attack angle, the projectile undergoes a large overall deformation at $15\sim 30^\circ$, and the overall plastic bending occurs at 1600 m/s. The reason is: an overturning moment is generated on the projectile under asymmetric loads, leading to the bending stress on the cross-section of the projectile; the critical section of the projectile reaches the yield limit under the combined action of axial compressive stress and bending stress; at this time, the plastic hinge line is formed in this section, leading to overall plastic bending deformation of the projectile; when the incident velocity is large enough, such as 2400 m/s, the critical section breaks directly, such as No. 30 and 31; however, when the initial attack angle is greater than 30° , only side erosion occurs during the high-velocity water entry of the projectile, and certain bending deformation also occurs when the incident velocity is high enough (No. 32); when the L/D ratio of the projectile is large enough, a large part of the projectile still undergoes deformation at low velocities and small attack angles; taking No. 33 as an example, the projectile with a L/D ratio of 10 at 1200 m/s undergoes bending when the attack angle is 8° . There are four failure modes of the projectile during its penetration into the liquid cabin: mushrooming, erosion, plastic bending and fracture. The failure phase diagram of the projectile penetrating the liquid cabin at different attack angles and initial velocities can be drawn. Figure 14 is the failure phase diagram of the penetration of the rod projectile with a L/D ratio of 4.8 into the liquid cabin.

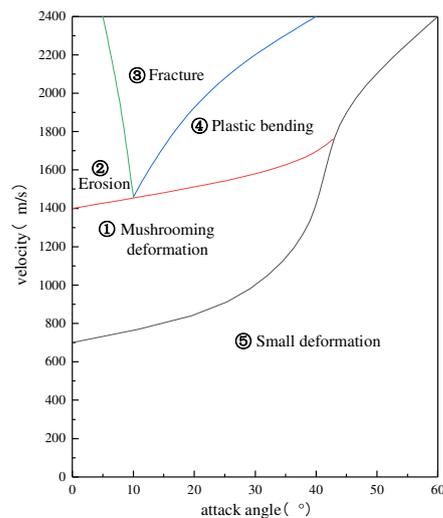


Figure 14. Failure phase diagram of the rod projectile with a L/D ratio of 4.8 penetrating into liquid cabin. ① Mushrooming ② Erosion (Condition 6) ③ Fracture (Condition 7) ④ Plastic bending (Condition 2).

In summary, at low projectile velocities, its penetration into the front plate is the main stage of failure, and at high velocities, the penetration into the liquid is the main stage of failure. The failure modes of the projectile are determined by its initial velocity, attack angle and L/D ratio. The projectile with an attack angle undergoes asymmetric deformation. When the attack angle is in the range of $15\sim 30^\circ$, overall deformation is most likely to occur. At low velocities, overall plastic bending deformation occurs and with the increase in velocity, fracture will occur. When the attack angle is greater than 30° , the failure mode of the projectile is side erosion, but with the increasing initial velocity of the projectile, the range of the attack angle that leads to the overall deformation of the projectile also expands. The larger the L/D ratio, the more easily the overall plastic bending occurs to the projectile.

3.3. Analysis of Pressure Load Characteristics of Liquid Cabin

Figure 15 shows the pressure load curve of the bottom of the middle position of the water tank measured in Test 3. As shown in the figure, the pressure loading on the side wall of the projectile during water entry can be divided into three stages: the initial shock wave stage, the cavitation loading stage, and the cavity collapse stage.

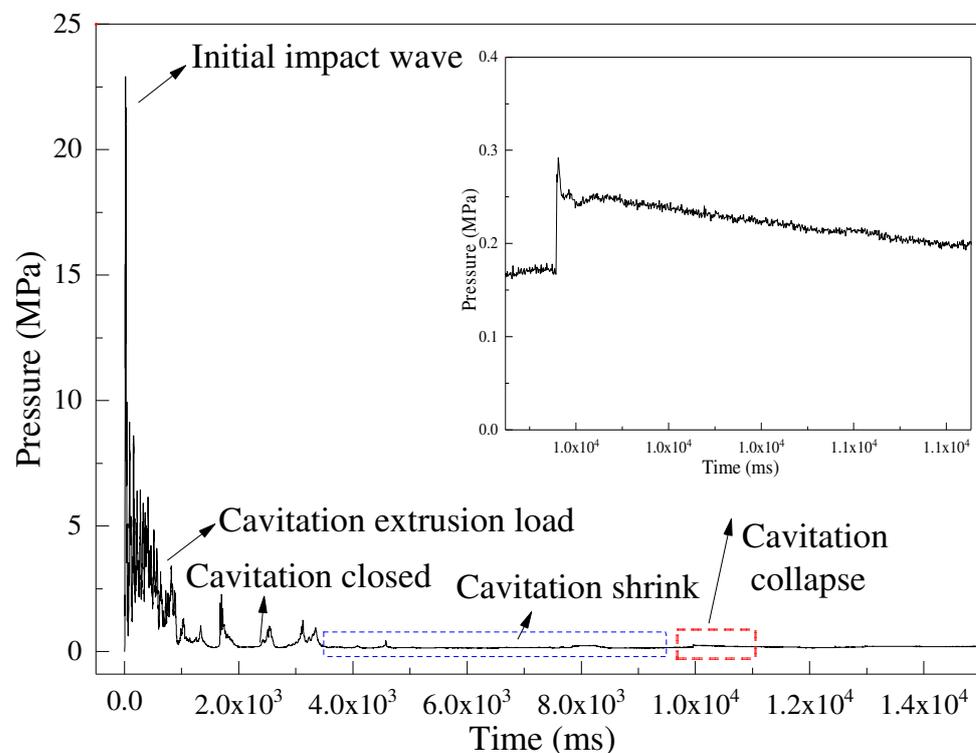


Figure 15. Load curve of measuring points in Test 3.

In the initial shock wave stage, the projectile perforates the front plate and then penetrates into the water, causing a huge acceleration of the previously static liquid relative to the projectile. This acceleration generates a shock wave that propagates in the water in an arc (Figure 16a). Peak value of initial pressure and cavitation load are shown in Table 7. The initial shock wave has the largest peak pressure, which is 19.1 MPa in Test 1 and 22.9 MPa in Test 3 through measurement. Since the L/D ratio and initial velocity of the projectiles in the two tests are different, the ratio of their kinetic energy is 0.821, and the ratio of the two initial shock wave peak pressures is 0.83. This indicates that under the same water tank structure and projectile shape, the initial shock wave peak pressure has a linear relationship with the kinetic energy of the projectile. The larger the kinetic energy, the larger the initial peak pressure.

Then, in the cavitation loading stage, the projectile penetrates into the water at a high velocity and displaces the water, which converts the kinetic energy of the projectile into the kinetic energy of the water. A cavity is formed on the moving path of the projectile, and the water keeps squeezing the water tank due to cavity expansion, causing the cavitation load. In Test 1, the specific impulse of the initial shock wave measured on the side wall is 450 MPa·us, and that of the cavitation load is 2516 MPa·us; in Test 3, the specific impulse of the initial shock wave is 502 MPa·us, and that of the cavitation load is 2617 MPa·us. The peak pressure of the cavitation load is much smaller than the initial shock wave load. However, due to its long duration, the specific impulse of the cavitation load is about five times that of the initial shock wave. Therefore, the cavitation load is the main load causing the side wall failure of the water tank. The ratio of the specific impulse of the initial shock wave to the kinetic energy ratio of the projectile is basically the same. As the kinetic energy increases, the specific impulse of the initial shock wave increases significantly. The reason

is: the initial shock wave propagates at a speed in the water close to the speed of sound, and the initial shock wave is affected by the rarefaction waves in all directions; thus, the shock wave pressure decays quickly with a short action time, and the shock wave duration is the same. Therefore, the larger peak value of the shock wave leads to larger specific impulse, but there is little difference between the cavitation load and the specific impulse.

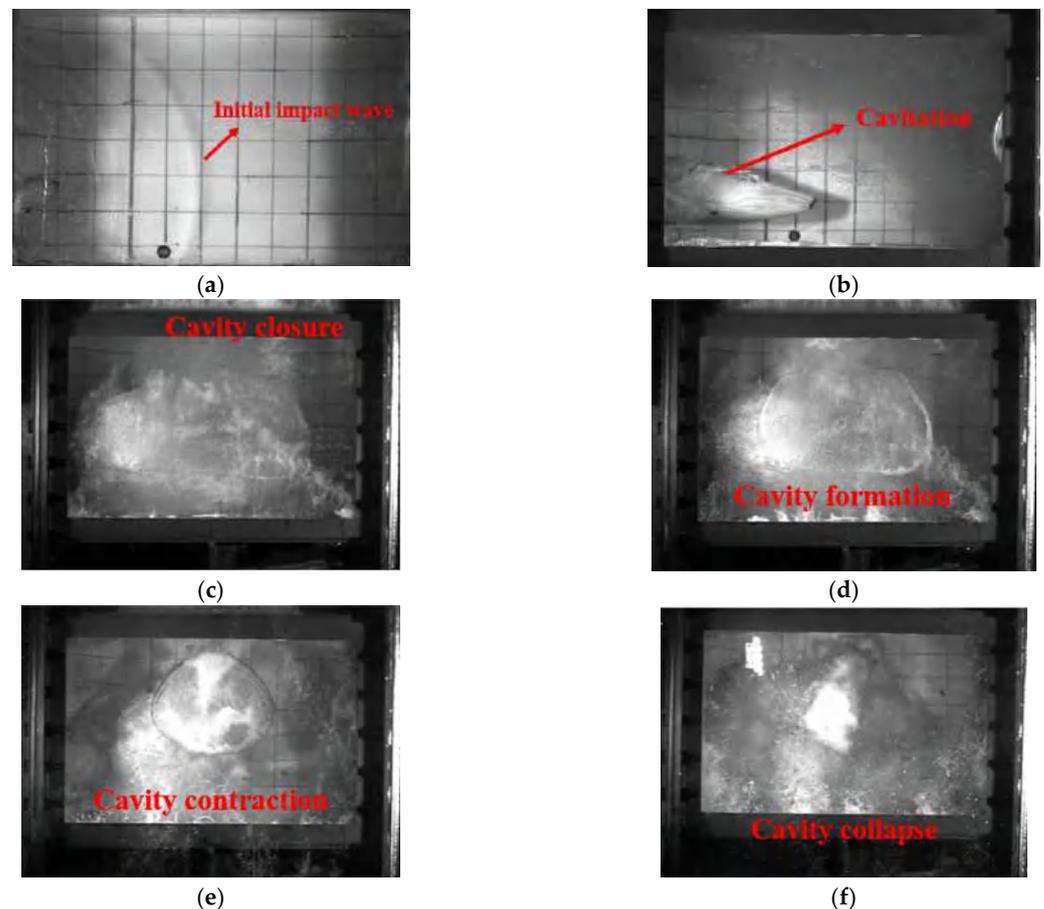


Figure 16. Pressure loading stage. (a) 100 us; (b) 250 us; (c) 2300 us; (d) 3300 us; (e) 8000 us; (f) 11,300 us.

Table 7. Peak value and specific impulse of initial pressure and cavitation load.

Test No.	Kinetic Energy (J)	Initial Peak Pressure (MPa)	Specific Impulse of Initial Pressure (MPa·us)	Peak Cavitation Load (MPa)	Specific Impulse of Cavitation Load (MPa·us)
1	15,432	19.1	450	8.3	2516
3	18,799	22.9	502	9.1	2617

As the kinetic energy of the projectile is continuously converted into the kinetic energy of the water, the water level rises. Under the influence of the backflow of disturbed water during cavity expansion, the air flow between the cavitation bubble and the atmosphere is cut off, and the cavity is closed at the entry hole. At this time, there are lots of gas and water vapor in the cavity, hence the formation of the cavitation bubble, which has the largest potential energy. Due to the pressure difference between the inside and outside of the bubble, its wall contracts, and the bubble gradually shrinks; in this process, the potential energy of the bubble is converted into kinetic energy, and negative pressure is generated in the water tank, which lasts for a long time, but the absolute value of the load is small; thus, it barely causes damage to the structure. With the contraction of the bubble, the gas in the bubble is compressed, leading to the increase in pressure and the formation

of a high-pressure region, and then the bubble collapses, followed by bubble expansion again. At this time, shock waves are generated, and most of the energy in the bubble is converted into impact energy. Thereafter, the bubble keeps expanding and collapsing until all energy is dissipated. Due to the existence of the free surface in the water tank and the pressure sensor located at the bottom of the side wall of the water tank, the measured shock wave load for bubble collapse is relatively small (Figure 15). The cavity collapse pressure measured by Disimilea was very large, even exceeding the initial shock pressure. On the one hand, because the measuring points were arranged near the ballistic axis, and the cavity collapse also occurred on the ballistic axis, the measured pressure was large; on the other hand, the liquid-filled tank used was a closed tank with a large volume, resulting in a large cavity and higher pressure value generated during cavity collapse.

To investigate the effect of attack angle on the loads during the projectile penetrating the liquid cabin, No. 29–33 are selected for investigation, and the loads on the side walls of the water tank and at the horizontal distance of 7.5 cm between the front/rear plates and the center are measured. According to Figure 17, as the attack angle of the projectile increases, the initial shock wave pressure load on the side walls of the tank tends to increase first and then decrease; meanwhile, due to improved attack angle, the area of the projectile in contact with water increases, resulting in greater water disturbance, so the load on the front plate keeps increasing; on the contrary, the pressure load on the rear plate decreases with the increasing attack angle for the following reasons: when the projectile enters the water at the velocity of 2400 m/s; meanwhile, the wave velocity in the water is only 1500 m/s, and the projectile separates from the shock wave after water entry for a period of time; at this time, the projectile yaws, and the measuring point on the rear plate is far away from the projectile axis; the increasing attack angle leads to greater projectile yaw and farther measuring point from the axis, and thus the pressure load on the rear plate decreases with the increasing attack angle. The initial shock wave propagates far away in a hemispherical shape, and the pressure along the wave arc decreases with the angle of the wave moving away from the ballistic axis. The measuring point on the front plate is more than 90° away from the axis, and the measuring point on the rear plate is near the axis, so at small attack angles, the pressure load on the rear plate is greater than that on the front plate; as the attack angle increases, the pressure load on the front plate increases but decreases on the rear plate, with the former exceeding the latter. The specific impulse change in the initial shock wave loads on the front and rear plates is consistent with the variation law of the shock wave pressure peak, but the specific impulse change of the cavitation load on the front plate is not significant. The reason is: as the projectile enters the water, a large cavitation region is generated, resulting in the measuring point on the front plate entering the cavitation region only after being subjected to cavitation loading for a short period of time, not the complete cavitation load.

3.4. Analysis of Failure Modes of Liquid Cabin

Figure 18 presents the deformation morphology and failure diagrams and scanned contours of the front and rear plates before and after the tests. As illustrated, the front plate of the water tank undergoes shear plugging failure and thin film bulging deformation during the projectile penetration into the water tank at a high velocity. When the high-velocity projectile impacts the steel plate at an attack angle with water medium as the “dynamic support”, the support of water improves the rigidity of the steel plate. As the front plate is thin, the impact on the target plate by the high-velocity projectile causes adiabatic shearing. The light blue color at the edge of the perforation hole is caused by the release of a large amount of heat during the contact between the projectile and the target. Because projectile penetration is at an attack angle, the shape of the hole is not circular, similar to the shape of the projectile nose, but rectangular, similar to the shape of the side of the projectile. The initial shock wave is generated after the projectile impacts the water and enters it. As the shock wave is close to the front plate, it bulges outward after being impacted. Then, due to the long cavitation, the water in the tank moves around and squeeze

the front plate, leading to the wide bulging deformation of the front plate. However, when Wu conducted the penetration test with the water tank, the front plate was depressed, which was caused by the negative pressure induced by bubble contraction on the front plate due to the closure of the tank. In the present paper, as the tank is not covered, there is no plate depression. Comparing the plate deflection, we see that the kinetic energy of the projectile increases, but the deflection of the front plate does not increase apparently. This is explained by the fact that the front plate mainly undergoes thin film bulging deformation caused by the cavitation load, and the specific impulse of the cavitation load in the three groups of tests is basically the same.

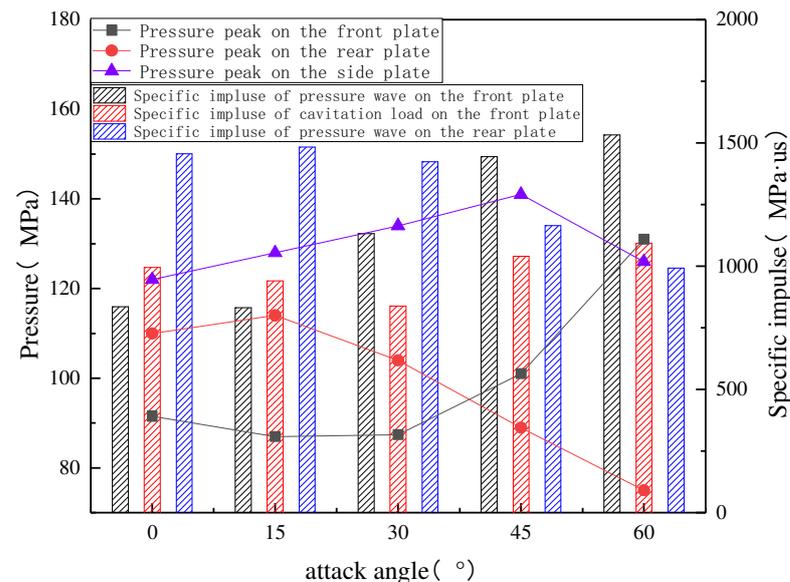


Figure 17. Peak value and specific impulse of pressure wave on each wall of the tank.

Due to the large thickness and stiffness of the rear plate, it has no obvious thin film bulging deformation. Since the projectiles used in Tests 1 and 3 have different masses and L/D ratios, under the same initial velocity, their kinetic energy varies, leading to very different failure modes of the rear plate. In Test 1, the rear plate mainly undergoes bulging and dishing deformation. Due to the long water domain, the incident shock wave generated after the water entry of the projectile is weakened to some extent when reaching the rear plate, having a small influence on the rear plate; as the projectile approaches the rear plate during penetration, a high-pressure region is produced at the position around the projectile nose as the projectile squeezes the water, and this high-pressure region acts on the rear plate, causing dishing deformation; then, the projectile impacts the rear plate, causing bulging deformation. After Test 3, the rear plate has annular breaches and radial cracks. The reason is: as the projectile moves in the water with an attack angle, it keeps overturning; it always has a large attack angle and a certain angular velocity upon reaching the rear plate; as the edge of the projectile nose first touches the target, which is similar to the penetration of a sharp-nosed projectile, the rear plate undergoes severe plastic deformation, and then the material is squeezed toward the cratering position by the sharp nose, causing annular breaches and radial cracks at the edge of the breaches; as the penetration continues, more cracks are generated and developed into petaling failure. In Tests 1 and 3, the kinetic energy ratio of the projectiles is 0.82, and the maximum deflection ratio of the rear plates is 0.60, which indicates that when the rear plate is approaching the ballistic limit, the impact force on the rear plate is increased by the projectile's pushing of the water; thus, the rear plate is subjected to the high pressure of the water and the impact of the projectile, resulting in two failures; the coupling effect of the two failures aggravates the damage to the rear plate.

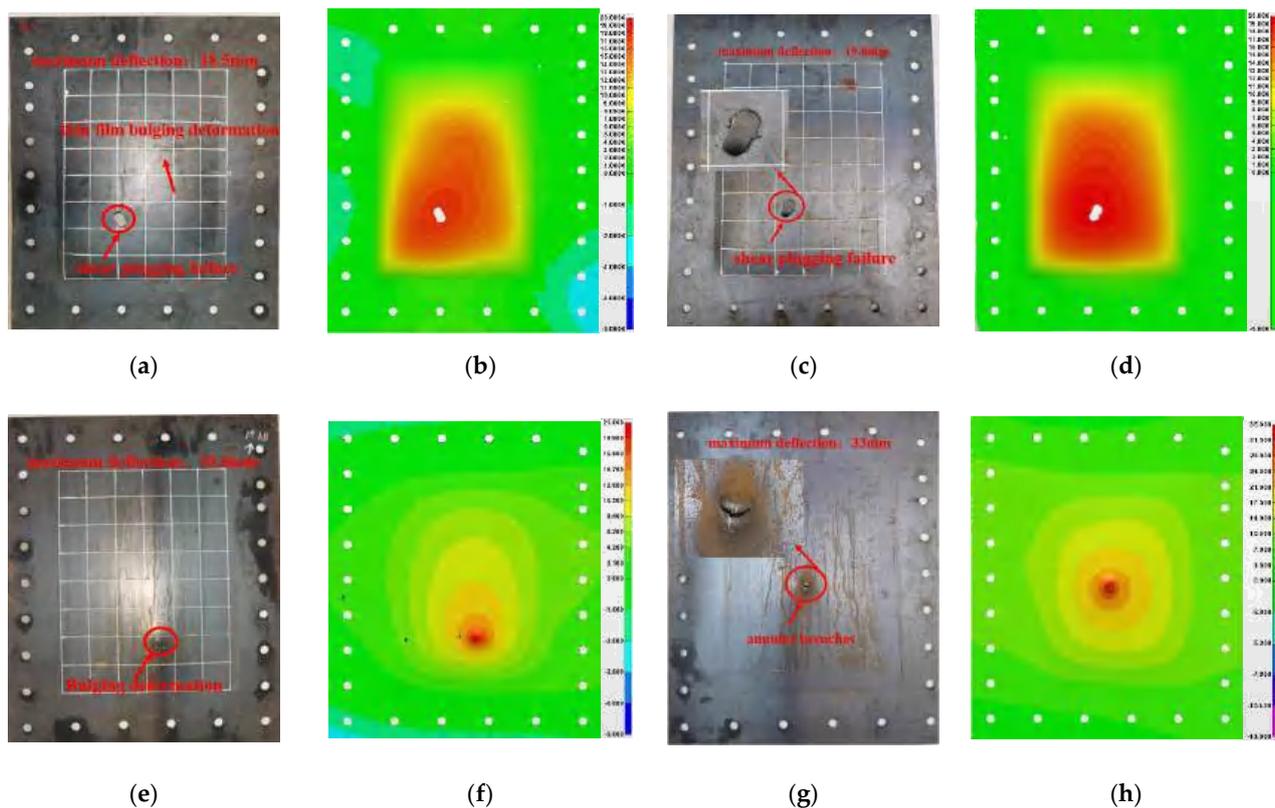


Figure 18. Deformation/failure diagram and deflection contour of the front and back plates. (the deflection unit: mm). (a) Deformation and failure diagram of the front plate in Test 1; (b) contour of deformation and deflection of the front plate in Test 1; (c) deformation and failure diagram of the front plate in Test 3; (d) contour of deformation and deflection of the front plate in Test 3; (e) deformation and failure diagram of the rear plate in Test 1; (f) contour of deformation and deflection of the rear plate in Test 1; (g) deformation and failure diagram of the rear plate in Test 3; (h) contour of deformation and deflection of the rear plate in No. 3.

Figure 19 shows the failure morphology of the liquid cabin after the penetration of the projectile with a L/D ratio of 4.8 and a velocity of 1600 m/s and at different attack angles (No. 19–23). As the attack angle increases, the breach in the front plate enlarges. During normal penetration, the front plate undergoes shear plugging–thin film bulging deformation, and the breach is circular. When an attack angle exists, the load applied by the projectile changes from a point load to a line load; thus, the area in contact with the target increases, and the shear plugging breach is rectangular. After the water entry of projectile, which is subjected to the shock wave pressure and cavitation load, the front plate undergoes bulging deformation; at the same time, the plate is subjected to the surface load of the water. Based on the classical yield line theory, the plastic hinge line in the rectangular plate is as shown in Figure 20, where the plate tears along the plastic hinge line. Since more water is displaced by the water entry projectile with an attack angle, the initial shock wave pressure and cavitation load on the front plate increase with the increasing attack angle: in the range of $15\sim 30^\circ$, the front plate shows “I”-shaped tear along the plastic hinge line; in the range of $45\sim 60^\circ$, the front plate is subjected to a greater load, and thus shows the “X”-shaped tear along the plastic hinge line.

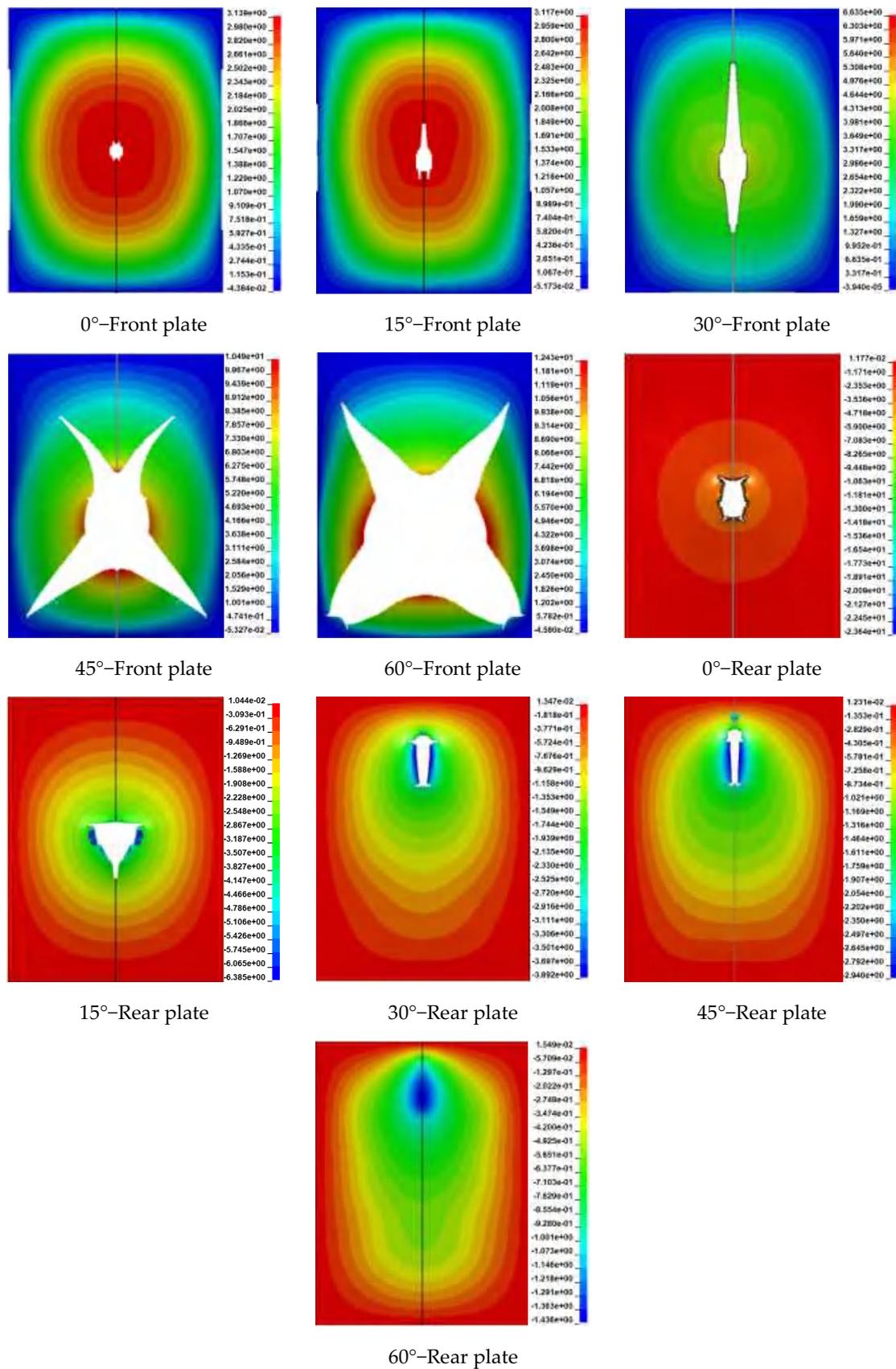


Figure 19. Failure morphology of the front and rear plates at different attack angles (deflection unit: cm).

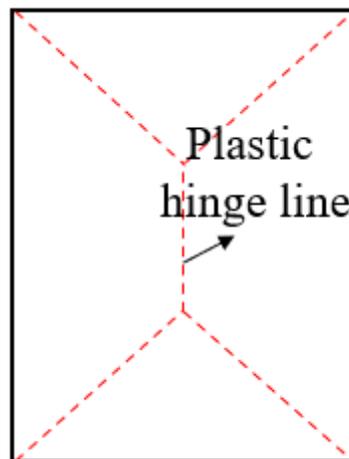


Figure 20. Plastic hinge line in the rectangular plate.

As the attack angle increases, the breach in the rear plate gradually decreases for the following reason: the projectile with an attack angle is subjected to greater drag in water; therefore, the projectile velocity decays faster; moreover, the front plate undergoes overall failure in advance under high attack angles; the liquid pressure is unloaded from the front plate, which reduces the damage to the rear plate; in the case of normal penetration, the rear plate undergoes dishing–petaling deformation; with the increase in the attack angle, the breach in the rear plate changes from the petal shape to the strip shape; in the range of $30\sim 45^\circ$, the rear plate undergoes dishing-“I” tearing; at 60° , the projectile cannot perforate the rear plate; therefore, the rear plate exhibits dishing–bulging deformation.

4. Conclusions

In this paper, an effective numerical calculation method was obtained through the model test of the projectile penetrating the liquid tank and the numerical calculation verification with the experimental data. The deformation and failure modes of the projectile after penetration at attack angles, the trajectory and attitude change of the projectile in water were explored, and the load strength on the side walls and the failure modes of the front and rear plates during the penetration of the projectile into the water tank were analyzed. The main conclusions are as follows:

- (a) There are four basic failure modes after the projectile penetrating the liquid cabin at attack angles: asymmetric mushrooming at the nose, side erosion, overall plastic bending and fracture. The overall plastic bending and fracture are mainly related to the L/D ratio, initial attack angle and initial projectile velocity; at low velocities, the main failure occurs during the penetration into the front plate; at high velocities, the main failure occurs during the penetration into the water.
- (b) In the case of the rod projectile with a L/D ratio of 2 and a velocity of 1600 m/s, the projectile was more prone to overall deformation at the attack angles in the range of $15\sim 30^\circ$. At low velocities, the overall plastic bending deformation occurred, whereas with the increasing velocity, fracture failure occurred; when the attack angle was above 30° , the failure mode of the projectile was side erosion, but as the initial velocity increased, the range of the attack angle that led to the overall deformation of the projectile also expanded. The larger the L/D ratio, the more easily the projectile undergoes overall plastic bending failure.
- (c) After the projectile entered the water at an attack angle, the overturning moment was generated due to the uneven force on the projectile, and the projectile was at a state of constant overturning. Due to the change of the position contacting with the water, the projectile overturned many times, and projectile yaw occurred; the yaw direction was mainly affected by the initial attack angle and projectile attitude.

- (d) After the water entry of the projectile, the side walls of the water tank were mainly affected by three stages of loading: the initial shock wave pressure loading, the cavitation loading and the cavity collapse loading. As the attack angle increased, the peak load of the initial shock wave pressure on the front plate increased gradually, whereas the initial shock wave pressure load on the rear plate decreased gradually.
- (e) After the penetration of the projectile into the water tank, the failure modes of the front plate were mainly shear plugging, thin film bulging deformation and tearing failure. At certain velocities, with the increasing attack angle, the front plate underwent tearing along the plastic hinge line; under small attack angles, the tear was “I” shaped, and under large attack angles, it was “X” shaped; the rear plate mainly underwent dishing-bulging deformation; when the rear plate approached the ballistic limit, annular breaches and circumferential cracks were produced.

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Article

Analysis of Damage of Typical Composite/Metal Connecting Structure in Aircraft under the Influences of High-Velocity Fragments

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Abstract: A two-stage light gas gun was used to conduct a high-velocity impact test on the aircraft's typical composite/metal connecting structure (CFRP/AL). The battle damage simulations used for the CFRP/AL connecting structure were carried out under different intersection conditions. Then, the damage morphology and mechanism of high-velocity prefabricated spherical fragments on typical structures, the dynamic process of hyper-velocity impact, and the formation of debris clouds on the secondary damage morphology of different component structures were investigated. Next, based on the X-ray computerized tomography (CT), the typical mode of different damage areas and evolution trends of CFRP under high-velocity impacts were explored. Finally, a simulation model was established for battle damages of typical structures by combining FEM methods, and structural components' energy dissipation capabilities for fragments under different velocities were analyzed. The study results provide a reference and model support for the rapid repair of battle-damaged aircraft and aircraft survivability design.

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Keywords: battle damage; X-ray tomography; two-stage light gas gun; high-velocity fragment

1. Introduction

The problem of damage repair of aviation weapon equipment on the battlefield has a long history. According to statistics, the number of battle-damaged aircraft is much greater than the number of battle-destroyed aircraft. Hence, rapid repair of battle-damaged aircraft can have a significant impact on the war situation by increasing operational intensity and ensuring sustained combat capability. Indeed, rapid repair of battle-damaged aircraft has attracted great attention as it is the most effective way to restore aircraft combat effectiveness [1].

The battle damage mode of aircraft is to study all possible conditions and combinations of threat sources causing battle damage. The battle damage simulation based on rapid repair is different from weapon effectiveness analysis, and the combat effectiveness and survivability analysis of aircraft. The purpose is to provide guidance on rapid repair techniques and to provide an aid for the analysis of rapid repair resource requirements, usually focusing on the analysis and research of battle damage of aircraft structures. The combat aircraft is the main target of all types of air defense weapons.

Currently, carbon fiber composites have been widely applied in advanced fighters, especially in wing panels, vertical fins, fuselage skins, and rudders [2]. Carbon fiber laminates are the most important aerospace composite structure. The damage of a composite structure under the impact of combat fragments is different from that of a metallic structure, and its damage mechanism is more complex than that of metallic material. On one hand, the reinforced fiber limits the expansion of damage under the composite structure's own load, and on the other hand, the damage modes (such as the delamination of composite laminate structures) are not found in aerospace metal structures. The impact of the combat

fragments on the composite laminate is a complex process of impact load evolution with time, structural deformation and damage extension. The impact damage of composite laminate is closely related to the laminate material's properties, lamination method, processing technology and the fragment's velocity, shape and quality. The damage mode and damage range of the laminate are different in different impact conditions. The possible damage mechanisms include fiber shear fracture, fiber tensile failure, matrix cracking, fiber matrix interface delamination and degumming. The damage mechanism and damage mode of a composite structure in a series of velocity ranges were investigated by numerical simulation and experiments. Thomas et al. conducted a hyper-speed impact test on CFRP/Al honeycomb composites and obtained the dynamic response of the structure with different thickness combinations [3]. Miao et al. conducted a hyper-speed impact test and impact damage test analysis to solve the protection effect problem of a spatial debris soft protection structure composed of a soft protection screen made of multilayer soft composites [4]. Phadnis et al. conducted a CFRP-Al/HC sandwich panel hyper-speed impact behavior analysis based on the finite element method. Coles et al. conducted ballistic tests on braided T300 carbon fiber/epoxy composite flat-plate specimens and 3D X-ray computer tomography (CT) was used to image and visualize the resultant damage inside the samples.

Modern aircraft are mostly multilayered thin-walled structures. Aircraft skin will often encounter penetrating damage under the action of high-velocity fragments, which has considerable penetration capabilities after penetrating the outer skin, and the secondary penetration of fragments will cause damage to the internal structure. The damage mode is closely related to the fragment's incident direction, incident velocity, and strike position [5].

The damage mode of aircraft composite/metal connecting structures under high-velocity impact has been little studied. Aiming at the carbon-fiber-reinforced laminate/Al alloy connection frame structure of a certain type of aircraft wing, the damage mode under the action of high-velocity spherical fragments (1600–2400 m/s) was investigated based on the numerical simulation model of simulated impact tests and the display dynamics of the two-stage light gas gun system.

2. Experimental

The multistage light gas gun is commonly used to achieve the high-velocity loading of small-sized projectiles and is often used in tests related to spacecraft collisions with space debris, and its technical indicators can achieve the requirements for fragment intrusion into aircraft structures [6]. The characteristic structure equal ratio test piece, fragment and the corresponding Sabot were prepared. Based on the two-stage light gas gun system, the simulated penetration of prefabricated spherical fragments into aircraft-typical vulnerable structures under different working conditions was realized.

2.1. Instrument

The two-stage light gas gun was used as the launcher, and a 10# steel ball (diameter = 8 mm) was used to simulate the prefabricated fragment of the missile, as shown in Figure 1. The target structure is fixed in the chamber by the fixture, adjusted to the corresponding strike position, and the intended test strike position is marked by the laser pointer, as shown in Figures 2 and 3.

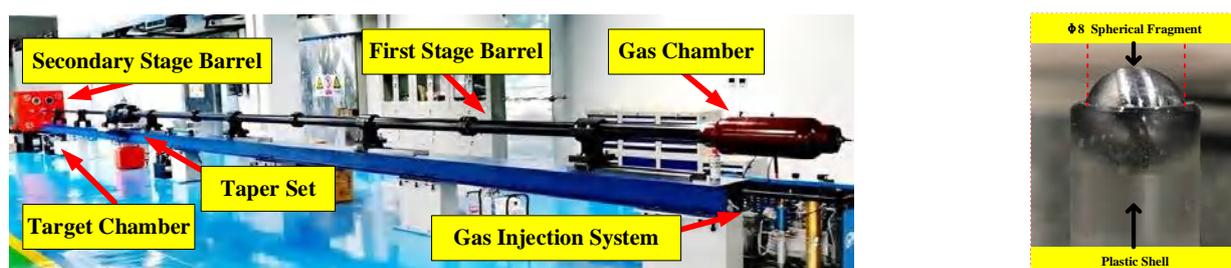


Figure 1. Two-stage light gas gun and simulation of spherical fragments.

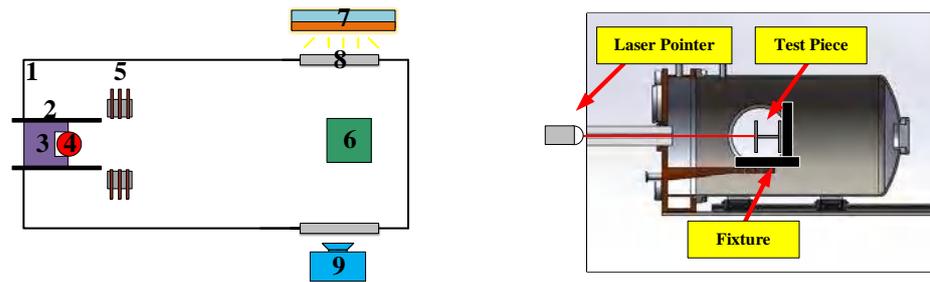


Figure 2. Schematic of the two-stage light gas gun: 1: target chamber; 2: gas gun barrel; 3: sabot; 4: projectile; 5: optical beam blocking (OBB) system; 6: target; 7: flash lamp; 8: windows; 9: high-velocity camera.

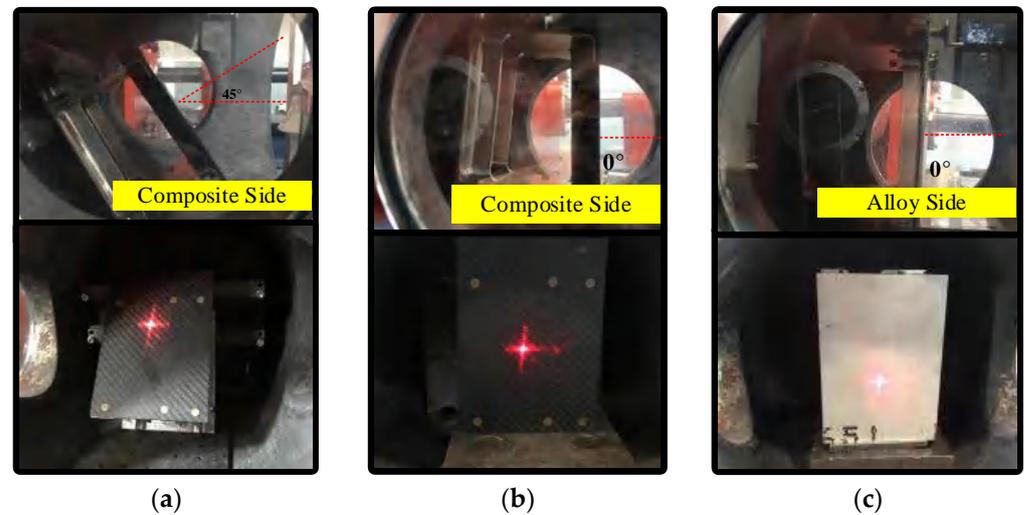


Figure 3. Tooling settings of the test piece. (a) Situation 1; (b) Situation 2; (c) Situation 3.

The target chamber is connected to the end of the secondary stage barrel, which is equipped with an optical beam blocking (OBB) sensor system to measure the exit velocity of the launched projectile. Three laser beams of similar intensity are placed in the projectile's path and directed to the photoelectric tube connected to the timer. The skimming of the projectile will block the laser beam, and the resulting pulse signal will be recorded on the timer by the photoelectric tube. The projectile velocity is obtained by comparing the time interval of the timer and the distance between the laser beams. The target chamber is equipped with bullet-proof glass observation windows on both sides and a high-velocity camera (Phantom V2512) for capturing the impact process, and a flash for the high-velocity camera to fill in the light. The high-velocity camera uses a 50 mm fixed-focus lens, and records at a pixel of 386×216 with a frame rate of 160 kfps, as shown in Figure 4.



Figure 4. Observation section setting of the target chamber.

The YXLON FF85 CT scanning system is equipped with two sets of radiation tubes and a large-sized flat detector for the detection of all types of damage on fiber-enhanced composites. The inspection sample is scanned layer by layer by emitting X-ray, and in combination with the analysis software, the composite can be inspected and global damage 3D modeling and rendering can be achieved. In this study, it is mainly used to implement the internal damage analysis and overall damage assessment of the composite, as shown in Figure 5.



Figure 5. YXLON FF85 CT scanning system.

2.2. Test Piece

The composite/Al alloy joint spacer frame structure test piece, as shown in Figure 6, consists of an Al alloy top skin, rib 1, rib 2, and with a composite bottom skin and stringer. The top skin is connected to rib 1 and rib 2 by 8 mm rivets, and the composite bottom skin is connected to rib 1 and rib 2 by high-locking bolts. The Al alloy is AL7075-T6 produced by Chalco Group, using the GB/T29503-2013 standard, and the composite material is T300/QY8911 epoxy resin-based carbon fiber unidirectional laminate. It contains bottom skin (45/−45/0/−45/0/45/0/45/0/45/−45/0/90/45/90/45/90/0/−45/45/0/−45/45/0/45/0/−45/0/−45/0/−45/45) and stringer (45/−45/0/−45/45/0/−45/0/−45/0/45/0/90/90/0/45/0/−45/0/45/−45/0/−45/45). The laminate was laid orthogonally and symmetrically with plain woven prepreg, as shown in Figure 7.

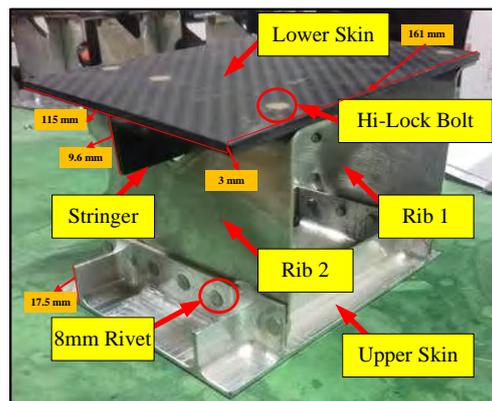


Figure 6. Test piece of composite/Al alloy connecting frame structure.

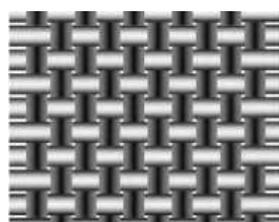


Figure 7. Schematic of composite layup method.

3. Damage Analysis

3.1. Analysis of Impact Process

A high-speed camera (frame rate 160 kfps) is used to capture the dynamic process of the intersection of the spherical fragments with the structure, as shown in Figure 8.

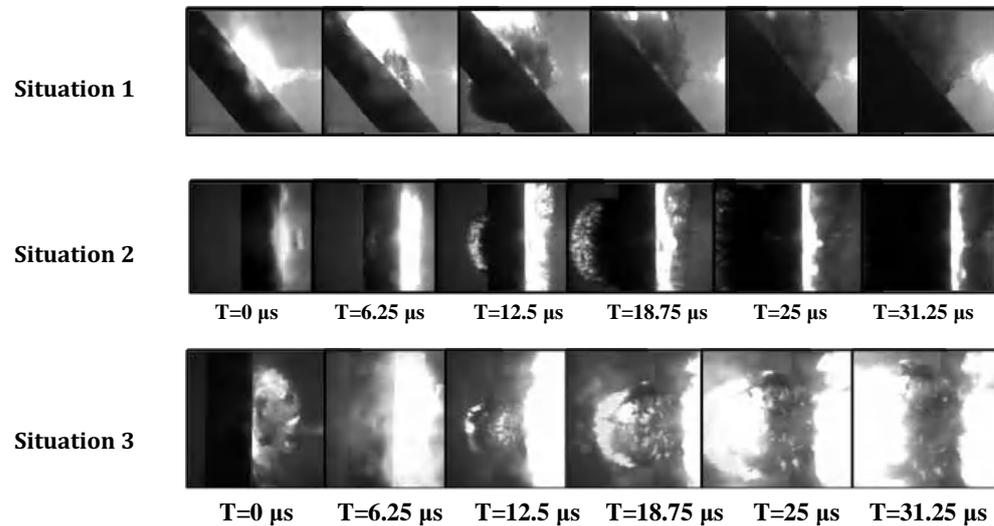


Figure 8. Impact process of test piece of composite/Al alloy lapping frame structure.

Situation 1: The fragment is incident from the bottom skin composite side at an incidence angle of 45° , and the measured fragment exit velocity is 2327 m/s. The moment the fragment intersects with the structure, a strong photo-thermal phenomenon is generated at the impact location, resulting in local overexposure of the high-velocity camera. Subsequently, the fragment penetrates the structure and forms a debris cloud on the outside of the structure, accompanied by a splash of massive fiber. As the invasion progresses, the composite laminate forms a bulge. The rapid rupture of the bulge forms a debris cloud, which is mainly composed of broken small-diameter carbon fiber particles. The outer debris cloud at the incident end expands radially along the vertical direction of fragment incidence, and its main components are bulk fiber debris and fragment metal particles, while the inner debris cloud at the incident end expands radially along the direction of fragment incidence, and its main components are small-diameter carbon fiber debris groups.

Situation 2: The dynamic intersection process of the fragment and structure in Situation 2 is similar to that shown in Situation 1. Due to the change in the direction of incidence, there is a difference in the expansion pattern of the debris cloud between the two. In Situation 2, the difference between the inner and outer debris cloud's highlight phenomena is more obvious, again due to the difference in their main components. The small-diameter carbon fiber debris group has a black main layer with dense spatial distribution and strong light absorption, and the photo-thermal phenomena captured by the high-velocity camera in the inner part of the incident section are weak.

Situation 3: The fragment intruded from the top skin Al alloy side, and the photo-thermal phenomena were stronger both inside and outside. In Situation 1/2, when the projectile intrudes from the composite side, unlike the metal side, the carbon fiber breaks up to generate a large amount of dusty debris cloud, and the debris cloud is dominated by the carbon fiber debris, which will cover the entire high-velocity camera capture area, and the radial velocity is larger than the incident velocity during the expansion of the debris cloud. It is because the composite is made out of anisotropic material that when the impact direction is perpendicular to the fiber layup, the fiber is susceptible to shear fracture, and its normal strength is much lower than that of the metal material, resulting in a difference in the morphology of the bulge formed in the impact process, and therefore, the morphology of debris cloud diffusion formed in the final bulge rupture is also different.

The shock photo-thermal phenomena in Situation 1/2 are not as obvious as in Situation 3, and the main component of the firelight is the high-temperature metal fragments. Due to the violent friction between steel spherical fragments and Al alloys during the impact penetration process, the temperature is extremely high, resulting in the appearance of small metallic debris and the release of part of the heat accumulated by friction in the pattern of luminescence.

3.2. Composite Damage Analysis

In Situation 1/2, the damage morphology of the structural composite top skin and the stringer intrusion by the high-velocity spherical fragments are shown in Figures 9 and 10, respectively. The damage broken hole is mainly ellipsoidal under the 45° oblique impact of the fragment in Situation 1. In Situation 2, under the positive impact of fragment 0°, the damage broken hole is mainly in the pattern of regular spherical rupture. In Situation 1/2, there is fiber spalling on the outer surface of the incidence. The stringer part is thin and shows a band-like broken hole under the shearing effect in both Situation 1/2.



Figure 9. Typical shear damage mode of composite bottom skin.



Figure 10. Typical damage mode of composite stringer.

In Situation 3, the fragment is incident from the Al alloy metal side. According to Figure 11, the fragment formed a debris cloud consisting of a large number of metal particles when it penetrated the top skin. The debris cloud still has a high kinetic energy, and the secondary damage formed at the bottom skin, which is mainly in the pattern of small-diameter broken holes and surface spalling, is widely distributed.

The test piece composite part was scanned by CT, and the damage morphology feature images of the composite test piece were obtained for each directional interface, layer by layer, and then the 3D view of the damage of the test piece composite part under each working condition was obtained by image rendering, as shown in Figure 12.



Figure 11. Damage mode of secondary damage of composite bottom skin.

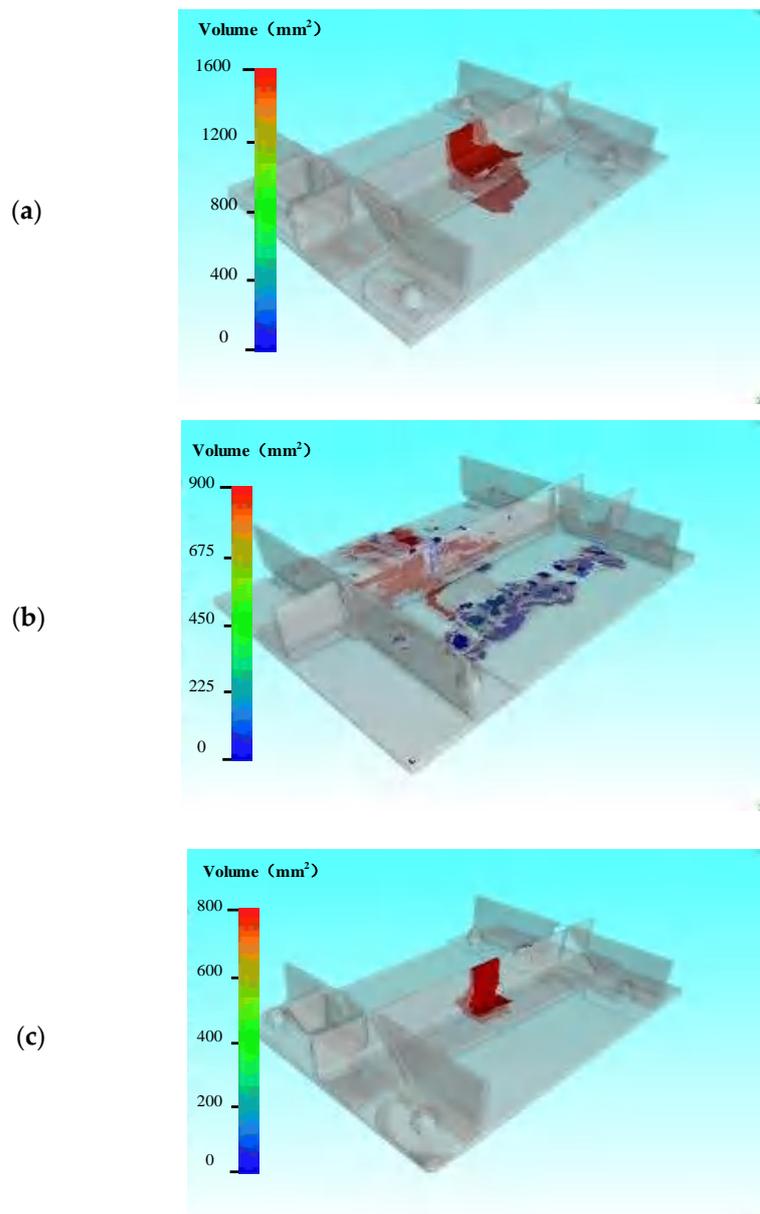


Figure 12. 3D CT damage player diagram. (a) Situation 1; (b) Situation 2; (c) Situation 3.

The section of composite laminate damage area under high-velocity impact of spherical fragments is mainly in the pattern of a combination of cylindrical and circular truncated cones, as shown in Figure 13. In his study on the impact damage of aramid laminate,

Reddy pointed out that the damage aperture of thick plates decreases slightly along the thickness direction and then increases rapidly, while the aperture of thin plates expands in the shape of a circular truncated cone, and Cantwell et al. used the combined area of a cylindrical and circular truncated cone [7–9]. In the early stage of projectile penetration, the front side of the laminate would generate a shear-plugging hole similar to the shape of the projectile contact area, where the matrix material is crushed and loses its support to the fiber, and shear failure occurs between the fiber and the surrounding fiber due to the presence of a large velocity gradient, which is called the shear failure zone (A). With the continuation of the penetration process, the projectile velocity decreases, the target plate bends, and with the continuous expansion of the bending deformation, fiber tensile failure occurs first in the outermost layer of the back side of the impact, and the tensile failure expands from the outer layer to the back side with a certain crack inclination angle and produces delamination, forming a fiber tensile failure zone behind the shear failure zone (B). Meanwhile, there is a small delamination damage zone around the shear failure zone/fiber tensile failure zone (C). In this test, the delamination zone of laminate caused by the high-velocity impact of spherical fragments is limited, and the fiber in the zone is mainly recoverable deformation without significant fiber damage.



Figure 13. Distribution of damaged area.

The projection of the laminate damage area in the direction of fragment incidence shows different morphological features in different damage areas, as shown in Figure 14. Herein, the position of the blue line is the position of the broken hole section in the vertical plane. On the upper surface of the shear failure zone, the main body of the broken hole section is circular. The laminate-free surface is strongly impacted by spherical fragments, and there are more striated fiber-stripping areas around the circular rupture holes. As the penetration progresses, the breach section inside the shear failure zone is mainly regular circular, and the area of the spalling zone decreases. At the shear failure zone/fiber tensile failure zone intersection, the breach section has common features of both. The shape of the breach section is transformed from a shear failure-oriented circle to a fiber tensile failure-oriented square. As the penetration progresses, the breach section in the fiber tensile failure zone appears as a regular square.

High-speed fragmentation spherical fragment intrusion CFRP, in the process of damage formation, can be divided into four stages as shown in Figure 14. In shear and tensile damage, the compression wave generated by the spherical fragment acting on the target plate propagates faster than the projectile movement, and the strong tensile wave formed by the reflection of the compression wave on the back of the target plate meets the projectile, and the meeting point is the interface between shear damage and tensile damage. When the tensile stress is greater than the bonding strength of the fibers and the substrate or the tensile strength of matrix material, tensile stress is induced at the defective area and is accompanied by partial delamination. Eventually the fibers fracture and splash under the impact, which is in line with the phenomenon captured by the high-speed camera.

As shown in Figure 15, from left to right are the upper surface, shear failure zone, area boundary, tensile failure zone, and lower surface of the broken hole morphology, respectively. The evolutionary pattern of breach section morphology shows a similar pattern under different velocities. Notably, as the fragment impact velocity decreases, the size of the laminate-free surface spalling area decreases significantly.

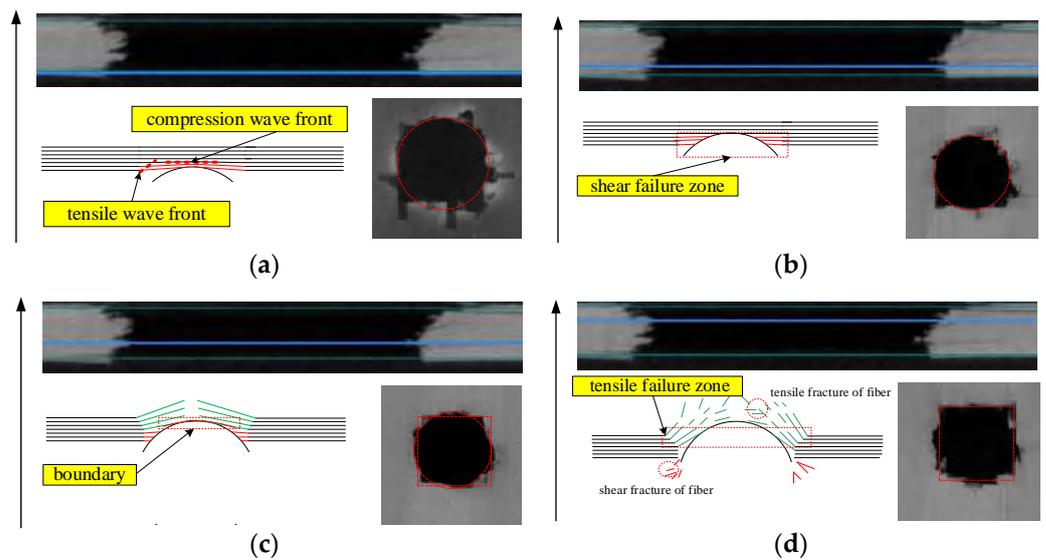


Figure 14. Morphological features of the CFRP damage process. (a) Impact crater extrusion, (b) Shear intrusion, (c) Tensile intrusion, and (d) fiber fracture splashing.

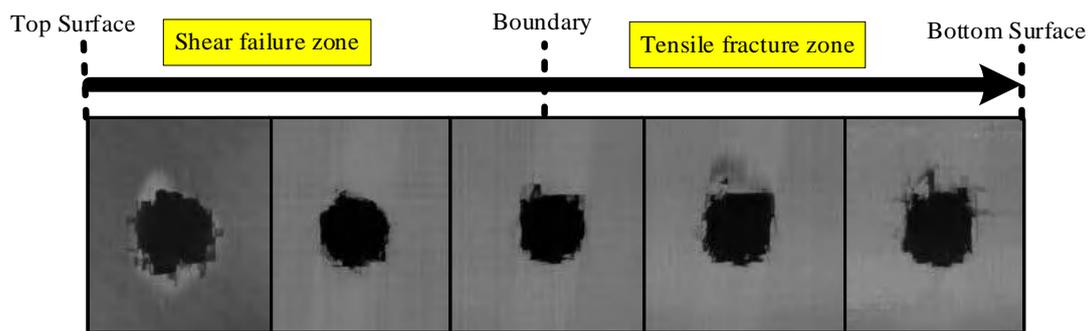


Figure 15. Morphology of damaged area in CFRP.

3.3. Analysis of Al Alloy Damage

Under the Situation 1 incidence condition, the fragment penetrated the bottom skin and a debris cloud hit the rib of the structure. The fragment's main body with small-diameter debris intruded to generate an ellipsoidal rupture hole. The thinner part of the rib is torn and fractured by the impact, as shown in Figure 16.



Figure 16. Top skin damage in Situation 1.

For Situation 2's incidence conditions, four sets of velocity gradient variables were set in the range of the actual fragment velocity. The debris cloud composed of small-diameter

carbon fiber particles formed by the fragment penetrating the bottom skin composite side does not have the energy to generate a distributed secondary image on the top skin side of the Al alloy. The top skin side damage is mainly caused by residual fragment penetration. The spherical fragments are separated into two parts by the erosion of composite laminate and the shearing effect of stringer, forming shear holes in the top skin. From the contact between the fragment and the top skin, the annular shear stress formed by the impact is much larger than the panel's ultimate strength, the annular shear zone then gradually accumulates and expands to the back of the panel to generate an annular shear surface. The shear punching is completed to generate a bulge, the radial tensile stress at the back of the back panel bulge then rises, with the depth of the intrusion, the tensile stress accumulates and expands, when the accumulation exceeds the ultimate strength of the panel, and the bulge fractures and breaks rapidly, forming a petal-shaped irregular fracture. The greater the initial velocity of the fragment, the greater the reaction force received in the impact process, and the farther apart the two separated parts will be. As the velocity decreases, the two holes are connected and eventually generate a single hole, as shown in Figure 17.

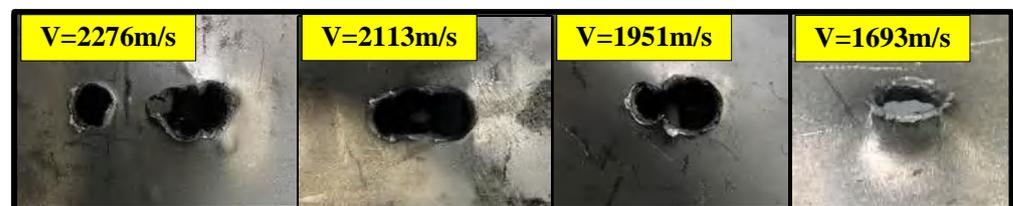


Figure 17. Top skin damage in Situation 2.

Under Situation 3's incidence conditions, the top skin Al alloy exhibits a relatively regular spherical shape in the entire damage area under the high-shearing effect of the fragment's high-velocity impact, and its area is basically the same as the area of the orthogonal projection of the spherical fragments. The reinforced ribs develop a columnar erosion zone under the high-shearing effect of the fragment, and the edges show the typical metal cutting marks under the high-shearing effect and the ablation marks caused by the accumulated heat of impact, as shown in Figure 18.



Figure 18. Top skin damage in Situation 3.

3.4. Numerical Modeling

In order to compensate for the limitation of the number of experiments, a fragment impact composite/metal connecting structure model based on LS-DYNA was established. In composite modeling, the *PART_COMPOSITE keyword is used to define the basic physical parameters such as the thickness of each layer of the composite carbon fiber laminate component, and the layup direction. In order to better reflect the loss between different layers independently in the modeling, and to take into account the computational scale and efficiency, a modeling scheme of one 2D shell cell layer is used instead of four to five actual layers, i.e., five 2D shell cell layers (i.e., elements in the meshing) are used in the simulation modeling instead of 25 actual layers of composite material for the bottom skin

components, and five 2D shell unit layers are used instead of 22 actual layers of composite material for the stringer components.

The *MAT_COMPOSITE_DAMAGE model is a composite constitutive model commonly used on shell units, containing physical quantities such as density, fiber elastic modulus in three directions, Poisson’s ratio, shear modulus and longitudinal tensile strength, and transverse tensile strength, and can be used to define various orthogonal anisotropic materials with brittle fractures, which is applicable to the T300/QY8911 epoxy resin-based carbon fiber unidirectional laminate in this study. The single-layer carbon fiber laminate thickness direction size is much smaller than the other direction size. Hence, it can be analyzed according to the plane stress problem, considering only the in-plane stress state, ignoring the plane normal upward stress. The stress–strain relationship can be expressed as:

$$[\sigma] = [S][\varepsilon] \tag{1}$$

$$\frac{1}{[S]} = \begin{bmatrix} \frac{1}{E_{11}} & \frac{\nu_{12}}{E_{11}} & \frac{-\nu_{31}}{E_{33}} & & & \\ -\frac{\nu_{12}}{E_{11}} & \frac{1}{E_{22}} & \frac{-\nu_{23}}{E_{23}} & & & \\ \frac{-\nu_{31}}{E_{33}} & \frac{-\nu_{23}}{E_{23}} & \frac{1}{E_{33}} & & & \\ & & & \frac{1}{2G_{12}} & 0 & 0 \\ & 0 & & 0 & \frac{1}{2G_{23}} & 0 \\ & & & 0 & 0 & \frac{1}{2G_{31}} \end{bmatrix} \tag{2}$$

where σ is the stress, ε is the strain, E is the elastic modulus, ν is the Poisson’s ratio, and G is the shear modulus.

The *MAT_COMPOSITE_DAMAGE model uses the Chang–Chang failure criterion and has the following four failure modes:

- (1) If $\sigma_{aa} > 0$, fiber is in the stretched state, when satisfied:

$$\left(\frac{\sigma_{aa}}{X_T}\right)^2 + \beta\left(\frac{\sigma_{ab}}{S_c}\right) - 1 \geq 0 \tag{3}$$

Herein, $E_a = E_b = \nu_{ba} = \nu_{ab} = G_{ab} = 0$ and the fiber undergoes stretching failure;

- (2) If $\sigma_{aa} < 0$, the fiber is in compression, when the following conditions are met:

$$\left(\frac{\sigma_{aa}}{X_C}\right)^2 - 1 \geq 0 \tag{4}$$

Herein, $E_a = \nu_{ba} = \nu_{ab} = 0$ and the fiber fails in compression;

- (3) If $\sigma_{bb} > 0$, the fiber matrix is in a stretched state, when the following conditions are met:

$$\left(\frac{\sigma_{aa}}{Y_T}\right)^2 + \left(\frac{\sigma_{ab}}{S_c}\right) - 1 \geq 0 \tag{5}$$

Herein, $E_a = \nu_{ba} = G_{ab} = 0$ and the fiber matrix undergoes a stretching failure;

- (4) If $\sigma_{bb} < 0$, the fiber matrix is in compression, when the following conditions are met:

$$\left(\frac{\sigma_{bb}}{2S_C}\right)^2 + \left[\left(\frac{Y_C}{2S_C}\right)^2 - 1\right]\frac{\sigma_{bb}}{Y_C} + \left(\frac{\sigma_{ab}}{S_C}\right)^2 - 1 \geq 0 \tag{6}$$

Herein, $E_a = \nu_{ba} = \nu_{ab} = G_{ab} = 0$ and the fiber matrix fails in compression.

The Johnson–Cook constitutive model, Mie–Gruneisen equation of state, and maximum tensile stress damage criterion were used for the Al alloy in the structure. The Johnson–Cook model [10] differs from the common plastic theory in that it characterizes the material response to impact and penetration through parameters such as processing